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MIDAIR COLLISIONS IN U. S. CIVIL AVIATION - 1968

A SPECIAL ACCIDENT PREVENTION STUDY



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NATIONAL TRANSPORTATION SAFETY BOARD

The National Transportation Safety Board was created by an Act of Congress in 1966, which simultaneously established the Nation's first Department of Transportation headed by a Cabinet-level Secretary. The Safety Board is headed by five Members appointed by the President and approved by the Senate.

Unlike the other functioning segments of the Federal Government that were brought together under the Department of Transportation, the Safety Board is autonomous.

The Safety Board is charged with a continuing across-the-board review of the general safety picture in all modes of transportation. Among other responsibilities it has broad powers in the investigation and cause determination of transportation accidents and may review on appeal the suspension, amendment, modification, revocation, or denial of any certificate or license issued by the Secretary of Transportation or by any Administrator under him.

The Safety Board's authority extends specifically over U.S. civil aviation, maritime, railroad, highway, and pipeline safety. Except for civil aviation, the Board delegates the greater percentage of its accident cause determination directly to the agencies under the Department of Transportation that are principally concerned. However, the Board can assert its authority in any of these transportation fields and makes a final determination of cause in unusual or catastrophic accidents in all modes of transportation. Moreover, Board Members or designated staff, participate frequently as "official observers" during accident investigations. By exercising its authority of recommendation, the Safety Board is continuously involved in the prevention of transportation accidents and enhancement of safety.

In the field of civil aviation safety, the Board conducts its own investigations and determines the cause of all air carrier accidents, all fatal lightplane accidents, helicopter and air taxi accidents and other special accidents or incidents it may desire to take over. It has delegated to the Federal Aviation Administration of the Department of Transportation the authority to investigate accidents involving nonfatal helicopters, aerial applicators, amateur-built and restricted aircraft, and nonfatal light plane accidents, except those involving air carriers (including air taxi operators) and midair collisions. However, the Board reserves to itself the right to determine the cause. The Board is staffed for its aviation safety operations by virtue of having taken over the Bureau of Safety of the Civil Aeronautics Board.

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MIDAIR COLLISIONS IN U. S. CIVIL AVIATION

1968

A Special Accident Prevention Study



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Mideir Collisions

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SUMMARY

The study showed that during calendar 1968 there were 38 midair collisions involving 76 aircraft, a 46-percent increase in the number of midair collisions over the 1967 figure. In every 1968 midair collision accident, a general aviation aircraft was involved. In three cases, an air carrier was involved with a general aviation aircraft. In one case, a military aircraft was involved with a general aviation aircraft. In the remaining 34 instances, the collisions were between two general aviation aircraft.

Twenty-four of the 38 collisions resulted in 71 fatalities—
all occupants of general aviation aircraft. The 1968 fatality figure
is 55 percent lower than the 1967 midair collision fatality figure.

There were no fatalities in the air carrier aircraft in 1968, but
in 1967 the three air carrier aircraft accounted for the vast
majority of the fatalities. There were three air carrier aircraft
involved in 1967 midair collisions and three in 1968 midair collisions.

Most of the 1968 midair collision accidents occurred at or near an uncontrolled airport, below 5000 feet, in Visual Flight Rules (VFR) weather during the summer months and on the weekend.

The traffic in the airspace involved in the 1968 midair collision accidents was not congested, and the closure rate between the aircraft involved was well below the cruise speed of the aircraft involved.

The air traffic control system was a factor in approximately 20 percent of the collisions, as was dual instruction. However, the majority of the midair collision accidents in 1968 occurred at uncontrolled airports in low-density traffic. The major problem in the midair collision accidents was the failure of the pilot to adhere to the "see and be seen" concept--a concept that well may be, at least in high-density terminal areas, on its way to becoming outmoded, unsafe, and incompatible with saturated operating environments.

Several recommendations on pilot awareness, pilot and controller education, air traffic control procedures and operations, airport pattern identifications, anti-collision lights, and aircraft certification process are included in the report.

INTRODUCTION

The midair collision hazard continues in spite of marked improvements in aircraft design, the air traffic control system and associated equipment. The human element continues to be a major factor; pilots are not adhering to safe flying techniques and/or existing flying rules which may now stand to be reviewed for improving compatibility with human performance capabilities and the operating environment. The midair collision problem is further complicated by the phenomenal growth, in the recent past, of the U. S. civil aviation fleet in both numbers of aircraft and hours flown. In this light, extension of separation criteria to aircraft operating VFR within a control zone may now be warranted. air traffic control system needs now to assume a more extensive responsibility for the safe operation of aircraft under the jurisdiction of the tower--within this airspace.

It is estimated that within the next 10 years, three times the present number of passengers will fly. About 10 times more cargo will be carried. Variations in air carrier speed and passenger carrying capacity will increase threefold. The number of general aviation aircraft will double. FAA towers will control three times the present traffic, and IFR traffic at airports will more than double. In an environment of a mix of jumbo jets and light aircraft, of

supersonic transport and STOL aircraft, the midair collision threat will tend to increase in geometric proportion. It is conceivable that in the near future a single midair collision could result in the loss of a thousand lives.

Improvements in collision avoidance equipment are under development, but unfortunately many of these are not yet in production, are economically prohibitive, or are still drawing board concepts. Additionally, studies are being made to seek improvements in the ATC system.

To examine every related cause of midair collisions would necessitate an in-depth study of every segment of the overall safety system. It would include such items as airport planning and development, airport congestion, the control of the airways system, weather reporting, radar surveillance control of air traffic, communications, rules and regulations, design of aircraft, aircrew training, and all of the factors that relate to human errors. Such an examination is far beyond the scope of this very limited study which deals specifically with the analysis of data concerning the 38 midair collisions of 1968 and the conclusions which can be drawn from such a study.

The study includes a survey of existing midair collision
literature, a review of the 1968 midair collision accident reports,
and an analysis of numerous automatic data processing readouts
on various pertinent midair collision factors.

FACTS & DISCUSSION

A. Data

Standard automatic data processing computer readouts and selected data readouts of the 38 collisions were analyzed and evaluated. Additionally, a case-by-case review of the investigators' reports was conducted to provide further insight into each collision accident for factors not recorded in the automatic data processing readouts. Witness statements, detailed radio communications, wreckage photos, terrain or airport drawings, and other items were referred to where relevant.

The following accident data combinations were studied:

- Airport Proximity by Injury Index
- Total Pilot Hours and Hours in Type
- Type of Operation by Injury Index
- Briefs of Accidents Midair Collisions 1968
- Cause/Factor Table
- First Phase of Operation by Injury Index
- First Type of Accident by Aircraft Damage
- Altitude of Occurrence
- Kind of Flying by Injury Index
- Month of Occurrence by Type of Weather Conditions
- Type of Operator by Conditions of Light
- Phase of Operation by Conditions of Light

^{*}See Appendices 1 thru 8

B. Weather

Selected weather data available from locations nearest to the collision sites were reviewed in detail. In all cases, weather conditions were VFR (ceiling 1,000 feet and visibility 3 miles) or better, as far as could be determined. The weather was considered to have been clear in nine cases, with only high clouds reported in 10 others, while middle and/or low clouds existed in the remaining cases. There were 16 cases where clouds prevailed at or below 5,000 feet (above ground level), but in only four of these were ceilings (broken or overcast) in existence. Haze, or haze and smoke, were likely to have been in the area in six instances. Although precipitation was not reported over the immediate accident site in any case, there probably was precipitation in the general area in 11 cases, and it would have been showery in nature.

C. ATC & Operations

ATC

Air Traffic Control service was involved in seven of the 1968 midair collisions. In all seven, traffic congestion, the tower controller's visibility limitations, and inadequacies of VFR traffic flow procedures were found to contribute to the chain of events leading up to the collisions. In one case involving an air carrier aircraft, human performance limitations on the part of the controller were considered to have been contributory. However, the final responsibility to see and avoid other aircraft under visual conditions continues to rest with the pilot.

Integrating aircraft into the traffic pattern from different positions around the airport was a factor in the midair collision problem. For example, Aircraft A was practicing touch-and-go landings, Aircraft B reported 5 miles out and was given instructions to report straight in final at 2 miles. Aircraft C also in the traffic pattern was told by the control tower operator to report downwind. Aircraft B was then cleared to land following Aircraft A. Aircraft B reported that he did not have his traffic in sight. Aircraft C was then given sequence number two to land following Aircraft B. It was at this point that C overtook and collided with B. While the tower was remiss in issuing clearances to aircraft before their

spacing was assured, additionally, the pilots involved should have taken corrective action when they did not have each other in sight.

Aircraft C, although cleared for a long straight in final approach, was in fact breaking into the normal traffic pattern flow. Whenever the normal flow of the traffic pattern is so penetrated, both the pilots and the controller must reassess their mental images of the existing traffic situation.

In still another case, Aircraft A was cleared to land. Aircraft B was following too close to Aircraft A and the tower controller instructed Aircraft B to go around. Aircraft B complied with the controller's instruction. Meanwhile Aircraft C was progressing toward the downwind leg of the traffic pattern. Aircraft D was on final and was the object of the controller's attention. Aircraft B, on go-around, collided with Aircraft C as Aircraft C was entering the downwind leg. Had the controller reassessed the out-of-flow problem created by the go-around traffic, he might have alerted the two aircraft to each other's presence and also assured proper spacing of the aircraft.

Midair collisions are very rare at airports where the traffic flow is directed in a positive and orderly manner. In this regard in 1955 the CAA (Civil Aeronautics Administration) made a study which traced the ground tracks of aircraft in the traffic pattern at certain airports. Such studies may be in order at this time for

all controlled airports so as to understand and minimize out-of-flow problems while standardizing controllers' reactions to out-of-flow conditions when they do occur.

Of course, any attempt to standardize traffic flow entry and departure routes must consider the pilots who are not familiar with the local landmarks. For example, in one midair collision accident, a Canadian pilot, unfamiliar with the area, was asked for his position by the tower. The pilot did not respond. Shortly thereafter the pilot was involved in a midair collision. It is logical to assume that the pilot's attention was diverted from looking outside for traffic to perhaps looking at the chart and/or ground landmarks. A local VFR approach chart, if available, might have prevented this collision.

In still another case, adherence to a standard entry procedure might have prevented the collision where an airliner and a small trainer were instructed to enter the traffic pattern in the same area. The controller who directed these aircraft did not verify if either aircraft had the other in sight. The controller in this case was indecisive in assigning a runway to the airliner, thus allowing the captain to choose his own course of action. This led to multiple communications which might have been better used in spacing traffic in a positive manner.

Positive air traffic control is sometimes affected by unusual topographical features such as mountains. In one midair collision accident, the controller was restricted in routing aircraft in and out of his area because of the high terrain. In a case in point, an airliner was required to make a steep descent which may have increased the in-cockpit duties of the crew to the detriment of observing local traffic. Also, high terrain dictated the number of access corridors available to the controller, thus increasing congestion in a limited airspace. A comprehensive study of airports of this nature may reveal needed improvements in directing VFR traffic in controlled airspace.

Not all of the tower controller's problems are in aviation.

Often an airport is in an area of smoke-producing factories.

Investigation has shown that smoke may have been a factor in two pilots' inability to see each other. The smoke was definitely a factor limiting the tower controller's vision. The controller in this case reported the smoke obscuration of the crosswind leg as a daily occurrence. Entry and departure routes to such airports should be adjusted until positive action against air pollution is taken in those areas.

Another procedural problem was present in a midair collision accident where Aircraft A was engaged in a practice-instrument

approach training flight in VFR conditions. The pilot of Aircraft A conducted his practice approach on the proper approach control frequency throughout. The approach controller did not require the pilot to change to local tower control frequency at the outer marker, as is the general practice. Aircraft B was approaching the same runway on local control frequency. In this case neither pilot knew, from radio communications, of the other's existence. They were occupying the same airspace under different controllers. True, they were VFR and should have seen one another, but insistence by the controllers on standard procedures might have prevented this collision.

Operations

Twenty-four of the 38 midair collisions occurred over or in the vicinity of an airport. Twenty of these occurred while approaching to land. Two occurred during the takeoff phase and two occurred when one of the aircraft was descending to the airport. Figure 9 is a diagram showing an imaginary funnel closing in on the approach end of a runway. The two collisions that occurred during the takeoff and the two that occurred during the descent to the airport are not shown. It will be seen that the number of midair collisions increased on final approach getting worse at position of flare-out.

The area shown in Figure 9 is not only the area of greatest concentration of aircraft but also that part of the traffic pattern where a pilot is engaged in the most vulnerable and critical part of the flight—the landing. Increased attention is given to the landing gear, flaps, attitude control, and airspeed. He is concerned with wind speed and direction and possible traffic on the ground. He endeavors to position his aircraft at such an altitude and distance as to enable an approach best suited to the prevailing conditions. He may have to be attentive to instructions from air traffic control and be concerned with positioning his aircraft in the right sequence to land. At times the weather conditions may force him to divide his attention and consequently interfere with his piloting duties.

The airport pattern (Figure 9) conditions are such that aircraft of different configuration, flown by pilots of varying skill, close on each other without being seen by either pilot. Unlike the surface highway, there is no dividing line and a separate lane for the fast and the slow, but rather three dimensions with which to contend, and a journey that must end in a single lane.

It takes two to collide but only one need be careless. Statistics in 1968 show all are vulnerable, from the pilot with 15,000 hours to the pilot on his first solo. Instructor pilots seem more vulnerable than the average pilot. Fifty percent of the collisions around the

airport involved flight instructors. Perhaps it is because they spend more time in the traffic pattern and have additional activities and related distractions in teaching and monitoring their students' actions.

Fourteen of the midair collisions occurred away from an airport under the following circumstances:

- (a) Three involved agricultural aircraft. One during a ferry flight and two on-the-job aerial spraying.
- (b) One occurred where one of the aircraft was engaged in instructional training.
- (c) One occurred where both aircraft were engaged in instructional training.
- (d) One occurred where one of the aircraft was engaged in instrument flying training.
- (e) One occurred between aircraft being flown in formation beyond pilot abilities, with one pilot under the influence of alcohol.
- (f) One occurred between aircraft on pleasure flights and both pilots under the influence of alcohol.
- (g) One occurred in normal cruise where the cockpit side window curtain of one was found to have been drawn.
- (h) One occurred between aircraft of the same organization while herding horses.
- (i) One occurred between aircraft of the same organization while spotting fish.
- (j) One occurred between two gliders while soaring in the same thermal.
- (k) One, involving a military aircraft, occurred in the vicinity of a military training area.

(1) One occurred in Alaska where the silhouette of a crossing aircraft blended with the snow covered background.

From Figure 8, it will be seen that 25 of the midair collisions involved pilots with more than a total of 1,000 hours pilot-time.

However, 32 of the collisions involved pilots with less than 100 hours in type as opposed to seven who had less than 1,000 hours in type. From the statistics on total pilot-time, an inference can be drawn that while all pilots are equally vulnerable to collisions, experienced pilots were involved in significantly high numbers.

A more significant inference is that pilots with less time in type are more vulnerable to collisions because of their greater attention to cockpit details and to flying the plane.

Limitation of the pilot's capability to detect visually other aircraft and the importance of the contrast aspects as they relate to visual detection were recognized and considered in the analysis of data in this study.

D. Collision Avoidance Systems

The airlines have been searching for a suitable midair collision avoidance system (CAS) for over a decade. During this period a number of technological schemes were submitted and tested, each having its own unique limitations. Although none were considered to be satisfactory, this flurry of technical activity serves as a basis for the eventual development of time-frequency technology, a methodology that holds substantial promise. The traffic scheme currently envisioned is a cooperative one, that is, all participating aircraft must have similar equipment aboard.

The Air Transport Association of America has been instrumental in the formulation of the technical specifications for the time-frequency system and its technical working group issued a set of such technical standards on June 30, 1967. Basically, these technical guidelines provide for an indication of the hazard presented by the detected aircraft, a determination as to whether a maneuver is required and an indication of the evasive action necessary to maintain safe separation. The time-frequency message format consists of range, range rate, and altitude.

This CAS is presently in the hardward development stage with flight evaluation planned for 1969. The projected costs of the system are about \$15,000 to \$30,000 per aircraft and, unless unusual difficulties are encountered during the tests, the installation

cycle may start as early as the end of 1970 and extend to full fleet protection by 1972.

While the general aviation fleet could incorporate and command all of the sophisticated protection a time-frequency CAS system offers, the relative equipment costs, in light of the present state of the art, would preclude widespread acceptance among that segment of the fleet most in need of such protection.

Various technical innovations over the years have been oriented primarily toward enhancing visual conspicuity of the aircraft, e.g., paints and paint patterns, visual aids, exterior lighting, etc.

While each of these various schemes may have some unique individual merits, their collective impact is not considered to be adequate in resolving the present and projected midair collision hazards.

In July 1968, the FAA's National Aviation Facilities Experimental

Center (NAFEC) completed an evaluation of certain low-cost

collision avoidance ground training equipment to determine how a

pilot's search under normal VFR conditions might be enhanced by improve

scan/search techniques. Basically the project utilizes a 35mm slide

projector which displays the intruder aircraft. Both the azimuth

and the slide exposure time are controlled, as well as the intruder

aircraft's wing span dimensions and viewing distances. FAA is

presently developing a small number of these "view/scan" packages

to offer to flight training schools for experimental purposes.

In conjunction with FAA, two automatic pilot-warning indicators
(PWI) are being considered by the National Aeronautics and Space
Administration, one at the NASA Langley Research Center and the
other at the NASA Electronics Research Center. Both are doing
PWI simulation work.

The Langley PWI, with a transmitter-receiver in the "protected" aircraft and a transponder in the "intruder" aircraft, is basically a cooperative Doppler radar. Preliminary flight tests were completed in February, March, and April of this year and further tests are scheduled for this coming July and August. The Electronics Research PWI (9 month contract for hardware let in August 1968) uses the Xenon strobe discharge beacons mounted on all aircraft using the system, supplemented by silicon photoelectric detectors in the protected aircraft. This system appears to be the "least cooperative" of all PWI's since purchase of only the emission portion, the strobe discharge beacons, would significantly enhance visual conspicuity.

Recent discussion with FAA regarding the status of PWI

technology indicates that there are perhaps six to eight manufacturers

presently conducting significant research and development of

infrared/strobe light devices. Some of these manufacturers hope

to perform prototype flight tests this summer and foresee market

delivery as early as the end of the current calendar year.

Additional data on state of the art and chronology of development of anti-collision/collision avoidance systems are included in Appendix 9.

Cockpit Visibility

In most midair collision accidents, the probable cause was thermined to be: "Pilot in command failed to see and avoid other aircraft."

A detailed analysis of the 1968 midair collisions was undertaken to determine if the pilot failed to see other aircraft because of poor visibility due to cockpit design.

Part 23 of the Federal Aviation Regulations Airworthiness
Standards does not spell out minimum design criteria regarding
cockpit visibility in general aviation aircraft. Therefore, the
analysis of cockpit visibility of 1968 midair collision accidents
centered around the reported facts and circumstances concerning
each accident. In some cases, without survivors or witnesses,
the physical evidence was not conducive to an objective analysis
of the visibility problem.

The cockpit visibility study was further complicated by the unorthodox behavior of some of the pilots. There were several cases where (1) pilots ignored standard traffic pattern procedures, (2) pilots flew under the influence of alcohol, (3) pilots flew unauthorized formation flights, and (4) one student pilot took a friend for an airplane ride. Under such circumstances, one can assume that good cockpit visibility would be of little use to the aforementioned pilots. Unfortunately, such pilots as described

above do occupy the airspace and therefore must be considered in the cockpit visibility study.

Particular attention was directed to comparing the convergence angles with the flight maneuvers of the two aircraft, in an attempt to determine if cockpit vision was restricted by the aircraft structure and therefore a contributing factor in the collision.

In any discussion of this nature, two schools of thought immediately assert themselves. One group maintains that the high-wing airplane affords the best cockpit visibility and the other group favors the low-wing airplane. This report does not presume to settle that controversy; however, it will indicate that both sides have merit and limitations. Table 2 was compiled to show angles and phase of operation by high- and low-wing aircraft involvement in each 1968 midair collision accident.

There were two high-wing aircraft for each one low-wing aircraft involved in the 1968 midair collisions. This generally follows the aircraft population figures for high- and low-wing aircraft.

In considering midair collisions, one tends to think of two aircraft closing head-on at a fast closure speed. However, as indicated on the chart, there were only two head-on collisions in 1968. In one case it was determined that the instructor pilot a could have seen the other aircraft. Presumably the instructor

inside the cockpit. It is possible that the pilot of the other aircraft, flying straight and level, could not have seen the instructor's
airplane, as it was under the nose of his airplane.

The majority of collisions occurred between aircraft converging on a near 0° angle as shown in Table 2. Of the 13 collisions in the 0° to 10° angle range, it was noted that the final approach and landing was by far the most hazardous phase of the flight. In several of these accidents, one aircraft overtook the other from the top and rear. On the surface, this situation may indicate a cockpit visibility problem; however, one should consider that both airplanes, if normal procedures were followed, were within sight of each other earlier in the traffic pattern or on the initial approach to the airport. There were several cases where surviving pilots admitted they saw the other traffic, but had lost sight of it somewhere in the traffic pattern or mistook another airplane for the one previously sighted.

Flight instructors would, normally, be expected to exercise a higher than average degree of vigilance. However, 10 flight instructors were involved in midair collisions. If they are indeed more vigilant, one might surmise that cockpit visibility deficiencies account for such flight instructor involvement. On the other hand, a cockpit visibility problem is not indicated when the instructor

and student fly in and out of a <u>busy</u> uncontrolled airport safely on a cross-country trip only to come to the home airport and collide with the only other aircraft in the air for several miles.

In almost all the midair collision accidents, it was determined that at least one pilot was in a position to see the other aircraft, and in many cases it was determined that both pilots could have seen each other's aircraft. Cockpit obstructions such as window posts, opaque sun visors, and instrument panel shields restrict the pilot's view.

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Conclusions

While there was no evidence of adverse weather having been a significant factor in any of the 38 midair collision accidents, haze and/or smoke were likely to have been in the area in six instances; precipitation, showery in nature, was probably in the general area in 11 cases. All 38, however, occurred during daylight hours under VFR conditions. It was found in eight cases where, during descent, one aircraft overtook another, low and at slow closure rates, that inherent aircraft design restrictions to vision reduced the pilots ability to see the other aircraft. The pilot's lack of compensation for this in his maneuvers was a significant factor in nearly all the midair collisions.

It was noted that most collisions occurred in areas and periods of greatest general aviation activity, and that the most likely time and place for collisions to occur would be on bright clear Sunday afternoons in August at uncontrolled airports.

It was also noted that two of every three collisions occurred at an airport, that the pattern of collisions started during the descent to the airport, that it gradually increased while entering the landing pattern, and that the most critical period is the final turn-in right up to the airport threshold and flare-out. It is during this critical period that four of every five collisions at or near airports occurred. It was also noted that the experienced and inexperienced

pilots were equally vulnerable and 50 percent of collisions at or near airports involved a flight instructor. Also, inexperience in type of aircraft being flown rather than total pilot-time was noted in the greater portion of them.

Generally, the collisions that occurred away from an airport involved the following pattern: two pilots, a team on the same mission and aware of each other's presence, flew into each other.

Examples of these are: a pair of ferry pilots, a pair herding horses, a pair spotting fish, a pair of aerial applicators spraying, a pair endeavoring to fly formation beyond their skill, a pair flying formation while under the influence of alcohol. There were a few exceptions to this pattern. For example, a silhouette of a crossing aircraft was unobserved against the snow background in Alaska. A pilot flying in the vicinity of a military training area was involved in a collision with one of a military formation whose leader's attention was drawn away from forward visual scanning.

In analyzing these cases, it can be said with justification that well applied air traffic control (where these services are provided), and pilot vigilance could have prevented the collisions which occurred at or near airports. With equal justification it can be said that most of the collisions that occurred away from the airports could have been avoided if pilots had not flown under the

concept; had they been aware of their limitations so as not to attempt to fly in formation beyond their skill; and had they not maneuvered into a blind spot when temporarily losing sight of their companion aircraft.

It was noted that two of every three collisions resulted in fatalities. There were three 1968 collisions involving air carriers.

Fortunately, none of these proved fatal to occupants of the air carrier aircraft. For this reason, although the 1968 midair collisions increased by 46 percent, there were 55 percent fewer total fatalities than in the year 1967.

Much can be done to eliminate the 1968 type collision. Consideration must now be given to assuring more positive VFR traffic separation, guidance and control both at our major passenger airports and in areas and periods of greatest general aviation activity; to improving radar surveillance; to improving certain air traffic control system procedures and provisions; to reemphasize the education of pilots, their instructors and controllers with regard to collision avoidance awareness; and to developing practicable and economically feasible collision avoidance devices for general aviation aircraft.

G. Recommendations

The National Transportation Safety Board, after review and analysis of the 1968 midair collision accident data, recommends that pilots of every rating make themselves aware of the increasing threat of midair collisions in civil aviation, to civil aviation, itself, to themselves, and to the lives and property of others.

In addition to reemphasizing and instilling the need for constant collision avoidance pilot awareness and alertness, certain basic common sense practices need to be reemphasized with regard to alcohol involvement and inadequate preflight preparation for flying in pairs or formation flying.

All pilots should become aware of and exercise every precaution against the midair collision potentials at controlled high density terminal arrival and departure areas, as well as at uncontrolled low-density traffic general aviation airports.

All pilots should renew their emphasis on well-disciplined, good and precise flying techniques and habits, (see Appendix 10 of this report); and should compensate for the inherent design restrictions to vision of the aircraft being flown.

Toward the prevention of midair collision accidents, the Safety

Board recommends that the pilot organizations such as the Air Line

Pilots Association, Aircraft Owners and Pilots Organization, Allied Pilots

Association, Flight Safety Foundation, National Air Taxi Conference,

tional Business Aircraft Association, National Pilots Association,

c. --and the numerous publishers of aviation periodicals each

make an effort to publicize the findings of this study of midair

collisions in 1968, as well as promoting, wherever appropriate,

the principles of collision avoidance awareness on the part of

pilots.

The Safety Board also recommends that the owners and operators of airports, and other responsible local, municipal, county and State authorities undertake to assure that VFR approach and departure traffic pattern procedures are established at every airport. Further, that such procedures be clearly identified and made known to pilots.

The Safety Board further recommends that the manufacturers of general aviation aircraft direct their attention to the need for increased visual conspicuity of small, as well as large, airplanes.

Notwithstanding these recommendations addressed to the aviation community, the National Transportation Safety Board recommends that the Administrator of the Federal Aviation Administration:

1. Undertake an educational program to make both pilots
and controllers more aware of the midair collision problem,
and to make pilots aware that most midair collisions occur
at or near airports in clear weather and in daylight hours.

- 2. Establish a continuing program to assure indoctrination and continuing awareness on the part of all pilots to the midair collision potential and avoidance techniques (i.e., "see and be seen" concept, descent, turn, and climb maneuvering techniques, etc.).
- 3. Examine more stringently all pilot applicants for their external cockpit vigilance, with particular attention to pilots who are tested for flight instructor ratings.
- 4. Provide special warning and guidance to pilots who are required by the nature of their operations to fly in pairs.
- 5. Inform all certificated flight instructors of the high statistical significance of their involvement in midair collisions.
- 6. Encourage all instructor pilots to notify the control tower operator, at airports where a tower is manned, regarding first solo flights, and require the tower operator to advise other traffic in the pattern about such flights.
- 7. Conduct detailed traffic flow studies for all high-volume general aviation controlled airports with a view to improving the VFR traffic flow techniques of the ATC personnel.
- 8. Designate climb and descent corridors for high-performance aircraft at high-density airports.

- Irrespective of the provisions contained in Part 91 of
 the Federal Aviation Regulations, establish standard entry,
 departure, and go-around procedures for each uncontrolled
 airport.
- 10. In cooperation with ESSA, develop and produce VFR approach and departure charts for selected airports with a high volume of traffic.
- 11. In addition to the requirements of Section 91.89 of Part 91 of the Federal Aviation Regulations, develop a requirement for the installation of surface pattern indicators (for day and night) at smaller airports which would define specific patterns, particularly the base leg and the final approach.
- 12. Reevaluate visual conspicuity standards for all civil aircraft.
- 13. Consider the establishment of requirements for the installation and day and night operation of high-intensity white flashing lights on all civil aircraft.
- 14. Support the expeditious development of low-cost Collision Avoidance Systems for all civil aircraft.

MIDAIR COLLISIONS 1968

	State State	<u>Date</u>		Place & State	Date
43	rla.	1/19/68	20.	Woodbury, Tenn.	7/31/68
	in, Ind.	2/18/68	21.	Rock River, Wyo.	8/1/68
ig a	madale, Wash.	2/24/68	22.	Lexington, Neb.	8/3/68
	n Jose, Calif.	2/25/68	23.	Milwaukee, Wis.	8/4/68
	Je nai, Alaska	3/16/68	24.	Chesapeake, Va.	8/6/68
Çî.	Puyallup, Wash.	3/24/68	25.	Cleburne, Texas	8/14/68
47.	St. Louis, Mo.	3/27/68	26.	Hayward, Calif.	8/14/68
. 8.	Columbus, Ohio	4/10/68	27.	Barre Plains, Mass.	8/18/68
9.	Kansas City, Mo.	4/10/68	28.	Calif. City, Calif.	8/24/68
1 0.	Pascagoula, Miss.	4/13/68	29.	Beaumont, Texas	8/28/68
11.	Bakersfield, Calif.	5/19/68	30.	Flushing, N.Y.	9/8/68
12.	Phoneton, Ohio	5/20/68	31.	Bitter Creek, Wyo.	9/16/68
13.	Norman, Oklahoma	5/30/68	32.	Hamilton, Ohio	9/15/68
14.	Denver, Colo.	6/12/68	33•	Miami, Fla.	10/5/68
15.	Indian Springs, Nev.	6/20/68	34.	Dixon, Calif.	10/24/68
16.	Reseda, Calif.	7/3/68	35•	Calipatria, Calif.	10/27/68
17.	Danbury, Conn.	7/7/68	36.	Santa Paula, Calif.	12/8/68
18.	Crows Landing, Calif.	7/16/68	37•	Sandwich, Ill.	12/8/68
19.	Wyandanch, N. Y.	7/28/68	38.	Rome, Georgia	12/20/68

Table 1

	COIV.	TOTALS		ŗ	?			m			+			ω '	
	DESCENT	DESCRIPT												T ,	1
	DAKE OFF INTEN	DAKE OFF INTEN	GMT(T)												
	CRUISE	TRAFFIC PATTERN CTRCLE													
	CRUISIE	DESCENT									1				
BY ACCIDENT	CRUISE	CLIMB											,		
PHASE OF OPERATION BY ACCIDENT	CRUISE	CRUTSE	1	1	1						П				7
PISA	LANDING LEVEL OFF	LANDING LEVEL OFF	3		1										
; ;	TRAPFIC	CINCLE T/P CIRCLE	τ							1					
	LANDING FINAL APPROACH	CRUTSE						1							
	LANDING FINAL APPROACE	LANDING FINAL APPROACH	1	1	3	T		T,		1					1
AIRCRAPT TYPE OF WING			иси иси 6	TOV TOV 2	LOW HIGH 5	HIGH HIGH 1	TON TON	LOW HIGH 2	EDCH RICH	LOW LOW 2	LOW HIGH 2	HICH HIGH 1	NOI NOI	LOW HIGH 1	нгси итси 3
CONVERGENCE ANGLE			ed I	0-10			11-30		1	31-50°	1		51-70"	1	22

TARIF 2

NOTE: 8 Accidents omitted from this study for lack of evidence.

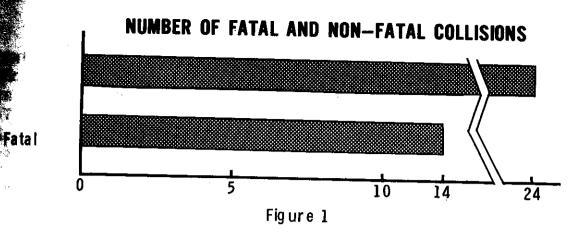
171-180

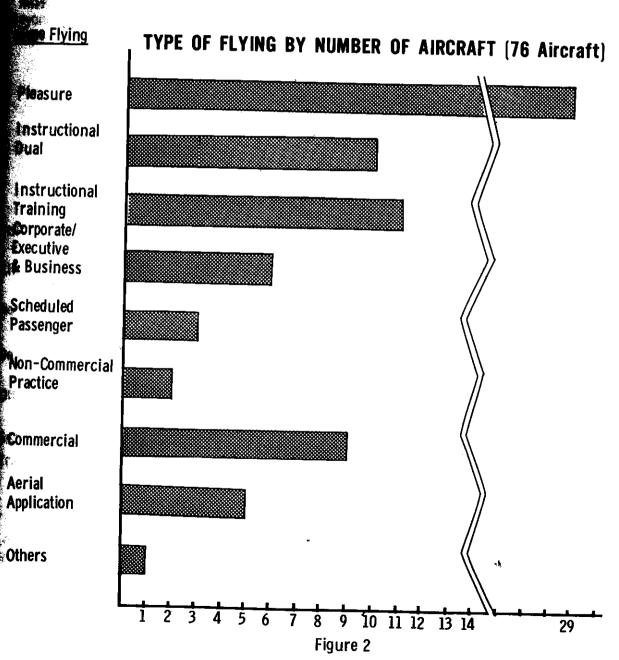
91-110

FON

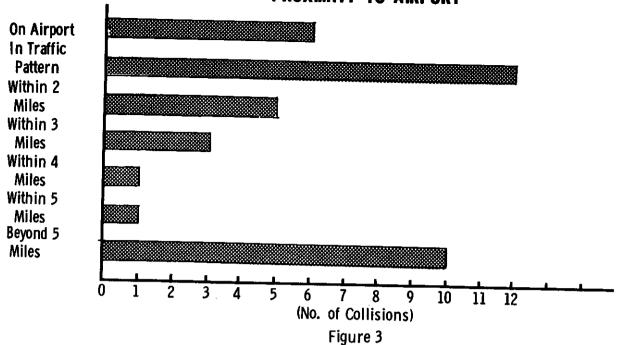
MOT

71-90°

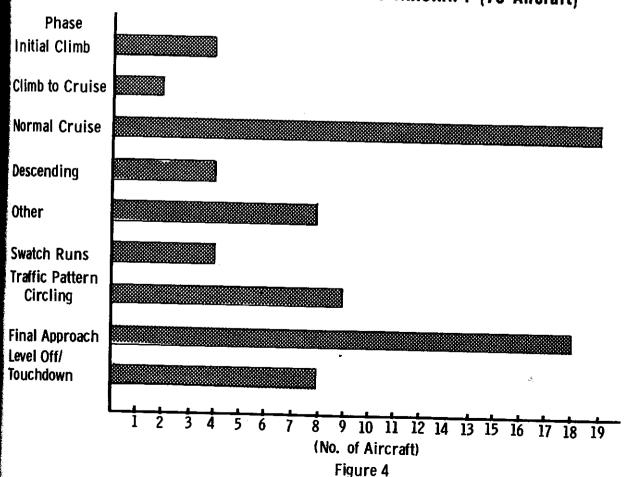




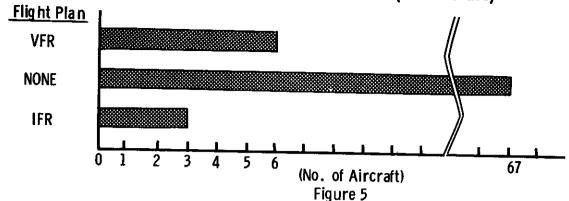




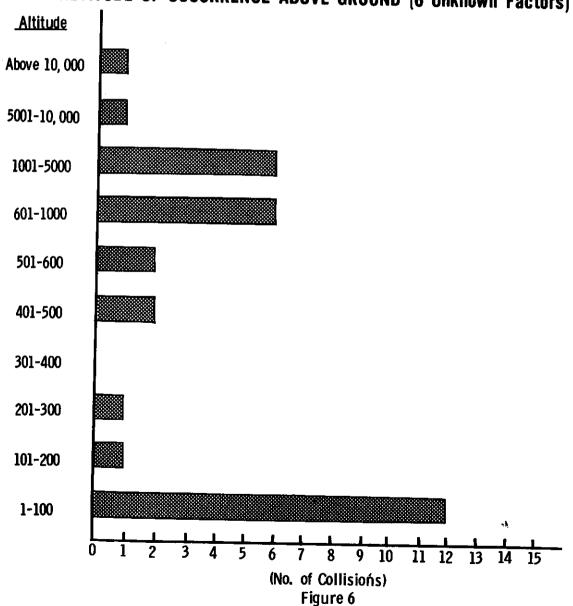
PHASE OF OPERATIONS BY AIRCRAFT (76 Aircraft)



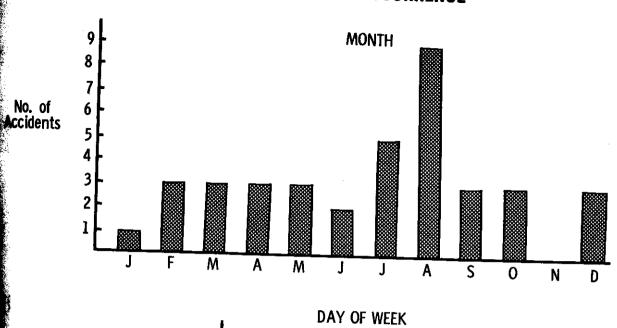
FLIGHT PLAN BY AIRCRAFT (76 Aircraft)



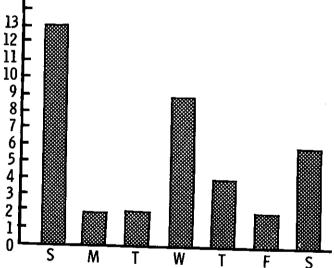
ALTITUDE OF OCCURRENCE ABOVE GROUND (6 Unknown Factors)



TIME OF OCCURRENCE







No. of Accidents

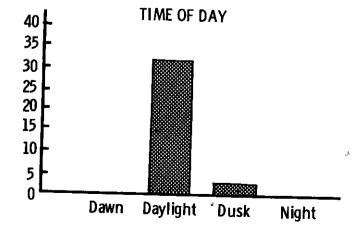


Figure 7

- 41 -

PILOT TIME

Hours

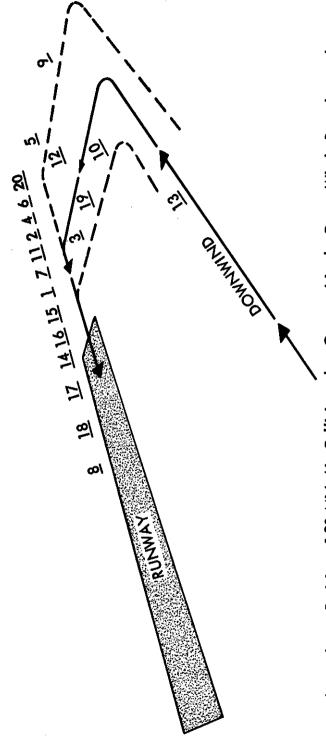
Total Time

ime In Type

1-90	100-199	200-299	300-399	1000 or More
14	8	7	16	25
32	11	4	8	7

Pilot at Controls. Total Time & Time-in Type (Total Time of 6 not known. Time-in type of 14 not known)

Figure 8



Approximate Position of 20 Mid-Air Collisions that Occurred in the Down-Wind, Base-Leg and Final Approach.

Docket Numbers

15. 3-2378	3-3983 3-4004	3-4803 (Flight Instructor)	3-4836 (" ")	
15.		<u>.</u> 6	20.	
8. 3-2044 (Flight Instructor)	3-2054 (Flight Instructor)	=	=	
(Fligh	(Fligh	- 	<u>-</u>	
3-2044	3-2054 (3-2123 (3-2125	3-2167
œ o		12.	<u>.</u>	14.
	3-0566 (Flight Instructor)	=		
	(Fligh	<u>=</u> /		
3-0136	3-0566	3-1996 (3-2000	3-2001
~	i n ∀	; ;	6	ζ.

Figure 9

NATIONAL TRANSPORTATION SAFETY BOARD

DEPARTMENT OF TRANSPORTATION

WASHINGTON, D. C. 20591

BRIEFS OF ACCIDENTS

INVOLVING

U. S. CIVIL AVIATION

MID-AIR COLLISONS

1968

- 47	DOCKET DATE	DATE	DOCKET DATE LOCATION	AIRCRAFT DATA	INJURIES F S M/N	FLIGHT N PURPOSE	AIRCRAFT DATA INJURIES FLIGHT PILOT DATA F S M/N PURPOSE	
-	E-0008	11ME - 0850 TYPE OF ACCIDENT	E-0008 8/14/68 CLEBURNE, TEX TIME, - 0850 TYPE OF ACCIDENT COLLINION WITH ABSERTED		CR- 1 0 0 PX- 0 0 0 OT- 1 0 0	CR- 1 0 0 MISCELLANEOUS PX- 0 0 0 FERRY OT- 1 0 0 PHASE OF OPERATION TABLE CELT MODERA	CESSNA 188A CR- 1 0 0 MISCELLANEOUS CONMERCIAL, AGE 40, 4000 H-K962X PX- 0 0 0 FERRY TOTAL HOURS, 150 IN TYPE, DAMAGE -DESTROYED 0T- 1 0 0 PHASE OF OPERATION	00.

UNDER INVESTIGATION

FIRE AFTER IMPACT REMARKS- OPERATOR, ACFT FERRYING CO. COLUMBIA, S.A. REGISTRY. DEMOLISHED BY FIRE AND IMPACT.

COMMERCIAL, AGE 26, 1250 FOTAL HOURS, 60 IN TYPE.	
_ ,-	PHASE OF OPERATION INFLIGHT NORMAL CRUISE
CESSNA 188A H-K965X DAMAGE -DESTROYED	RCRAFT BOTH IN FLIGHT
E-0008 8/14/68 CLEBURNE,TEX TIME - 0850	TYPE OF ACCIDENT COLLISION WITH AIRCRAF

UNDER DIVERTICATION

FIRE AFTER IMPACT REMARKS- OPERATOR,ACFT FERRYING CD. COLUMBIA,S.A.REGISTRY.DEMOLISHED BY FIRE AND IMPACT.

DOCKET	DATE LOCATION	AIRCRAFT DATA	INJURIES	FLIGHT	PILOT DATA
1-0051	6/12/68 NR.DENVER,COLO TIME - 0719 TYPE OF ACCIDENT COLLISION WITH AIRCRAFT	CESSNA 337 N-2212X DAMAGE -SUBSTANTIAL BOTH IN FLIGHT	CR- 0 0 PX- 0 0 65 PHASE	MISCELLANEOUS POLICE PATROL OF OPERATION FIGHT DESCENDING	COMMERCIAL, AGE 36, 5000 TOTAL HOURS, 200 IN TYPE.
	PROBABLE CAUSE MISCELLANEOUS-PERSONNEL - REMARKS- LANDED AT ARPI,NO FI	- PILOT OF OTHER AIRCRAFT FURTHER DAMAGE.	CRAFT		
1-0021	6/12/68 NR.DENVER,COLO TIME - 0719 OPERATOR - UNITED AIR LINES, TYPE OF ACCIDENT COLLISION WITH AIRCRAFT	BOEING 727 N-7086U DAMAGE -MINDR INC.	CR- 0 0 7 PAS PX- 0 0 55 OT- 0 0 2 PHASE OF OP	0 7 PASSG S-D 0 55 0 2 PHASE OF OPERATION INFLIGHT DESCENDING	AIRLINE TRANSPORT, AGE 43, 16000 TGTAL HOURS, 2400 IN TYPE.
	PROBABLE CAUSE PILOT IN COMMAND - FAIL REMARKS- LANDED AT ARPT,NO F	LED TO SEE AND AVOID OTHER AIRCRAFT FURTHER DAMAGE.	THER AIRCRAFT		
3-0136	1/19/68 MIAMI:FLA TINE - 1715 TYPE OF ACCIDENT COLLISION WITH AIRCRAFT	CESSNA 150 N-3940J DANAGE -SUBSTANTIAL BOTH IN FLIGHT	CR- PX-0 OT-0	0 1 NONCOMMERCIAL 0 0 PRACTICE 0 1 PHASE OF OPERATION LANDING FINAL APPROACH	PRIVATE, AGE 31, 104 TOTAL HOURS, 7 IN TYPE.
•	PROBABLE CAUSE PILOT IN COMMAND - FAILED PILOT IN COMMAND - FAILED FACTOR HISCELLANEOUS-PERSONNEL - REMARKS - N3940J STRUCK N7018X.		AND AVOID OTHER AIRCRAFT ON APPROVED PROCEDURES, DIRE IF OTHER AIRCRAFT HAD FLOWN RT HAND PATTERN 1	TO SEE AND AVOID OTHER AIRCRAFT TO FOLLOW APPROVED PROCEDURES, DIRECTIVES, ETC PILOT OF OTHER AIRCRAFT N3940J HAD FLOWN RT HAND PATTERN TO UNCONTROLLED ARPT.	•
3-0136	1/19/68 MIAMI:FLA TIME - 1715 TYPE OF ACCIDENT COLLISION WITH AIRCRAFT	CESSNA 150 N-7018X DAMAGE -MINOR BQTH IN FLIGHT	CR- 0 0 1 IN PX- 0 0 0 DT- 0 0 1 PHASE OF UP LANDING	INSTRUCTIONAL TRAINING OF OPERATION DING FINAL APPROACH	STUDENT, AGE 42, 31 TOTAL HOURS, ALL IN TYPE.
	PROBABLE CAUSE PILOT IN COMMAND - FAILED TO SEI FACTOR MISCELLANEOUS-PERSONNEL - PILOT REMARKS- CESSMA M3940J STRUCK, CESSMA	FAILED TO SEE AND AVOID OTHER IEL - PILOT OF OTHER AIRCRAFT TRUCK, CESSNA, NJOÌAX,N3940J. MAD	AND AVOID OTHER AIRCRAFT JF OTHER AIRCRAFT (ZOIBX-N3940.) HAD ELOWN RI-	E AND AVOID OTHER AIRCRAFT OF OTHER AIRCRAFT N7018X-N3940J HAD FLOWN RI-HAND PATTERN ON HIS APPROACH TO FINAL-	APPROACH TO FINAL.

				F S M/N	PURPOSE	TEON BRAK
	3-0550	2/18/68 FRANKLIN,IND TIME - 1715 '	STINSON 108 N-8905K NAMAGE DESTROYED	CR- 1 0 0	NONCOMMERCIAL PLEASURE	PRIVATE, AGE 36, 164 TOTAL HOURS, 33 IN TYPE.
		TYPE OF ACCIDENT COLLISION WITH AIRCRAFT BO		PHASE LAN) OF OPERATION UDING FINAL APPROACH	
		PROBABLE CAUSE MISCELLANEOUS-PERSONNEL - PILOT OF OTHER AIRCRAFT REMARKS- N9516M OVERTOOK AND DESCENDED INTO N8905K.N9516W FLEW NORMAL PATTERN.N8905K MADE	PILOT OF OTHER AIRC DESCENDED INTO N8905K	AIRCRAFT 905K.n9516W FLEW N	ORMAL PATTERN.N8905K	MADE LONG, LOW APPROACH
	3-0550	2/18/68 FRANKLIN,1ND TIME - 1715	a 19	CR- 1 0 0 PX- 0 0 0	INSTRUCTIONAL TRAINING	STUDENT, AGE 25, 18 TOTAL HOURS, ALL IN TYPE,
		TYPE OF ACCIDENT COLLISION WITH AIRCRAFT BC	DAMAGE -DESTROYED BOTH IN FLIGHT	1 1 C PHASE LAN	Δ.	
		PROBABLE CAUSE PILOT IN COMMAND - FAILED TO SEE AND AVOID OTHER AIRCRAFT REMARKS- N9516W OVERTOOK AND DESCENDED INTO N8905K.N9516W FLEW) TO SEE AND AVOID DI DESCENDED INTO N8905K	HER AIRCRAFT •N9516W FLEW N	GRMAL PATTERN.N8905K	NORMAL PATTERN.N8905K MADE LONG,LOW APPROACH
40	3-0566	2/25/68 SAN JOSE,CALIF TIME - 1826	ധര	CR- 0 0 1	NDNC OMMERCIAL Pleasure	PRIVATE, AGE 37, 267 TOTAL HOURS, 76 IN TYPE,
_		TYPE OF ACCIDENT COLLISION WITH AIRCRAFT 80	DAMAGE -SUBSTANTIAL BOTH IN FLIGHT	°E	PERATION TRAFFIC PATTERN-) ^r
	ند	PROBABLE CAUSE PILOT IN COMMAND - FAILED	TO SEE AND AVDID OTHER	HER AIRCRAFT		
	3-0566	2/25/68 SAN JOSE,CALIF TIME - 1826		CR- 0 0 1	NONCOMMERCIAL PLEASURE	CONMERCIAL, FL. INSTR., AGE 34, 672 TOTAL HOURS, 600 IN TYPE,
		TYPE OF ACCIDENT COLLISION WITH AIRCRAFT BO	DAMAGE -SUBSTANTIAL BOTH IN FLIGHT	O DHASE	PERATION TRAFFIC PATTERN	CIRCLING
		PROBABLE CAUSE HISCELLANEOUS-PERSONNEL -	PILOT OF OTHER AIRCRAFT	IAFŦ		
	3-0951	3/24/68 PUYALLUP,WASH TIME - 1136	172	CR- 1 0 0 PX- 1 0 0	NONCONNERCIAL PLEASURE	PRIVATE, AGE 48, 692 TOTAL HOURS, 162 IN TYPE.
		TYPE OF ACCIDENT COLLISION WITH AIRCRAFT BO	DAMAGE -DESTROYED BOTH IN FLIGHT	2 0 PHASE LA	OPERATION 1G FINAL APPROACH	
		PROBABLE CAUSE PILOT IN COMMAND - FAILED MISCELLANEOUS-PERSONNEL - FACTOR	TO SEE AND AVOID OTHER PILOT OF OTHER AIRCRAFT	ER AIRCRAFT Aft		
		PILOT IN COMMAND - FAILED TO FOLLOW APPROVED PROCEDURES, DIRECTIVES, ETC REMARKS- ROTH PLTS FLEW OPPOSITE DATTEDNS EATLEN TO SEE EAST ATTENDATED	TO FOLLOW APPROVED P	ROCEDURES, DIRE	CTIVES*ETC	:::::::::::::::::::::::::::::::::::::::

DOCKET	DATE LOCATION	AIRCRAFT DATA	INJURIES F S M/N	FLIGHT PURPOSE	PILOT DATA	
3-0951	3/24/68 PUYALLUP,WASH TIME - 1136 TYPE OF ACCIDENT COLLISION WITH AIRCRAFT BO	CESSNA 150H N-6543S DAMAGE -DESTROYED OTH IN FLIGHT	CR- 2 0 0 IN PX- 0 0 0 OT- 2 0 0 PHASE OF OP LANDING		CONMERCIAL FL. INSTR., 2319 TOTAL HOURS, 41	. AGE 35.
	PROBABLE CAUSE MISCELLANEOUS-PERSONNEL - PILOT IN COMMAND - FAILED FIRE AFTER IMPACT REMARKS- BOTH PLTS FLEW OPPOSI	RSONNEL - PILOT OF OTHER AIRCRAFT)) - FAILED TO SEE AND AVOID OTHER AIRCRAFT FLEW OPPOSITE PATTERNS,FAILED TO SEE EACH OTHER.N6069A FLEW IMPROPER PATTERN	RAFT IHER AIRCRAFT 10 SEE EACH OTHEI	R.N6069A FLEW IMPRO	PER PATTERN.	
3-1055	3/16/68 NR.KENAI,ALAS TIME - 1040 TYPE OF ACCIDENT COLLISION WITH AIRCRAFT B	PIPER PA-18 N-232T DAMAGE -DESTROYED BOTH IN FLIGHT	CR- 1 0 0 NOI PX- 1 0 0 1 OT- 2 0 0 PHASE OF OPI	0 0 NONCOMMERCIAL 0 0 PLEASURE 0 0 PHASE OF OPERATION INFLIGHT NORMAL CRUISE	PRIVATE, AGE 27, HOURS, 95 IN TYPE.	AGE 27, 104 TOTAL N TYPE.
	PROBABLE CAUSE PILOT IN COMMAND — FAILE MISCELLANEOUS-PERSONNEL — FIRE AFTER IMPACT REMARKS— BOTH PLTS FAILED TO	FAILED TO SEE AND AVOID OTHER AIRCRAFT EL - PILOT OF OTHER AIRCRAFT D TO SEE EACH OTHER.N232T WHITE,BLENDED WITH SNOW COVERED BACKGROUND.	THER AIRCRAFT CRAFT WHITE, BLENDED W	ITH SNOW COVERED BA	CK GROUND.	
3-1055	3/16/68 NR.KENAI.ALAS TIME - 1040 TYPE OF ACCIDENT COLLISION WITH AIRCRAFT	PIPER PA-18 N-37921 DAMAGE -DESTROYED BOTH IN FLIGHT	CR- 1 0 0 NOP PX- 1 0 0 6 DT- 2 0 0 PASE OF OPI) NONCOMMERCIAL) PLEASURE) OF OPERATION =-LIGHT NORMAL CRUISE	PRIVATE, AGE 27, HOURS, 99 IN TYPE.	AGE 27, 138 TOTAL N TYPE.
	PROBABLE CAUSE PILOT IN COMMAND - FAIL! MISCELLANEOUS-PERSONNEL - REMARKS- BOTH PLTS FAILED TO	FAILED TO SEE AND AVOID OTHER AIRCRAFT IEL - PILOT OF OTHER AIRCRAFT ID TO SEE EACH OTHER.N232T WHITE,BLENDED WITH SNOW COVERED BACKGROUND.	THER AIRCRAFT CRAFT WHITE, BLENDED V	IITH SNOW COVERED B/	C K GROUND .	
3-1578	5/19/68 NR.BAKERSFIELD,CAL TIME - 1415 TYPE OF ACCIDENT COLLISION WITH AIRCRAFT	IF AERONCA 7AC N-3537E DAMAGE -DESTROYED BOTH IN FLIGHT	CR- 1 0 0 NO PX- 1 0 0 OT- 2 0 0 PHASE OF OP INFLIGHT	0 O NONCOMMERCIAL 0 O PLEASURE 0 O PLEASURE PHASE OF OPERATION INFLIGHT OTHER	PRIVATE, AGE 26, HOURS, 135 IN TYPE	AGE 26, 203 TOTAL IN TYPE.
	PROBABLE CAUSE PROBABLE CAUSE FACTOR FACTOR PILOT IN COMMAND - ATTEMPTED OPERATION BE MISCELLANGOUS ACTS, CONDITIONS - ALCOHOLIC ALSCENANTION ELYING-BLOOD ALCOHOL LEVEL	MISJUDGED CLEARANCE ATTEMPTED OPERATION BEYOND EXPERIENCE/ABILITY LEVEL NDITIONS - ALCOHOLIC IMPAIRMENT OF EFFICIENCY AND JUDGEMENT 10.81.000 ALCOHOL LEVEL 63 MG PCT., DVER TOOK, STRUCK PIPER N57253.	ID EXPERIENCE/AB PAIRMENT OF EFF MG PCT_DVER TOO	ILITY LEVEL ICIENCY AND JUDGEME K•STRUCK PIPER NS72	¥F 53.	

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DOCKET	DATE LOCATION	AIRCRAFT DATA	INJURIES	S M/N	FL IGHT PURPOSE	PILOT DATA	
3-2000	4/10/68 KANSAS CITY,MO PI TIME - 1730 DA TYPE OF ACCIOENT COLLISION WITH AIRCRAFT BOTH	PIPER PA-24 N-9210P DAMAGE -DESTROYED H IN FLIGHT	CR 1 0 PX 0 0 DT 0 0	0 NO 0 1 1 SE OF OP	0 0 NONCOMMERCIAL 0 0 BUSINESS 0 1 PHASE OF OPERATION INFLIGHT DESCENDING	PRIVATE, AGE 51, HOURS, 152 IN TYPE.	2323 TOTAL
	PROBABLE CAUSE PILOT IN COMMAND - FAILED TO MISCELLANGOUS-PERSONNEL - PI FACTOR AIRPORT CONDITIONS - OTHER REMARKS- FCTY SMOKE OBSCURS TWR N	TO SEE AND AVOID OTHER PILLOT OF OTHER AIRCRAFT VIEW OF RNWY 35 X-WIND	HER AIRCRAFT RAFT WIND TRFC. P	IFT PLT NO	DTHER AIRCRAFT RCRAFT X-WIND TRFC. PLT N6386S FAILED TO SEE,AVOID N9210P.	E. AVGID N9210P.	
3-2001	8/18/68 BARRE PLAINS,MASS TA TIME - 1415 DA TYPE OF ACCIDENT COLLISION MITH AIRCRAFT BOTH	TAYLORCRAFTBC12-D N-43118 DAMAGE -SUBSTANTIAL BOTH IN FLIGHT	CR- 0 PX- 0 OT- 0	1 NO 0 0 1 1 1SE OF OP	0 1 NONCOMMERCIAL 0 0 PLEASURE 0 1 PHASE OF OPERATION LANDING FINAL APPROACH	COMMERCIAL, AGE HOURS, 83 IN TYPE.	AGE 28, 201 TOTAL YPE.
	PROBABLE CAUSE PILOT IN COMMAND - FAILED TO MISCELLANEOUS-PERSONNEL - P:	TO SEE AND AVOID OTHER AIRCRAFT PILOT OF OTHER AIRCRAFT	IHER AIRCRA	.F.T			
3-2001	8/18/68 BARRE PLAINS,MASS PITINE - 1415 TYPE OF ACCIDENT COLLISION WITH AIRCRAFT BOTH	PIPER PA-22 CR- N-59451 DAMAGE -SUBSTANTIAL OT- H IN FLIGHT	000	1 IN 0 0 1 I ASE OF OP	0 1 INSTRUCTIONAL 0 0 TRAINING 0 1 PHASE OF OPERATION LANDING FINAL APPROACH	STUDENT, AGE 42, HOURS, 33 IN TYPE.	43 TOTAL
,	PROBABLE CAUSE PILOT IN COMMAND - FAILED TO MISCELLANEOUS-PERSONNEL - P	TO SEE AND AVOID OTHER AIRCRAFT PILOT OF OTHER AIRCRAFT	THER AIRCRU	PF			
3-2044	7/7/68 DANBURY,CONN P TIMĘ – 1649 D TYPE OF ACCIDENT COLLISION WITH AIRCRAFT BOTH	PIPER PA-28 N-4397J DAMAGE -SUBSTANTIAL H IN FLIGHT	CR- 0 PX- 0 OT- 0	0 1 0 0 0 2 PHASE OF	1 INSTRUCTIONAL PRI'S SOLO HOU! SOLO SOLO HOU! SE OF OPERATION LANDING LEVEL OFF/TOUCHDOWN	VATE, RS, 105	AGE 18, 133 TOTAL IN TYPE.
	PROBABLE CAUSE PILOT IN COMMAND - FAILED TO FACTOR MISCELLANEGUS ACTS, CONDITIONS REMARKS- N4397J LANDED ON N51622.	TO SEE AND AVOID OTHER AIRCRAFT 1S - SUNGLARE 22. COLLIDED DURING LNDG FLARE, BOTH AIRBORNE.	THER AIRCR LNDG FLAR	AFT E. BOTH	AIRBORNE.	:	

BRIEFS OF ACCIDENTS

DOCKET	DATE	LOCATION	AIRCRAFT DATA	INJURIES F S M/N	FL 16HT	PILOT DATA
3-2044	TINE - 1649 TYPE OF ACCI	NBURY,CONN DENT MITH AIRCRAFT	PIPER PA-22 N-51622 DAMAGE -SUBSTANTIAL BOTH IN FLIGHT	CR - 0 PX - 0 0T - 0	0 2 INSTRUCTIONAL COM 0 0 DUAL 151 0 1 PHASE OF OPERATION LANDING LEVEL OFF/TOUCHDOWN	COMMERCIAL, FL. INSTR., AGE 21, 1510 TOTAL HOURS, 350 IN TYPE, DOWN
	PROBABLE C. MISCELL/ REMARKS - N4	PROBABLE CAUSE MISCELLANEOUS-PERSONNEL - REMARKS- N4397J LANDED ON N53	PROBABLE CAUSE MISCELLANEOUS-PERSONNEL - PILOT OF OTHER AIRCRAFT REMARKS- N4397J LANDED ON N5162Z. COLLIDED DURING LANDING FLARE,	RAFT LANDING FLARE	:• BOTH AIRBORNE•	
3-2051	7/16/68 CRI TIME - 0600 TYPE OF ACCI	DWS LNDNG,CALIF DENT WITH AIRCRAFT	PIPER PA-25 N-45747 DAMAGE -DESTROYED BOTH IN FLIGHT	CR- 0 0 1 PX- 0 0 0 OT- 0 0 1 PHASE	1 COMMERCIAL COM 0 AERIAL APPLIC TOT 1 SE OF OPERATION INFLIGHT STARTING SWATH RUN	COMMERCIAL, AGE 35, 1218 TOTAL HOURS, 314 IN TYPE, RUN
	PROBABLE CA PILOT IN REMARKS- PL	CAUSE IN COMMAND - EXERCI PLT OF N4574Y FAILED	SED POOR JUDGHENT TO ASSURE FIELD WAS	CLEAR OF	OTHER ACFT BEFORE STARTIN	STARTING SWATH RUN.
3-2051	7/16/68 CROWS TIME - 0600 TYPE OF ACCIDENT COLLISION MIT	LNDNG,CALIF H AIRCRAFI	FAIRCHILD WM-62C N-68444 DAMAGE -\$UBSTANTIAL BOTH IN FLIGHT	CR- 0 PX- 0 0T- 0	O 1 COMMERCIAL O 0 AERIAL APPLIC O 1 PHASE OF OPERATION INFLIGHT SWATH RUN	COMMERCIAL, AGE 33, 2400 TOTAL HOURS, 2100 IN TYPE.
	PROBABLE CAUSE MISCELLANEO REMARKS - SPRAY	PROBABLE CAUSE MISCELLANEOUS-PERSONNEL:- REMARKS- SPRAYING ALFALFA.PLT	PILOT OF OF N4574Y	RAFT Swath Run Beli	OTHER AIRCRAFT INITIATED SWATH RUN BELOW POWER LINE, OUT OF RISING SUN.	RISING SUN.
3-2052	7/28/68 WYANDA TIME - 1926 TYPE OF ACCIDENT COLLISION WIT	NCH, NY. H AIRCRAFT	PIPER PA-23 N-5980Y DAMAGE -DESTROYED BOTH IN FLIGHT	CR- 1 0 0 PX- 3 0 0 UT- 4 0 0 LANI	0 0 NONCOMMERCIAL PRIVATE, 0 0 PLEASURE HOURS, 3 0 0 PHASE OF OPERATION LANDING TRAFFIC PATTERN-CIRCLING	PRIVATE, AGE 46, 456 TOTAL HOURS, 35 IN TYPE. -CIRCLING
	PROBABLE CAUSE PILOT IN COMMA MISCELLANEOUS— FIRE AFTER IMPACT	BABLE CAUSE PILOT IN COMMAND FAILED MISCELLANEOUS-PERSONNEL E AFTER IMPACT	ED TO SEE AND AVOID OTHER AIRCRAFT PILOT OF OTHER AIRCRAFT	HER AIRCRAFT Raft		

BRIEFS OF ACCIDENTS

			D CLUING			4044004004004104444444
DOCKET	DATE	LOCATION	AIRCRAFT DATA I	INJURIES F S M/N	FLIGHT PILL PURPOSE	PILOT DATA
3-2052	7/28/68 WYANDANCH,NY TIME - 1926 TYPE OF ACCIDENT COLLISION WITH AIRCRAFT	NCH+NY H AIRCRAFT	PIPER PA-28 N-7162R DAMAGE -DESTROYED BOTH IN FLIGHT	CR- 1 0 0 NONCONMER PX- 3 0 0 PLEASUR OT- 4 0 0 PRATION PHASE OF OPERATION LANDING TRAFFI	CIAL E C PATTERN-	PRIVATE, AGE 25, 47 TOTAL HOURS, 5 IN TYPE. -CIRCLING
3-2054	PROBABLE CAUSE PILOT IN COMMAND FAI MISCELLANEOUS-PERSONNEL 5/20/68 PHONETON*OHIO	AUSE N COMMAND - FAILE ANEOUS-PERSONNEL - PHONETON, OHIO	ED TO SEE AND AVOID OTHER AIRCRAFT PILOT OF OTHER AIRCRAFT CESSNA 120 CF- 2 0 CF- 2 0 CF- 2 0	00	NGNCOMMERCIAL AGE 6 Pleasure type.	AGE 63, 355 TOTAL HOURS, 158 IN TYPE.
	TYPE OF ACCIDENT COLLISION WITH AIRCRAFT	T TH AIRCRAFT	DAMAGE -DESTROYED BOTH IN FLIGHT	SE I) OF OPERATION -LIGHT NORMAL CRUISE	
	PROBABLE CAUSE PILLOT IN COMMAND FACTOR PILOT IN COMMAND REMARKS-'N3773Q,TURNING		FAILED TO SEE AND AVOID OTHER AIRCRAFT INADEQUATE PREFLIGHT PREPARATION AND/OR ON DOWNWIND LEG, WAS STRUCK BY CF-KUE IN		PLANNING RT REAR.CF-KUE PLT UNFAMILIAR WITH TRAFFIC.	ILIAR WITH TRAFFIC.
3-2054	5/20/68 TIME - 12 TYPE OF AC	PHONETON,OHIO :00 :CIDENT ION WITH AIRCRAFT	BEECH 95-C55 N-37739 DAMAGE -DESTROYED BOTH IN FLIGHT	CR- 1 0 0 NC PX- 0 0 0 OT- 2 0 0 PHASE OF DE LANDING	NCOMMERCIAL CORP/EXEC SERATION TRAFFIC PATTERN-	COMMERCIAL, FL. INSTR., AGE 38, 7941 TOTAL HOURS, 572 IN TYPE. CIRCLING
	PROBABLE CAUSE Miscellaneous-Personnel Remarks- N37739,Turning on	IS-PERSONNEL	- PILOT OF OTHER AIRCRAFT DOWNWIND LEG HAS STRUCK BY	KCRAFT SK BY CF-KUE IN RI	AAFT BY CF-KUE IN RT REAR.CF-KUE PLT UNFAMILIAR WITH TRAFFIC.	ILIAR WITH TRAFFIC.
3-2055	6/20/68 INDIAN SPRNGS,NEV TIME - 0816 TYPE OF ACCIDENT COLLISION WITH AIRCRAFT	IN SPRNGS+NEV UT ITH AIRCRAFT	MDDNEY M20A N-1092B DAMAGE -DESTROYED BOTH IN FLIGHT	CR- 1 PX- 4 OT- 0	NCOMMERCIAL PLEASURE ERATION NORMAL CRUISE	PRIVATE, AGE 45, 668 TOTAL Hours, 125 in type.

The second second

PROBABLE CAUSE MISCELLANEGUS-PERSONNEL - PILOT OF OTHER AIRCRAFT

TYPE OF ACCIDENT COLLISION MITH AIRCRAFT BOTH IN FLIGHT

DOCKET	DATE	LUCATION	AIRCRAFT DATA	INJURIES	ES F S M/N	FL IGHT PURPOSE	PICOT DATA
3-2055	6/20/68 INDIAN SPRNGS,NE TIME - 0816 TYPE OF ACCIDENT COLLISION WITH AIRCRAFT	 >	REPUBLIC F-105 U-SAF624422 DAMAGE -MINDR BOTH IN FLIGHT	CR- 0 PX- 0 0T- 5	0 0 2 NO 0 0 0 5 0 0 PHASE OF OP INFLIGHT	NONCOMMERCIAL PRACTICE OPERATION GHT NORMAL CRUISE	AGE UNKNOWN, UNKNOWN TOTAL HOURS, UNKNOWN IN TYPE.
	PROBABLE CAUSE PILOT IN COMMAND - FAILED REMARKS- LANDED SAFELY AT USAF	MAND - FAILED SAFELY AT USAF) TO SEE AND AVOID OTHER AIRCRAFT 8 BASE.	THER AIR	CRAFT		
3-2056	8/6/68 CHESAPEAKE,VA TIME - 1335 TYPE OF ACCIDENT COLLISION WITH AINCRAFT	EAKE,VA H AINCRAFT	PIPER PA-22 N-59292 DAMAGE -DESTROYED BOTH IN FLIGHT	CR- PX- 0 0T- 1	0 0 IN 0 0 0 0 0 0 PHASE OF OP INFLIGHT	INSTRUCTIONAL TRAINING OF OPERATION LIGHT NORMAL CRUISE	STUDENT, AGE 16, 27 TOTAL HOURS, 25 IN TYPE.
•	PROBABLE CAUSE MISCELLANEOUS-PERSONNEL	:-PERSONNEL -	PILOT OF OTHER AIRCRAFT	CRAFT			
3-2056	8/6/68 CHESAPEAKE,VA TIME - 1335 TYPE OF ACCIDENT COLLISION MITH AIRCRAFT	EAKE, VA H AIRCRAFT	CESSNA T-210H N-6908R DAMAGE -DESTROYED BOTH IN FLIGHT	CR- PX- OT-1	0 0 0 0 0 0 PHASE OF INFLIG	NONCOMMERCIAL CORP/EXEC OF OPERATION LIGHT CLIMB TO CRUISE	COMMERCIAL, AGE 30, 407 TOTAL Hours, 94 in Type.
	PROBABLE CAUSE PILOT IN COMMAND - FIRE AFTER IMPACT	AND - FAILED:	TO SEE AND AVGID OTHER AIRCRAFT	THER AIR	CRAFT		
3-2122	8/3/68 LEXINGTON.NEBR TIME - 1955 TYPE OF ACCIDENT COLLISION WITH AIRCRAFT	TON:NEBR H AIRCRAFT BO	AERONCA 7AC N-2966E DAMAGE -DESTROYED TH IN FLIGHT	0 - TO 0	1 0 CO 0 0 2 1 PHASE OF OL LANDING	1 O COMMERCIAL 0 O AERIAL APPLIC 2 1 PHASE OF OPERATION LANDING FINAL APPROACH	COMMERCIAL,FL.INSTR., AGE 54, 15550 TOTAL HOURS, 1000 IN TYPE.
	PROBABLE CAUSE PILOT IN COMMAND - FAILED TO MISCELLANEOUS-PERSONNEL - PI FACTOR MISCELLANEOUS ACTO. COMPITIONS	AND - FAILED -PERSONNEL -	TO SEE AND AVOID OTHER AIRCRA PILOT OF OTHER AIRCRAFT US	THER AIRC	AIRCRAFT		

DOCKEY	AVA.	ATRCKAF F DAYA " IN.	INJURIES F S M/N	FLIGHT PURPOSE	PILOT DATA
3-2125	7/3/68 RESEDA,CALIF TIME - 1541 TYPE OF ACCIDENT COLLISION WITH AIRCRAFT BOT	68 STANTIAL	CR- 0 0 2 PX- 0 0 0 OT- 2 0 0 PHASE OF	2 NONCOMMERCIAL AIRLINE 0 CORP/EXEC 15391 TO 0 SE OF OPERATION LANDING TRAFFIC DATTEDA_CYDC1 TAC	AIRLINE TRANSPORT, AGE 45, 15391 TOTAL HOURS, 285 IN TYPE.
	PROBABLE CAUSE PILOT IN COMMAND - FAILED TO SEE MISCELLANEOUS-PERSONNEL - PILOT O REMARKS- BOTH PLTS FAILED TO SEE EACH	TO SEE AND AVOID OTHER AIRCRAFT PILOT OF OTHER AIRCRAFT E EACH OTHER. N8394H LANDED SAF	R. AIRCRAFT FT .ANDED SAFELY	ED TO SEE AND AVOID OTHER AIRCRAFT PILOT OF OTHER AIRCRAFT SEE EACH OTHER. N8394H LANDED SAFELY,N7188S CRASHED IN RESIDENTIAL AREA.	r cincling RESIDENTIAL AREA.
3-2166	RDCK RIVER, WYO 1920 ACCIDENT LION WITH AIRCRAFT BOTA	BEECH K35 CR- N-3069C PX- DAMAGE -DESTROYED OT- H IN FLIGHT	NO4	O O NONCOMMERCIAL O O PLEASURE O O PHASE OF OPERATION INFLIGHT NORMAL CRUISE	COMMERCIAL, AGE 51, 500 TOTAL HOURS, UNKNOWN IN TYPE,
	PROBABLE CAUSE PILOT IN COMMAND - FAILED TO SEE AND AVOID OTHER AIR: MISCELLANEOUS-PERSONNEL - PILOT OF OTHER AIRCRAFT MISSING AIRCRAFT-LATER RECOVERED REMARKS- RECOVERY DATE-8/2/68.RIGHT WINDOW CURTAIN CLOSSES	TO SEE AND AVOID OTHER PILOT OF OTHER AIRCRAFT ED LIGHT WINDÓW CURTAIN CLO	AIRCRAFT T OSEB		
3-2166	B/1/68 ROCK RIVER, WYO CESSNA TIME - 0920 N-82362 TYPE OF ACCIDENT COLLISION WITH AIRCRAFT BOTH IN FL PROBABLE CAUSE PILOT IN COMMAND - FAILED TO SEE MISCELLANEOUS-PERSONNEL - PILOT ON MISSING AIRCRAFT-LATER RECOVERED REMARKS- RECOVERY DATE-8/2/68.	210 CR- 2 PX- -DESTROYED GT- 1GHT AND AVDID OTHER 1F OTHER AIRCRAFT	1 0 0 3 0 0 2 0 0 PHASE OF INFLIG	0 0 NONCOMMERCIAL 0 0 PLEASURE 0 0 PHASE OF OPERATION INFLIGHT NORMAL CRUISE	PRIVATE, AGE 42, 434 TOTAL HOURS, 255 IN TYPE.
3-2167	8/14/68 HAYWARD.CALIF CESSNA 150 TINE - 1758 N-7819E DAMAGE -SUB ÉYPE OF ACCIDENT COLLISION MITH AIRCRAFT BOTH IN FLIGHT	A 150 CR- 9E PX- E -SUBSTANTIAL OT- -LIGHT	0 0 1 0 0 0 0 0 1 PHASE OF 1	INSTRUCTIONAL TRAINING OPERATION G FINAL APPROACH	STUDENT, AGE 20, 20 TOTAL HOURS, 14 IN TYPE.
	PROBABLE CAUSE MISCELLANEOUS-PERSONNEL - PILOT FACTOR TRAFFIC CONTROL PERSONNEL - ISSU FAILURE TO ADVISE OF UNSAFE AIRPOR REMARKS- N8329P WAS CLEARED NO 2 TO L	PILOT OF OTHER AIRCRAFT ISSUED IMPROPER OR CONFLICTING INSTRUCTIONS AIRPORT CONDITION INADEQUATE SPACING OF AIRCRAFT 2 TO LAND WHEN THERE WERE 2 ACFT AHEAD OF IT ON F	NFLICTING INS EQUATE SPACIN RE 2 ACFT AHE	PILOT OF OTHER AIRCRAFT ISSUED IMPROPER OR CONFLICTING INSTRUCTIONS AIRPORT CONDITION INADEQUATE SPACING OF AIRCRAFT 2 TO LAND WHEN THERE WERE 2 ACFT AHEAD OF IT ON FINAL APPROACH.	IPPROACH.

DOCKET	DATE LOCATION	AIRCRAFT DATA	INJURIES	FLIGHT	PILOT DATA
3-2167	68 HAYWARD,CALIF - 1758 - ACCIDENT ARCRAFT BOT	PIPER PA-24 N-8329P DAMAGE -MINOR	CR- 0 0 1 PX- 0 0 0 1 DT- 0 0 1 PHASE C	DN do L	PRIVATE, AGE 32, 305 TOTAL HOURS, 87 IN TYPE.
	PROBABLE CAUSE PILOT IN COMMAND - FAILED FACTOR TRAFFIC CONTROL PERSONNEL - FAILURE TO ADVISE OF UNSAFE REMARKS- N8329P WAS CLEARED NO	FAILED TO SEE AND AVOID OTHER AIRCRAFT NNEL - ISSUED IMPROPER OR CONFLICTIN UNSAFE AIRPORT CONDITION INADEQUATE S RED NO 2 TO LAND WHEN THERE WERE 2 ACF	THER AIRCRAFT OR CONFLICTING INADEQUATE SPA RE WERE 2 ACFT	TO SEE AND AVOID OTHER AIRCRAFT ISSUED IMPROPER OR CONFLICTING INSTRUCTIONS AIRPORT CONDITION INADEQUATE SPACING OF AIRCRAFT 2 TO LAND WHEN THERE WERE 2 ACFT AHEAD OF IT ON FINAL SZD FOKA-4 CR-0 0 I NONCOMMERCIAL	APPRGACH. PRIVATE, AGE 32, 155 TOTAL HOURS, 77 IN TYPE.
	ACCIDENT SION WITH AIRCRA CAUSE IN COMMAND - LLANEOUS-PERSONN SUCCESSFUL LOG	N-204FZ N-204FZ DAMAGE -SUBSTANTIAL OT 0 0 0 DAMAGE -SUBSTANTIAL OT 0 PHASE FT BOTH IN FLIGHT FAILED TO SEE AND AVOID OTHER AIRCRAFT AT ARPT-CIRCLING LT IN THERMAL.	O O O PHA	9 H	
3-2168	8/24/68 CALIF.CITY,CALIF TIME - 1250 TYPE OF ACCIDENT COLLISION WITH AIRCRAFT BOT	GLASFUEGEL H301 N-301F DAMAGE -DESTROYED 30TH IN FLIGHT	CR- 0 0 1 PX- 0 0 0 0T- 0 0 PHASE INF	0 1 NONCOMMERCIAL 0 0 PLEASURE 0 1 PHASE OF OPERATION INFLIGHT OTHER	COMMERCIAL, AGE 40, 455 TOTAL Hours, 150 in Type.
٠	PROBABLE CAUSE PILOT IN COMMAND - FAILED TO SEE AND AVOID OTHER A MISCELLANEOUS-PERSONNEL - PILOT OF OTHER AIRCRAFT REMARKS- PLT PARACHUTED SAFELY. CIRCLING LT IN THERMAL.	FAILED TO SEE AND AVOID OTHER IEL - PILOT OF OTHER AIRCRAFT SAFELY. CIRCLING LT IN THERMAL	AND AVOID OTHER AIRCRAFT JF OTHER AIRCRAFT .NG LT IN THERMAL.		
3-2375	5/30/68 NORMAN,OKLA TINE - 0700 TYPE OF ACCIDENT COLLISION WITH AIRCRAFT	CESSNA 3378 N-2308S DAMAGE -SUBSTANTIAL BOTH IN FLIGHT	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	0 1 NONCOMMERCIAL 0 1 PLEASURE 0 1 PHASE OF OPERATION TAKEOFF INITIAL CLIMB	COMMERCIAL, AGE 40, 944 TOTAL HOURS, 57 IN TYPE.
	PROBABLE CAUSE PILOT IN COMMAND MISCELLANEOUS-PERSONN FACTOR AIRPORT CONDITIONS	EXERCISED POOR JUDGMENT IEL - PILOT OF OTHER AIRCRAFT OTHER	RCRAFT		

DOCKET	DATE LOCATION	N AIRCRAFT DATA	INJURIES F S M/N	FLIGHT PURPOSE	PILOT DATA	
3-2375	5/30/6B NORMAN,OKLA TIME - 0700 TYPE OF ACCIDENT COLLISION WITH AIRCRAFT	CESSNA 150F N-8396G DAMAGE -SUBSTANTIAL T BOTH IN FLIGHT	CR- 0 C PX- 0 O OT- 0 O PHASE	1 INSTRUCTIONAL 2 SOLO OF OPERATION (EOFF INITIAL CLIMB	STUDENT, AGI HOURS, ALL IN	AGE 25, 9 TOTAL IN TYPE.
	PROBABLE CAUSE PILOT IN COMMAND - FAILE FLIGHT INSTRUCTOR - INADE MISCELLANEOUS-PERSONNEL - FACTOR AIRPORT CONDITIONS - OTHE	O TO SEE EQUATE SU PILOT O	AND AVOID OTHER AIRCRAFT IPERVISION OF FLIGHT IF OTHER AIRCRAFT ANCE OVER OTHER ARPT TRAFFIC	•TERRAIN OBSTRUCTII	ONS BETWEEN RNW	YS.
3-2378	9/8/68 FLUSHING.NY TIME - 1455 TYPE OF ACCIDENT COLLISION WITH AIRCRAFT PROBABLE CAUSE PILOT IN COMMAND - FA	CESSNA 402 CR- N-8283F DAMAGE -SUBSTANTIAL OT- FT BOTH IN FLIGHT FAILED TO SEE AND AVOID OTHER	- 0 0 1 - 0 0 6 - 2 2 0 PHASE OF LANDIN	I - APPR	COMMERCIAL, Passg Total Hours, 1 Oach	AGE 47, 6805 169 IN TYPE.
3-2378	9/8/68 FLUSHING,NY TIME - 1455 TYPE OF ACCIDENT COLLISION WITH AFRCRAFT PROBABLE CAUSE MISCELLANEOUS-PERSONNEL	PIPER PA-28 CR-N-8828W PX-DAMAGE -DESTRUYED OT-I BOTH IN FLIGHT	1 0 0 1 2 0 0 0 7 PHASE OF LANDII	COMMERCIAL AIR TAXI - PASSG OPERATION NG FINAL APPROACH	COMMERCIAL, TOTAL HOURS,	AGE 54. 4000 UNKNOWN IN TYPE.
3-2396	9/15/68 HAMILTON,OHIO TIME - 0855 TYPE OF ACCIDENT COLLISION WITH ÁIRCRAFT	CESSNA 150 N-3314J DAMAGE -SUBSTANTIAL F BOTH IN FLIGHT	CR- 0 0 1 Px- 0 0 1 OT- 0 0 1 PHASE OF LANDIN	NONCOMMERCIAL PRI PLEASURE HOU OPERATION G LEVEL OFF/TOUCHDOWN	VATE, RS, 46 I	AGE 48, 350 TOTAL N TYPE.
	PROBABLE CAUSE PILOT IN COMMAND - FA REMARKS- CESSNA N3314J DES	CAUSE IN COMMAND - FAILED TO SEE AND AVOID OTHER AIRCRAFT CESSNA N3314J DESCENDED ON TO THE TOP OF NS9441 IN-CONTROLLED ABOT	HER AIRCRAFT	i con		

			1		TUST 10	PILOT DATA
DOCKET	DATE	LOCATION	AIRCRAFT DATA	INJURIES F S M/N	PURPOSE	
3-2396	9/15/68 HAI TIME - 0855	68 HAMILTON, OHIO - 0855	CESSNA 150 N-5994T DAMAGE - SU"STANTIAL	CR- 0 0 PX- 0 0 OT- 0 0	INSTRUCTIONAL TRAINING OPERATION	STUDENT, AGE 28, 90 TOTAL HOURS, 65 IN TYPE.
	TYPE OF ACCIDENT COLLISION WIT	E OF ACCIDENT COLLISION WITH AIRCRAFT	BOTH IN FLIGHT	LANDIN	LANDING LEVEL OFF/TOUCHDOWN	NAO
	PROBABLE C MISCELL	PROBABLE CAUSE MISCELLANEOUS-PERSONNEL	- PILOT OF OTHER AIRCRAFT	CRAFT		
3-2443	9/16/68 BI TIME - 1100	BITTER CREEK,WYG	INTERSTATE S-1A N-37455 DAMAGE -DESTROYED	CR- 0 1 0 PX- 0 0 0 GT- 1 1 0 PHASE OF	25 E	CONMERCIAL. AGE 21, 457 TOTAL. HOURS, 300 IN TYPE.
	TYPE OF ACCIDENT COLLISION WIT UNDER INVESTIGAT REMARKS- N48100	TYPE OF ACCIDEN! COLLISION WITH AIRCRAFT UNDER INVESTIGATION REMARKS- N48100 STRUCK LEFT		INFLIGHT HERDING HORSES.LOW	GHI OIMEK OW LEVEL FLT,ROLLIN(BOTH IN FLIGHT WG,N37455. BOTH PLTS HERDING HORSES.LOW LEVEL FLT,ROLLING TERRAIN.NO RADIO COMM
3-2443	9/ TI 9YT	16/68 BITTER CREEK,WYO ME - 1100 E OF ACCIDENT COLLISION WITH AIRCRAFT	INTERSTATE S-181 N-48100 DAMAGE -DESTROYED BOTH IN FLIGHT	CR- 1 0 0 COI PX- 0 1 0 DT- 0 1 0 PHASE OF OPI INFLIGHT	0 0 COMMERCIAL 1 0 OTHER 1 0 OPERATION PHASE OF OPERATION INFLIGHT OTHER	COMMERCIAL, FL. INSTR., AGE 47, 16103 TOTAL HOURS, 700 IN TYPE.
,	PROBABLE CAUSE PILOT IN COI PILOT IN COI MISCELLANEOI PILOT IN COI REMARKS - N48101	BABLE CAUSE PILOT IN COMMAND - FAILE PILOT IN COMMAND - DIVER MISCELLANEGUS-PERSONNEL - PILOT IN COMMAND - INADI	ED TO SEE / TED ATTENT PILOT OF QUATE PREF	ND AVOID OTHER AIRCRAFT ION FROM OPERATION OF AIRCRAFT - OTHER AIRCRAFT LIGHT PREPARATION AND/OR PLANNING BOTH PLIS HERDING HORSES.LOW LEVEL		FLT, ROLLING TERRAIN.NO RADIO COMM
3-3188		8/28/68 BEAUMONT.TEX TIME - 1005	CESSNA 1506 N-2654J DAMAGE -DESTROYED	CR- 0 0 2 PX- 0 0 0 0T- 0 0 1 PHASE OF	INSTRUCTIONAL DUAL F OPERATION	COMMERCIAL, FL. INSTR., AGE 53, 500 TOTAL HOURS, ALL IN TYPE.

TIONAL COMMERCIAL, FL. INSTR-, FREE 500 TOTAL HOURS, ALL IN TY	TAL CLIMB		SARNE HO NOLLLUNGSENE EN MONTON COLUMN
CR- 0 0 2 INSTRUCTIONAL px- 0 0 0 DUAL OT- 0 0 1 DERATION	TAKEOFF INITIAL CLINB	CRAFT	
ROYED		OBABLE CAUSE PILOT IN COMMAND - FAILED TO SEE AND AVOID OTHER AIRCRAFT MISCELLANEOUS-PERSONNEL - PILOT OF OTHER AIRCRAFT	- HIGH VEGETATION
8/28/68 BEAUMONT.TEX TIME - 1005	TYPE OF ACCIDENT COLLISION WITH AIRCRAFT BOTH IN FLIGHT	FPROBABLE CAUSE PILOT IN COMMAND - FAIL MISCELLANEOUS-PERSONNEL -	FACTOR CONDITIONS - HI
3-3188			

AIRPORT CONDITIONS - HIGH VEGETATION AIRNAY 02, COLLISION AT INTERSECTION OF RNWYS. REMARKS- N3536J TOOK OFF RNWY 30, N2654J TOOK OFF RNWY NAT INTERSECTION OF RNWYS.

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		OKIETS -	UF ACCIDENTS				
DOCKET	DATE LOCATION	AIRCRAFT DATA	INJURIES F S M/N	FL IGHT PURPOSE	PILOT DATA		
3-3188	8/28/68 BEAUMONT,TEX TIME - 1005 TYPE OF ACCIDENT COLLESION WITH AIRCRAFT	CESSNA 150 N-3536J DANAGE -DESTROYED BOTH IN FLIGHT	CR- 0 0 1 C PX- 0 0 0 OT- 0 0 2 PHASE OF 0 TAKEGFF	0 1 COMMERCIAL 0 0 POWER/PIPELINE 0 2 PHASE OF OPERATION TAKEOFF INITIAL CLIMB	COMMERCIAL, TOTAL HOURS,	AGE 37,	3220 PE.
	PROBABLE CAUSE PILOT IN COMMAND - FAILED TO SEE AND AVOID OTHER AIRCRAFT MISCELLANEOUS-PERSONNEL - PILOT OF OTHER AIRCRAFT FACTOR AIRPORT CONDITIONS - HIGH VEGETATION REMARKS- TALL CROPS BETWEEN RNWYS.UNCONTROLLED ARPT.BOTH RNWYS IN USE,WIND 5	ILLED TO SEE AND AVOID OTHER PILOT OF OTHER AIRCRAFT HIGH VEGETATION IN RNWYS.UNCONTROLLED ARPT.BO	HER AIRCRAFT RAFT 1-BOTH RNWYS IN	KTS	FROM 45 DEGS.		
3-3356	7/31/68 WDDDBURY,TENN TINE - 1637 TYPE OF ACCIDENT COLLISION WITH AIRCRAFT BG PROBABLE CAUSE PILOT IN COMMAND - FAILED MISCELLANEOUS-PERSONNEL -	CESSNA N-5631E DAMAGE TH IN FL	1 0 0 0 0 0 0 0 0 PHASE INF	INSTRUCTIONAL TRAINING OF OPERATION LIGHT NORMAL CRUISE	STUDENT, HOURS, ALL	AGE 27, 42 TO In Type.	TOTAL
3-3356	7/31/68 WOODBURY,TENN TIME - 1637 TYPE OF ACCIDENT COLLISION MITH AIRCRAFT BO PROBABLE CAUSE PILOT IN COMMAND - FAILED MISCELLANGOUS.DERSONNEL	CESSNA 150 N-5873E DAMAGE -SUBSTANTI TH IN FLIGHT TO SEE AND AVOID		O 1 INSTRUCTIONAL O 0 TRAINING O 0 PHASE OF OPERATION INFLIGHT NORNAL CRUISE	STUDENT, /	AGE 20, 51 TOTAL IN TYPE.	4
3-3398	EMERGENCY CIRCUMSTANCES - SPENGENCY CIRCUMSTANCES - SPENGENCY CIRCUMSTANCES - SPENGENCY -	S P E C A	0 0 0 SE OF INFLIG	COMMERCIAL COMME AERIAL APPLIC TOTAL OPERATION HT PROCEDURE TURNAROUND	COMMERCIAL, Total Hours, Round	AGE 25, 1400 UNKNOWN IN TYPE.	T ← P E •

UNDER INVESTIGATION

- 62 -

FIRE AFTER IMPACT REMARKS- N3280H OVERTOOK AND STRUCK N5847H FROM THE REAR.

STUDENT, AGE 33, 40 TOTAL HOURS, ALL IN TYPE.	
CR- 1 0 0 NONCOMMERCIAL PX- 1 0 0 PLEASURE .07- 2 0 0 DEBATTOM	INFLIGHT NORMAL CRUISE
PIPER PA-16 N-5847H DAMAGE -DESTROYED	BOTH IN FLIGHT
3-3570 10/27/68 CALIPATRIA,CALIF TIME - 1640	TYPE OF ACCIDENT COLLISION WITH AIRCRÁFT
3-357	à

UNDER DIVESTICATION

FIRE AFTER IMPACT REMARKS- N3280H OVERTOOK AND STRUCK N5847H FROM THE REAR.

34 TOTAL		
PRIVATE, AGE 53, 234 TOTAL HOURS, 94 IN TYPE.		
PRIVATE, HOURS, 94	CHDOWN	
NONCOMMERCIAL PLEASURE	PHASE OF OPERAILON LANDING LEVEL OFF/TOUCHDOWN	
~0 C	ASE UF	
000	Ī	
7X - T		
CESSNA 150E CR- 0 0 1 NONC N-3984U PX- 0 0 0 PI DAMAGE -SUBSTANTIAL 0T- 0 0 2	BOTH IN FLIGHT	
12/8/68 SANDWICH, 11L TIME - 1500	TYPE OF ACCIDENT COLLISION WITH AIRCRAFT	UNDER INVESTIGATION
3-3983		

	AIRCRAFT DATA INJURIES FLIGHT PILOT DATA	3-3983 12/8/68 SANDWICH, ILL FAIRCHILD 24 CR- 0 0 1 NONCOMMERCIAL PRIVATE, AGE 40, 180 TOTAL TIME - 1500 N-81330 PX- 0 0 1 PLEASURE HOURS, 60 IN TYPE. TYPE OF ACCIDENT COLLISION WITH AIRCRAFT BOTH IN FLIGHT UNDER INVESTIGATION	
	FL IGHT	F S M/N PURPOSE	
BRIEFS OF ACCIDENTS	INJURIES	S M/N PURPOSE	
SKIEPS	AIRCRAFT DATA INJURIES	FAIRCHILD 24 CR- 0 0 1 N-81330 PX- 0 0 1 DAMAGE - SUBSTANTIAL DT- 0 0 1 BOTH IN FLIGHT LANDI	
****************************	E LOCATION	3-3983 12/8/68 SANDWICH, ILL TIME - 1500 TYPE OF ACCIDENT COLLISION WITH AIRCRAFT UNDER INVESTIGATION	
	DOCKET DATE	3-3983 12/8 11ME TYPE (CQI	

PRIVATE, AGE 34, 220 TOTAL	noons) of In IPPE.	CHOOMN
NONCOMMERCIAL PLEASURE	***************************************	LANDING LEVEL OFF/TOUCHDOWN
-0	1 20	ANDI
00	OH O	<u> </u>
11	ı	
CESSNA 182K CR- 0 0 1 N-2812R PX- 0 0 0	DAMAGE -SUBSTANTIAL O	
12/20/68 ROME,GA TIME - 1645	TYPE OF ACCIDENT	UNDER INVESTIGATION
3-4004		

REMARKS- N2812R WAS LOWER ACFT.N2812R FASTER, DVERTODK N8103S.

STUDENT, AGE 32, 24 TOTAL	NADOH)
INSTRUCTIONAL TRAINING	PHASE OF OPERATION LANDING LEVEL OFF/TOUCHDOWN
~0	SE OF
00	O X
d x	<u> </u>
CESSNA 150F CR- 0 0 1 1 N-8103S PX- 0 0 0	UAMAGE -SUBSTANTIAL O BOTH IN FLIGHT
3-4004 12/20/68 ROME,GA TINE - 1645	TYPE OF ACCIDENT COLLISION WITH AIRCRAFT
3-4004	N.

REMARKS- UNCONTROLLED ARPT.N01035.LANDED ON TOP OF N2812R

COMMERCIAL, FL. INSTR., AGE 19, 726_TOTAL HOURS, UNKNOWN IN	:TPE. N-CIRCLING
CR- 2 0 0 INSTRUCTIONAL PX- 0 0 0 DUAL OT- 3 0 0	PHASE OF OPERATION LANDING TRAFFIC PATTERN-CIRCLING
CESSNA 1508 N-1192Y DAMAGE -DESTROYED D	
•	COLLISION WITH AIRCRAFT UNDER INVESTIGATION
3-4444	

REMARKS- N3420S STRUCK N1192Y FROM REAR. BOTH ACFT HAD TURNED FROM CROSSWIND TO DOWNWIND LEG.

	142 TOTAL TYPE.	39, 12000 IN TYPE.	2. 18600 HN IN TYPE.	•• AGE 26• 00 IN TYPE•	200 TOTAL TYPE.
	UNKNOWN IN T	AGE 4000	AGE 42.	FL.INSTR HOURS, 2	AGE 35. UNKNOWN IN
PILOT DATA	AL PRIVATE. A HOURS, UNKN PATTERN-CIRCLING	(IND LEG. COMMERCIAL, TOTAL HOURS,	COMMERCIAL. Total Hours,	COMMERCIAL, FL. INSTR., 1518 TOTAL HOURS, 200	PRIVATE, Hours, unk
FL IGHT PURPOSE	DNCOMMERCI PLEASURE PERATION TRAFFIC	HAD TURNED FROM CROSSWIND TO DOWNWIND LEG. CR- 0 0 1 COMMERCIAL COMMERCI PX- 0 0 0 FISH SPOTTING TOTAL HO OT- 1 0 0 PHASE OF OPERATION INFLIGHT OTHER	O O COMMERCIAL O O FISH SPOTTING O PHASE OF OPERATION INFLIGHT OTHER	O INSTRUCTIONAL O DUAL O E OF OPERATION NFLIGHT NORMAL CRUISE	NONCOMMERCIAL PLEASURE
INJURIES F S M/N	CR- 1 0 NC PX- 2 0 0 OT- 2 0 0 PHASE OF OI LANDING	HAD TURNED FROM CI CR- 0 0 1 CO PX- 0 0 0 OT- 1 0 0 INFLIGHT	CR- 1 0 0 CO PX- 0 0 0 OT- 0 0 1 PHASE OF OP INFLIGHT	CR- 2 0 0 IN: PX- 0 0 0 0 OT- 1 3 0 PHASE OF OP! INFLIGHT	CR- 0 1 0 PX- 1 2 0
AIRCRAFT DATA IN	CESSNA 182 C N-3420S DAMAGE -DESTROYED D	Y FROM REAR. BOTH ACFT CHAMPION TECA N-96468 DAMAGE -SUBSTANTIAL BOTH IN FLIGHT	PIPER PA-18A N-7771D DAMAGE -DESTROYED BOTH IN FLIGHT	CESSNA 177 N-29407 DAMAGE -DESTROYED GTH IN FLIGHT FLT.COLLIDED APPROX	CESSNA 1823 N-3306F
DATE LOCATION	12/8/68 SANTA PAULA,CAL TIME - 1450 TYPE OF ACCIDENT COLLISION WITH AIRCRAFT E	FIRE AFTER IMPACT REMARKS- N3420S STRUCK N1192) 4/13/68 NR.PASCAGOULA,MISS TIME - 0635 TYPE OF ACCIDENT COLLISION WITH AIRCRAFT	4/13/68 NR.PASCAGOULA.MISS TIME - 0635 TYPE OF ACCIDENT COLLISION WITH AIRCRAFT B BEMARKS - UNDER INVESTIGATION.	10/5/68 MIAMI,FLA CESSNA 177 TIME - 1437 DAMAGE -DESTROYED TYPE OF ACCIDENT COLLISION WITH AIRCRAFT BOTH IN FLIGHT REMARKS- INSTRUMENT TRAINING FLT.COLLIDED APPROX	10/5/68 MIAMI,FLA Time - 1437
DOCKET D	3-4444 12 3-4444 11 1 YP	FIR REM 3-4803 4/	3-4803 4 TY	3-4836 1 1 TY	3-4836

TYPE OF ACCIDENT COLLISION WITH AIRCRAFT BOTH IN FLIGHT

	AIRCRAFT DATA INJURIES FLIGHT PILOT DATA F S M/N PURPOSE	CONVAIR 580 CR- 0 1 2 PASSG S-D AGE UNKNOWN, UNKNOWN TOTAL N-4364S PX- 0 0 8 HOURS, UNKNOWN IN TYPE. DAMAGE -SUBSTANTIAL OT- 3 0 0 CLINES, INC. PHASE OF OPERATION BOTH IN FLIGHT DESCENDING
, , , , , , , , , , , , , , , , , , , ,	INJURIES FLIGHT F S M/N PURPOSE	CR- 0 1 2 PASSG S-D PX- 0 0 8 L OT- 3 0 0 PHASE OF OPERATION INFLIGHT DESCENDING
	AIRCRAFT DATA	CONVAIR 580 CR- 0 1 2 N-4364S PX- 0 0 8 DAMAGE -SUBSTANTIAL 0T- 3 0 0 IRLINES, INC. BOTH IN FLIGHT
	DOCKET DATE LOCATION	8/4/68 NR.MILWAUKEE,WIS CONVAIR TIME - 0848 N-4364S OPERATOR - NORTH CENTRAL AIRLINES, INC. TYPE OF ACCIDENT COLLISION WITH AIRCRAFT BOTH IN FLI
	DOCKET	3-9005

REMARKS- UNDER INVESTIGATION. CONVAIR LANDED WITH CESSNA IMBEDDED IN ITS FUSELAGE.

AGE 19. UNKNOWN TOTAL HOURS, Unknown in Type.
CR- 1 0 0 PX- 2 0 0 TROYED 0T- 0 1 10 PHASE OF OPERATION INFLIGHT NORMAL CRUISE
CESSNA 150 N-8742S DAMAGE -DESTROYED BOTH IN FLIGHT
3-9005 8/4/68 NR.MILWAUKEE,WIS TIME - 0848 TYPE OF ACCIDENT COLLISION WITH AIRCRAFT UNDER INVESTIGATION

REMARKS- UNDER INVESTIGATION. CONVAIR LANDED WITH CESSNA IMBEDDED IN ITS FUSELAGE.

COMMERCIAL, FL. INSTR., AGE UNKNOWN, UNKNOWN TOTAL HOURS, UNKNOWN IN TYPE.	
COMMERCIAL, FL. INS UNKNOWN, UNKNOWN UNKNOWN IN TYPE.	TTERN-CIRCLING
CR- PX- OT-	PHASE OF OPERATION LANDING TRAFFIC PATTERN-CIRCLING
CESSNA 150 N-8669G DAMAGE -DESTROYED	BOTH IN FLIGHT
	COLLISION WITH AIRCRAFT UNDER INVESTIGATION
3-9006-	

REMARKS- UNDER INVESTIGATION. CESSNA CRASHED IN PARKING LOT, DC-9 CONT APPCH AND LANDED.

AGE UNKNOWN. UNKNOWN TOTAL Hours. Unknown in Type.	-	
DOUGLAS DC-9 CR- 0 0 5 PASSG S-D N-9702 PX- 0 0 37		PHASE OF OPERATION LANDING FINAL APPROACH
3-9006 3/27/68 ST.LOUIS,MD DOUGLAS DC-9 TIME - 1800 N-9702	OPERATOR - OZARK AIR LINES, INC.	ITPE UP ACCIDENT COLLISION WITH AIRCRAFT BOTH IN FLIGHT UNDER INVESTIGATION
3-9006		

REMARKS- UNDER INVESTIGATION. CESSNA CRASHED IN PARKING LOT, DC-9 CONT APPCH AND LANDED.

246

154

C. S. CLVIC PVIRILOR

	FATAL	SERIOUS	MINOR	NONE	UNKNOWN		TOTAL
PILOT COPILOT	36	4-	7	29	† 	•	76
DUAL STUDENT CHECK PILOT	9		-) <i>per</i>			ေထ
FLIGHT ENGINEER NAVIGATOR				-			-
CABIN ATTENDANT Extra crew				90			v 0 (
PASSENGERS	27	7	-	110			145
TOTAL	11	12	6	154		ABOARD	246
OTHER-AIRCRAFT							
GRAND TOTAL	11	12	œ	154			776

TOTAL ACCIDENTS FATAL ACCIDENTS స్థిజ్ఞ INVOLVES INVOLVES

NOTE: * INCLUDES ONE ACCIDENT INVOLVING TWO FOREIGN REGISTERED AIRCRAFT ON U. S. SOIL, FATAL ACCIDENT INVOLVING TWO PILOT FATALITIES,

KIND OF FLYING BY INJURY INDEX

INJURY INDEX

	کے	AL.	*10US	OR OHE				
OF FLYING	44	50	14,	40		RECORDS	ACCIDENTS	PERCENT
STRUCT I ONAL								
JAL .	7		1	1			_	
pro				2		9	9	11.84
IECK						2	2	2.63
MINING	5		1	4				
HCOMMERC I AL						10	9	13.16
EASURE	18		3	8				
MCTICE	1			1		29	21	38.16
SINESS	2	1		_		2	2	2.63
MPORATE/EXECUTIVE	3					3	3	3.95
NAL SURVEY						3	3	3, 95
MPANY FLIGHT								
HER								
WERC I AL								
NAL APPLICATION	2		2					
OCIATED CROP CONTROL ACTIV		1	-			4	2	5.26
E CONTROL		_				1	1	1.32
CLIATED FIRE CONTROL ACTIV								
IAL MAPPING/PHOTOGRAPHY								
IAL ADVERTISING								
ER AND PIPELINE PATROL			1					
H SPOTTING	2		-			1	1	1.32
TAXI-PASSENGER OPERATIONS	2					2	1	2.63
TAXI-CARGO OPERATIONS	•					2	1	2.63
TRUCTION WORK								

RACT/CHARTER-CARGO-DOMEST

NACT/CHARTER-PASSENGER-DU

DULED PASSENGER SERVICE DULED CARGO SERVICE CHEDULED/CHARTER REVENUE CHEDULED/CHARTER REVENUE

RACT/CHARTER-CARGO-INTERN

ACT/CHARTER-PASSENGER-IN

KIND OF FLYING BY INJURY INDEX

INJURY INDEX

en in serious minor home

OF FLYING

R	2						2	1	2.
NR									
ELLANEOUS									
RIMENTATION									
NSTRATION									
Y	2						2	1	2.
CH AND RESCUE									
SHOW/AIR RACING									
CHUTE JUMP									
CHUTE JUMP IN CONNECTION									
NG GLIDERS									
ING CLOUDS									
ING									
CE PATROL				1			1	1	1.
OTHER PUBLIC FLYING									
R.	2			1			3	3	3.
/NR									
os	48	2	8	18		7	16		
ENTS	24	1	Ĭį.	9				38	
i									

UDES ONE ACCIDENT/TWO FOREIGN REGISTERED AIRCRAFT ON U.S.SOIL

63.2 2.6 10.5 23.7

RECORDS ACCIDENTS PERCEN

AIRPORT PROXIMITY BY INJURY INDEX TABLE
MID AIR COLLISIONS 1968
U.S.CIVIL AVIATION **

INJURY INDEX

AIRPORT PROXIMITY	<i>b</i> ,	14.33	50014	NONIA SOLIA	RECOROS	ACC IDENTS	PERCENT
ON AIRPORT	y 01	r	r a	₩ 80	12	•	15.79
ON SEAPLANE BASE						•	
ON HELIPORT							
ON BARGE/SHIP/PLATFORM							
IN TRAFFIC PATTERN	12	2	8	80	24	12	31.58
WITHIN 1/4 MILE						!	
WITHIN 1/2 MILE							
WITHIN 3/4 MILE							
MITHIN 1 MILE							
MITHIN 2 MILES	60		7		10	ĸ	13,16
WITHIN 3 MILES	4		8		ý	6	7.89
WITHIN 4 MILES	2				2	-	2.63
MITHIN 5 MILES	2				~	-	2.63
BEYOND 5 MILES	18			2	20	10	26.32
UNK/NR							
OTHER							
RECORDS	84	~	•	81	76		
ACCIDENTS	24	-	#	σ.		38	

* INCLUDES ONE ACCIDENT/TWO FOREIGN REGISTERED AIRCRAFT ON U.S.SOIL

PERCENT

63.2 2.6 10.2 23.7

FIRST PHASE OF OPERATION BY INJURY INDEX

INJURY INDEX

FIRST	
ERATIONAL	PHASE

FATAL SERIOUS MINOR PAR

RECORDS ACCIDENTS PERCENT

STATIC	

STARTING ENGINE/S

IDLING ENGINE/S

ENGINE RUNUP

IOLING ROTORS

PARKED-ENGINES NOT OPERATING

OTHER

TAXI

10 TAKEOFF

FROM LANDING

OTHER

GROUND TAXI TO TAKEOFF

GROUND TAXI FROM LANDING

GROUND TAXI, OTHER

MERIAL TAXI TO TAKEOFF

MERIAL TAXI TO/FROM LANDING

MERIAL TAXI, OTHER

JAKEOFF

WN

HITIAL CLIMB

ERTICAL

UNNING

MORTED

MORTED

BORTED

THER

#FL IGHT

LIMB TO CRUISE

MAL CRUISE

ESCENDING

DLDING

WERING

MER-ON DESCENT

MOROTATIVE DESCENT

ROBATICS

- 73 ~

2

2

19

19

2 2.63 12 25.00

5.26

5.26

FIRST PHASE OF OPERATION BY INJURY INDEX

INJURY INDEX

FIRST OPERATIONAL PHASE	ERTA	SERIO	WINO	HOHE		RECORDS	ACCIDENTS	PERCE
BUZZING								
UNCONTROLLED DESCENT								
EMERGENCY DESCENT								
LOW PASS								- 19
OTHER	6		2			8	4	10.
EN ROUTE TO TREAT CROP								ų
EN ROUTE TO RELOADING AREA								ļ
SURVEY FIELD/AREA								1
STARTING SWATH RUN			1			1	ı	1
SWATH RUN			1			1	1	14 14
FLAREOUT FOR SWATH RUN								į
PULLUP FROM SWATH RUN	1					1		16 1
PROCEDURE TURNAROUND	1					1	1	14
CLEANUP SWATH								
MANEUVER TO AVOID OBSTRUCTION								, i
RETURN TO STRIP								4
LANDING								
TRAFFIC PATTERN-CIRCLING	7			2		9		11
FINAL APPROACH	9	2	2	4		17	9	22
INITIAL APPROACH								j
FINAL APPROACH	1					1		1
LEVEL OFF/TOUCHDOWN				8		8	3 4	10
ROLL								1
ROLL-ON/RUN-ON								
POWER-ON LANDING								
POWER-OFF AUTOROTATIVE LANDIN								
GO-AROUND								;
MISSED APPROACH								
OTHER								
UNK/NR								
RECORDS	48	2	8	18		7	6	
ACCIDENTS	24	1	ją.	9			3	8
PERCENT	63.2	2.6	10.1;	23.7	-			

INCLUDES ONE ACCIDENT/TWO FOREIGN REGISTERED AIRCRAFT ON U.S.SOIL

MONTH OF OCCURRENCE BY TYPE OF MEATHER CONDITIONS MID AIR COLLISIONS 1968 U.S.CIVIL AVIATION

rof occurrence	THE	RECORDS	ACCIDENTS	PERCENT
01	2	2	1	2.63
02	6	6	3	7.89
03	6	6	3	7.89
04	6	6	3	7.89
05	6	6	3	7.89
06	4	4	2	5.26
07	10	10	5	13.16
08	18	18	9	23.68
09	6	6	3	7.89
10	6	6	3	7.89
l 1				
12	6	6	3	7.89
EOROS	76	76		
CIDENTS	38		38	
RCENT	100.0 .0 .0			

NCLUDES ONE ACCIDENT/TWO FOREIGN REGISTERED AIRCRAFT ON U.S.SOIL

TYPE OF OPERATOR BY CONDITIONS OF LIGHT NID AIR COLLISIONS U. B. CIVIL AVIATION * 1968

TYPE OF OPERATOR	Court Carl	ichi Next	TWILCHT DARK	I BEICHT WHI LEFORT	ęÞ	RECORDS	ACCIDENTS	PERCE
FLYING SCHOOL	6	0	4 4	44		6	4	7.(
CORPORATE/EXECUTIVE	8					8	7	10.
AERIAL APPLICATOR	5					5	3	6.5
PRIVATE OWNER	20	2				22	18	28.9
AIR TAXI OPERATOR	2					2	1	2.6
FIXED BASE OPERATOR	16	2				18	15	23.6
FEDERAL-PUBLIC AIRCRAFT	1					1	1	1.
STATE-PUBLIC AIRCRAFT	1					1	1	l,
MUNICIPAL-PUBLIC AIRCRAFT								
CIVIL AIR PATROL								
AIRCRAFT MANUFACTURER								-
FLYING CLUB (MILITARY)								i
FLYING CLUB	5					5	5	6.
INTRASTATE CARRIER								
CONTRACT CARRIER	1					1	1	14
								1
OTHER	5					5	d ₄	6.
UNK/NR	2					2	1	24
								4
RECORDS	72	4				76		
ACCIDENTS	36	2					34	8
PERCENT	.0 94.7	5.3	.0 .0	•0				

^{*} INCLUDES ONE ACCIDENT/THO FOREIGN REGISTERED AIRCRAFT ON U.S.SOIL

PHASE OF OPERATION BY CONDITIONS OF LIGHT MID AIR COLLISIONS 1968 U.S.CIVIL AVIATION

	Or why Dr. richi Lander, Dr. ret Brech, Rebert Brech, Brec
FIRST	IN IGHT ITM TO THE WORLD
RATIONAL PHASE	OAWH DAYLICHT WICHT DE LICHT BERNOW CORN
	* * * * * * *

RECORDS ACCIDENTS PERCENT

•				
STATIC				
STARTING ENGINE/S				
IDLING ENGINE/S				
ENGINE RUNUP				
IDLING ROTORS				
PARKED-ENGINES NOT OPERATING				
OTHER				
IXAI				
TO TAKEOFF				
FROM LANDING				
OTHER				
GROUND TAXI TO TAKEOFF				
GROUND TAXI FROM LANDING				
GROUND TAXI, OTHER				
AERIAL TAXI TO TAKEOFF				
AERIAL TAXI TO/FROM LANDING				
AERIAL TAXI. OTHER				
TAKEOFF				
RUN				
INITIAL CLIMB 4	4	4	2	5.26
VERT ICAL				
RUNNING				

INFL IGHT

ABORTED
ABORTED
OTHER

 CLIMB TO CRUISE
 2
 2
 2.63

 NORMAL CRUISE
 19
 12
 25.00

 DESCENDING
 4
 3
 5.26

HOLDING HOVERING

POWER-ON DESCENT

AUTOROTATIVE DESCENT

ACROBATICS BUZZING

FIRST PHASE OF OPERATION BY CONDITIONS OF LIGHT CONDITIONS OF LIGHT

FIRST OPERATIONAL PHASE	on the online	oust ru	ILICHT D'	PAT O	JENGT OF	HT RED			RECORDS	ACCID
UNCONTROLLED DESCENT	▼ -	•								
EMERGENCY DESCENT										i
LOW PASS								8	4	10.5
OTHER	6									1
EN ROUTE TO TREAT CROP										ļ
EN ROUTE TO RELOADING AREA	•									!
SURVEY FIELD/AREA								1	1	1.1
STARTING SWATH RUN	1							1	1	1.4
SWATH RUN	1									1
FLAREGUT FOR SWATH RUN								1	1	ւ
PULLUP FROM SWATH RUN	1							1	1	1.
PROCEDURE TURNAROUND	1									ļ
CLEANUP SWATH										
MANEUVER TO AVOID OBSTRUCTION										
RETURN TO STRIP										
LANDING								9	.6	11.
TRAFFIC PATTERN-CIRCLING	. 5	4						17	9	22.
FINAL APPROACH	17							_		
INITIAL APPROACH								1	1	1,
FINAL APPROACH	1							8	4	104
LEVEL OFF/TOUCHOOWN	8									
ROLL										
ROLL-ON/RUN-ON										ļ
POWER-ON LANDING										ļ
POWER-OFF AUTOROTATIVE LANDIN										į
GO-AROUND										Į
MISSED APPROACH										
OTHER										
UNK/NR										
	72	4						76		}
RECORDS	36	2							3	38
ACCIDENTS	.0 94.7	5.3	.0	.0	•0					

INCLUDES ONE ACCIDENT/THO FOREIGN REGISTERED AIRCRAFT ON U.S.SOIL

PERCENT

- 1955: At a joint public meeting of Institute of Radio Engineers and Radio Technical Commission for Aeronautics, ATA requested industry to propose, or produce, a collision avoidance system (CAS). Airline analysis of CAS problem and "inventor's chart" was also circulated among hundreds of inventors, engineers, and manufacturers. No response received.
- 1956: Midair collision over the Grand Canyon stimulated interest, resulted in a flood of ideas to ATA.

In July, 1956, ATA sponsored a symposium in Washington,
D. C., bringing together experts, engineers and inventors
to compare airline requirements against then-current
technology.

As an outgrowth of that symposium, Collins Radio, in September, 1956, submitted the first formal proposal to the airlines for a noncooperative pilot warning indicator (PWI) system, which they believed could later be developed into a CAS. Two million dollars worth of airline orders were placed with Collins; but, in the development work that followed, Collins discovered that normal aircraft movements in flight prevented their airborne Doppler radar from making reliable collision prediction, or could create erroneous predictions in a significant percentage of cases. This eliminated the CAS feature of

the Collins proposal. So the proposal was withdrawn while Collins continued its analytical work, which has been carried on to this day.

- 1958: Development of the Tau concept (Tau is range divided by range rate) by Bendix Radio's Dr. J. S. Morrell, following their publication in the midfifties of the first accurate description of the fundamental physics of the airborne collision avoidance problem.
- 1959: The ATA Collision Avoidance Committee (formed in 1956) continued to monitor and encourage CAS investigations until 1959, when FAA created its Collision Prevention Advisory Group (COPAG). Thereafter, ATA took part in CAS work as a member of COPAG.
- 1960: Collision between two fighter aircraft being tested by their manufacturer, McDonnell Aircraft Corporation, sparked an all-out effort by McDonnell to develop a CAS for use in their flight test area.
- 1962: In July 1962, the FAA's Airborne CAS symposium, held in Washington, D. C., gave the first industry-wide briefing on the state of CAS investigations, and received widespread support from all segments of aviation. Bendix Radio described a CAS based on the Tau concept and using one-way ranging, with the ground-bounce technique. With ATA encouragement, Collins Radio outlined a method for testing by computer simulation the features of any

proposed CAS technique. This was a major contribution, since it offered for the first time a quick and minimum-cost method of choosing the most promising techniques from among a wide variety of theoretical concepts.

- 1963: In the fall of 1963, FAA gave Collins a contract to study CAS techniques by simulation.
- 1965: A report of the Collins study was forwarded to FAA in mid-1965. In essence, Collins told FAA that the simple time-frequency system is the most promising, that it works quite well for the en route case, but has some limitations in the terminal area--where aircraft are most likely to be maneuvering in flight.

At an ATA meeting in April 1965, Collins reported on the results of its studies and McDonnell Aircraft showed the equipment they planned to use in their flight test operation later that year. The airlines were sufficiently impressed with McDonnell's system to invite their experts to brief the Airlines Operations Conference in September 1965. After this briefing, the Conference charged the ATA Air Traffic Control (ATC) Committee with monitoring the development of collision avoidance systems and their relationship with the air traffic control system.

1966: Intensive review of the application of time and frequency techniques to CAS by ATA staff and ATC Committee. Four manufacturers--McDonnell, National, TRG, and Collins--briefed ATC Committee.

October 1966: ATA issued expanded statement of airline policy on CAS, with detailed listing of functional requirements for a CAS meeting airline needs. Document circulated widely throughout industry and government, particularly to electronics manufacturers with experience in time and frequency techniques.

December 1966: ATC Committee of ATA formed CAS Technical Working Group, to study technical details of proposed CAS concepts and prepare draft technical description of system that will meet airline requirements.

1967: First meeting of CAS Technical Working Group in January 1967, for 3 days. Group continued regular monthly meetings thereafter (sometimes twice a month) for a total of 37 meeting days between January 1 and June 30, 1967.

June 14, 1967: ATA asked FAA to have the necessary frequencies designated for CAS.

Technical description of a CAS that will meet airline requirements completed June 30 and distributed to industry, government, and airline early in July.

July 12, 1967: ATA asks FAA to use system technical description:

--- to test, by simulation, the interaction of CAS and ATC in order to verify that operation of such a CAS will not adversely affect ATC.

- --- as a starting point for developing a common national system for airborne collision avoidance systems.

 FAA asked to give early attention to insuring compatibility of civil and military use of CAS techniques.
- --- to begin the efforts that will ultimately be needed to secure international agreement on a common international system for airborne CAS.

CAS Equipment for General Aviation

By **W.** R. Lewis

This paper discusses minimal collision avoidance equipment which, we believe, has a good probability of being palatable to general aviation. The ideas presented herein are the result of analyzing the ATA description recently circulated by ARINC.

As stated by ATA in their July 12 "Fact Sheet" on CAS,

"the airlines seek a common system - and that means a system
satisfying all users, both civil and military." The statement
could have gone further to point out the fact that of the three
user groups the airlines are in the minority, in number of aircraft. I believe the statement could be interpreted to mean that
the system <u>must be</u> common to all users in its basic function
in order to provide satisfactory usefulness to any one group.

The ATA committee has accomplished a monumental task in forging the CAS concepts to date; however, as ATA states, "a great deal of effort...to simplify the system, reduce equipment cost...and make the advantages of CAS more widely available" must be undertaken.

The only statement in the whole ATA "Fact Sheet" with which

I have the slightest quarrel is: "It is recognized that a system
of the type now being worked on will be initially beyond the
reach of most of general aviation." It is my personal conviction,
and others at the Wilcox Company in general agree, that with

proper attention to defining the minimum equipment to provide only

the <u>basic function</u>, general aviation can, and will, rapidly become equipped to "play in the system."

The functional minimum, which is referenced, concerns:
minimum required timing accuracy, a simple and direct method
of resynchronization, use of lower numbered time slots, the
simplest method of generating and transmitting the altitude signal,
a loop gain consistent with aircraft speed, a single transmitter
frequency, and the elimination of all on-board memory and computation
except that required for the simplest of escape maneuvers.

Each of the factors mentioned in the above paragraph will be discussed later in the light of each factor's effect on any aircraft in the system as well as on the aircraft carrying the minimum equipment.

Timing Accuracy

The timing accuracy, which is proposed in the ATA draft, is one part in 10⁸ when participating in the CAS environment. This is roughly equivalent to ±10-foot error in the distance measurement or ±5-knot error in the closing rate measurement. For the computation of possible collision courses for general aviation aircraft, this accuracy is more than sufficient. It is, however, fairly practical to achieve and is considered the minimum accuracy consistent with adequate protection for the high-speed aircraft.

To maintain a short-time stability of one part in 10^8 over a 3-second period is not exceedingly difficult; the major problem is determining the start of the 3-second period (epoch) with an accuracy equal to, or greater than, the short-time stability.

If we assume that the epoch start can be accurately established by some other means, and also assume that the crystal's instability is cumulative over the 3-second period, it follows that aircraft with minimum CAS equipment should be assigned the first message slots in the 3-second interval in order that the problem is not compounded. Aircraft which carry atomic clocks or time standards better than the minimum would be assigned slots later in the epoch.

As mentioned above, the determination of the epoch start must be accomplished by some means other than the basic long-time crystal stability. The ATA draft of CAS presupposes some automatic resynchronization to be supplied the minimum CAS equipment. Systems have been operated by Sierra Research, McDonnell Company, and others which have demonstrated the feasibility of resynchronization methods as well as automatic error correction preceding the actual synchronization. The technique of resynchronization with another aircraft or with a ground station will allow the design of relatively inexpensive oscillator circuitry which will be capable of maintaining the desired accuracy. As long as the system parameters are specified in such a manner as to allow the presently envisaged digital circuitry in the resynchronization loop, the manufacturing complexity will not be a major obstacle.

Loop Gain

The transmitter power requirement and receiver calculations in Attachment 3 to the ATA draft are believed to be influenced by airline type aircraft, especially when one considers the 40-nautical-mile minimum operation capability and the 5 db cable

loss between the antenna and RF plumbing. The latter indicates a cable length of 40 to 50 feet which is not required by the small aircraft. The figure is estimated to be about 2 db, including effect of antenna VSWR, for both the receiver and the transmitter, or a difference of 3 db.

The operational range of the small aircraft can also be reduced; however, not in direct proportion to the aircraft's speed because the small aircraft must protect, and be protected against, the high speed supersonic craft. Assuming the speed of the small aircraft to be negligible in comparison to the supersonic aircraft, the range could conceivably be halved. This would establish the power at one-fourth that specified in the ATA draft, or about 380 watts. If one also excludes the 3 db cable loss, the 380 watts could drop to 190 watts and still provide 20-nautical-mile coverage. A transmitter providing 500 watts ±3 db at 1,600 kHz, similar to the present ATC transponders, presents few design and manufacturing problems and will provide better than the 20-mile coverage when cooperating with an ATA specified receiver.

The receiving levels are analyzed similarly; however, the 3 db advantage one obtains via the shorter RF cable is nullified by other factors. In inexpensively produced receivers, it is difficult to realize the 1/2 db RF circuit loss and 7-1/2 db receiver noise factor. The RF preselector and duplexer is presently conceived as being similar to the interdigital stripline filters on a solid teflon base as used in present Wilcox ATC

transponders and DME receivers for general aviation. This technique provides small, light, stable, and very inexpensive RF elements, however, at the expense of about 5 db more front-end loss.

Comparing the minimum CAS receiver to the receiver as calculated in Attachment 3 of the ATA report, we estimate the net difference to be approximately 4 db.

In practice, this can probably be tolerated. When the minimum CAS equipment is cooperating with a supersonic aircraft which has a 1,440-watt transmitter, the minimum receiver would provide service at about 25 nautical miles. This would be comparable to the small aircraft's transmitting range working into the same supersonic aircraft's receiver. When replying or cooperating with another minimum equipment, the range for both would be 12 to 15 nautical miles. We believe this would be satisfactory for aircraft whose speed is below 300 knots.

Single Transmitter Frequency

Although ATA's proposed collision avoidance system allows for the use of four transmitting frequencies, the present draft allows the minimum equipment to be located on one common frequency. Therefore, excluding the problem of maintaining frequency via a precision crystal oscillator, the transmitter is equivalent to the present ATC and DME transmitters being used in general aviation equipment.

Data Capability

The minimum CAS equipment should have the capability to handle only that data required to perform its basic function. There are

two parts to its basic function: (1) to provide other aircraft with information, and (2) to process received information to only that degree required to exercise an adequate escape maneuver.

An SST aircraft might want to know many things from the smaller aircraft; however, if the SST, via a more complicated computer and memory system, extracts all information available from a set of minimum data provided by the smaller aircraft, the complexity of the minimum equipment can be reduced. As the "name of the game" is "Tag", where one intruding untagged aircraft spoils the whole game, it is of the utmost importance to see to it that all computer and memory functions which have any possibility of being removed from the small aircraft be placed at the ground station or on board the larger aircraft.

Using this philosophy, the only information which needs to be transmitted is (1) the timed range burst from which range and range rate can be determined, and (2) data from some on-board measurement by which another aircraft can determine if he is, or shortly will be, on the same flight level.

The ATA draft seems to allow this minimum set of information, yet infers that even the minimum CAS system should be capable of exchanging altitude and altitude rate information. If the term "altitude information" means only the raw data which is proportional to air pressure or air density, then no data reduction need be accomplished on the small aircraft. A safe flight level can be established in terms of pressure or density in lieu of "feet of separation" by each aircraft. If the ground station or the other aircraft desires to know the small aircraft's altitude, the

computation and correction to mean sea level should be accomplished at the end of the circuit where the information in that form is desired. The pressure or density information should come from some means completely independent of the aircraft's altimeter or on-board data system.

As a sidelight, the pressure of density sensing element should be capable of providing an absolute physical measurement which is established in manufacture by some fixed mechanical configuration, preferably requiring no calibration in either manufacture or in the field. The measurement should also be capable of being encoded by simple logic circuitry having no moving parts. Several physical properties of air vary with pressure and/or density which meet the above ground rules. I do not believe that enough effort has yet been spent in the determination of an optimum method of deriving data which varies with the various flight levels. A part of this lack of development can be traced to our shortsighted desire to have flight levels expressed in feet and then to compound this conversion by reducing to some fictitious mean sea level.

I must certainly agree with the person in another former ARINC session who stated that "feet were made to walk on and should have nothing to do with flying."

The ATC transponder, equipped for automatic altitude reporting, has been around for quite a few years; however, only a very small percentage are yet equipped with the altitude encoder. Unless the "flight level" data input is designed and manufactured as an

integral part of the CAS equipment and enjoys a design reliability compatible with the CAS function, it can never be considered a successful device. The Wilcox Company is presently exploring several air pressure and/or density sensing elements which can produce outputs to be encoded by simple solid-state logic. The sensitivity, accuracy, and stability promise to be all that is desired.

Conclusions

Present-day technology has demonstrated its ability to provide collision avoidance protection. Many of the military's station-keeping projects and McDonnell's operating EROS system attest to this.

The cost of the airborne equipment to implement such systems has, to date, been very much above what the average small aircraft owner will pay for such protection. The high cost can, however, be reduced if the system is designed to accept basic minimum inputs.

Also, the Wilcox Company is convinced that the general aviation user will equip his aircraft to fit the basic system if, in return, the system provides him with that degree of protection which he would expect after spending about the same amount of money on a new car.

COLLISION AVOIDANCE AND THE PILOT

"See and be Seen" Concept

Seeing is a full-time job for every pilot regardless of the type of aircraft being flown. A pilot must visually scan in all directions, constantly.

Keep your windscreen and windows clean and also keep them clear of obstructions, such as solid sun visors and window curtains.

Obstructions to Pilot Visibility Inherent in Aircraft Design

In many instances, the pilot's view is restricted by the inherent design of the aircraft. A window frame, fuselage structure, a wing, a wing strut, or a nacelle, create a blind spot. On some aircraft the forward fuselage restricts the view in front of and below the aircraft. On low-wing aircraft, the pilot's view is restricted below the aircraft; and on high-wing, above the aircraft. Blind spots due to aircraft design are inevitable, but recognizable, and can be compensated for by the pilot.

Never let down, turn, or climb into a blind area. When letting down, turning, or climbing it is advisable to make a slight left or right turn, or an 'S' turn or a rolling maneuver, whichever is appropriate and practical. Also, where applicable, look for converging shadows on the ground or on the cloud cover.

Radar Advisories

When there is less than 3 miles visibility, file an IFR flight plan or stay out of controlled zones. If operating

under marginal visibility flight conditions, take advantage of radar advisories. Contact the appropriate controller (radar), give your identification, position, altitude, heading, destination and type of flight plan. When advised of traffic by the controller, respond in effect with "negative contact" or "have in sight" rather than an ambiguous "Roger."

Vigilance should not be relaxed even though radar traffic service is being provided.

Converging Traffic /

When your aircraft is at a constant angle with another aircraft, or the image of the other aircraft on your windscreen is not moving, a collision is imminent. To estimate the altitude of an intruder aircraft, compare the relative position of the target to the horizon. When the target is at the horizon, it is at your altitude. If the target is lower than the horizon, it is at an altitude lower than yours. A target above the horizon should be higher than you.

Once you have spotted an aircraft, don't concentrate on it to the exclusion of other aircraft. Keep track of known traffic, but continue to look for others.

Visual Scanning

The proper technique for daylight visual scanning is for the pilot to systematically move his head and eyes over the entire area of visibility. Using this technique, any contrast or movement in the area of sight will be readily noted by the pilot. Visual scanning at night requires a different technique.

The pilot should depend almost entirely on his peripheral vision.

He should, without staring for more than a few seconds at a time,
look first in one area without moving his eyes and then to other

area and so on. Any light in the area scanned will be noted.

An excellent aid, both in daylight and at night, to the pilot in visual scanning is the high-intensity flashing white light. If you pilot an aircraft equipped with such a light, for your own protection, it is suggested that the light be on at all times while the aircraft is in flight.

Designated Altitude

Always fly at the designated altitude, and remember, even thousands plus 500 feet altitudes westbound and odd, plus 500 feet altitudes eastbound. Below 3,000 feet (AGL) you're on your own. Update your altimeter setting as often as practicable.

High-Density Areas

When flying cross country, avoid high-density areas unless landing. When approaching an airport, call the tower at least 15 miles out and give your aircraft type, "N" number, position, and your intention. If en route, keep 3,000 feet or higher over the airport or well clear, laterally, and call the tower when clear of the "local traffic" area. If landing, be precise in the pattern. Make your turn precisely into the final approach course and stay in line with the centerline of the runway, especially where there are parallel runways. Remember, 65 percent of midair collision accidents occur around airports.