Time (Lcl) - 0803 EDT A/C Reg. No. N4973V 5/28/90 DELAND, FL File No. = 0641 AIRFRAME/COMPONENT/SYSTEM FAILURE/MALFUNCTION Occurrence #1 Phase of Operation APPROACH - MISSLD APPROACH (IFR) Finding(n) 4. ELECTRICAL SYSTEM, ELECTRIC WIRING - CHAFED 2. ELECTRICAL SYSTEM, ELECTRIC WIRING - SHORTED 3. ELECTRICAL SYSTEM, CIRCUIT BREAKER - OPEN 4. FIRE WARNING SYSTEM, POWERPLANT - FALSE INDICATION 5. PROPELLER FEATHERING - PERFORMED - PILOT IN COMMAND (CF1) 6. GEAR RETRACTION - NOT POSSIBLE -____ _____ FORCED LANDING Occurrence #2 Phase of Operation DESCENT - EMERGENCY Occurrence #3 NOSE OVER Phase of Operation LANDING - ROLL Finding(s) 7. TERRAIN CONDITION - SOFT ----Probable Cause----The National Transportation Safety Board determines that the Probable Cause(s) of this accident was: THE CHAFED AND SHORTED ELECTRICAL WIRING BETWEEN BUS BARS, WHICH CAUSED A PARTIAL ELECTRICAL FAILURE, RESULTED IN A FALSE FIRE WARNING INDICATION, AND PREVENTED THE RESTART OF THE RIGHT ENGINE.

Brief of Accident (Continued)

PAGE 2

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	National Transportation Safety Board M I A 9 0 F A 1 3 5							5					
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		enue operation conducted u		9, 135)	
54	- 55		56		
1 Scheduled 2 Non-scheduled	d 1	Domestic	1 Passen 2 Cargo	iger 3 Passenger/ 4 Mail contrac	
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1 Personal	-	4 Execut	ive/corporate 7	Other work use	
2 Business		5 Aerial a	application 8	Public use 10	Positioning
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Flight Plan/Itinerary					·
58 Flight Plan filed					
1 None 2 59 Itinerary-Last Departu	VFR 3	X IFR 4 IFF 60 State 61 Airport I.D.		pany(VFR) 6 Military	State 64 Airport I.D.
1 Same as accider			(If "local", mark X		
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A DAYTONA		TL DAD	A		
Weather Information	n				
65 Source	6	7 Sky/Lowest Cloud Condi	llon 68 Lowest	Ceiling	69 Visibility
1 Accident site (Pilot/	witness)	1 X Clear		one	(decimals)
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Facility	i	3 Thin broken		vercast	70 Temperature
A Facility Identifier D	AB	4 Thin overcast 5 Partial obscuration		bscured	79 Fahr.
66 Time of Weather Obse	ervation				71 Dew Point
		A Ft. AG		FL AGL	
<u>0831</u> (local)		· · · · · · · · · · · · · · · · · · ·			<u>73</u> Fahr.
72 Wind Direction	73 Wind Speed	74 Gusts 7	5 Altimeter 7	6 Weather Conditions (at accident site)	77 Precipitation
150 Degrees(Mag.)	5 Kts.	Kts.	29.76 Hg 1		1 Yes
Narrative				VMC 2 X IMC	2 X NO
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persons involved.)	The mornation 3			chocking on the character of t	megnity of the
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Cessna	T-303, N4	1973V, register	red to Embry	y-Riddle	
		—		making a force	
landing	g, follow:	ing shutdown o	Luc zight	engine due to	a
fire warning. Instrument meteorological conditions					
prevailed at the time and an instrument flight rules					
clearance had been received. The airplane received					
substantial damage and the commercial rated flight					
instructor and private rated dual student received minor					
injuries. The flight criginated at Daytona Beach, Florida,					
on May	on May 28, 1990 at 0730.				

	The inst	tructor pilot :	stated that	after departur	the
right a	alternato	r fail light i	lluminated,	and the altern	ator
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National Transportation Safety Board PRELIMINARY REPORT AVIATION ACCIDENT/INCIDENT

NTSB Accident Incident Number

Na, rative (continued)

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an NDB approach a missed approach was initiated. Shortly after power was applied for the missed approach the right engine fire warning activated. The right engine was shut down, per the pilot's checklist. At this time, it was also noted that the low voltage light was illuminated. The airplane would not climb after the right engine was shut down, and he verified that he had "cleaned up the airplane." The airplane would not maintain level flight, and while descending through 600 feet they entered the clouds. They broke out of the clouds at 150 to 170 feet and flew toward a The student asked if he should extend the gear and field. the instructor stated no. Touchdown in the field was made and the airplane nosed over. A small postcrash fire occurred in the left engine compartment.

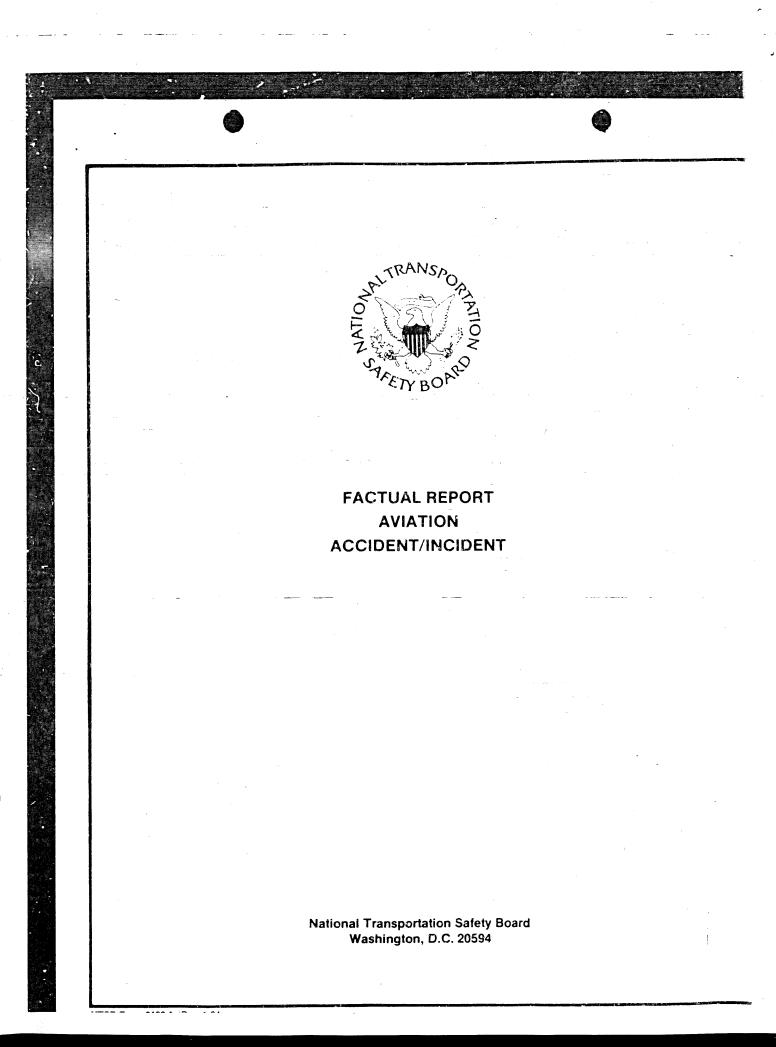
Examination of the airplane after the accident by FAA inspectors indicated the landing gear was extended and locked at touchdown. The landing gear handle was in the up position and the flaps were extended to 10 degrees. The right propeller was in the feathered position and the left propeller was in the normal flight position with no rotational damage. The investigation continues.

Administrative Data			
79 Notification From		Local Time (24 hour clock)	82 Time Zone
CROCKETT, FAA, FSDO ORL	ANDO, FL 5/28/90	0930	EDT
3 FAA District Office/Coordinator	84 Other Federal Agencles Involved In In	ivestigation	
	1 FBI 3 DEA	5 Customs	
CROCKETT, SDO ORLANDO, FL	2 USCG 4 DOD	A Specify	
nvestigator(s) Assigned		· · · · · · · · · · · · · · · · · · ·	
35 Investigator-In-Charge	86 Form Preparation Date (Nos. for M, D, Y	() 87 Form Receipt Date (Fo	or NTSB use only)
EFFREY L KENNEDY	5/30/90	5/31/90	
8 Other NTSB Personnel Assigned			
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NTSB Form 6120.19A (1-84) This form supersedes NTSB Form 6120.19(Rev. 1-83)

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	National Transportation Safety Board ACCIDENT FILE CONTENTS	PAGI	012NG
Tanspe	ortation Mode - Real Aviation C. H. Ghway - C. Pipeline - C. Intermodal - C. Marine - C. Rahroad	NTSR E115 - N	 L
IDENTI 	HEATION OF ACCIDEN) DELAND, FLORIDA CESSNA T303; N4973V 5-28-90 MIA-90-F-A135		24422400000000000000000000000000000000
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1.	Accident File Contents, NTSB Form 6120 3	-	
2.	Factual Report, Aviation Accident/Incident, NTSB For 6120.4 w/Supplements "A, B, E, I, M, and S"	m 30	-
3.	Pilot/Operator Aircraft Accident Reports, NTSB Form 6120.1/2w/Attachments	2.6	
4 -	Sheriff's Department Report	4	
5.	Airplane Maintenance Records	24	
6.	Excerpts from Cessna T303 Information Manual	- 5	
7.	Telephone Conversation Slip - Cessna Aircraft	12	
8.	Postaccident Fire Warning Tests	5	
9.	Release of Aircraft Wreckage, NTSB Form 6120.15	π -	
10.	Statement of Party Representatives to NTSB Investigation	2	
11.	Photographs (14) w/Negatives (4)	1	7
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NTSB Form 6120.4

Instructions

Unless otherwise stated in the instructions or on the form, all data fields must be completed. Each data field requires either a direct entry or the entry of one or more x's in appropriate blocks that best describe the mishap circumstances. Multiple entry fields may require two or more responses. Enter all applicable responses in multiple entry fields. When the selections offered are inappropriate, a two digit "other" code shall be entered in the space that follows the word "other." Do not make additional remarks in the margins as the automated data processor is not programmed to accept them. Any information which is needed to outline the sequence of events which preceded the occurrence, to support probable cause determination or which is pertinent to crashworthiness studies should be addressed in the narrative report.

"Other" Codes

01	Limited access to and/or limited time available at site.
02	Aircraft not recovered/missing.
03	Part/component not recovered/not located.
04	Aircraft too badly damaged to determine.
05	Part/component too badly damaged to determine.
06	Information not pertinent to accident/incident.
07	Applicable personnel could not provide information or information not available to applicable personnel.
08	Applicable personnel would not provide information.
09	Not installed.
10	Records not located/not available.
11	Information not entered on NTSB Form 6120.1.
12	- See narrative-report.

Supplements

The following accident scenarios are provided to assist investigators in selecting the report forms which should supplement the basic NTSB Form 6120.4.

1. A Cessna 172 collided with a snowbank during landing goaround at an airport. Weather was not a factor. The pilot said there was no powerplant or control malfunction. The pilot and one passenger received minor injuries. The pilot had recently been certificated as a private pilot.

Complete supplemental forms F (Training and Proficiency), Q (Airport) and S (Occupant list). A "Limited" investigation should be completed.

2. A PA-31, being operated by two pilots under FAR 135, crashed into a tower while being vectored to intercept the localizer at the destination airport. The PA-31 struck the tower while being operated at an assigned altitude. Flight was in IMC. There were two fatal injuries and three serious injuries. CFR personnel responded and treated the injured.

This accident requires an onscene investigation. Thus supplement A (Wreckage documentation), B (Cockpit documentation) and I (Crash kinematics) are required. Supplements E (Second pilot), F (Training and Proficiency) and U are required because of the two pilot FAR 135 operation (even though proficiency may not be at issue). S is needed to list the occupants; T, to document the CFR activity and P, to cover the possible ATC involvement. R (Meteorology) is required to document the weather conditions. Copies of supplements K and L would be required to document injury/toxicology and seat/restraint damage information, respectively.

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4 Aircraft Registration Number	5 Flight Numb	er	aircraft.	ision between enter reg. no for other aircraft	6	Aircrall Reg	istration N	lumber	7 Fligtis Number: A Citie
N4973V 8 Nearest City/Place		9 SI	itate 10 Zip Code (First 5 numt e		ers	onn 1	1 Acciden	t Site Elev	ation
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16 Narrative Statement of Facts, Conditions and Circumstances Pertinent to the Accident/Incident

HISTORY OF THE FLIGHT

On May 28, 1990, at 0803 (all times eastern daylight time), a Cessna T303, N4973V, registered to and operated by Embry-Riddle Aeronautical University, nosed over while making a forced landing near Deland, Florida, following loss of electrical power and shutdown of the right engine due to a fire warning. Instrument meteorological conditions prevailed at the time and an instrument flight rules clearance had been received. The airplane was destroyed, and the commercial-rated flight instructor and private-rated dual student received minor injuries. The flight originated at Daytona Beach, Florida, on May 28, 1990 at 0715.

The flight was a dual instructional flight conducted in accordance with Federal Aviation Regulation 14 CFR Part 141. The instructor pilot stated that during taxi to take off at Daytona Beach, the right alternator fail light illuminated. He noted electrical system load was below 30 amps. During pretakeoff checks, the light went out and remained out. The flight departed with the dual student flying the airplane. They received radar vectors to the Deland non directional beacon (NDB). The right alternator fail light again illuminated. The instructor noticed that no circuit breakers were open and that total electrical system load was less than 30 amperes. They shut down the right alternator by turning off the control switch.

The flight entered holding at the Deland NDB and after four complete holding patterns, performed the NDB approach to runway 30 at the Deland Municipal Airport. Upon reaching the missed approach point, the dual student applied climb power and initiated a missed approach. The right engine fire warning activated at this time and the electrical system low voltage light was

Additional Persons Participating in this Accident/Incident Investigation (Name, addres	ss. affiliation, Continue on page 2 if necessary)
Ken Crockett, FAA, FSDO, Orlando, FL	Jack Haun, ERAU, Daytona, FL
Dale Carter, Continental Engines, Mobile, AL	Joe Foole, ERAU, Daytona, FL
Rick Basco, Cessna Aircraft, Wichita, KS	A.C. Tucker, ERAU, Daytona, FL
Lyle Sunderland, ERAU, Daytona, FL	
James Teski, ERAU, Daytona, FL	

Investigaled By:							
17 Date (Nos. for M, D. Y)	18 Agency	19 Name/Signature	Kenniky				
10-22-90	NTSB (MIA)	JEFFREY L. KENNEDY	1				

NTSB Form 6120.4 (Rev. 1-84)

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also noted to be illuminated. The instructor pilot took control of the airplane. Upon climbing to 700 feet, the right engine was shut down and secured per the emergency checklist.

The airplane would not climb or maintain altitude after the right engine shut down even though full power was applied on the left engine. The instructor verified that they had placed the landing gear and wing flap handles to the retracted position. He did not remember hearing the landing gear retract or noticing what the landing gear light indications were. They broadcast a mayday call on the Deland unicom frequency, Daytona Beach Approach frequency, and on the emergency frequency 121.5. They also placed emergency code 7700 in the transponder.

As the flight descended through 600 feet, it entered a local area of low clouds. The student was asked to restart the right engine at this time. All controls were set for the start; however, when the starter switch was engaged the engine would not turn over. As the flight descended through 150 feet, the ground became visible and the instructor flew toward a field in preparation for a forced landing. The student asked if the landing gear should be lowered and the instructor said no due to the low airspeed. As the airplane touched down-in the field, they realized the landing gear was down. As the airplane rolled out, the nose landing gear bogged down in the soft dirt and the airplane nosed over.

The occupants exited the airplane through the left rear cabin door. As they exited, they noted that the right engine fire warning was still sounding and that a small fire had begun in the area of the left engine. After approximately 20 minutes, a sheriff's department helicopter arrived at the scene. The helicopter had been launched by the Daytona Beach Approach Control after receiving the emergency code 7700 on radar.

PERSONNEL INFORMATION

The pilot-in-command flight instructor holds a commercial pilot certificate with airplane single engine land, airplane multiengine land, and instrument airplane ratings. This pilot holds a flight instructor certificate with airplane single engine land, airplane multiengine land, and instrument airplane ratings. Information furnished by the operator indicates this pilot had accumulated 1,520 total flight hours, 100 flight hours in the Cessna T303, and 930 flight hours as a flight instructor. This pilot holds a second-class medical certificate, issued on February 23, 1990, with the limitation that correcting lenses be worn while exercising the privileges of the certificate. This pilot received a biennial flight review in accordance with Federal aviation regulations on September 20, 1989.

Attach additional pages as necessáry (Page 2a. 2b. 2c. atc. ,

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The dual student holds a private pilot certificate with airplane single engine land and airplane multiengine land ratings. Information furnished by the operator indicated this pilot has accumulated 183 total flight nours, with 48 flight hours in the Cessna T303. This pilot holds a first-class medical certificate, issued on April 24, 1990, with no limitations. This pilot received a biennial flight review in accordance with Federal aviation regulations on August 21, 1989. For additional personnel information, see data fields 87 through 145 of this report, Supplement E, and attachments.

AIRCRAFT INFORMATION

The airplane was a Cessna Aircraft Company model T303, serial No. T30300285, registration No. N4973V, manufactured on June 15, 1984. The airplane was equipped with Teledyne Continental TSIO-520-AE, 250-horsepower engines. Logbook records indicated the airplane had accumulated 2,542 total flight hours. The airplane received a phase six inspection in accordance with the Embry-Riddle Aeronautical University approved airplane inspection program on May 17, 1990, 42 flight hours before the accident. The left and right engines had accumulated 4,352 total hours, with 445 flight hours since overhaul. For additional airplane information see data fields 39 through 70 of this report.

WEIGHT AND BALANCE INFORMATION

Postaccident calculations indicated the airplane weighed 4,848 pounds at the time of the accident. The center of gravity was located at 151.0 inches aft of the datum. The weight and balance condition was within allowable limits for the airplane.

METEOROLOGICAL INFORMATION

The Daytona Beach Regional Airport 0831 surface weather observation was: clear skies, visibility 12 miles, temperature 79° F, dewpoint temperature 73° F, winds 150 degrees at 6 knots, and altimeter 29.76 inHg. Daytona Beach is located 15 miles east of the accident site.

The pilot-in-command stated that at the time of the accident a low cloud layer which extended from 150 feet to 600 feet above the surface was present in the area around the Deland Municipal Airport.

COMMUNICATIONS

The flightcrew stated that during the descent for the forced landing they attempted to transmit a mayday call on the Deland Airport unicom frequency, Daytona Beach Approach Control Frequency, and the emergency frequency 121.5-

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16 Narrative Statement of Facts, Conditions and Circumstances Parlinent to the AccidenVincident (continued)

They also put the emergency code 7700 in the air traffic control transponder. None of the voice calls were received by ground stations. The code 7700 from the transponder was received by the Daytona Beach Approach Control, which notified search and rescue.

After the accident, the No. 1 communications radio was found set to frequency 122.8, the Deland unicom frequency. The No. 2 communications radio was found set to frequency 125.35, the Daytona Approach frequency. The transmit selector was set to the No. 2 communications radio.

WRECKAGE AND IMPACT INFORMATION

The airplane came to rest inverted in a field approximately 3 miles east of the Deland Municipal Airplane. On site examination of the crash site was conducted by a Federal Aviation Administration airworthiness inspector. Tracks caused by the three main landing gears of the airplane extended from west to east for approximately 50 feet. At this point the airplane was lying inverted with the nose pointing to the west. All components of the airplane were found attached or lying adjacent to the airplane. The airplane structure and flight control systems were intact with no evidence of failure or malfunction.

The left propeller was found in a low pitch setting with chordwise scratches across the face of the blades. Each blade was bent aft slightly at the outboard ends. The right propeller was found in the feathered position. One blade was free to rotate in the hub.

The right engine showed no evidence to indicate the occurrence of a fire in the engine area or of excessive heating which would have caused activation of the right engine fire warning. The left engine had sustained minor fire damage to the accessory area as a result of a postcrash fire.

The main circuit breaker panel had five circuit breakers open after the accident. They were, the left bus isolation, right bus isolation, bus tie, landing gear indicator, and air conditioning compressor breakers. The landing gear handle was found in the landing gear retracted position after the accident. The landing gears were found extended and locked. The flap control handle was found in the flaps 10° position and the wing flaps were found in the 10° position. The flap handle was damaged during the accident.

Detailed examination of the airplane, systems, and engines, was accomplished by the NTSB after recovery of the airplane. Examination of the engine assemblies, ignition systems, and fuel systems, revealed no evidence to indicate precrash failure or malfunction. The left and right starter motors operated normally and rotated their respective engines. The wiring circuits for the starters had normal continuity after the accident.

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15 Narra	live Statement of Facts, Conditions and Circumstances Perlinent to the Ac	cident/Incident (continued)

The right engine fire warning loop within the engine compartment was separated in the accident. The wiring for the system tested normal after the accident. The left engine fire warning system wiring tested normal after the accident.

The left and right alternators were tested on a test bench after the accident. The left alternator was intact and had normal output of 28 volts and 95 amperes. The right alternator case was cracked and the drive shaft was bett as a result of impact damage. This alternator put out normal voltage and 50 amperes before the test drive belt slipping off the bent drive shaft. All alternator circuitry was tested and found to have continuity after the accident. The alternator control units were tested by installation on another airplane and were found to operate normally.

Examination of the aft side of the main circuit breaker panel revealed that an electrical power supply jumper wire between two strips of the left brs was burned. This wire was found to have been caught between the breaker panel chassis and the protective liner on the airplane structure. Further examination revealed a chaffing mark on the wire insulation in which wiring with arching damage was exposed. Arching damage was present on the grounded metal chassis of the breaker panel at the point the damaged wire was found to rest on the panel. The left pilot arm rest also had <u>slight</u> soot damage in an area that is adjacent to the burned wire on the panel.

The flap operating motor was tested after the accident and found to draw 2 amperes with no load on it. The flap motor and flap control circuitry had continuity after the accident. The gear operating motor was found to draw 5 amperes unloaded. The gear motor and control circuitry was found to have continuity after the accident. The left bus isolation, right bus isolation, and bus tie circuit breakers were tested after the accident. Each was found to open when a load of between 50 and 60 amperes were placed on them. The battery was found to have leaked all electrolite after the accident and could not be tested.

MEDICAL AND PATHOLOGICAL INFORMATION

The flight instructor and dual student received minor injuries as a result of the accident.

FIRE

A postcrash fire erupted in the area of the left engine turbocharger which was resting in dry grass after the accident. This fire caused damage to the engine accessory area of the left engine.

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16 Narrative Statement of Facta, Conditions and Circumstances Pertinent to the Accident/Incident (continued)

TESTS AND RESEARCH

The Pilot's Operating Handbook for the Cessna T303 states under section 3, emergency procedures, electrical power supply malfunctions, that:

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"due to component tolerances and low signal levels, it may be normal for one ALT OFF light to be illuminated when the total load on both alternators is less than 30 amps."

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The handbook states in section 7, airplane and system descriptions, that:

"a 50 amp main bus tie and 60 amp isolation circuit breakers are incorporated into the electrical system to automatically isolate the main bus bars from the battery supply current in the event of a fault of one or the other of the main buses. In such a case, the bus tie circuit breaker, labeled bus tie, would first become overloaded and open the circuit connecting the main bus bars together. If the fault remained uncorrected, the isolation circuit breaker of the affected bus bar would open, disconnecting that bus from battery supply. Without battery current, the ACU of the affected main bus shuts down the respective alternator, preventing further damage to the main bus system until correction of the fault."

Wiring diagrams for the Cessna T303 indicate that power for the engine starter motors, flap operating motor, and landing gear motor, are supplied by the left and right main bus system. As described above if the bus isolation breaker is open, no electrical power is supplied to the main bus system. This would prevent electrical power from reaching the engine starter motors, flap operating motor, and landing gear operating motor.

Performance charts contained in the Cessna T303 information manual indicated that for the conditions present at the time of the accident, the airplane should have a rate of climb of 170 feet per minute with one engine shut down, landing gear retracted, and flaps retracted. The charts do not give figures for rate of climb degradation do to extended landing gear and flaps. The instructor on N4973V stated after the right engine was shutdown, the airplane descended at approximately 200 feet per minute. (See attached excerpts from the Cessna T303 information manual.)

After the accident Embry-Riddle Aeronautical University personnel reported that on two previous occasions Cessna T303 airplanes operated by the university experienced false fire warnings due to electrical system malfunctions. They stated, in occurrences in which the bus tie circuit breaker opens and a bus isolation circuit breaker opens the engine fire warning for

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	NTS8 Accident/Incident Number
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FACTUAL REPORT	
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ative Statement of Facts, Conditions and Circumstances Pertinent to the Accid	ent/Incident (continued)
Cessna Aircraft Company engineers were not	ified of this false fire warning
occurrence. They were able to simulate the fal a Cessna T303 airplane at the factory. They the false warning and the need for a diode to system to prevent the occurrence during	se fire warning occurrence using further identified the cause of be installed in the fire warning
occurrence. They were able to simulate the fal a Cessna T303 airplane at the factory. They the false warning and the need for a diode to system to prevent the occurrence during malfunction.	se fire warning occurrence using further identified the cause of be installed in the fire warning times of electrical system
occurrence. They were able to simulate the fal a Cessna T303 airplane at the factory. They the false warning and the need for a diode to system to prevent the occurrence during malfunction. Examination of the maintenance records fo	se fire warning occurrence using further identified the cause of be installed in the fire warning times of electrical system or N4973V revealed that the last
occurrence. They were able to simulate the fal a Cessna T303 airplane at the factory. They the false warning and the need for a diode to system to prevent the occurrence during malfunction.	se fire warning occurrence using further identified the cause of be installed in the fire warning times of electrical system or N4973V revealed that the last hel was pulled loose from the
occurrence. They were able to simulate the fal a Cessna T303 airplane at the factory. They the false warning and the need for a diode to system to prevent the occurrence during malfunction. Examination of the maintenance records fo recorded time the main circuit breaker par airplane side wall was on January 4, 1990	se fire warning occurrence using further identified the cause of be installed in the fire warning times of electrical system or N4973V revealed that the last hel was pulled loose from the

Attach additional pages as necessary (Page 2a, 2b, 2c, etc.)

-	ansportation Safety Board TUAL REPORT AVIATION		1/Incident Number 9 10 1/F1/11/13 151
Airport/Approach/Landing Info 5 Airport Name 26 Airp Iden A Other	ort 27 Accident Location 2 Itilfier 1 Off airport/airstrip 2 On airport 3 On airstrip	Go to block 39) 8 Distance From Airport Center (Nearest SM) SM A Other	29 Direction From Airport °mag A Other
 b VFR Approech/Landing (Multiple end) 1 None 2 Traffic pattern 3 Straight-in 4 Valley/terrain following 5 Go around 6 Touch and go 7 Full stop 8 Stop and go 9 Simulated forced landing 10 Forced landing 11 Precautionary landing A Other 	1 None 2 ADF/NDB 3 SDF 4 VOR/TVOR 5 VOR/DME 6 TACAN 7 ILS-complete 8 ILS-localizer 9 ILS-backcourse 10 RNAV 11 MLS	12 LDA 13 ASR 14 PAR 15 Sidestep 16 Visual 17 Contact 18 Circling 19 Practice A Other	32 Runway Used Identifier 30 A Other 33 Runway Length 6003 Feet A Other 34 Runway Width 100 Feet A Other 35 Airport Elevation 200 Ft. MSL A Other
	1 Dry 2 Wet 3 Ice covered 4 Snow-dry 5 Snow-wet 6 Snow-crusted 7 Snow-compacted 8 Vegetation 9 Water-calm 10 Water-choppy	11 Water—glassy 12 Rubber depose 13 Soft 14 Rough 15 Slush covered 16 Holes A Other	its
Aircraft Information Aircraft Manufacturer	40 Aircraft Model/Series	41 Serial No. 730300.385	42 Certificater Maximum Gross Weight 5 / 75
CESSNA 3 Typo of Aircraft 1 Airplane 5 Blimp 2 Helicopter 6 Ultral 3 Glider 7 Gyro 4 Balloon A Specify		A Other	A Other 45 Home Built 1 Yes 2 No A Other

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Aircraft Inform	nation (continued)											
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lf not Engine powered. go to block 59	53 Engine Manufactur	LISIC	Engine Model and Series 55 Engin 27520-520-AE A-2 B			25Ø	ted Power Horsepower Lbs. Thrust		per of Engines			
If 3 or more	Engine Time (Hours)		otal Time	В	Time Since	Inspect	ion	C Time O	Since			D Other
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Supp. C	58 Engine No. 2	435		-	-42		- 445			·		
3 Other a	-		3 🗙 🗛	nual hour P	airworthiness		61 Date Last inspection Performed (Nos. for M. D. Y) 5-17-90 A Other			62 Time Since Inspectio		
64 Source of Ma	Intenance Information				65 Hazard		teriais	Emergency L			2	A
1 Tach 2 Flight 3 Hobbs	4 5 6	Estima	oxs Records ite Operator Report		on Airci 1 No A (Type)	1		Transmitter 67 Installed) Yes	No	Other
66 Hazardous M	Aterial Spill/Factor	Other	<u></u>		B Other			66 Required	l 			d
1 Yes 2 No A Other								69 Operated	ocatio	1 X 1		
	ator Information		1				· · · · · · · · · · · · · · · · · · ·	of accider				<u></u>
71 Registered Air Name Eng	craft Owner lY-RIDDLEAELON		L UNIVERS		72 Address			L AIRPO			14	
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NTSB Form 6	120.4 (Rev 1-84))	~	Page 4

National Transportation Safety Board FACTUAL REPORT AVIATION	NTSB Accident/Incident Number
FACTUAL REPORT	NTSB Accident/Incident Number
FACTUAL REPORT	
	MIIA910VFA1/13151
Owner/Operator Information (continued)	
77 Operator Status of This Aircraft 78 Pilot Status 1 Owner 4 Borrower 1 Owner 2 Lessee 5 Unauthorized 2 Lesse 3 Renter A Other 3 Renter	ee 5 Unauthorized
Type of Certificate(s) Held	79 None (X (30 to block 83)
B0 Air Carrier Operating Certificate (Check all applicable) 81 Operating Certificate 1 Flag carrier/domestic (121) 4 Large heliccoter (127) 2 Supplemental 5 Commuter air carrier 3 All cargo (418) 6 On-demand air taxi	
Regulation Flight Conducted Under	
83 Regulation Flight Conducted Under 1 14 CFR 91 (only) 4 14 CFR 105 7 14 CFR 1 2 14 CFR 91D 5 14 CFR 121 8 14 CFR 1 3 14 CFR 103 6 14 CFR 125 9 14 CFR 1	33 11 14 CFF 129 (Foreign (lag)
Type of Flight Operation Conducted	
(Complete 84a, b, c ONLY if flight was a revenue operation conducted unc	der 121, 125, 127, 129, 135)
84a 84b 1 Scheduled 1 2 Non-scheduled 2	84c 1 Passenger 3 Passenger/cargo 2 Cargo 4 Mail contract ONLY
(Complete 86 ONLY if 84a, b, c is not applicable)	
36 4 Executive/corporate 7 1 Personal 4 Executive/corporate 7 2 Business 5 Aenal application 8 3 Instructional (Including air carrier training) 6 Aenal observation 9	Other work use 10 Positioning Public use Ferry A Specify
First Pilot Information	
B7 Name (Last, First, Initial) 88 Pilot Certificate No. <u>RAPNANO, MICHAEL</u> J.	89 Street Address <u>P. O. Bcx</u> 10167 A Other
$\begin{array}{c c} \textbf{30 City} \\ \hline D \ D \ Y \ To NA \ BEACH \\ \hline A \ Other \end{array} \qquad \begin{array}{c c} \textbf{51 State} \\ \hline FL \end{array} \qquad \begin{array}{c} \textbf{92 Date of Birth (Nos. for N)} \\ \hline \hline q \ -7 \ -66 \\ \hline A \ Other \end{array}$	$\begin{array}{c c} M, D, Y \end{array} \begin{array}{c} \textbf{93 Age} \\ \hline \begin{array}{c} 2 \\ \hline A \\ \end{array} \end{array} \begin{array}{c} Yrs. \\ \hline A \\ \end{array} \begin{array}{c} \textbf{94 Sex} \\ 1 \\ \hline \end{array} \begin{array}{c} Male \\ \hline Female \end{array}$
95 Sent Occupied 96 Principal Profession 1 Left 1 2 Right 2 3 Center 3 Other-military 9 Student	rancher 1 Student Flight Engineer 2 Privatz 7 Military 3 Commercial 8 None

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	First Pilot Information	(continued)	Multiple entry	- biecka 98	-1021				1/		
-	98 Ratings-Airplane	99 Rotorcraf	l:Glider/LTA	100 Instri	ument Rating	101 Instructo	or Rating(s)				
	1 None 2 X Single engine and	1 X Ne		3	None		one	6 Glide			
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	4 Single engine sea		snip		-	4 H	elicopter				
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	102 Ground instructor		ing Endorseme	nt This	104 Months	Since Check/I	Endorsement	105 Biennial Figm Peries			
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	139 Shoulder Harness Used			sy Perform Yes	ed (This priot)			Yes	1.00 -0001		
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Pilot Information (continued)			
142 Person at Controls	143 Simulated Instrument Filght	144 Vision Restricting Device User	145 Second Pilot
1 Pilot in command 4 Non-pile 2 Second pilot 5 No one		1 Yes 2 X No	1 Yes (Lan second pira
3 Both pilots A Other	2 No A Other	A Other	2 No
Flight Itinerary Information			
155 Last Departure Point (Multiple entry)	157 Destination (Multiple el	ntry) 158 Flight P	n Filed (Meultiple arth
1 Same as accident/incident location A Airport identifier <u>DAB</u>			one sual Fright Rules 1 ⁰¹⁵
B City/Place DAYTONA BEAC	2 Local flight <u>H</u> A Airport Identifier		istrument Flight Fue
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156 Time of Departure	C State-		ompa ⊳y ∋∿⊊R)
A Time $07/5$ C Other	D Other	· · · · · · · · · · · · · · · · · · ·	hiltary (VF=)
B Time Zone <u>EDT</u> 159 Type of Clearance	160 Airspace	A Other	
2 VFR 7 Cruise 3 Special VFR 8 Traffic Adv 4 FR 9 VFR Flight 5 Special IFR Following A Other	4 Control zone 5 Airport advisory 6 Positive control	11 Restricted area area 12 Military Operating Arianea area 13 Student Jet Training	
	7 Terminal control		s.
3 Jet airway 3 S 4 Control airway 4 R 5 Colored airway 5 D	tandard instrument departure 8	VR route (military) Estimation IR route (military) 1 SR route (military) 2 Refueling route (military) A ter	t Two Way Commune ablished None Yes Facility Identifier DAB TOWER Other
Aircraft Loading Information			
164 Fuei on Board at Takeoff (Multiple entr) 1 Estimated 2 Verified A Gallons or B 700 Pounds	165 Fuei Types (Multiple entry 1 80/87 2 100 low lead 3 100/130 4 115/145	5 Kelosene 9 Ma 6 JP 3, 4, 5, 6 10 Au	ture omotive Hice additive anost if
C Other 165 Aircraft Weight at Takeoff (Multiple enti-		167 Aircraft CG at Takeofi (Multo	le entry i
1 At or below max cert. gross take		1 X Within limits	5
2 Above max certified gross takeof		2 Exceeded fwd limit	6
3 Estimated		3 Exceeded aft 'imit 4 Exceeded lateral limit	A Other
4 Verified A Other 168 Aircraft Weight at Accident (Multiple en	tryi	169 Alrcraft CG at Accident (Muth	ple ectry :
1 Same as takeoff	·· ; /	1 Same as takeoff	_
2 At or below max cert. gross take	off weight	2 Within limits	6 X San 7 7
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Aircraft Loading Info	ormation (contin	nued)			
170 Load Description Atu 1 None 2 Passengers	it ple entry) 3 Cargo 4 Towing glide	5 Towing banner r 6 Other external	7 Parachutist 8 Water	s 9 Chemic 10 Livestoc	
Weather Information					
2 National Weath 3 Flight Service S	ieting - Go to btock 18 er Service (NWS) Station r Automated Tel, WX	6 7 8 7 7 Co 8 TV	mpany mmercial weather se radio weather itàry		81 Method of Briefing Multiple entry: 1 Disenson 2 Disectore 4 Disectore 4 Disectore 4 Disectore 5 Disectore 4 Disectore 5 Dis
1 Weather not pe 2 Full 3 Partial—limited 4 Partial—limited A Other			to block 185)	C Elevation D Distance from	$\frac{233}{100} = \frac{923}{22} = \frac{100}{22}$
185 Basic Weather Condit 1 Visual Meteorol 2 Vithstrument Met A Other		is (IMC) 2 Dayli 3 Nigh	n 1 ght 2 t (Dark) 3 t (Bright) 4 5 A	//Lowest/Cloud Cond Clear Scattered Thin broken Thin overcast Partial obscuratio ————————————————————————————————————	n A S
189 Visibility (decimals) A 12. SM B RVR Feet C RVV SM D Other	190 Temperature 79 F A Other 191 Dew Point 73 F A Other	192 Wind (From) 1 Variable A /S D Magneti B Other	Vana	and A	Kts. A Other
197 Restrictions to Visibilit 1 None 2 Haze (H) 3 Dust (D) 4 Smoke (K) 5 Fog (F) 6 Ice fog (IF) 7 Ground fog (Gf 8 Blowing spray (9 Blowing dust (E 10 Blowing snew (11 Blowing sand ()	1 2 3 4 5 6 7 8 9 8 9 8 9 8 9 8 9 8 9 8 9	ype of Precipitation None (Go to block 200) Rain (B) Snow (S) Hait (A) Rain showers (RW) Freezing rain (ZR) Snow shower (SW) Drizzle (L) Ice pellets (IP)	11 Snow 12 Freez 13 Ice ci	r pellets (S₽) r grains (SG) ring drizzte (ZL) rystals (IC, ellet show∉r (IPW)	199 Intensity of Precare 1ught 2Whoderaw 3eavy A Cther

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T					T			and a second		
1 None 1 2 Minor 2 3 Substantial 3					203 Da 1 2 3 4 5	Non Res Res Con	Property e dence dential area imercial bidg iclersi	6 Arcort facili 7 Treet 8 Cropsili 9 Ferice 10 Writes poles 11 Other prope	-	
204 Injury Index (Most critical in) 1 None 2 Minor		Senous	4 Fatal	je I						
Injury Summary	A	B	C Minor	D None	E			i i i i i i i i i i i i i i i i i i i		
205 First Pilot	Fatal	Serious		Hone			217 Classification			
205 Co-pilot		+ + +	++++	+ + +		+4	1 US R	egistered Aircraft on	US S:	
207 Dual Student		+-+	+ + + 7			+1		ries and Possession	s, or	
208 Check Pilot		╂╌┾╾┾╼				-++-	International Waters 2 U.S. Registered Aircraft on Fo			
209 Flight Engineer	┝─┼━╌┼──			+			Soil	egistered Antrais on	Foreign	
210 Cabin Attendants			+ + +-+				······	egistered Aircraft op	erated D.	
211 Other Crew		<u> </u>		+				n Operator		
212 Passengers						4 Foreign Registered Aircraft on US				
213 TOTAL ABOARD			1 3	1		15	Soll T	erritories or Possess	ons	
214 Other Aircraft			1-1-10	¥+		19		y Aircraft		
215 Other Ground	-			++			6 jAircrat	t nor Registered		
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216 GRAND TOTAL			12			171				
Part Failure/Incorrect Part	LAA				B ar b ar-	<u></u>				
220 Part Failure/Malfunction (Mul-	tiple entry)			12	21 Incor	rect Par	(Multiple entry)	-		
1 None		4 P	art compone	ent #3	1	None		4 Part com	onent=1	
2 Part component #1		A Other			2	Part co	mponent #1	A Other		
3 Part component #2					3	Part co	imponent #2			
		A Part/(Component #	#1	В	Part/Co	mponent #2	C Part/Cor	nponeπ≍	
222 Part Name	12	UMPER	WIRE							
223 ATA Code		2450							,	
224 Manufacturer		CESSA								
225 Mfg. Part #		~/n								
226 Mfg. Model #		N/N	1							
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227 Serial # 228 Part Condition 229 Total Time		HAFFED 2542				·				
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523302		1 🛛 Yes 2 🖾 No		X Yes	Manufa	Caurer 94LF4	Model's
A Other	A Other	A Other			A Othe	and the second re-	A Other
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7 Propeller Type (M	_	Ground Adjustable/	variable pitch				Actication!
1 💭 Wood 2 🔀 Metal		Reversable Full automatic feath					
3 Composite		Eull manual featheri			A Ot		
4 🖾 Constant sp	eed-controllable pitch A (·	1		
Landing Gear	9 Nose/Tail	10 Left Main		11 Right Main		For Ro	otorcraft o
Positions (If fixed gear,		1 🗌 Up	-	1 🖸 ປາ	1	Balloo	n accide
go to block 12)	2 🖾 Down 3 🔲 Intermediate	2 🔯 Down 3 🔲 Interme	diate	2 🛛 Down 3 🗍 Interme		to bloc	:k 20.
go to 2.00.12,	A Other	A Other	UI315	A Other	diate .	4	
Controi Surface	12 Left Trailing Edge	13 Right Trailing	Eda	14 Speed Brake	-	15 Spoile	
Positions	12 Len Trailing Loge Flap	Fiap	Eage	14 Speed Brake	alled		r Not Insale:
	1 🗖 Up	1 🗖 Up		2 Stowed			Stowed
	A Extended 10 dec	· · ·	10 - deg	3 🗌 Deploye	d		Deployed
•	8 Other	B Other	•	A Other		and the second	Deployed 16
				-		A Oth	ier -
Trim Tab Positions	16 Lett Aileron 1	17 Right Alleron 1 Not Inst	- 11	18 Rudder 1 🗌 Notinst			tor/Stabiles
(Multiple entry)	2 X Neutral	2 X Neutral	alled		alleo		Neutral
	3 🗍 Up	3 🗍 Up		3 Left		2 🗋	
	4 Down	4 Down		4 🗖 Rigta		з 🗖	Down
	A deg.	A de	g.	A deg		A	deg.
	B Other	B Other		B Other		BOth	her and
Cargo Restraint	20 Cargo Restraint Installed	(Multiple entry) 21	Cargo Restrain	t Used (Multiple ent	(y) 22 Can	o Restrai	nt Falled /M.
System	1 🛛 None (Go to block	26)	1 None (G	o to block 26)	<u> </u> 1 [None	
	2 Cargo net		2 Cargo ne			Cargo net	
	3 Straps/tie down		3 Straps/te	e down] Straps/ti	ie cown
	A Other			t and/or cenier	A O		00.040.0
Computed Weig	ht and Balance Informa	accident II	ight. (Otherv	vise go to block	32)	annano	nis die er
Takeoff							
	Center of Gravity 26	CG Range (Multiple e	ntry)				
-	A % MAC or	1 At takeoff weigh		А	%	MAC to	% MAC
	B Inches	2 At max gross w		B			Inches
l						32 Fue	On Board
Accident			میں				Estimated
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	A% MAC or	1 At takeoff weigi	ht A	% MAC to	% MAC o	AT AT	otal gailors
	B Inches	2 At max gross w		Inches to)ther

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	Àircra	t (conti	nued)			1. T.				$\langle l \rangle$					
	Fuel on	Board at Ac	cident	DT	ank Constru	iction		F Sp	illsafe	Fittings		H Fuel	Leakage/I	Rupure	
Fuel Tanks	A Gallons Estimated	B Gallons Verified	C Other	1 Wet Wing	2 Bladder	3 Metal	E Other	1 Yes	2 No	G Other	1 None	2 Line	3 Fitting	4 Ta**	Ctr.
33 Left Wing	50			X					X		X				
34 Right Wing	50			X					X		X				
35 Left Tip	-				•					-	ľ N				
36 Right Tip	-													j.	
37 Fuselage				· .				-	-						\$
38 (Specify)													-		
41 Fuel Found In #1 E	ngine (Mult	iple entry)	A			42 Fi	el Fou	nd in #	2 Eng	ine (Mu	ltiple er	ntry j			
: 🔲 None		7 🖾	Filter(s)		1						7 🛛 F	liter(s)		
2 🖾 Lines			Selecto							_			elector val		
3 🖾 Gascolator				nanifold spider 3 Gascolator/strainer 9 G Fuel man								s.			
4 🔯 Carburetor.		10 🗆	Accum	ulator ti	ank						or 1	0 🗖 🗛	ccumulate	or tatik	
5 🖾 Engine drive								-							
6 🖾 Auxiliary fu		A 01	-	-			- <u>-</u> - Aı	· · · ·	T			4-Othe		-	
43 Flight Controls, Evidence or		rame/Struct		ence of	in-ragnuse	paration	/ranure	•		Propelle of In-Flig	r,Evidea	œ	4S Powerp of In-Fi		
Operational Failure		None			7 🗖	Right s	tah/ele	vator			on/Failu	re	Malfun		
or Malfunction		-	(Comp	ete Supp. G) 8 Vertical fin rudder						1 🛛 Ye	s		1 🛛 Yes		
(Multiple entry)	_	General di			1					2 🔯 No			2 🖾 No		
1 🔀 None	4	Left wing	-							A Other A O			A Othe	er	
2 Pitch contro	5	Right wing	3	11 🔲 Cabin/cargo door						1					
3 🔲 Roll control	6	Left stab/e	levator		A Ot				1 .						-
4 🔲 Yaw control	47 Fuel	, Evidence	of Impr	oper Gr	ade or Con	taminal	ion 4	8 O%,	Evide	nce of	mprop	r Grad	e or Conta	mitatik	'n
A Other	(Mu	itiple entry,						(M u	itiplə	entry)					
	1 🛛	& None		3	Contar	nination	n [1	🕄 No	one		з [Contan	ninzion	1
	2 🛛		r grade	A	Other			2 [] Im	proper g	grade	A (Other		
Emergency Locat	or Transm	litter (EL1													
51 ELT Manufacturer			52 EL1	Model	No.	0 1			5				ion(s) (Mul	tiple end	τ γ !
DORNE MAR	GOWN				ELT 2	201					Cockpit				
A Other				Other					_	2			5 🖸 Ra		
53 ELT Battery Type	· —		54 ELT	Batter	Expiration	Date () か	vos. lor	м. D,	71		Tailcon -		6 🗖 Si		ί ι
1 🛛 Alkaline					<u>~~~</u>	¥		-		٩×	Empenr	age	A Other		
2 Cadmium	5 🗖 Li		AO	ther											
3 Nicad	A Other														
56 ELT-Reason for N		6 🔲 Bati				11 🗖	Water	submo		•	с П т.		factorily a	1101 700	dent
1: 🔀 Operated e 2 🔲 Insufficient						_	Unitin						rection alte		
· · = ·		8 S Fire					Shield						device still		
3 Improperar 4 Battery dea						_	Shield	-					witch off	1112-22/192	J
					sconnected										
5 Battery cor					sconnected						Other		-		

National T	ransportation Safety Board	NTSB Accident/	Incident Number
· · · · · · ·	CTUAL REPORT AVIATION	m. J.A.	70FAI/35
Supplement B — Cockpil Aircratt 1 Cockpit Secured, Readings Not P	and the second	M: J A Cockpit/Instrument Panel Destroy	
Cockpit Instrument Indication	Enter direct in appropriate c	ategory	
Flight In	struments	Engine/System	m Instruments
ltem	Reading/Setting	ltem	Reading/Setting
· · · · · · · · · · · · · · · · · · ·			
• ••••			
; 			
Comm/Nav	/ Equipment	Miscel	laneous
ltem	Frequency/Remark	ltern	Remark
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NTSB Form 6120.4 Supplement B (1-84)

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National	Transportat	ion Safety	Board		NTSB Acc	ident Incid	lent Number
· · · · ·	CTUAL F AVIATI	ION			m T	A.9.1	DEAL3E
a second s	all (contin	ued)					and the second
2 🖾 Glide slope 3 🖾 HSI 4 🔲 Flight director 1 5 🔲 RMI A 6 🖾 RNAV	s Installed (Mui 7 🗍 LORAN C 9 🖾 DME 9 🖾 ADF 0 🖾 Marker be 0 Other	Omega INS Bacons	2 🔲 Eng 3 💢 No A Other	t installed gaged t engaged	5 Digital Electro Nav/Com Dia 1 Not inst 2 X Installe A Other	plays talled d	6 Primary Altimeter Type 1 Counter-pointer 2 Drum-pointer 3 X 3-pointer 4 2-pointer A Other
1 🔀 Yes 2 🔲 No	Radar Altimeter 1 Yes 2 S No A Other	r Installed	3 🔲 Insta	nstalled lled-not used lled-used	d titlude encoding	0 Attitude Indicator Installed 1 ☑ Yes 2 □ No A Other	
1 🔀 Pressure vacuum system 2 🔲 Pressure vacuum system 3 🔲 Electrical 4 🗔 Standby indicator with al A. Other	4 🔲 Standby indicator with alternate power source				ig Indicator 1	1 🗆 N 2 🛄 ir 3 🔀 ir	r Radar/Detection Equipment lot installed nstalled-on nstalled-off nstalled, on/off unknown
14 Type Weather Radar/Detection 1 Storm scope 2 Electrical/System Switches	Black and with		3 🖾 Color		-A		ible (Go to block 56)
Switch/Item	Not 1 Installed	2 On	3 011		vitch Positions N	oi Pertiner	nt (Go to block 56) nt Setting Remark
20 Electrical Master 21 Battery 22 #1 Gen/Alternator 23 #2 Gen/Alternator 24 Inverter	X	X	X				
25 Avionics Master 28 Pitct Heat 29 Ice Detection 30 Propeller Deice/Anti-Ice 31 Windshield Deice	X		X X X				
32 Windshield Anti-Ice 33 Airframe Deice 36 Cabin Air/Fan 37 Cabin Heater 38 Air Conditioning 39 Cabin Pressure Attitude							
40 Cabin Pressure Temperature 41 Crew Oxygen 42 Cabin/Passenger Oxygen 45 Taxi Lights			X				
46 Landing Lights 47 Rotating Beacon		X					

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Nationa	Transportation S	Safety Board		NTSB Accident/Incident Number
F.	ACTUAL REP AVIATION			MIJIA19101FW1/13151
Supplement B—Cockp Aircra	il Døcumentatio It (continued)	òn, Single ár	ıd Twin Recip	procating Engine and Unpowered
Electrical/System Switches (
Switch/Item	1 Not 2 Installed	On 3 Off	A Other	Pertinent Setting Remark
49 Navigation Lights 50 Instrument Panel Lights		<u></u>		
51 Cockpit/Storm Lights 52 Cabin Lights 53 ELT Remote		- X - X		
Engine Controls-No. 1 Engir	ne		56 🗌 Engine (Control Positions Not Pertinent (Go to bicc+ 55)
 57 Throttle Position Not installed Full forward Midrang- Idle A Other 1 Alte.nate Air Not installed Open I Closed Midrange Other 	3 🔲 Midrange	ease (Low pitch) e ease (High pitch) lled	 59 Mixture Not instal Full rich Full rich Midrange Midrange Milde cuto A Other 63 Magneto Switcl Not instal Not instal Not instal Both Left Right Off Start A Other 	2 Fuilton 3 Fartial 41 Off A Cther
Engine Controls-No. 2 Engi				gine Control Positions Not Pertinent (Go to clock 74
66 Throttle Position 1 □ Not installed 2 □ Full forward 3 □ Mictrange 4 ☑ Idle A Other	3 🗖 Midrange	ase (Low pitch)	3 🛛 Midrarige	2 Eull on 3 Partial
70 Alternate Air 1 Not installed 2 Open 3 X Closed 4 Midrange A Other	71 Cowl Flaps 1 Not insta 2 Open 3 🖾 Closed 4 Midrange A Other		72 Magneto Switc 1 □ Not insta 2 □ Both 3 □ Left 4 □ Right 5 ☑ Off 6 □ Start A Other	

Nat	Ional Transportation Safet FACTUAL REPOR AVIATION	-	-	NTS	9 Accident/Incident	Number	
Alle	ckpit/ Documentation, S rail/ (continued) (Indicators, Flight Controls/I	$\rho \rightarrow \gamma$			To the local	FIAI/13151 a Unpowered	
Landing Gear Control I Not installed Z X Up G Down G Off	75 Landing Gear Indicator 76 1 Not installed 2 Up 3 Down 4 Transit/unsafe	76 Trailing Edge Flap System 1 Not installed 2 Manual 3 Electric 4 Hydraulic A Other			relling Edge Flap Contro! Not installed Up Down _/ rieg Other	78 Trailing Edge Fap Indicator 1 Not installed 2 Up A Down Jeg	
A Other 9 Speed Brake Control 1 🖾 Not installed 2 🗋 Stowed 3 🗋 Deployed A Other	A Other 30 Spoiler Control 1 X Not installed 2 Stowed 3 Deployed A Other				B Other B Other Throwover Control Yeke/Position Not installed Left Right Intermediate A Other		
3 Elev/Stab Trim Control (Multiple entry) 1 Not installed 2 Manual 3 Electric A Other	84 Elev/Stab Trim Indicator 1 ☐ Not installed 2 ☐ Up 3 ☐ Down 4 ⊠ Neutral A Other	S5 Aileron Trim Control 6 (Multiple entry) 1 1 1		1 🛛 2 🖵 3 🗖	Right Neutral	87 Rudder Trim Indicator 1 Not installer 2 Left 3 Right 4 Neutral A Other	
Fuel Selector Position(s)	(Multiple entry) 7 Forward 8 Aft 9 External tank 10 Between tanks 11 X-feed left to right 12 X-feed right to jeft			_ `	89 Fuel Boost Pur 1 Not insta 2 On 3 High 4 Low 5 Other		
 Fuel Boost Pamp, Engin 1 Not installed 2 On 3 High 4 Low 5 Q Off A Other 	e #2 91 Fuel Transfer Pum 1 🖾 Not installed 2 🗋 Off A On (tank to B Other		1 🛛 Not 2 🗌 Lock		2	mer Engine #2 Not installed Locked Unlocked Other	
	· · ·						
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Nat	ional Transportation S	-	NTSB Accident/Incident Number	
	AVIATION		MIA90F413	5
Supplement E —	Second Pildt Informa s	Nion		
1 Copilot 2 🔀 2 Name (Last. First. Initial)	Dual student 3 Safety pi 3 Pilot Cer		5 None (Pilot-Rated Passenger) A Other	
<u>FOLKERTS, DAR</u> A Olher	A Other		A Other	
S City DAYJONA BEAC A Other	H FL	7 Date of Birth (Nos. ior 	M, D, Y) B Age 9 Sex 1 M Male 2 □ Femal	e
10 Seat Occupied (Multiple of 1 🛛 Left 4 🔲 Fr 2 🔲 Right 5 🛄 Ri 3 💭 Center A Other	ront 1 🗍 Pilot-civilian ear 2 🗍 Pilot-military	4 Aircraft mechanic 5 Business 9 6 Lawyer	7 Doctor/dentist 10 Clergy 13 Farr 8 Police 11 Teacher 14 Reti 9 Student 12 Engineer A Other	ner/Fancher red
12 Certificate(s) (Multiple en 1 Student 2 Ø Private 3 Commercial 4 Airline Transport 5 Flight Instructor 6 Flight Engineer	try) 7 ☐ Military 8 ☐ None 9 ☐ Foreign A Other	 3 Ratings—Airplane 1 □ None 2 ☑ Single engir 3 ☑ Multiengine 4 □ Single engir 5 □ Multiengine 	1 🖾 None e land 2 🖬 Helicopter land 3 🗍 Gyroplane e sea 4 🛄 Airship	ile entry)
15. Instrument Rating 16 (Multiple entry) 1 1 X None 2 Airplane 3 Helicopter	2 Airplane SE 6 0 3 Airplane ME 7 1	Gyroplane Glider nstrument airplane	Isound Instructor 18 Type Rating/Endorsement Basic Aircraft Advanced 1 Yes Instrument 2 X No (Go to block 20) None A Other	t This
19 Months Since Check/End This Aircraft A Other	orsement 20 Biennial Flig 1 🛛 Yes 2 🗌 No A Other	nt Review 21 Months 9 A Othe	Since Last BFR 22 BFR (or equivalent) Aircraft Ma Months A Make CECSNA B Model 172 C Other	ke/Wodel
23 Medical Certificate 2 1 □ None 2 ☑ Class 1 3 □ Class 2 4 □ Class 3 A Other	 4 Medical Certificate Validity 1 X Valid medical-no waive 2 Valid medical-with waix 3 Non valid medical for the formation of the formation o	ers/limitations A Other	redical certificate $ \frac{25 \text{ Date of Last Medical (Nos. 10)}}{4 - 24 - 94} $ A Other	- D, M, Y) -
26 Medicel Linstation 1 🖾 None 2 🛄 Vision A Specify B Other	27 Medical Walver 1 🖾 None 2 🗌 Vision 3 🔲 Hearing A Specify B Other	28 Statemeni of Demons 1 🗌 Yes 2 🖾 No A Other	29 Correcting Lenses (Multiple of 1 2 Not required 1 2 Not required 2 Required to be in possed 3 Required, not in possed 4 Required to be worn 5 Required, not worn 6 Worn at time of accided A Other	ssion
	FAA	5 🔲 Investigator's Estim 6 🔲 Relative	te 7 🗌 Other Person A Other	/ Page

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Natic	FACT		tion Safety REPOR				NTSB Ac	cident/incl	dent Numi	bү r	
	-			-			MI	A 9	O[F]	Ā1/3	50
Supplement E – Se	cond Pi	lot int	Sec. 1970	and the second			1. A.	1	¥.	4	
Flight Time	A All A/C	B This Ma & Moo		Airplane	E Night	F Instru Actual	ument G Simulated	H . Rotorcraft	l Glider	J Lighter Than Air	X Other Code
35 Total Time	183	48	135	48	19	4	60				
36 Pilot in Command (PIC)	79	5	74	5	5						
37 Instructor										-	
38 This Make/Model	7.9	(<u>1) 2</u> 75			-					1-	
39 Last 90 Days	21	21		21	2	1	12				
40 Last 30 Days	13	13		13		1	2				-
41 Last 24 Hours			· · · · ·								
42 Landings—Last 30 Days	Ali Aircraft-	Day	43 Landings	—Last 90 Days	-All Aircr	aftNighi	44 Lan Day		l 90 Days~	-This Make/	Modei
A Other // 45 Landings—Last 90 Days—	This Make/	Model—	A Other / 46 Seatbeli				A Oth	er / ¦ stbeit Uso			
Night			1 🛛 Yes 2 🗌 No	5			11	X Yes			
A Other //	a 49 Shou	ider Hari	A Other	50 Autopsy	Performs	d — (This		Other 51 Toxico	logy Perio	ormed (T	his Pilot)
1 🖾 Yes 2 🗖 No A Other	1 🛛 2 🗖 A Oti	Yes No		1 Yes 2 🔀 No A Other				1 🔲 Y 2 🗖 N A Othe	es io		
						-					
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Natio	onal Transportation Safety Board		NTSB Accident/Incident Numbe	er
-	FACTUAL REPORT AVIATION	·		
pplement I—Crash	Kinemalice		MJAI9OFK	A1/315
ccident Site Geographic	Coordinates-Latitude (Multiple entry)	2 Accident Site Ge	eographic Coordinates—Longitud	de (Multiple entry)
North	A _ 29 deg 04 minutes	1 🔲 East	A 081 dec	g. <u>17</u> minutes
2 🔲 South	B Other	2 🔀 West	B Other	
1 None 2 Rock face 3 Rigid structure 4 Rocks to 1' diam 5 Rocks 1'-2' diam. 6 Rocks > 2' diam	er in sequence. Multiple entry.) 7 🖾 🖾 Ground 8 📄 Dirt bank 9 💭 Scrub tree 10 💭 Trees.limbs to 6'' diam. 11 💭 Trees.limbs 6''-9'' diam 12 💭 Trees limbs 9''-12'' diam	14 🔲 Frangible ap	opproach aid 20 Wa oble approach aid 21 Wi obstacle 22 Po	•
Terrain at Principal Impac 1 None 2 Wei cultivated soil 3 Dry cultivated soil 4 Dry packed clay 5 Boggy swampy	6 Packed snow 11 7 Loose snow 12 8 Concrete 13 9 Asphalt 14	Dry sod Wet sod Water Tundra Dirt	16 - Rock 17 - Ice 18 - Mud 19 - Sand A Other	
rincipal Impact Kinem	atics			
1 0-15 6 2 15-30 7 3 X 30-45 8 4 X 45-60 9 5 60-75 10	irrect or mark estimated range) 75-90 11 210 plus knots 90-120 A	6 Flight Path Angle 1 □ Up 2 ⊠ Down 3 ⊠ 0-5 4 □ 5-10 5 □ 10-15	7 🔲 20-25 🔋 🗛	nge) 60-90 Degrees Other
			·	
Pitch Attitude 1 Down 2 Up ADeg.	Nose Down Angle With Horizon	0 🗌 15 🔲 30 [45 60 75	B or Other
	90 1 75 60 45 30 15 1			
Roll Attitude At Impact (Ent	er direct or mark estimated range.)			1
	Aircraft Rolled Left		Aircraft Rolled Right	
Roll 1 🔲 Left 2 🗍 Right			• • •	B or Other
ADeg.	* * * * * * *	* * * Y	* * * *	
	90 🗌 75 🔲 60 🗌 45 🗌 30 🗌 15 🗌] 0 🗹 15 🗖 30 F] 45 🛛 60 🗋 75 🗂 90 🗂	

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	Natio	nal Tran	sportat	ion Saf	ety Boar			N	TSB Accide	nt/Incian	t Numbe	r	an and an
		FACT A	UAL I VIAT		RT					arci	L L	1/13	IE 1
upplement 1 Yaw Attitude at Impa	منبدهم		S. I. S.	C.C. Links					NIZIA	7 [C]	/- //	v P Š	
					, 								
1 Di Nose left 2 Di Nose right A		Aircraft Yawed Left			ed Left			Aircraft	Yawed Righ	I	-	or B Other	
		90 🗌 75 🗌 60 🗌 45 🔲 30 🛄 15			-								
1 Terrain Angle 1 X Level A Up de B Down C Other		11 Principal Impact Ground Scar Length 1 X None A feet B Other				i .	None Other	÷.,	nd Scar Dep	. 1.		otally Dest	
Cockpit Damage // Destroyed Collapsed X Part collaps 4 Distorted Fuseiage Split	5 [6 [ect 7 [<u>A</u> (Burnt Intact None Other		1 🗆 D 2 🖵 C		5 6 7 <u>19 Fus</u>	Burnt	apse (Es	2 🗌 C 3 🔲 F	estruyed Collapsed Part collaps Distorted	sed	5 🗍 Burn 6 💭 Intac 7 🔀 None A Other	t
1 🔀 No (Go to l 2 🔲 Longitudina		_				A	Horizonta		inches			i	nches
1 X No (Go to L 2 Congitudina 3 Circumferei A Othe:	il.	A	Other			в	Horizonta Vertical Other	al	inches inches	B	Horizonta	il inct	
2 Congitudina 3 Circumferei A Othei	il ntial					BC	Vertical	al	inches	B	Horizonta Vertical	inct	
2 Congitudina 3 Circumferei A Othei	il ntial ta	Type C	of Exit	в	1	B C Operable	Vertical Other	al	E Fire Damag	e F	Horizonta Vertical Other In 1	inct	nes nage
2 Congitudina 3 Circumferer A Oine: pproved Exit Da Exit Loc: .on	il Intial ta	ř Type c	of Exit	Other	1	B C C Operable	Vertical Other		E Fire Damag	A I B C C C	Horizonta Vertical Other	inct	nes nage
2 Congitudina 3 Circumferei A Otne. pproved Exit Da	il ntial ta	Type C	of Exit	1	1	B C Operable	Vertical Other	al	E Fire Damag	e F	Horizonta Vertical Other In In	inct	nes nage
2 Uongitudina 3 Circumferer A Otne. pproved Exit Da Exit Loc: .on 21 Cockpit-Left	il ntial ta	Type C	of Exit	Other	1	B C Operable	Vertical Other	1 Yes	E Fire Damag	e F	Horizonta Vertical Other In In	inct	nes nage
2 Cockpit Right	il ntial ta	Type C	of Exit	Other	1	B C Operable	Vertical Other	1 Yes	E Fire Damag	e F	Horizonta Vertical Other In In	inct	nes nage
2 Cockpit-Left 2 Cockpit Right 23 Long tube 21 Cockpit Right	il ntial ta	Type C	of Exit	Other	1	B C Operable	Vertical Other	1 Yes	E Fire Damag	e F	Horizonta Vertical Other In In	inct	nes nage
2 Cockpit-Left 22 Cockpit Fight 23 1L 24 1R	il ntial ta	Type C	of Exit	Other	1	B C Operable	Vertical Other	1 Yes	E Fire Damag	e F	Horizonta Vertical Other In In	inct	nes nage
2 ☐ Longitudina 3 ☐ Circumferer A Oine: pproved Exit Da Exit Loc: .on 21 Cockpit-Left 22 Cockpit Right 23 1L 24 1R 25 2L	il ntial ta	Type C	of Exit	Other	1	B C Operable	Vertical Other	1 Yes	E Fire Damag	e F	Horizonta Vertical Other In In	inct	nes nage
2 Cockpit-Left 23 1L 24 1R 25 2R 21 Cockpit-Left 22 Cockpit Right	il ntial ta	Type C	of Exit	Other	1	B C Operable	Vertical Other	1 Yes	E Fire Damag	e F	Horizonta Vertical Other In In	inct	nes nage
2 ☐ Longitudina 3 ☐ Circumferer A Otne. pproved Exit Da Exit Loc: .on 21 Cockpit-Left 22 Cockpit Right 23 1L 24 1R 25 2L 26 2R 27 3L	il Ita	Type C	of Exit	Other	1	B C Operable	Vertical Other	1 Yes	E Fire Damag	e F	Horizonta Vertical Other In In	inct	nes nage
2 ☐ Longitudina 3 ☐ Circumferer A Otne: pproved Exit Da Exit Loc: .on 21 Cockpit-Left 22 Cockpit Right 23 1L 24 1R 25 2L 26 2R 27 3L 28 3R	il Ita	Type C	of Exit	Other	1	B C Operable	Vertical Other	1 Yes	E Fire Damag	e F	Horizonta Vertical Other In In	inct	nes nage
2 Cockpit-Left 2 Cockpit-Left 22 Cockpit Right 23 1L 24 1R 25 2L 26 2R 27 3L 28 3R 29 4L	il Ita	Type C	of Exit	Other	1	B C Operable	Vertical Other	1 Yes	E Fire Damag	e F	Horizonta Vertical Other In In	inct	nes nage
2 □ Longitudina 3 □ Circumferer A Otne. pproved Exit Da Exit Loc: .on 21 Cockpit-Left 22 Cockpit Right 23 1L 24 1R 25 2L 26 2R 27 3L 28 3R 29 4L 30 4R	il Ita	Type C	of Exit	Other	1	B C Operable	Vertical Other	1 Yes	E Fire Damag	e F	Horizonta Vertical Other In In	inct	nes nage
2 □ Longitudina 3 □ Circumferer A Othe. Approved Exit Da Exit Loc: .on 21 Cockpit-Left 22 Cockpit Right 23 1L 24 1R 25 2L 26 2R 27 3L 28 3R 29 4L 30 4R 31 5L	il Ita	Type C	of Exit	Other	1	B C Operable	Vertical Other	1 Yes	E Fire Damag	e F	Horizonta Vertical Other In In	inct	nes nage

	Y		
	National Transportation Safety E FACTUAL REPORT AVIATION	Board .	SB Accident/Incident Number $2 \mathcal{F}' \mathcal{A} 9 0 \mathcal{F} \mathcal{A} 1$
Supplement Crash Site Pla	I-Crash Kinematics (continued)		
36 Sketch of Cra	sh Site — Show distribution of major components, fire	area, obstacles strikck, occupants	s, and magnetic norm. Sketch is "TVO
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National 7	Fransportation Sa	fety Board	I	NTSB	Accident/Inc	icent Number		
FA	CTUAL REPO AVIATION	DRT			•			
				$n_1 2$	A191	OFAI	35	
Supplement M—Search/A	escue/Firelightin	g/Medical	Treatment.	and the second second second				
Search and Rescue 1	None Conducted (Go t	o block 16)				San da se angele an		
2 Type of Search Conducted (Multip 1 ☑ Air 3 □ S 2 □ Ground 4 □ !r	ea			ncy Notified	r M. D. Y.			
A Other	r		C Other					
5 Aircraft/Occupants Located A <u>5-28-90</u> (Nos. for M. D. Y) B <u>Q&26</u> Local time C Other	· · · _ ·	No	ved in Search		Military or Co 1 🗌 Yes 2 🔀 No A Other	डी Guard Person	DEI 114034	
 9 Distress Call Transmitted (Multiple entry) 1 None transmitted 2 2 Prior to accident 	10 Cistress Call Recei (Multiple entry) 1 None receive 2 Prior to accid	ed	11 Method o 1 [] EL1 2 [] HF 3 [] VH	radio		1ultiple entry) 7 D Vesual sign schoke/fire		
3 After impact accident A Other	3 🔽 After impact A Other		4 UHF radio 5 X Visual sighting of wrecka 6 X Visual sighting of occupa			3 SAR satellite ge 9 2 ATC computer inner		
2 Condition of Aircraft Occupants a 1 A Located alive 2 Located deceased 3 Located alive-died later 4 Died awaiting rescue 5 Located alive-trapped	6 Able to 7 Left sce 8 Left sce	assist with loc ene-successfu ene-unsucces			· · [1	Weather Condition Most Servere Ter Chill Condition De A Terricherature B Wind chill fact C Other	nseiture preg Sea 74	
	onducted (Go to block							
		140 Election	1 I I - I 8 -		a limite Acais	ted 21 Sim Fich		
17 Firedghting Unit Notified 18 Fl (Nos. for M, D, Y) A A	rat Firefighting Unit rrived Local time	19 Firefight Respond (Multiple 1 Aı	ding e entry)	20 Firefightin Evacuatio 1 2 Yes 2 1 No	n	A Other		
17 Firedghting Unit Notified 18 Fl (Nos. for M, D, Y) A A	rrived	Respond (Multiple 1 🔲 Ai	ding e entry) rport unicipal ilitary	Evacuatio	n	· · · ·	3905778d	
17 Fire.ighting Unit Notified (Nos. Ior M. D. Y) 18 Fi A	Local time	Response (Multiple 1 Ai 2 M 3 M	ding e entry) rport unicipal ilitary	2 . No	n	· · · ·		
17 Fire.ighting Unit Notified (Nos. Ior M. D. Y) 18 Fi A	Local time	Response (Multiple 1 Ai 2 M 3 M	ding e entry) irport unicipal ilitary r	2 . No	n	A Other		
17 Fire.ighting Unit Notified (Nos. Ior M, D, Y) 18 Fi A	Local time	Respond (Multiple 1 al 2 M 3 M A Other	ding e entry) rrport unicipal ilitary r	Evacuatio	n	C Useo		
17 Fire lighting Unit Notified (Nos. Ior M. D. Y) 18 FI A	Local time	Respond (Multiple 1 al 2 M 3 M A Other	ding e entry) rrport unicipal ilitary r	Evacuatio	n	C Useo		
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National Transportation Safe	y Board		NTSB Accident/Incider	nt Numiber
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Supplement M—Search/Rescue/Firelighting/	Medical T	reatment (co	a short the second of the seco	
Search and Rescue, Firefighting, Medical Treat	nent			-
31 Medical Units Responding (Doctor or paramedic) (Multiple 1	entry)		scue, Firefighting, Medical T	reatment Units H
 2 Airport 3 A Municipal 4 Other paramedics 		2 🔀 No A Other	mpiele (lems 35-31)	
5 🔲 Military A Other				
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Search and Rescue. Firefighting: Medical Treatment Units Hampered by	1 Searc	h and Rescue	2 Firefighting	3 Medical 1
36 Weather/Visibility			- -	
37 Terrain				
38 Darkness		<u> </u>	- 10 - 110-110 - 1 -	
39 Unfamiliarity With Terrain				
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NTSB Form 6120.4 Supplement M (1-84)

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Other Occupants	в	с			F		1	H Degree o	f Injury
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sacribe What Occurred In Chron rrain And Include A Sketch Of W Departure, Intended Destination	/reckage Distribution If Pe	umstances Leading rtinent. Attach Entral	To The Accident And Sheets If More Space Is	The Nature Of The Accid Needed State Point Of De	ent Describe The parture, Time
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06-08-90 Signature Of Person Filing Rep					
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TEL NO:904-239-6459 -29-190 1:54 ID:EPAU ¤9⊒1 P08 PILOT ADVISORY DATE OF FULL NAVE Michael John Rapiano BIRTH 09/07/66 ADDRESS PO Box 10167 Daytona Beach, FL 32120 TELEPHONE 904-767-0734 PILOT CENTIFICATE Commercial, Flight Instructor PHOT JERT. 136-524361 MITINGS ASMEL-IA (Both) BIENNIAL FLIGHT REVIEW DATE 9/20/89 (Rating) DATE & CLASS OF LAST MEDICAL 02/23/90 (200) EXAMINER COOK (1-7585-1) OCCUPATION Flight Instructor PAST 90 150 (200rox) TOTAL FLIGHT TIME 1500 TOTAL PIC/OR SOLD 1300 DAYS IN EREST IN AVIATION, SUCH AS BUSINESS, PLEASURE, COMMERCIAL Commercial TYPE OF AIRCRAFT FLOWN 172, 172RG, MZOJ, PA28 1404hvough 181, PA44-180, CT303, AC680 IF YOU ARE A STUDENT PILOT, GIVE INSTRUCTOR'S NAME AND ADDRESS

AIRCRAFT DIFORMATION

HODEL Cruseder 303 IN THIS MAKE/HODEL 120 hrs
HODEL Crusader 303 DI THIS MAKE/MODEL 120 hrs
OFTNER'S NAME Embry - Riddle Acrowaut University Inc
OWNER'S ADDAFF35 Regional Airport, Daytona Beach, 32014
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REMARKS PERTAINING TO OUCURRENCE (Continue on reverse) See attached written report

Muchael & Romer 05/29/90 SIGNATURE DATE

Michael J. Kepueno

On May 28th at approx. 6:55 and walked through the gate arts the name to do a extra-training flight in the Cripacies. On this flight we were to do a NDB geprach and a single-engine opprach. as I gprasch the averaft of dis my preflight. I cliented into the cockpit and my strilert and I briefed about the flight. We were going to do holding at the NDB followed by a I.S. when listening to atis we learned of a westerly operation and decided to hold at Deland followed fy the NDB 30 DED them rade vectors to the back course. Before enting the cochpit & noticed some problem with the precuois flight (power loss) and checked the MEL itens (heater vert far inop - ARC a runber 2 new niop). We had no trouble during engine start and werything was normal. The student completed the instrument cockpit chuck and called deamance, after cleanance readback (which I did) are ranged out and tapies to end of the ramp. Colled ground control and proceeded to November six intersection. During the taxe the right alternator hight care on we noted alternator output on the left was below 30 aups. Doning the engine cyster and atternation check the light istunquished itself. after He unup we turned to face traffic and called for our take off clearance. We were cleared for takeoff and during the coll of noted inquite indications being normal. We proceeded with an 44

carly turn to avoid wake furbulence from a departed Caster DC-9, contacted dipartine and we were then getting vectors to the Deland radio beacon. During this time the regist alternator light illiminoted and we followed the checklist no breakers were popped at the trie and to oling procedure ceres to tan off the alternator. 'e thin placed the volt-annater to the lift alternation and output was normal- we continued. We extered holding at Deland and held for approx. four circuts we asked for the NDB 30 ad we were cleared for the approch. My student ashed if this would be single - enquire, I replied to doe to the chance of entering IMC. at the MDA the studied brought up the thottles when we got the fire wanning light on the ught ungine - I innediately took control of the aucraft and executed the missed apprach. at an altitude of approximately 700 also we started to shot down the right enquire per the checklist. This is also when I noticed the low vottage light of felt that the engine fire was the most antical energoncy after executing the securing checklist the airciaft woold not chink - verify that I die in fact clean up the ancieft , I checked my procedures. During this time I also made a mayday call on 1228 MHz, then on 125.35 MHz and finally on 121.5 MHZ. Donig this time the student was told to squak 1700 on the transponder. Durig the tie of the go-arand

I was at the controls of the averaft and the studiet was on the checklist - We intered the clouds at approximately 600 MISL at this time it was primarily flying the aircraft and also telling the sticked to look outside for curval information as well as procedures that I wanted him to complete. Such as pulling the energony crossfeed and tighting his safely belt. at approximately 200'he spotted trees it was not intil about 170-150'we started to get below the fig. We saw a fide to our left and it was not possible to make. Up ahead we were heading for power lines, then we saw a field and went for it. The studiet ask if I wailed the gear down and I said no" at this time the aircraft was starting to role because of our angreed and I then retarded the throttle (gooding) and then inpact as we shiddled across the field the aircraft turned up side down I felt the window break and the dist in my face. as we stopped I noticed that there was a fire Darwin asted if I was ok, I saw, yes" and he said he was ok. We crawled of of the aircraft and rap towards the canter of the field worried about a fire. approve ten ninisted later we heard the sheriffs helicoptu ---

after the crash of also noted that the fire warning how was still blaring and that the fire lights including low voltage and right alt off were

Ilominated throughout the incident. When I make my radio calls I used the # 2 comm and the last frequency put in was 121.5 not sure which preset that was, I used # 2 because it was closest to my hand while I was flying the A/C.

Statement Given by Michael J. Rapuano CFI, Multiengine & Instrument Airplane

About 0655 was when I went through the gate on the ramp maybe a little bit after 0700. It was around that time frame. Walked out to the aircraft and started my pre-flight of the airplane, checking all you usually cover on a preflight, fuel, oil, gear, the control surfaces and the overall view of the airplane to make sure its untied, unchocked. Then when I crawled through the cabin I could only get the discrepancy board since it was right there saw that over the weekend it had a problem with the power of the aircraft, powerloss on both engines. Made note of it, crawled in the airplane, noted the MEL item of the aircraft, none of them having to do with situation that was going to occur. We, uh, me and my student were talking about the activities we had to go through that day, with some extra training flights on a failed pre-prog and there was one incomplete item that was done and that was an NDB approach and he failed the single-engine landing. Told the student what we would probably do is go out to, uh, we were talking about going to New Smyrna and hold and that was when I called Operations to find out if they still had that car show or the airplane show down in New Smyrna, we decided to go to DeLand and that was also when we found that they wera using a westerly operation. When we arrived at the airport there was some fog over the west side of the field but that was moving west, ATIS was calling it to be about a mile visibility out toward the west but it was clearing up in Daytona so we were just going __at Deland and to do an NDB approach both engines then come back and do the Do that single back course. engine. We finished our briefing, started the aircraft up, that was when we called clearance the way we get our clearance for Ξ local IFR flight. Everything was OK, we ramped out, called ground control. As we were taxiing out the right open airlight came on. Checked the voltage, checked the power of the left alternator and found out that they were both were indicating less than 30 amps total and according to the Cessna manual that was normal and during the runup that cleared up. When we checked the alternators they still weren't paralleling correctly, again a good parallel in both the alternators but we were still less than 30 amps total amperage on the system the right alternator light came on. During the check Proceeded normal takeoff, contacted departure it went out. control, we were going to radar vectors to the NDB and then finally got cleared for holding. During that time the right alternator went out again the open air light came on. Checked the amperage on both alternators and decided to shut down the right alternator and just work off the left cause the left was still under 30 amps. At that time no breakers were popped an average run. Went through the procedures with that, I think it was the second or third paragraph or section down on the checklist in the airplane. We continued to fly, went through about four circuits in the holding pattern got clearance for the approach, started our descent out cf 3,000 down to 2,000, leveled off, did the procedure turn, coming back inbound started descent. Started to level off at the NBA to get above it at around 600 feet. As the student brought the power up the at the same what happened is the right fire warning light came on. Immediately I took control of the aircraft and executed a missed approach going through my procedures making sure bringing the flaps and gear up and to make sure that the propellers and the

forward going around. At that time also the low voltage light came on. Somehow we had the right alternator off low voltage and the engine fire light. I felt that the most critical emergency at that time was the engine fire. So we went to the right engine we went to the fire checklist and went through our procedures then we went to securing the engine checklist and secured the right engine. At this time went to BYSE, the airplane was not climbing. Due to the fact that we were not climbing and we were just at the top of the cloud layer 600 feet top to bottom and it was solid fog. I also gave a PIREP to the university that it was solid dense fog west of I-95 and that was earlier in the flight. As we secured the engine we did not get a climb out of the aircraft, wondering why we weren't getting climb out of the aircraft due to the only two passengers on board and we only had roughly 700 lbs. of fuel on board, but the aircraft was not climbing. I went through and I checked the gear handle and the flap handle to make sure that the gear and flaps were up. At this time we entered the soup, we entered the clouds. I was flying the aircraft now, the student was We, at that time made two mayday calls, doing the checklist. mayday, mayday, mayday with the aircraft number 4973 Victor on the Unicom hoping that somebody would hear us or find us nearby. Could not receive anybody. Immediately went to 2535, made two calls on that frequency. Did not hear anybody. I checked the volume and the switch position and they were both normal. Went to 21.5 and made two calls on that using the same calls as previous. I told the student to squawk 7700 on the transponder. At this time I was still flying the aircraft and we were coming up about 200 feet. About 200 feet I had to start the student start looking outside. He started to see trees just below us but we could still not see ahead of us. I had the student tighten his restraints and pull the We broke out of the clouds emergency crossfeed shutoff. approximately 170-150 feet, noticed a field to our left. We were too low to make that field, to make a turn towards that field. Then looking in front of us we saw powerlines and we knew we weren't going to clear the powerlines. I say about another, I'd say about a quarter mile ahead of us was when we noticed the field was directly in front of us. The student asked if I wanted the gear down and I said no. I did not know what type of field it was, all I know is it was a field. I had to turn to avoid the tree. During that turn the aircraft started to roll so I had to retaid the throttle on the good engine to go ahead and maintain and control the aircraft. It was at that time we didn't have enough time. I remember pulling the throttles back. This time I couldn't remember if I pulled the mixtures back nor I can't remember if I put the flaps down also. I had my hands on the throttles and I had my hands, my left hand on the throttles and my right hand on the yoke during the impact. I remember the impact hitting. We slid I couldn't tell you how long. The aircraft flipped over on its back and the cabin windshield broke allowing dirt and glass to fly

into the cockpit. We came to a rest in the field and the student asked me if I was OK. I replied yes I was OK and I asked him if he was OK and he said he was OK. AT this time I saw fire. Could not determine which side of the aircraft the fire was on, but I remember seeing the fire. Told the student that we were going to have to get out of the airplane. I unlatched my seatbelt, got out of the aircraft. I remember using the instrument panel to help me push myself through. Crawled on the cabin. The student was Lue to the fire we did not grab the first aid following me. equipment. Felt that we should just leave the aircraft. Opened the cabin door, crawled out, and went to the center of the field away from the aircraft. At that time we sat down, we talked about we were both OK, we both walked out of the airplane. My neck was starting to stiffen up on me and I had cuts to my head. They were bleeding, so I took off my shirt and used that as a bandage around About ten minutes later we noticed the sheriff's my head. department helicopter was circling to the south of us about a half mile and then immediately either they saw or they directed in on our ELT is what they said. They came right to the field and found us and that is where the incident's at.

Can you back up a little bit, Mike? Before you started to have the problems, what all did you have on, landing lights, perhaps?

Negative, what we had on were our strobes and beacons. That was it. As far as external lighting the rest there was just the did not have radar on, no need for that because of the weather that day. Both number one radios and number one radio and number one com, I mean nav, number two radio and number two nav and transponder.

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I went, my first thing was to do was to get full power in the engine so I went to the mixtures, props, and throttles and I brought the flaps and gear up at that time.

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My hand was blocking the red transition light. Could not remember if I the gear or if I felt the gear hit the wells. But I remember when I did bring the handle up and I raised my hand did not see unsafe indication nor the green lights.

?

Technically.

Flown the aircraft before ... that low...?

Crusader 5 has been written up including myself on numercus occasions about the right alternator going out. I flew it twice before with similar situations and had not had this happen.

? ...had a situation like that before where... similar .. aircraft perform ..?

When working out the performance we should have gotten around 170 in a climb. Other situations where we weren't able to climb was when we had three students on board and full fuel. Or trying to say meaning myself and two students with full fuel and aircraft did have airconditioning which depletes fuel.

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No, no, going through the checklists there was nothing in there of a fire checklist or securing of a ...

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FRAU MAY-29-190 200 10:EPAU TEL NC: 204-239-6455 #941 FO1 PILOT ADVISORY LATE OF BIRTH 10-24-61 FULL NAME Darwin Folkerts ADDRESS 601 Bill France Blud. Apt#405 Daytona Reach FL 32114 TELEPHONE 316-697-4755 PILOT CERTIFICATE 512-78-3201 Private PILOT CERT. #5/2-78-3201 RATINGS Airplane Single of Multi engine han BIENNIAL FLIGHT REVIEW DATE CONSTRUCTOR 8-21-89 DATE & CLASS OF LAST MEDICALH-24-90 1- Class EXAMINER Dr Langford OCCUPATION Student PAST 90 TOTAL FLIGHT TIME 183. 3 _____ TOTAL PIC/OR SOLO____ DAYS IN EREST IN AVIATION, SUCH AS BUSINESS, PLEASURE, COMMERCIAL Student TYPE OF AIRCRAFT FLOWN T-303 Crusader IF YOU ARE A STUDENT PILOT, GIVE INSTRUCTOR'S NAME AND ADDRESS _____

AIRCRAFT INFORMATION

IDENTIFICATION NUMBER Nº 4973 V MAKE <u>Cessna T-303 crusader</u> MODEL <u>T-303 Crusader</u> DI THIS MAKE/HODEL <u>SOhrs</u> OWNER'S NAME <u>Embry Riddle Aeronaut University Inc</u> CWNER'S, ADDRESS <u>Regional Airport Daytona Beach, FL 32014</u>

REMARKS PERTAINING TO OUCURRENCE (Continue on reverse) SQ attached written report

DATE 5-29-90 SIGNATURE Danie L. Followts

Statement of Darwin Folkerto may 28, 1990 Q 1030 I anued at ERALL at 6:10 am May 28, 1996 for a FA 315 Dual. The Dual wasan X-18 and "I checked the Wx for the local and surveyinding les Conditions. The flight was able to be made with the current lix conditions. I checked in at the Julight desk, and chase C-5 because all deavorin while working. I went out to the aupland riddid a complete predictenspection. I found nothing was with the amplore, and I was ready to go at 7:00a My Instructor come out to all andone and he chicke the ail and feel repping. I wonted to ge to New Sminn Black, but due to the fly in there, d. was decided that it would be better to go to Dehand instead. I called partona Chorarce and received a local JFR Olaronce to Jaco at Deland, to be followed by the NDB 30 correct. our original demande toos Cleared to the Delard amport, usa reactors, then after departure turn left to a hearing of 230, and climb and maintain 2000 mst, and departure fuquere was 123, 35 and squark 02114 I believe. I then call riddle flight date and ramped ant devil. ounder for was ago, I then taken to force I and called herfor ground. Grand control cleand us to runway 25P. I then taked to running 25 R. I stopped at Abushia. to complete my ground run up. When I was doing my engre unich, I noticed the Right atternation ling was on, but the amp meter showed that the right atternation was working properly, B. Cfore toberff The Right alternation wat wint and.

We were given cleanance by Waltone Tours for lapedity - We were tald to term To a hading of DO after departure. The topic/ 100 mounal in Too we town of a a loan of 210° and town too usto Connet Dayton deportane an 125,35. We contacted baijana deportante. All instruction dese called ERAU flight operations to Call in a PIREP. While dimining the 2000, Baisona Neparture Tala ins To proceed direct to Pickand NPB when alike and to continue and climberto 300 MS. Our house instructions were ita hald with east on this He Cours to with left turns and TO EFC at 1000 Z. We porceded direct to the debond MDE. The Court for hording would be aduct with I reached the deland NDB and turned to my authourd heading of 136°. We did several laps of tolding. The WX conditions were great at an attetude, We then decaded to do the approach ento deland. We were cloned for the approach, and told to descend down to Dood lefore timing to an authord track on the appres we crossed the NDB at 2000'and tured to ain authound heading. We continued the outhand leg for 3 minutes and we did a 90°, 270° procedure tur. I stanted the descent and lowered the londing gon. We descended down to our MDA of 600: While in holding we saw the East light come on and my instructor tak me to ture the right atterne off. we monitored to left alternation on the comp 116 moter.

I leveled off at the MDA at 600'. We were right on tap of the fag larger. We had not reached the missed approach point with, when the Night engine fire light come and the handand on My instructor toob control of the cuplane HE exacted the engine and procedures I tost the emergency check list and and statist lices g. it to my instruction Pight and it like was at idiaprojection was feathing the methic was at idle atop- the court find was closed. The nort alterateunsput to alf. The right maga were light to off. the night fuel selecter was placed in off. with all affithis dord, we tould not clerk, in the during departure Contration 125, 35, he called from daytora. He then could an 1215 and the atte Emergrey. All maning it. My Instructor told mite Aut the tron spinder or 1700 which I did. We went still going down, so my instructor turned to the east so that we could get away from dolord. We slawly descended into the Loop of clouds, noticed a field but we were passed it and content made it to that field. we wert loop into the Clouds, and my instructor told me to try and Tot the right engine. I turned the master and fuel on mitude full ich and unfeithed IG prop-I Att 26 Staten Cuition but it underthan de angé. at this poil, my historican sour a free and der Tr- lad in-160 and

De litelle grand in a linge livel, sol high allitude. When we lit, the sing flips and adjonde on its Top - We wire lot alle to a Contempsion and sun ponder These Line small finger the left side of the amplane. We Lest gal and a decirci from the plane. We writte in the field with the sharps through separted up. Mit water we seen midentertan and taking the respective Inde and lance I was alight and rode with my instructor to the poter. I then rolled FRAU to Celetion And what we ping on.



Daytona Beach, FL 32114 904-239-6000, FAX 904-239-6459, Telex: 258052 SAND UR

STATEMENT

05-28-90

When I arrived at the site with A.C. Tacker, the aircraft had been righted by the Fire Department, the windshield was broken out and the circuit breaker panel and electrical switch panel were intact and clearly visible. The following conditions were noted:

- 1. Both left and right isolation circuit breakers were in the tripped/pulled (off) position.
- The buss tie circuit breaker was in the tripped/pulled (off) position.
- 3. The landing gear selector handle was in the up position.
- 4. The landing gear was down.
- 5. The right prop was feathered.

The following electrical system characteristics are presented for your consideration.

The system is a "split" type design meaning there are two independent main busses powered by the left engine alternator and the right engine alternator. The elec trical loads are divided up equally between the two busses. Normally, both busses are tied together in parallel via the left and right isolation breakers and the battery buss and via the buss tie breaker. This arrangement allows for load and source sharing from one side to the other. Based on the position of the circuit breakers, if the right alternator had failed, there would be no electrical power on the right main buss. When the engine is feathered, it can not rotate the alternator, thus the result would be a failed alternator.

With no power on the right main buss, any electrical item attached to that buss would be useless due to lack of power. Some key items that would be without power are:

- Right fire detection system no power gives a fire alarm; power is necessary to keep the alarm in silence.
- 2. Landing gear pump relay and power. No power will not allow the gear to be retracted. It will go down due to gravity but not up.
- Right starter motor contactor no power. Will not be able to energize the contactor, thus unable to get power to the starter motor.

In conclusion, based on the above, with the right engine feathered and the gear down, there was no way the gear could be retracted and/or the engine started without resetting those circuit breakers.

(SIN Lee. G.J. Nelli

Flight Standards

World Leader In Aviation Higher Education

Embry-Riddle Aeronautical University Daytona Beach Campus Flight Technology Department

HAZARD & MISHAP WORKSHEET

CASE # Date 5/27 Time (LCL) l(d)1) Location Local Practice Area - Sector 2) Aircraft Type and "N" Number T303 N49731 3) Solo/PIC, Dual) Observer (Circle One); Name TOHN KIRK 4) Stdnt. #/Ph. #/Box # 157-56-4073, 766-9427 "Course 3152 Student's Instructor WENDY L. RICHTER 1.p. 1_097 Describe Hazard or What Happened 5) experienced Aua Phsine NPSS Pins 1Dante d to 191151 (In O Straight ina 12 1a in d tà panad want -45 NS 1 Ne drela tive Checcost a r Q 112 out 1a incide (Use Additional Paper if Necessary) None noted 6) Brief Description of Aircraft Damage Reported by FOR MAINTENANCE USE 7) Describe Mechanical Defect/Maintenance Factor/Damage 8) Describe Action Taken Signature Yellow Copy Forward to Safety Engineer Come Formand to Chad.

May 27, 1990

Dear Mr. Joe Poole:

On May 26th, 1990, I encountered engine problems in N4973V. I was on a FA315 training flight with my student John Kirk practicing traffic pattern operations and VMC demonstration. After approximately 1.5 hours of flight, difficulties occurred.

We were at 3500 feet, 2.5 miles east of Flagler County Airport when we noticed poor aircraft performance in the clean configuration. I increased power to 24"MP and checked the position of the gear and flaps. The aircraft was still performing poorly with no increase in airspeed above 120 kts. It felt as if heavy drag was acting upon the aircraft. All engine instruments were operating normally, with no other discrepencies found. At that time, I increased the mixtures, propellers, and throttles to full power and both engines abruptly wanted to quit. I immediately turned directly to Flagler airport and told my student to turn the auxillary fuel pumps on the high position. Both engines responded favorably and aircraft performance increased with minor misfiring. We contacted Daytona Approach and requested runway 16 at Daytona Beach Regional Airport, and advised them we were experiencing dual engine roughness. Daytona Approach asked if we wanted to declare an emergency, and we replied negative. I landed the aircraft at DAB without incident.

Due to high altitude, I was permitted to land at either Daytona, Ormond, or Flagler at all times. This condition, and the availability of emergency equipment made Daytona Beach Airport a more favorable choice for landing.

I hope this letter will help answer any questions you may have concerning the engine problems I encountered in N4973V on Saturday afternoon, May 26, 1990.

Sincerely,

Windy & Reacher

Wendy L. Richter

Embry-Kiddle Aeronautical University Daytonal Beach Campus Flight Technology Department HAZARD & MISHAP WORKSHEET CASE 9/30/89 Time (LCL) /300 1) Date -2) Location Flader Aircraft Type and "N" Number 49 3) 4) Solo/PIC, Dual, Observer (Circle One); Name noss Stdnt.#/Ph.#/Box # Ccurse Student's Instructor I.P. 4 5). Describe Hazard or What Happened Hn. alternator friled Stalls wh :12 doina Flag arou 70 25 GD 400 Warning abou fice G n ennine. amp C Cicc P Gr +4, 6, 20 MARGANCY G م. ان ا nsn SUA lation reaker So Nere right terna 10002 H. 2 C He 10-9-85 (Use Additional Paper if Necessary) Brief Description of Aircraft Damage_ 6) Reported by FOR MAINTENANCE USE 7) Describe Mechanical Defect/Maintenance Factor/Damage ne AD. 8) SIAN eL ce o Describe Action Taken 12 Signature CXS D'

Embry-Kiddle Aeronau Cal University Daytona Beach Campus Flight Technology Department HAZARD & MISHAF WORKSHEET CASE # Date 8/19/16 Time (LCL) 1300 1) Location 10 AIL SW A DAD, 1600 FT / TELAWATED AT 12/201 2) Aircraft Type and "N" Number C T303 / 5529V 3) Solo/PIC, (Dual, Observer (Circle One); Name 4) Course 207-1 Stdnt.#/Ph.#/Box # Student's Instructor MIREN I.P. # 075 Describe Hazard or What Happened The Flight Degate Dar VFIZ 200 proceed 5) TOWAR ? the "320" Deposition - 2 pareximately Emiles ^ INTO THE Flight the LEFT ENGINE FIRE WARNING Light Iluminated th Gina here Similed . The Left evenue was Suna FILL IN Flight " chuck list. The Trans priver FACIEG 500 500 -10 Denzature control was Lov.sc the Deland UNICON Was centacted ~10 IZ DEFETTEMENT As the Final CENTRET F121. Larding alune uns (Tarran) winting lights D.D Net .1k ~ ac-105 Ninon 5.7 211.20 TOWER The ThroThes LATTE , mi HOLN TO Sourd 1 (Jow) DinsT RESISTING TO Pumpinia Tix year Inavil IT There was preserve in the Sister a Do to The Sin, LUENINCE Encine Fizz 1-25 Lavied Despit- The yest 3/0 vour Incedent (housing Fize un juit 1222 ~ 1 1 112 (Use Additional Paper if Necesser) 511 Santure) and Exacuted 6) Brief Description of Aircraft Damage 1 por the draw The Bache Reported by FOR MAINTENANCE USE Describe Mechanical Defect/Maintenance Factor/Damage 7) aller 8) | Describe Action Taken Signature Yellow Copy Forward to Safety Engineer White Copy Forward to Chairman

Addle Aeronautical University fight Technology Department INVESTIGATION Summary of Investigation: Incident occurred on described. 1) no endence of fire a geor malfunction. Investigation uduly promil Subsequent duplicition an. electrical mallisty Vation clearly identifier 1011 C.1 ittime replacing lung tie preshere and battones Ter 2) Recommended Action: no additional action Investigator (Please Print) Date 3) Action Taken if Appropriate . Signature of Action Agency Distribution: Student's File () Flight Standards () Instructor's File () Other (Secretary's Initial Yellow Copy Forward to Safety Engineer White Copy Forward to Chairman

- ____ הבקרבדים הרשת F.CS - DESCHADINGTRIET 2 TO FPON n ku=08+1990 j 10:02 HAGE _____ OF _____ PA **VOLUSIA COUNTY SHERIFF'S OFFICE** NAL REPORT YES X NO 3 SUPPLEMENTARY REPORT 2 :ONE Incident Type incident Number ;tim Gr. 6093 EN scie. A Narrative T Storen/Recovered Property Listing TT Investigative Leads opientient E Other __ Evidence/Property Release 📃 Vehicle Release ilains Narrative - Descriptions - Circumstances Sizien -____ Code V FI Ersboy 32014 U Michae ZIIC Jethr F Duch . 4361 SSAN 090766 136 52 Eng. and CEIL IILMET 022390 c Bill Frince Blud #405 601 EL Žu C. 241 SSAN 512-78-3201 M Uulti Engine ۲f 2490 6 Losogtan ንግ le soa . <u>le</u> (en \mathbf{C} ERAIL CI ERAU the TER 0.720 6 0826 1600 FA9 <u>~___</u> (YZ)nei <u>G</u> hesp EVA 8200 inve 0.1.2.1 LDC1 Deintin and Se damage とつ NGIC T Entry Date: By/ Y: Clased <u>,</u> NCIC _ Cancelled Date: By/ Date 05/29/20 13 + 2C93 orting Officer Υ, roved by.

757055765860 FROM 19050 NW DISTRICT 2 ΤŪ =.03 101-08-1990 - 10: 🖨 PAGE _3 <u>34</u> 2: **VOLUSIA COUNTY SHERIFF'S OFFICE** NAL REPORT YES I NO I SUPPLEMENTARY REPORT ZONE incident Type ictim Incident Number 90 B¹7 6093 E. Staten/Recovered Property Listing II Investigative Leads 🗴 Narrative upplement 🕽 Svidence/Property Release 🛛 🔲 Vehicle Release C Other intains Narrative - Descriptions - Circumstances Stolen. Code Repres 1 Quadra Tear ۲ 24 2002: De eather CEMA 155.80 125.35 1.000.00 Ħ Ex Nase (rear Flass ncns Bt Frathered Eng. Γ¢ erma 270 uies 4 iasti 40 Scen 99.4 20 Ê 0 09 - + Fegi Flie (RE Fc いフ DVAV (Un 50 797 Se. 229 20 Ro lecated 1 DTCS ÷.
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 By/ X Closed By/ 3 1.0. * 2093 Date: 5/29/90 sporting Officer: 2 10.4: 2055 5130/90 proved by Date.

<u>readerraan</u> FROM. VOSO NU DISTRICT 2 ΤŪ TUN-08-1990 10:39 ⇒.n.: PAGE _____ DF ____ 740 **VOLUSIA COUNTY SHERIFF'S OFFICE** SINAL REPORT YES TO NO 23 SUPPLEMENTARY REPORT ZONE _ Incident Type Incident Number içtim 90 6023 Embr cide Stolen/Recovered Property Listing _ Investigative Leads 🖡 Narrative ipplement Evidence/Property Release 🔟 Vehicle Release D Other: ontains Narrative - Descriptions - Circumstances Stolen Code 200 ppre EDD Fal F d' Fe Fal A 0 JAU over Ċ LY.CC DIC I NCIC I Entry Date: ₿y/ Cases: 🖿 Not Assigned \equiv Inactive \equiv Exc.Cleared 🛪 Closed Status: \square Continued \square Unfounded \square Cleared/Arrest By/ C 10. #: 2093 Date: 05/29 90 Reporting Officer: _C Date: 05:30/90_ 10 #: _ 2035 Approved by:

822 5703 1 153 F.01 10H-08-1990 10: 🗭 FROM ΤQ асвони: БНЕТРИСТ В PAGE ______ OF __ 227 **VOLUSIA COUNTY SHERIFF'S OFFICE** INAL REPORT YES 2 NO I 23 SUPPLEMENTARY REPORT ZONE . 1.ctim lident Type Incident Number K:79 90 En С (ng)1 ---😧 Narrative 🛛 🚍 Stolen/Recovered Property Listing E investigative Leads upplement ⊇ Other 🗇 Evidence/Property Release 🛛 🙄 Vehicle Release onteins Code Narrative - Descriptions - Circumstances Resteres Stolen FAA Tuspect Fellow FA T SB the Copies pockage C TPDC b 32114 -William A. Martin Dean College of Aviation Technology EMBRY-RIDDLE AFRONAUTICAL UNIVERSITY Daytona Beach, FL 32014 Telephone 904/239-6821 Telex: 258052 SAND UR NCIC Cases: \Box Not Assigned \Box inactive Entry Date By/ □ Ext.Cleared K Closed = Ξ Status: C Continued C Untounced C Cleared/Arrest NCIC Cancelled Date: 8y/ 10, 2093 Date: DS 190 porting Officer 129 Date 05/30/90 10 + 2035 10ú proved by

AIRCRAFT FLIGHT/DISCREPANCY RECORD Parking Spot#: 5/17/90 Number of Landings Dater DAB OMN Fuel Load: 1/2 FLG NSB Time Dut: 0740 7 -SFB DED Tach Time: 1259.0 Time In: 0900 DTHER TOTAL DISCREPANCY: AIRCRAFT STATUS OK MAY 17 600 NUMBER IP/STUDENT NAMI CORRECTIVE ACTION: MAINTENANCE SUPERVISOR7MAINTENANCE COORDINATOR/MECHANIC Date: 5/.7 Parking Spot#: 4 4 Number of Landings DAB OMN Fuel Load: -/c/ Time Dut: 7, 4 CU FLG NSB DED SFB Tach Time: 17.60.2 Time In: /, aD OTHER TOTAL AIRCRAFT DISCREPANCY: STATUS ΞĬ MAY 17 RECD 104.024 NUMBER. IP/STUDENT CORRECTIVE ACTION: MAINTENANCE SUPERVISOP/MAINTENANCE COORDINATOR/MECHANIC

AIRCRAFT FLIGHT/DISCREPANCY RECORD 43 Number of Landings Parking Spot#: Date: DAB OMN 500 Fuel Load: FLG NSB Time Out: 1130 DED SFB 1230 161.0 Tach Time: Time In: TOTAL OTHER DISCREPANCY: AIRCRAFT STATUS DIK 8 gg in Cu-NAME IP NUMBER IP/STUDENT £ CORRECTIVE ACTION: 12631 SUPERVISOR/MAINTENANCE COORDINATOR/MECHANIC MAINTENANCE Parking Spot#: Number of Landings Dates OMN DAZ Fuel Load: FLG NSB Time Out: DED SFB 1262. 1:23 Tach Time: Time In: OTHER TOTAL AIRCRAFT DISCREPANCY: STATUS MAY 17 100 MARK STUDENT NAME & NUMBER CORRECTIVE ACTION: MAINTENANCE SUPERVISOR/MAINTENANCE COORDINATOR/MECHANIC 1-3

AIRCRAFT FLIGHT/DISCREPANCY RECORD Number Parking Spot#: Landinds Date: 5117190 of DAB OMN Fuel Load: FLG NSB 1500 2 Time Out: SFB DED 1640 Tach Times 1263.1 Time In: OTHER TOTAL 3 DISCREPANCY: AIRCRAFT STATUS OKMAY 17 997 Provin Pattonson OG PISTUDENT NAME & IP NUMBER IP/STUDENT CORRECTIVE ACTION: MAINTENANCE SUPERVISOR/MAINTENANCE COORDINATOR/MECHANIC Parking Spot#: Number of Landings Dates 5-1-7-90 43-DAB OMN Time Dut: 1700 Fuel Load: FLG NSB DED SFB Time In: 1330 Tach Time: 1264.0 OTHER TOTAL AIRCRAFT DISCREPANCY: STATUS MAY 17 GED CORRECTIVE ACTION: 64

AIF	CRAFT FLIGHT/DI	SCREFANCI RECU		
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MAINTENANCE SUPERVISO	R7MAINTENANCE_C Parking Spot			Landings OMN
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Date: <u>5/18/90</u> Time Dut: <u>0836</u>	Parking Spot	600#	Number of DAB	OMN
Date: 5/18/90	Parking Spot	#1 47	Number of DAB / FLG	0MN
Date: <u>5/18/96</u> Time Dut: <u>0836</u> Time In: 1230 AIRCRAFT DISCREPAND	Parking Spot Fuel Load: Tach Time:	600#	Number of DAB / FLG DED	OMN NSB SFB
Date: <u>5/18/96</u> Time Dut: <u>0836</u> Time In: 1230	Parking Spot Fuel Load: Tach Time:	600#	Number of DAB / FLG DED	OMN NSB SFB
Date: <u>5/18/96</u> Time Dut: <u>0836</u> Time In: 1230 AIRCRAFT DISCREPAND	Parking Spot Fuel Load: Tach Time:	600#	Number of DAB / FLG DED	OMN NSB SFB
Date: <u>S/18/96</u> Time Dut: <u>0836</u> Time In: <u>1230</u> AIRCRAFT DISCREPANC STATUS	Parking Spot Fuel Load: Tach Time: CY:	600#	Number of DAB / FLG DED	OMN NSB SFB
Date: <u>5/18/96</u> Time Dut: <u>0836</u> Time In: 1230 AIRCRAFT DISCREPAND	Parking Spot Fuel Load: Tach Time:	600#	Number of DAB / FLG DED	OMN NSB SFB
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Dates <u>S/18/90</u> Time Duts <u>0836</u> Time Ins <u>1230</u> AIRCRAFT DISCREPANC STATUS IP/STOPENT	Parking Spot Fuel Load: Tach Time: CY:	600#	Number of DAB / FLG DED	OMN NSB SFB

AIRCRAFT FLIGHT/DISCREPANCY RECORD Number of Landings Parking Spot#: Datei 5-18 V / . 3 DAB OMN Fuel Load: FLG NSB 1230 Time Out: DED SFB Time In: 1400 Tach Time: :271,0 OTHER TOTAL AIRCRAFT DISCREPANCY : STATUS MAY 18 RECTO 062 TUDENT NIMB NAME £ CORRECTIVE ACTION: MAINTENANCE SUPERVISOR/MAINTENANCE COORDINATOR/MECHANIC Parking Spot#: Number of Landings Date: 5-18 CAB OMN < 12 Time Out: 1500 Fuel Load: FLG NSB DED SFB Tach Times 2720 Time In: 700 TOTAL Z OTHER AIRCRAFT DISCREFANCY: STATUS mone 058 IP/STUDENT NAME G NUMBER CORRECTIVE ACTION: MAINTENANCE SUPERVISOR/MAINTENANCE COORDINATOR/MECHANIC

1 AIRCRAFT FLIGHT/DISCREPANCY RECORD Q, Parking Spot#: Number of Landings 5-18 Date: DAB OMN Time Out: 700 Fuel Load: FLG NSB DED SFB 68.2 800 Tach Time: Time In: OTHER TOTAL AIRCRAFT DISCREPANCY: STATUS OK 623 NAME IP NUMBER IP/STUDENT £ CORRECTIVE ACTION: MAINTENANCE SUPERVISOR/MAINTENANCE COORDINATOR/MECHANIC Date: Parking Spot#: Number of Landings DAB OMN Time Out: Fuel Load: FLG NSB 3/0 DED SFB Tach Time: Time Ing $IJ \eta c$ 11:15 TOTAL OTHER AIRCRAFT DISCREPANCY: STATUS AP/STUDENZ/ NAME a v CORRECTIVE ACTION: MAINTENANCE SUPERVISOP/MAINTENANCE COORDINATOR/MECHANIC

AIRCRAFT FLIGHT/DISCREPANCY RECORD Date: 5-18 Parking Spot#: Number of Landings DAB OMN Fuel Load: below 2 Time Out: 1 70.00 FLG NSB DED SFB 273.5 Tach Time: Time In: 1900 CTHER TOTAL DISCREPANCY: AIRCRAFT STATUS · , MAY 18 RECT NAME UDENT CORRECTIVE ACTION: SUPERVISOR/MAINTENANCE COORDINATOR/MECHANIC MAINTE Parking Spot#: Number of Landings Cates 5-19-90 DAB OMN Fuel Load: $\frac{3}{4}$ + FLG NSB Time Out: 0730 DED SFB Tach Time: 1275.5 Time In: 1005 OTHER Z TOTAL AIRCRAFT DISCREPANCY: I. RT OIL PRESSURE RUNS ON LOW SiDE OF GREEN ARC. STATUS Chins Ching 038 IP/STUDENT NAME & NUMBER PLEARE RECEILIE RETARK CORRECTIVE ACTION: IAINTENANCE SUPERVISOP/MAINTENANCE COORDINATOR/MECHANIC

AIRCRAFT FLIGHT/DISCREPANCY RECORD Number of Landings Parking Spot#: Date: 5/20/90 44 DAB OMN 3/4 Fuel Load: FLG NSB Time Out: 0800 DED SFE 7930 Tach Time: Time In: 12-16.3 OTHER TOTAL 5 DISCREPANCY : AIRCRAFT STATUS MAY 20 RECT CORRECTIVE ACTION: MAINTENANC AINTENANCE COORDINATOR/MECHANIC 5/20/90 Parking Spot#: Dates Number of Landings DAB OMN FLG Time Dut: Fuel Load: NSB 1200 DED SF2 Time In: 1330 Tach Time: 1277-1 OTHER TOTAL 5 AIRCRAFT DISCREPANCY: STATUS MAY 20 RECTO JMBER IΡ CORRECTIVE ACTION: MAINTENANCE SUPERVISOR/MAINTENANCE COORDINATOR/MECHANIC

AIRCRAFT FLIGHT/DISCREPANCY RECORD Parking Spot#: Number Dates = /21/20 of Landings 46 DAB OMN. 14 7.6 Fuel Load: FLG NSB Time Out: 800 DED SFB Tach Time: Time In: 1280,0 10:40 OTHER TOTAL AIRCRAFT DISCREPANCY: STATUS ac Oth. 1773 PAREL. NUMBER IP/STUDENT COPRECTIVE ACTION: MAINTENANCE SUPERVISOR/MAINTENANCE COORDINATOR/MECHANIC 5-21-90 Parking Spot#: Dates Number of Landings $\cdot b$ CIMIN DAB Time Dut: 1/.00 Fuel Load: FLG NSE Hobes DED SFB TIME IN: 140 1280.9 OTHER TUTAL DISCREPANCY: AIKCRAFT STATUS MAY 21 DED Ob mmrkle IP/STUDENT | NAME & NUMBER CORRECTIVE ACTION: MAINTENANCE SUPERVISOR/MAINTENANCE COORDINATOR/MECHANIC

AIRCRAFT FLIGHT/DISCREPANCY RECORD Number Parking Spot#: of Landings Dates Ļ 6 DAB OMN FLG Fuel Loads NSB Time Gut: 400 DED SFB Tach Time: 1530 Time In: OTHER TOTAL DISCREPANCY: AIRCRAFT STATUS ØК **IP/STUDENT** NUMBER NAMI IΡ CORRECTIVE ACTION: MAY 20 RECT SUPERVISOR/MAINTENANCE COURDINATOR/MECHANIC MAINTENANCE Parking Spot#: Number of Landings Dates DAB OMN FLG NSB_ Time Out: Fuel Load: DED SFS KÒÌ Time In: Tach Time: 2 OTHER TOTAL DISCREPANCY: AIRCRAFT STATUS 341 & NUMBER IP/STUDENT NAME CORRECTIVE ACTION: MAINTENANCE SUPERVISOP/MAINTENANCE COORDINATOR/MECHANIC

11 AIRCRAFT FLIGHT/DISCREPANCY RECORD Ar the mail was at 5/21/90 Number of Parking Spot#: Dates Landings DAS OPIN _700# Time Out: 1931) Fuel Load: FLG NSB DED SFB Tach Time: 1283.6 Time In: 2/00 OTHER TOTAL DISCREPANCY: AIRCRAFT STATUS MAY 21 RECT NUMBER LP/STUDENT NA CORRECTIVE ACTION: MAINTENANCE SUPERVISOR/MAINTENANCE COORDINATOR/MECHANIC Parking Spot#: Number of Landings Date: 5/22/90 DAB / CMN Fuel Load: 1/2 Time Out: CRIC-FLG NSE DED SFB Tach Time: 1284.4 Time In: 0920 OTHER DISCREPANCY: AIRCRAFT STATUS OK MAY 22 RECT. colitor 015 14 IP/STUDENT CORRECTIVE ACTION: IAINTEN UPERVISOR/MAINTENANCE COORDINATOR/MECHANIC

AIRCRAFT FLIGHT/DISCREPANCY RECORD Number of Landings Parking Spot#:-Date: 5-21-90 DAB OMN 1/2t FLG NSB Time Dut: 1220 Fuel Load: SFB DED 1281.5 Time In: 1530 Tach Time: TOTAL OTHER DISCREPANCY: AIRCRAFT MK STATUS Chino Chiq. NAME & **JP/STUDENT** IP NUMBER CORRECTIVE ACTION: 9.5821 MAY 21 RECTO MAINTENANCE SUPERVISOR/MAINTENANCE COORDINATOR/MECHANIC Number of Landings Parking Spot#: 46 Dates 21/96 OMN DAB 45 1630 Time Out: Fuel Load: FLG N/5B FULL DED SFB 1.82.6 1730 Tach Time: Time In: OTHER TOTAL AIRCRAFT DISCREPANCY: STATUS OK MAY 21 RECTO (297 IP/STUDEN CORRECTIVE ACTION:

AIRCRAFT FLIGHT/DISCREPANCY RECORD Parking Spot#: Number of Landings B\$ \$ 5/22 Dater 45 DAB OMN Fuel Load: 3/4FLG NSB Time Out: 1920 DED SFB 12856 Tach Timer Time In: 1(30 OTHER TOTAL DISCREPANCY: AIRCRAFT STATUS CORRECTIVE ACTION: MAINTENAN INTENANCE COORDINATOR/MECHAN Parking Spot#: Number of Landings Dates SIZZ OMN DAB Time Out: 1200 FLG NSB Fuel Load: DED SFB Time In: Tach Time: 1285.7 OTHER TOTAL DISCREPANCY: AIRCRAFT Did not fly atus Right fuel flow inop rano M] Ğ Repaired transducer connection - ops JOK CORRECTIVE ACTION: MAINTENANCE SUPERVISOP/MAINTENANCE COORDINATOR/MECHA

AIRCRAFT FLIGHT/DISCREPANCY RECORD Parking Spot#: Number of Landings 5/22/90 41 DAB OMN FLG Fuel Load: NSB Time Out: 1,00 DED SFB Tach Time: Time In: 1700 1286.2 OTHER TOTAL DISCREPANCY: AIRCRAFT STATUS OK. Her 1. Jou 004 NAME & IP NUMBER 12/STUDENT CORRECTIVE ACTION: SUPERVISOR/MAINTENANCE COORDINATOR/MECHANIC MAINTENANCE Dates Parking Spot#: L Number of Landings DAB OMN Time Out: FLG NSB Fuel Load: 730 DED SFB 1287.8 Time In: Tach Time: 430 OTHER TOTAL FF gauge still intom Hert AIRCRAFT DISCREPANCY: STATUS 130 & NUMBER STUDENT NAME CORRECTIVE ACTION: TAINTENANCE SUPERVISOP/MAINTENANCE COORDINATOR/MECHANIC

AIRCRAFT FLIGHT/DISCREPANCY RECORD Number of Landings Parking Spot#: 217 5/24 Date: DAB OMN Fuel Load: 5/4 FLG NSB / Time Dut: 1530 SFB DED Tach Time: 1,40,2 Time In: 1730 TOTAL 7 OTHER DISCREPANCY: AIRCRAFT STATUS (124 IP NUMBER IP/STUDEN CORRECTIVE ACTION: Noted MAINTENANCE SUPERVISOR/MAINTENANCE COORDINATOR/MECHANIC Number of Landings Parking Spot#s 5-24 43 Dates OMN DAB 6 NSB 400 -Time Dute 17-30 Fuel Load: FLG SFB DED 1291.7 Tach Time: Time In: 2000 TOTAL 7 OTHER DISCREPANCY: AIRCRAFT STATUS MAY 84 80 42 6 3 022 & NUMBER NAME TP/S-FODEN CORRECTIVE ACTION: MAINTENANCE SUPERVISOR/MAINTENANCE COORDINATOR/MECHANIC

AIRCRAFT FLIGHT/DISCREPANCY RECORD Parking Spot#: Number of Landings 43 5-77-20 DAB OMN Fuel Load: FLG NSB Time Out: 2030 600 DED SFB Tach Time: Time In: 2/00 1786.3 OTHER TOTAL DISCREPANCY & RIGHT FUEL FLOW INCP DURING AIRCRAFT STATUS FLIGHT EVERY ONCE in ALAYKE IT WOULD INDICATE, NO POWER COSS 0K, I D/ STUDENT NAME & IP NUMBER CORRECTIVE ACTION: REPLACES TRANSDUCER MAINTENANCE SUPERVISOR7MAINTENANCE COORDINATOR/MECHANIC Parking Spot#: Number of Landings Dates 44 5-24 DAB OMN Time Out: Fuel Load: FLG NSB 17770 650 CBS DED SFB 1445 Time In: Tach Time: 1244.2 OTHER TOTAL AIRCRAFT DISCREPANCY: STATUS ·OK Til 49 NAME & IP/SPODENI NUMBER CORRECTIVE ACTION: TAINTENANCE SUPERVISOP/MAINTENANCE COORDINATOR/MECHANIC

AIRCRAFT FLIGHT/DISCREPANCY RECORD Parking Spot#: Number, of Landings Date: 25190 DAB OMN 1/4 FLG Fuel Load: NSB Time Out: DED SFB 1294.9 1900 Tach Time: Time In: GTHER TOTAL DISCREPANCY: AIRCRAFT STATUS OK MAY 25 RECD Endut the 015 CORRECTIVE ACTION: SOR7MAINTENANCE MAINTENANCE COOPDINATOR/MECHANIC Parking Spot#: Number of Landings Dates 'Z(0 OMN DAB FLG NSB Fuel Load: Time Out: 1530 Z DED -SFB Time In: Tach Time: 1700 TOTAL ? OTHER apper arriver cource performance was very poor-fettas if the year; Flaps were down-lot of drag but all wes Rien. AIRCRAFT DISCREPANCY: STATUS Then increase throttles to try to get performance - both engines Feltasif They were going to got. Turned aux pumps on tigh-setter performance, bout still really rough. tolego allerge inst read fin MP., tock, stuel flow normal - no floot ation ulevely Relation CHT IP/STUDENT NAME & NUMBER CORRECTIVE ACTION: cleazer plugs - MAZ drops & static smooth tin limits - all inst. in green ŧ an engs at static - No loss of power noted TAINTENANCE SUPERVISOR/MAINTENANCE COORDINATOR/ME

AIRCRAFT FLIGHT/DISCREPANCY RECORD Parking Spot#: 43 Date: 5-25-90 Number of Landings DAB OMN 3/4 Fuel Load: FLG NSB Time Out: 0900 DED SFB Time In: 1030 Tach Time: 1292.6 TOTAL / OTHER AJRCRAFT DISCREPANCY: Right Alt. Light Illuminated Several times During Flight STATUS IP/STUDENT NAME & IP NUMBER Second wires at Rt Alt - replaced resistors CORRECTIVE ACTION Screw MAINTENANCE SUPERVISOR/MAINTENANCE COORDINATOR/MECHANIC 5-25.90 Parking Spot#: Number, of Landings Dates Ψų OMN DAB Fuel Load: NSB Time Dut: 1500 FLG DED SFB 16.45 Time In: Tach Time: 7939 OTHER TOTAL DISCREPANCY: AIRCRAFT MAY 25 RED STATUS SAME AS ABOVE. 014 IP/STOLLNT / NAME & NUMBER CORRECTIVE ACTION: MAINTENANCE SUPERVISOR/MAINTENANCE COORDINATOR/MECHANIC

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WHITE COPY - MAINTENANCE

YELLOW COPY - CUSTOMER

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AIRFRAME MAINTENANCE LOG MO

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SECTION 3 EMERGENCY PROCEDURES

- 4. Emergency Crossfeed Shutoff -- PULL TO CLOSE.
- 5. Auxiliary Fuel Pumps -- OFF.
- 6. Approach -- NORMAL with full flaps.
- 7. Touchdown -- NOSE HIGH. Hold nose off as long as possible during roll.
- 8. Mixture Controls -- IDLE CUT-OFF.
- Fuel Selectors -- OFF. 9.
- Battery and Alternator Switches -- OFF. 10.
- 11. Magneto Switches -- OFF.
- 12. Airplane -- EVACUATE.

LANDING WITH FLAT MAIN TIRE.

- 1. Landing Gear -- DO NOT RETRACT.
- 2. Fuel Selectors -- POSITION BOTH TO TANK ON SIDE OF FLAT TIRE.
- Airplane -- FLY as desired to lighten fuel loud on side of flat tire
- 3. Before Landing Checklist -- COMPLETE.
- 4 5. Approach -- NORMAL with full flaps.
- Landing -- WING LOW TOWARDS GOOD TIRE. Lower nose 6.
- immediately for steering. 7. Ailerons -- AWAY FROM FLAT TIRE to lighten load on tire
- Brakes -- APPLY ONLY ON GOOD TIRE. 8.

LANDING WITH FLAT NOSE TIRE

- 1. Landing Gear -- DO NOT RETRACT.
- 2. Passengers and Baggage -- MOVE AFT if practical.
- 3. Before Landing Checklist -- COMPLETE.
- 4. Approach -- NORMAL with full flaps.
- Touchdown -- NOSE HIGH. Hold nose wheel off as long as possible 5. during roll.
- 6. Brakes -- MINIMUM necessary.

ELECTRICAL POWER SUPPLY SYSTEM MAL-FUNCTIONS

LALT OFF OR RALT OFF ANNUNCIATOR LIGHT (AMBER) ILLUMINATED

NOTE

Due to component tolcrances and low signal levels, it may be normal for one ALT OFF light to be illuminated when the total load on both alternators is less than 30 amps.

CESSNA MODEL T303

SECTION 3 EMERGENCY PROCEDURES

Affected Alternator -- OFF.

2. Electrical Load -- REDUCE as required to extinguish LOW V (low volt) annunciator light. Significant nonessential loads are the combustion heater, vent fan, or external lights.

If alternator circuit breaker is tripped:

- 3. Bus Tie Circuit Breaker and Affected Isolation Circuit Breaker --CHECK and RESET if tripped.
- Select affected alternator on volt-ammeter, and turn on alternator while monitoring output.
- If output is indicated on ammeter, turn off alternator and continue or terminate flight with electrical load reduced to capacity of single alternator.
- 6. If no output is indicated on ammeter, turn off alternator, reset circuit breaker, and turn alternator on again.
- If circuit breaker trips again, turn off alternator and continue or terminate flight with electrical load reduced to capacity of single alternator.
- If circuit breaker does not trip and ammeter indicates within 30 amps of other alternator output, disregard ALT OFF light and have system checked prior to next flight.

If bus tie circuit breaker and affected isolation circuit breaker are tripped:

- 3. Pull affected alternator circuit breaker and reset bus tie and affected isolation circuit breakers.
- 4. If circuit breakers trip again, terminate flight as soon as practical. Equipment powered from the affected main bus will be inoperative (see figure 7-12, Electrical System)
- If circuit breakers do not trip, perform steps 3 through 8 outlined in the preceding "alternator circuit breaker tripped" procedure.

If alternator circuit breaker, bus tie circuit breaker and left or right isolation circuit breaker are not tripped:

- Vorify alternator field circuit breaker is not tripped; reset if 3. tripped.
- Select affected alternator on volt-ammeter, and turn on alternator.
- 5. If ammeter indicates within 30 amps of other alternator, disregard ALT OFF light and have system checked prior to next flight.
- 6. If ammeter indicates zero amps with other alternator loaded to capacity, turn off alternator and continue or terminate flight with electrical load reduced to capacity of single alternator.

SECTION 3 EMERGENCY PROCEDURES CESSNA MODEL T303

LANDING WITH A FLAT TIRE

In the event of a blowout on takeoff, do not attempt to retract the landing gear. The tire may be distorted enough to bind the gear within the wheel well and prevent later extension. Accomplish the landing in accordance with the appropriate checklist.

PRACTICE MANUAL GEAR EXTENSION

To perform a practice manual gear extension, it is necessary to first pull the LDG GEAR, PUMP circuit breaker prior to performing the Landing Gear Fails to Extend checklist.

ELECTRICAL POWER SUPPLY SYSTEM MAL-FUNCTIONS

Malfunctions in the electrical power supply system can be detected by periodic monitoring of warning lights, volt-ammeter, and circuit breakers; however, the cause of these malfunctions may be difficult to determine in flight. A broken alternator drive belt, defective wiring, malfunctioning equipment, defective battery, or a combination of these is most likely the cause of an electrical system failure. The electrical system is designed to be as fail-safe as practical, making in-flight failures improbable. However, an electrical system malfunction should be dealt with expeditiously to avoid an in-flight emergency.

LEFT OR RIGHT ALTERNATOR OR MAIN BUS MALFUNCTIONS

The ALT OFF light circuitry in the alternator control units monitors the output of each alternator and will cause an amber LALT OFF or RALT OFF light in the annunciator panel to illuminate when the alternator output is insufficient to supply power to the associated main bus. It should be noted that due to component tolerances and low electrical load, one alternator may be carrying all of the load and the other ALT OFF annunciator light will be illuminated when the total load on the system is less than 30 amps. Either alternator output can be monitored by selecting L ALT or RALT on the volt-ammeter.

In the event the L ALT OFF or R ALT OFF light illuminates with an electrical load of more than 30 amps, the probable cause is an alternator drive belt failure or malfunctioning alternator control unit. However, since a ground fault (short to airtrame) on the alternator output wires or main bus could result in illumination of one of the ALT OFF lights, the procedure specified in the checklist portion of this section should be followed to isolate the problem (see LALT OFF or RALT OFF Ambucator)

CESSNA MODEL T303

SECTION 3 EMERGENCY PROCEDURES

Light (Amber) Illuminated checklist). Other indications such as sudden light dimming, audible arcing, or momentary power interruption followed by ALT OUTPUT, BUS TIE, or ISOL (L or R) circuit breakers tripping should provide for immediate detection of a ground fault requiring immediate action

Illumination of the red LOW V light in the annunciator panel warning that the battery bus voltage level is low enough for power to be drained from the battery. The LOW V light can be used as a quick reference for electrical load adjustments to prevent battery discharge. Battery charge rate can also be monitored by selecting BATT on the volt-ammeter.

In the event it is necessary to terminate a flight with electrical power on only one of the main busses, it should be noted that the wing flaps are powered from the left main bus and the landing gear is powered from the right main bus; therefore, either a flaps-up landing or an emergency gear extension procedure will be required.

LOSS OF ALL ELECTRICAL POWER

The possibility of the loss of all electrical power is extremely inprobable due to the dual bus design and fail-safe features. However, in the unlikely event of this occurring, a procedure is available in the checklist portion of this section which provides a means to restore electrical power by utilizing the EM AL FIELD switch. This switch bypasses the battery contactor, bus and supply cables to provide excitation voltage directly to the alternator field switches from the battery. The battery switch should be turned off immediately and remain off, to conserve battery power for emergency alternator excitation in case there has been a partial battery failure or a battery bus fault. During operation with the battery switch off, the electrical noise level on the bus will increase and can cause degradation in the performance of avionics equipment. If an increase in electrical noise is detected, the battery supply system should be checked prior to the next flight.

PARTIAL AVIONICS POWER FAILURE

Primary power is supplied to the left and right avionics busses from the battery bus to optimize battery filtering action for avionics equipment. Circuit breakers protecting the supply cables are located in the left nacelle electrical compartment and cannot be reset in flight. In the event of an avionics power failure, there is a procedure in the checklist portion of this section which provides a means to restore electrical power to each avionics bus by utilizing the red-guarded EMER AVN (L or R) switch headed in the bett sidewall switch and circuit breaker panel

Conc.

ESCRIPTIONS

CESSNA MODEL T303

atic pressure alternate source valve, parking ing controls, and the landing gear lever with he center of the lower instrument panel incoremperature, oil temperature, and oil pressure e right side of the switch and control panel with position indicator, the cabin heating/deo compartment A control pedestal, extending nent panel to the floor, contains propeller installed), throttles, propeller and mixture rudder and aileron trim controls with position e. The lower portion of the pedestal contains fuel selector valve handles, fuel consefed nd the cowl flap controls. An oxygen shutoff i located on the right side of the pedestal.

e instruments, switches, and controls on this the description of the systems to which these

NT PANEL

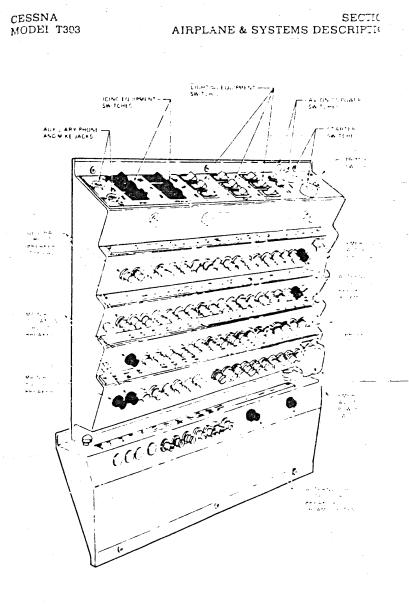
strument group may be installed directly in ager of copilot. Like the pilot's flight instrunel groups the attitude and directional indiced indicator to the left and the altimeter to arrangement. The remainder of the instrur and vertical speed indicator) are grouped t flight instruments-require a second, indestem for operation.

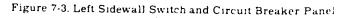
ND CIRCUIT BREAKER PANEL

al equipment switches and circuit breakers mounted on the left cabin sidewall adjacent ols on this panel are illustrated in figure 7-3.

Figure 7-4) is located at the top edge of the nt of the pilot. The panel contains 12 sepaninate green, amber or red when a specific d airplane system. A green colored lamp is or safe condition in the system. However, cates that a cautionary condition exists, diate corrective action. When a hazardous diate corrective action, a red lamp illumi-

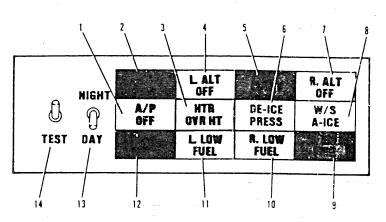
Original Issue





Original Issue

SECTION 7 AIRPLANE & SYSTEMS DESCRIPTIONS CESSNA MODEL T303



- 1. AUTOPILOT OFF LIGHT (AMBER) Indicates the autopilot has disengaged.
- 2. DOOR OPEN LIGHT (RED) Indicates that the forward baggage compartment door, the cabin entry door and or the emergency exit door are not secured safely for flight.
- 3. HEATER OVERHEAT LIGHT (AMBER) Indicates an abnormally high temperature has occurred in the combustion heater and it has been automatically shut off. Once the light illuminates, the heater will not operate until the overheat switch in the right forward nose section (accessible in nosewheel well) has been reset.
- 4. LEFT ALTERNATOR OFF LIGHT (AMBER) Indicates the left alternator is not supplying electrical current.
- 5. LOW VOLTAGE LIGHT (RED) Indicates electrical system bus voltage is less than 24.5 volts.
- WING AND STABILIZER DE-ICE SYSTEM PRESSURE LIGHT (GREEN) - Indicates pressure is being applied to the surface de-ice boots to inflate them.
- 7. RIGHT ALTERNATOR OFF LIGHT (AMBER) Indicates the right alternator is not supplying electrical current.
- 8. WINDSHIELD ANTI-ICE SYSTEM LIGHT (GREEN) Indicates that heating elements in the windshield anti-ice system are operating.

Figure 7-4. Annunciator Panel (Sheet 1 of 2)

7-14

Original Issue

CESSNA MODEL T303

- 9. RIGHT ENGINE FIEE perature condition or p compartment.
- 10. RIGHT LOW FUEL Lit right main fuel tank is
- LEFT LOW FUEL LIGE main fuel tank is 60 pc
- 12. LEFT ENGINE FIRE LI ature condition or pos compartment.
- 13. DAY NIGHT SWITCH indicator lamps for #th
- TEST SWITCH Tests c ing gear system position of landing gear, fire d switch can be used to warning tone.

Figure 7-4. Ann

The annunciator panel also TEST and DAY NIGHT. When lamps on the annunciator pane cator lights, providing a funct also sounds the warning horns stall warning systems. If desare ing tone can be silenced by turn NIGHT switch allows two trig lamps appropriate to either fay

GROUND CONTROL

Effective ground control w wheel steering by using the rud rudder pedal to steer right. We loaded steering bunges (which rudder bars) will turn the bass degrees each side of center. B degrees of turn may be increase minimum turning radius of a differential power, and nose wh figure 7-5.

Original Issue

SECTION 3 EMERGENCY PROCEDURES

CESSNA MODEL T303

4. Battery Switch -- OFF.

- Magneto Switches -- OFF. 5
- 6. Airplane -- EVACUATE.
- 7 Fire -- EXTINGUISH.
- 8. Fire Damage -- INSPECT, repair damage or replace damaged components or wiring before conducting another flight.
- ENGINE FIRE IN FLIGHT (RED ENG FIRE ANNUNCIATOR LIGHT ON)
 - Auxiliary Fuel Pumps -- OFF. 1. 2
 - Affected Engine:
 - a. Throitle -- IDLE.
 - b. Propeller Control -- FEATHER.
 - Mixture Control -- IDLE CUT-OFF. С.
 - d. Fuel Selector -- OFF.
 - Cowl Flap -- OPEN. e
 - Alternator Switch -- OFF f.
 - Magneto Switches -- OFF. g.
 - Emergency Crossfeed Shutoff -- PULL TO CLOSE. 4.
 - Cabin Heater -- OFF. 5.

3

- Operating Engine -- ADJUST POWER as required. 6.
- Airspeed -- ACCELERATE as required to find speed which will
- provide an incombustible mixture until fire is extinguished
- 7 Electrical Load -- REDUCE as required.
- 8. Land airplane as soon as practical.

NOTE

Refer to engine-out operational procedures and to Single-Engine Approach checklist in this section for further details.

ELECTRICAL FIRE IN FLIGHT

- 1. Battery and Alternator Switches -- OFF.
- 2. Vents -- CLOSED (to avoid drafts).
- 3. Cabin Heater -- OFF.
- 4. Fire Extinguisher -- ACTIVATE (if available).

WARNING

If an oxygen system is available, occupants should use oxygen masks until smoke and discharged dry powder clears. After discharging an extinguisher within a closed cabin, ventilate the cabin.

5. Avionics Power Switches -- OFF.

3-12

- CESSNA MODEL T303
 - 6. All Other Electrica

If fire appears out and flight

- 7. Battery and Altern
- Circuit Breakers --8.
- Radio Switches (9.
- 10. Avionics Power Sw

Radio Electrical Sv 11

- until short accuit
- Vents -- OPEN wh 12 guished.
- Cabin Heater OF 13.

CABIN FIRE

- 1. Battery and Alter
- 2. Vents -- CLOSED
- 3. Cabin Heater 0
- 4. Fire Extinguishe
 - - If an oxygen sys oxygen masks u clears. After disci cabin, ventate th

5. Land the airplane

WING FIRE

- 1. Pitot Heat Switch
- 2. Strobe Light Swi
- Navigation Light 3.
- Landing Light St 4.

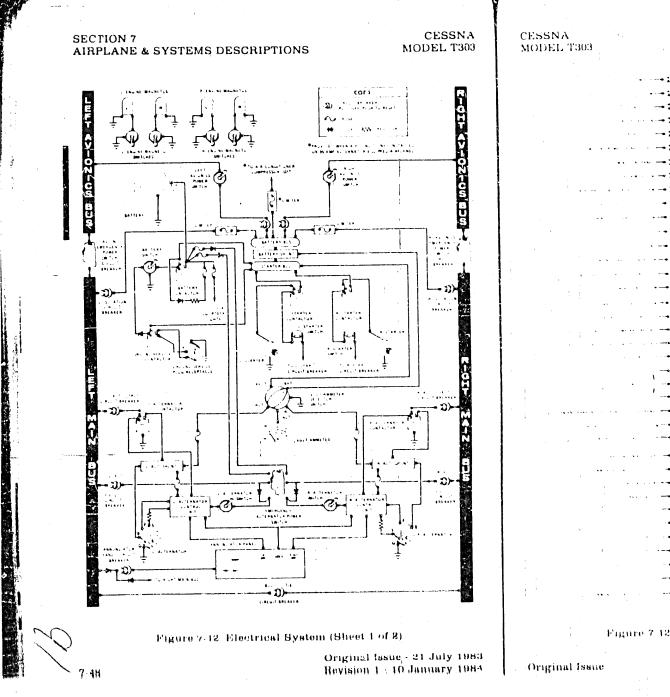
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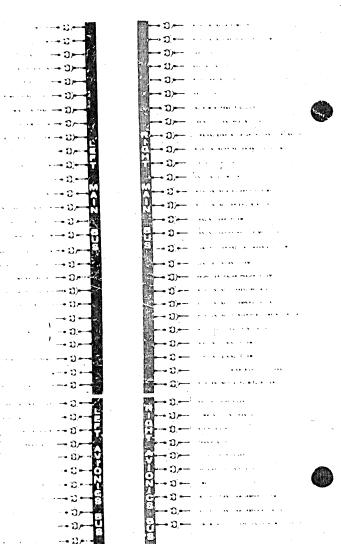
5. Land airplane as

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Original Issue





AIRPLANE & SYSTEMS DESCRIPTIONS

Figure 7-12 Electrical System (Sheet 2 of 2)

7-49

SECTION' 7

Time Date NATIONAL TRANSPORTATION SAFETY BOARD RECORD OF [] VISIT [] CONFERENCE OR [] TELEPHONE CALL 6-28-90 1400 Name(s) of Person(s) contacted or in conference and location Routing Symbol Initials ED HEIBE CESSNA AIRCRAFT COMPANY WECHETA, KS. Subject V4973V Digest HE IS WITH THE CESSNA ENGINEERING DEPATION HIS FLECTRICAL ENGINEERING GLOUP HAS EXAMINED A CESSNA T303 WHICH HAS JUST BEEN RETURNED TO CESSNA, THEY WERE ABLE TO SIMULATE THE ACTIVATION OF THE ENGINE FIRE WARNING WHEN A BUS ISOLATION CIRCUIT BREAKER AND A CRUSCISE RELAY CIRCUIT BREAKER ARE OPENED. THEY HAVE IDENTIFIED THE CAUSE OF THE FALSE FIRE WARNING IN THE ABOUE SCENARIC AND MOVE DEVELOPED A FIX WHICH CALLS FOR INSTALLATION OF A DIUDE IN THE CIRCUIT, THEY PLAN TO ISSUE A SECULCE LETTER AND SECULCE KIT TO ALL DWNERS OF T303 AIPLANES. Conclusions, Action Taken, or Required FOR REPORT Title Date Signatype (-28-91) / NTSB Form 1320.12 (5/70) ASI

1000 DAYTONA BEACH AU 904-252-010

let Center DADIONA BEACH

FACSIMILE COVER SHEET: TO: JEFF KENEDY NITSB 305-536-530 FROM: JIM Teski NUMBER OF PAGES INCLUDING THIS PAGE _____ MESSAGE: <u>RE T303</u> IF ALL PAGES ARE NOT RECEIVED, PLEASE CALL ME AT 904-____

561 PEARL HARBOR DRIVE . DAYTONA BEACH, FLORIDA 32114 . (904) 255-0471 . FAX (904) 252-070

ALTERNATOR SWITCHES - OFF (ENGINES NOT RUNNING) BUSS TTE BREAKER - PULLED LEFT ISOLATION BREAKER - PULLED 1. THROTTLES -UP 2. THROTTLES - BACK 3. LEFT FIRE DETECTION BREAKER - PULLED 4. RIGHT FIRE DETECTION BREAKER-PULLED RIGHT ISOLATION BREAKER - PULLED 5. THROTTLES -UP · 6. THROTTLES - BACK 7: LEFT E RE DETECTION BREAKER - PULLED 8. RIGHT FIRE DETECTION BREAKER - PULLED ANNUNCIATOR TEST - (BUS TIE & IJOLATION BREAKERS - IN) 9. BOTH FIRE DETECTION BREAKERS - IN 10, LEFT " " - OUT 11. RIGHT "____" - 2VT 121 BOTH "_____ " - OUT AFNC

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NATIONAL TRANSPORTATION SAFETY BOARD RELEASE OF AIRCRAFT WRECKAGE

ACCIDENT DENTIFICATION NUMBER MLA9SFA 37

PART I-RELEASE OF AIRCRAFT WRECKAGE

REGISTERED OWNER (name and address)

REGISTRATION NUMBER - N

EMBRY-RIDDLE AERONAUTICAL UNEVERSITY REGIONAL AIRPORT DAYTONA BEACH, FL. 32014 MAKE CESSNA LOCATION

MODEL

DATE OF ACCIDENT

T303 5-28-90 DELAND, FLOREDA The National Transportation Safety Board has X has not-r completed its investigation of the aircraft wreckage described above at wreckage except that listed on the reverse side is hereby released to the registered owner, or owner's representative, for appropriate destation, (If parts are retained, insert NONE.)

TITLE

NONE

SIGNATURE OF NTSB REPRESENTATIVE

(This section may be signed by a person, not the owner or owner's representative, who has knowledge of the disposition of the artait wreat age and its parts. Such signature does not place a responsibility for disposition of the wreckage upon that person.)

I HEREBY ACKNOWLEDGE:

Receipt of the above described aircraft wreckage.

Removal of the parts, if any, listed on the reverse side of this form.

SIGNATURE

TITLE

DATE

CATE

VERBAL TO MICHAEL BARRETT INSULANCE ADJUSTER REMARKS:

NTSB FORM 6120.15 (Rev. 5/79)



NATIONAL TRANSPORTATION SAFETY BOARD RECEIPT OF AIRCRAFT PARTS

ACCIDENT IDENTIFICATON

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REGISTRATION NUMBER		MAKE	MODEL
			-
DATE OF ACCIDENT	LOCATION		

The National Transportation Safety Board has retained, for further examination, those parts, pieces, or components listed below. When the examination is complete, they will be returned to:

OWNER OR OWNER'S REPRESENTATIVE-

ADDRESS

PARTS, PIECES, OR COMPONENTS RETAINED:

dge receipt of the materials by signin	g this form in the spaces designater below
TITLE	DATE

STATEMENT OF PARTY REPRESENTATIVES TO NTSB INVESTIGATION

Aircraft Identification:

Registration Number $N.49737$					
Make and Model CESSNA 7-35					
Location DELAND FLOETDA					
Date 5-28-90					

The undersigned hereby acknowledge that they are participating in the above-referenced aircraft accident field investigation (including any component tests and teardowns or simulator testing) on behalf of the party indicated adjacent to their name, for the purpose of providing technical assistance to the National Transportation Safety Board.

The undersigned further acknowledge that they have read the attached copy of 49 CFR Part 831 and have familiarized themselves with 49 CFR 8831.11, which governs participation in NTSB investigations and agree to abide by the provisions of this regulation.

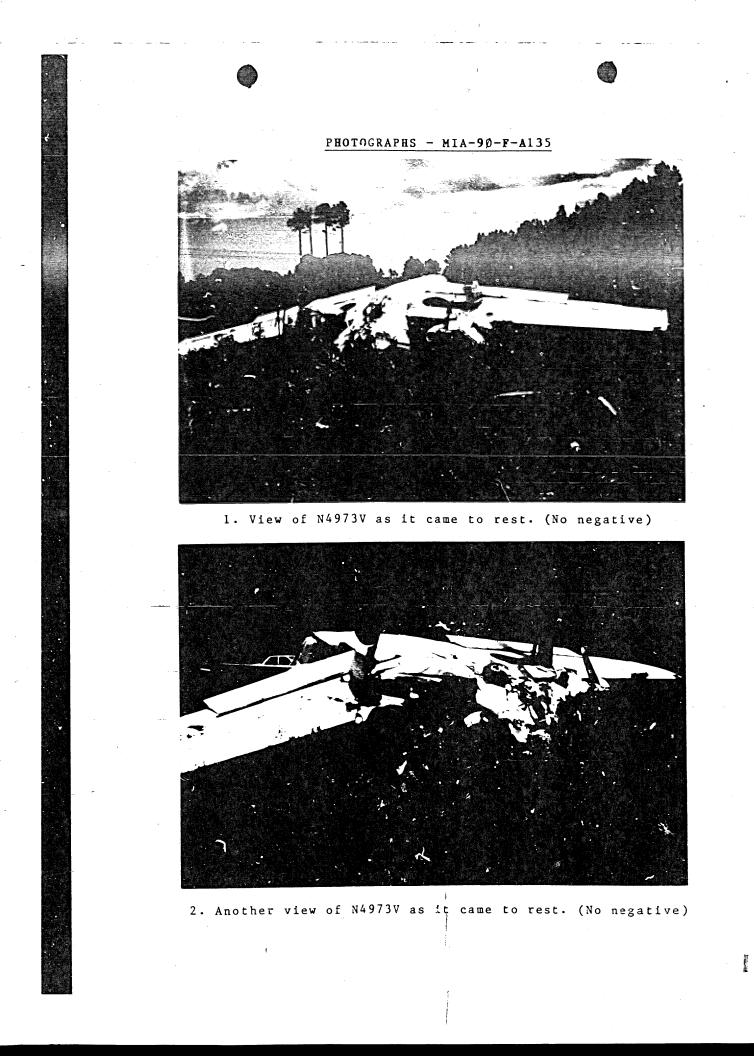
It is understood that a party representative to an investigation may not be a person who also represents claimants or insurers. The placement of a signature hereon constitutes a representation that participation in this investigation is not on behalf of either claimants or insurers and that, while any information obtained may ultimately be used in litigation, participation is not for the purposes of preparing for litigatict.

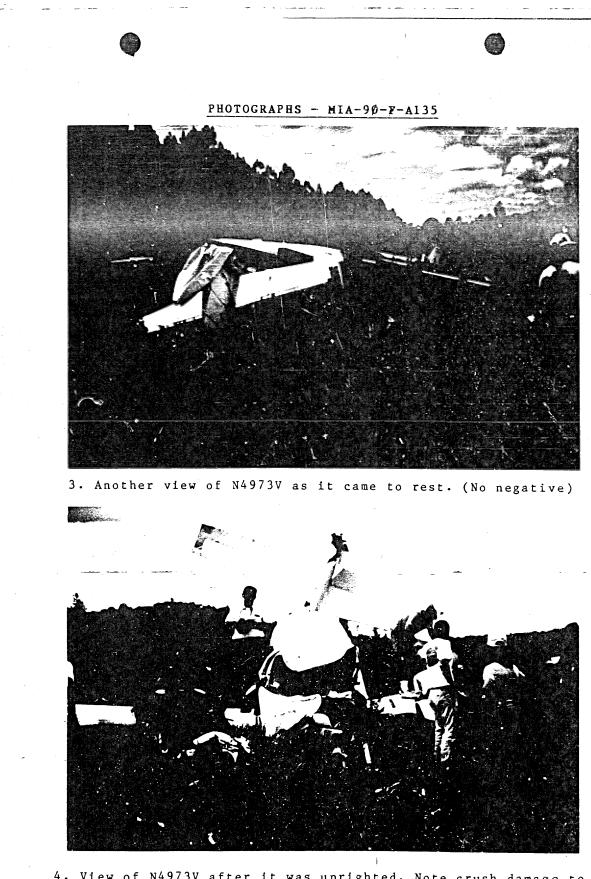
By placing their signatures hereon all participants agree that they will neither assert nor permit to be asserted on their behalf, any privilege in litigation, with respect to information or documents obtained during the course of and as a result of participation in the NTSB investigation as described above. It is understood, however, that this form is not intended to prevent the undersigned from participating in litigation arising out of the accident referred to above or to require disclosure of the undersigned's communications with counsel.

NAME (Printed)	PARTY	DATE
Pule Carte	(eledyne	5. <u>5.</u>
Fire x-co	DEFINIA AT	<u> </u>
LYLE SUNDERLA	W EPAU	6-5-20
	Pale Carte	Pale Carter Celedyne

Continued on reverse

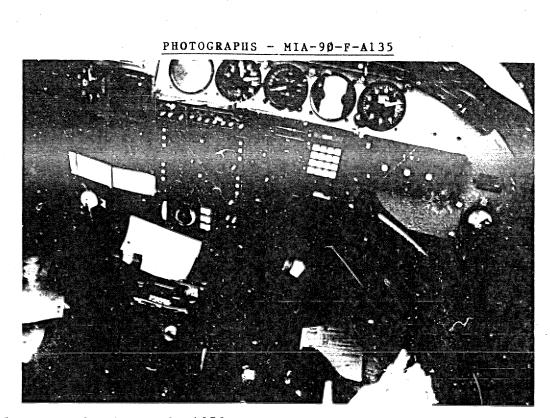
-2-SIGNATURE NAME (Printed) PARTY DATE Sames Fisk. ERALI 6-590 AACK ARUN ERAU G-5-92 JOECPOOLE ERAU G-5-92 A.C. TACKER ERIU 6-5-90 107



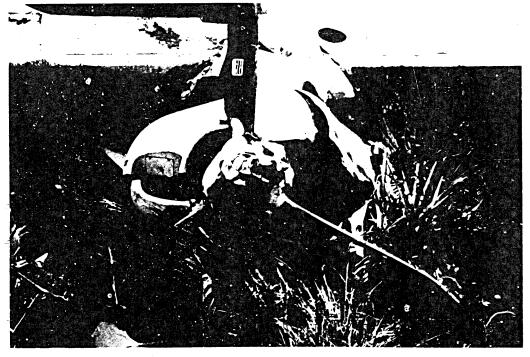


4. View of N4973V after it was uprighted. Note crush damage to cockpit, roof, and windshield. (No negative)

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5. View of cokpit of N4973V after the accident. Note position of gear handle, flap handle, and engine controls. (No negative)

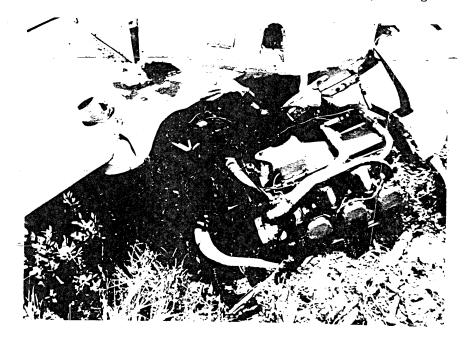


 Right engine and propeller from N4973V Note propeller in feathered position. (No negative)

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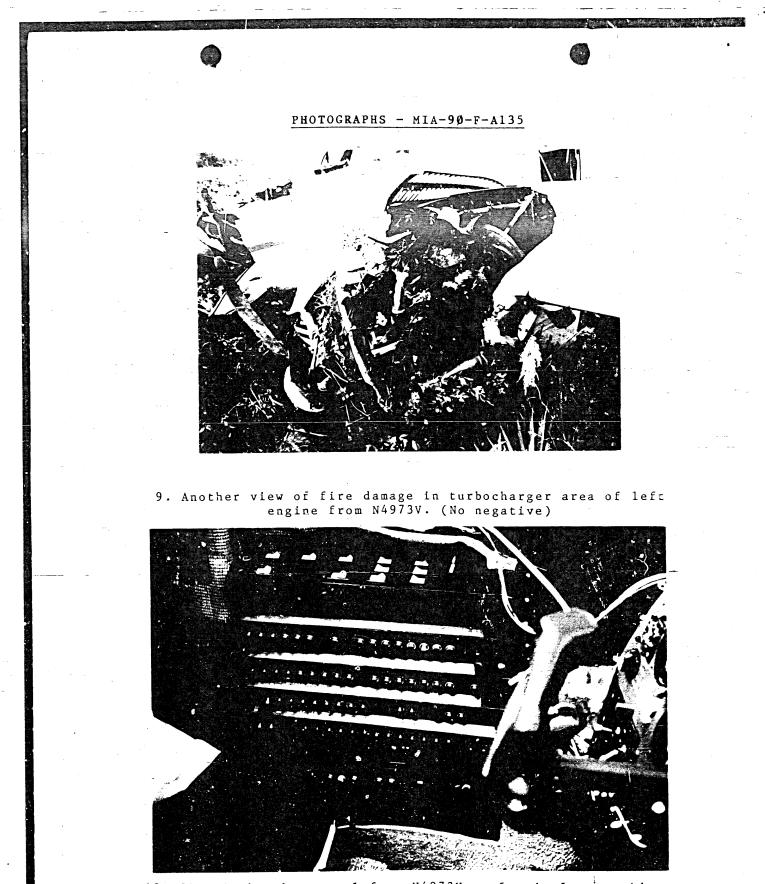


7. Left engine and propeller from N4973V. Note propeller in flat pitch with chordwise scratches on face. (No negative)



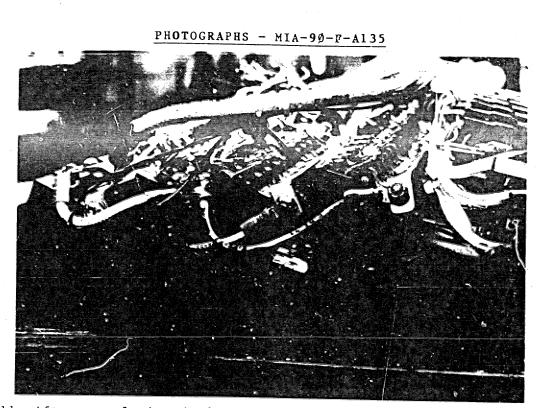
8. Left engine from N4973V. Note fire damage in area of turbocharger (No negative)

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10. Circuit breaker panel from N4973V as found after accident. Note left bus isolation, right bus isolation, bus tie, landing gear indicator and air conditioning compressor breakers are open. (No negative)

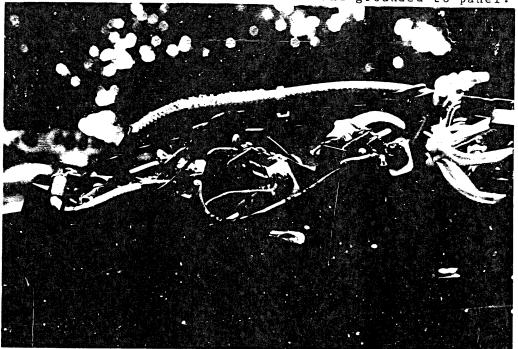
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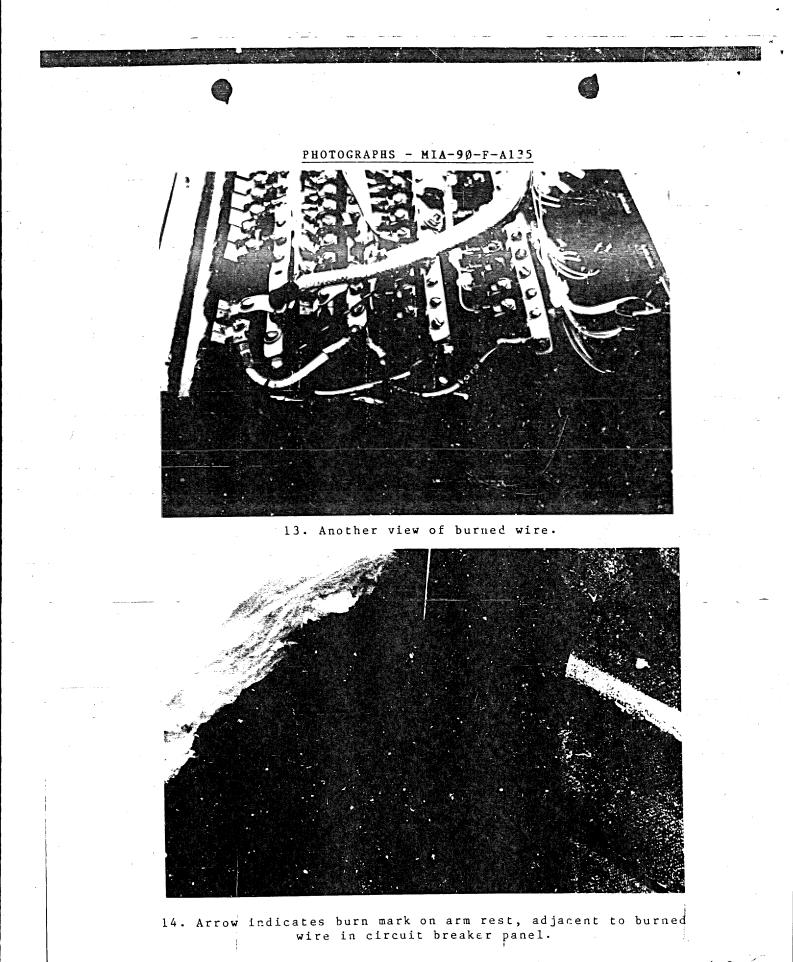
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ll. Aft area of circuit breaker panel from N4973V. Arrow indicates burned wire which connects left bus strips together and point on metal that wire was chaffed and grounded to panel.



12. Another view of burned wire. Arrow indicates chaff mark.



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