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SPECIAL STUDY

CARBURETOR ICE IN

GENERAL AVIATION

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NATIONAL TRANSPORTATION SAFETY BOARD Washington, D. C. 20591

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Special Study

CARBURETOR ICE IN GENERAL AVIATION

Adopted: January 19, 1972

NATIONAL TRANSPORTATION SAFETY BOARD
Washington, D. C. 20591
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16.Abstract

15. Supplementary Notes

The cartureter ice to which reciprocating engine installations are conceptible as a probable cause or factor in 3th general aviation accidents ever a recent impour period. In these accidents there were a retallities and the people injury out of a total of the percent about. Such losses as these can be removed through the exercise of preator pilet awareness and viriliance. Includes in this report are descriptions of conditions community to carbonater leins, moves of carbonater leins, and procedures for electmentally power loss and to carbonate through the believed that reduction of carbonater leins accidents in attainable through further pilet education, and that the most effective means of accomplication this would be for the Federal Aviation Aministration to send a carbonater loss gives used of its registered pilots.

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CARBURETOR ICE IN GENERAL AVIATION

I. INTRODUCTION

It is realized and acknowledged that much of the material in this report will be familiar to most pilots. However, the National Transportation Safety Board believes that pilots in general can benefit from fresh exposure to this material, and that some carburetor icing accidents may be avoided through such exposure.

During the latest 5-year period for which complete data are available, there was a total of 360 general aviation accidents involving carburetorice as a probable cause or factor. There were 40 fatalities and 360 persons were injured, 40 of them seriously, in these accidents. The number of persons exposed to death or injury in these accidents was (36; 47 mireraft were destroyed and 313 others substantially samages.

"Carturator ice," as used herein, means ice at may location in the induction systems of strength equipped with reciprocal that engines. The term is frealfilled. It is used in afterest accident research, which has many reclined for engine location location rates to introduce reflect the time of the engine of the form of the engine of the

Along it uncolars in it a that concept on the recommendation of the state of the st

These leaded resulting from earliester leing each terraness by an iter awareness and vigitance by pilots. The Carety Board has lengthese expected with earliester leing as one of the "unsectionry" ended factor. It is complayed for accidents. Unlike menanteal failure over write the pilot an little in-flight control, cariorster leing can be avoid a torque to proclude it are readily availables.

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II. DISCUSSION

Awareness of carburetor icing conditions sometimes requires extra effort because of unique combinations of weather, engine installation, and engine operation. However, through familiarity with certain general information, modified or supplemented as necessary for the particular aircraft model being operated, the informed pilot should be able to avoid carburetor icing troubles.

It is important for pilots to know the three categories of carbureter ice, and the manner in which each is formed. These categories are impact ice, fuel ice, and throttle ice.

Impact Ice

Impact ice is formed by the impingement of moisture-lader air at temperatures between 15° F. and 32° F. onto the elements of the induction system which are at temperatures below approximately 32° F. Under these conditions ice builds up on such components as the air scoop, heat valve, earburetor screen, throttle, and carburetor metering elements. Pilots should be particularly alert to such icing when they are operating in snow, sleet, rain, or clouds. The ambient temperature at which impact ice can be expected to build up most rapidly is about 25° F., when the supercooled moisture is still in a semi-liquid state.

Succeptibility to impact ice is greatly reduced by an induction system that has been designed to eliminate free water by means of inertia separation.

Fuel Ice

Fuel ice forms at and downstream from the point at which fuel in introduced, when and if any entrained moisture reaches a freezing temperature as a result of cooling of the mixture by fuel vaporisation. This ecoling pricess takes place in the aircraft induction system when the heat necessary for fuel vaporization is taken from the currending air, thus cooling the air. Then, since cooler air can held less water valor, the excess water is precipitated in the form of condensation. Further vaporization cooling freezes the condensate. When any structure, such as an adapter elbow, lies in the path of the water at time of freezing, lee accretion is initiated on that structure. If this condition continues and no anti-icing action is taken, the ice buildup will increase until the oistruction throttles the engine.

Visible moisture in the air is not necessary for fuel leing; only air of high humidity is required. This fact, coupled with the fact that fuel leing can occur at high ambient temperatures, makes this type of leing cometimes difficult for a pilot to believe unless he is fully mane of the fuel leing process. It can occur in no more than scattered clouds, or even in bright cumchine with no sign of rain.

The usual range of ambient temperatures at which fuel icing may be expected to occur is 40° F. to 80° F., although the upper limit may extend to as high as 100° F. A temperature of around 60° F. should be regarded as the most suspect. The minimum relative humidity generally necessary for fuel icing is 50 percent, with the icing hazard increasing as the humidity level increases.

Fuel ice is not a problem in systems designed to inject the fuel at any location beyond which the passage surfaces are maintained above freezing. Thus, injection of fuel directly into each cylinder obviously will preclude the possibility of such iding. In engines with centrifugal superchargers, the fuel is introduced at or downstream from the face of the impeller in a manner that will avoid splashback from the impeller blades so that fuel ice is not formed.

Throttle Tce

Throtthe ice is formed at or near a partly closed throtthe when water valor in the induction air condenses and freezes, due to the expansion cooling and lower pressure as the air passes the restriction imposed by the throttle. This temperature drop normally does not exceed 5° F. When the unbient temperature is above 37° F., then, the pilot need not be concerned with throttle icing as long as only air passes the throttle, such as in a fuel injection installation with the fuel introduced downstream from the throttle.

When there is a fuel-air mixture at the throttle, nowever, any ice formation would be attributable to water vager freezing from the ormalative effects of the fuel les and throttle ice plessmena. Tolast at the throttle that can occur at ambient temperatures much higher than 30°F. Throttle les is not a problem in some fael systems which are ussigned so that the throttle is leasted in a warmed region, such as between an engine-campe outpercharger and the cylinders, or a constant supply of heat is previded to the throttle accomply and dewastreer curfaces in some other manner.

Corcurator Tea Fernation and Provention

Any one or combination of these ice-forming situations may cause ices of power by restriction of insuction flew and interfer one with an appropriate fuel-air ratio. One reason it can be important to use carbureter that an anti-icer rather than a delect lies in the "vicious circus" aspect, especially in fact-forming conditions and when the ise tuiting might not be diagnosed at an early stage. An uncorrected cartureter ice especially in one, mean tens power, and thus reduces earlies to hear which may result in the formation of more ice. It is containly any present to guard against a bailing of carburster ice before deleting equality is 1 of.

For a terror conception of carturater les formation that might to expected in light-plane systems, the results of tests reported in Reference 1. At this report are summerized to follows:

Two typical light-plane installations were tested, one with a float-type carburetor, the other with a pressure-type carburetor. With the first, serious icing occurred up to carburetor air temperatures of 62°, 63°, and 93° F., and the lower limits of relative humidity of 80, 60, and 30 percent, for high-cruise, low-cruise, and glide-power conditions, respectively. With the pressure-type carburetor installation, the results were serious icing between carburetor air temperatures of 48° F. and 55° F. with relative humidity from 90 percent to 100 percent at low-cruise power, and up to approximately 75° F. with relative humidity greater than 32 percent at glide power. No serious icing occurred at the high-cruise power condition.

Carburetor air heaters in small aircraft are usually of the exhaust pipe cuff type. The exhaust-heated air is directed into the earburetor air duet as desired, so that with full carburetor heat the normal air duet is essemially closed off at the carburetor heat valve location.

It should be realized that partial carburetor heat can be worse than none at all under certain conditions. For example, the fuel/air mixture temperature might be at 25° F., with no heat applied, which normally would not be so conducive to bee-forming as if the temperature were brought up to 35° F., by means of partial heat. Full heat, of course, could be expected to raise the temperature out of the iding range entirely. At least with the smaller engine installations when there is no carburetor fir temperature or fuel-air mixture temperature instrumentation, the general raise should be to use full heat whenever may carburetor heat is applied. With the higher output engines and those employing superchargers, more discretionary use of full heat should be practiced because of the overheat and detenation hazard. Temperature instrumentation should be installed as a necessary reference to assist the plict in modulating the appropriate amount of heat.

Exceptive Use of Carburster Heat

Howerhortending the importance of using carburstor heat when hereograpy, the importance of guaraing against usual everuse should be recognized. This is based on the lower powers and higher cylinder temperatures that generally result when carbureter heat raised industion air temperature. For example, the lower power can be critical in case a switch go-around in required, and full carbureter heat at high-power levels and high-ambient temperatures can cause cylinder everheating and even detoint ion drange. It is noted that under high-power conditions carburator heat in rarely required.

There are exceptions to the rule that earthreter heat application results in lever power. In extremely cold and any worther, with no loing potential, the age of a little carturefor heat may accountly increase power to a small extent receipt of improved fuel vaporization. This reversal of the ageat, newsver, would not occur in most discalibles.

From the above, it can readily be seen that induction temperature instrumentation serves not only to assist the anti-icing effort, but also to protect the engine from overheat damage.

Appraisal of Carburetor Icing Fotential

Cognizance of the prevailing humidity is basic to the important awareness of the possible carburetor icing hazard. Even though relative humidity is less than 50 percent at takeoff, one cannot be completely confident that he will not encounter a carburetor icing atmosphere some time during his flight. Whenever the pilot has reason to suspect high or marginal humidity, he should utilize the best means available to maintain cognizance of prevailing humidity levels.

When the aircraft is equipped with induction temperature instrumentation, humidity level awareness is somewhat less essential.

Operational Indicators of Carburetor Ica

Carburetor ice should be considered immediately as the possible cause of a power 'oss. With a fixed pitch propeller installation, a power loss obviously is indicated by an engine speed reduction. When there is a manifold pressure gage provided, a reduction in manifold pressure would show up along with the engine speed reduction. With a constant speed propeller installation, however, only the manifold pressure would be decreased.

Another way an iced carburetor condition might be first noted is through development of a slight nosedown attitude. Upon trim adjustment to level flight, an engine speed reduction might then be noted, again assuming a fixed pitch propeller.

Finally, an icea carburetor might cause engine roughness although in come cases roughness will not show up until the engine is close to complete stoppage.

Susceptibility to leing

The susceptibility to carboretor ice varies greatly among the various aircraft models. For example, an engine installation employing a float-type carboretor and having the fuel introduced upstream from the throttle valve, would be the most susceptible to carboretor icing troubles. At the opposite end would be an installation with direct cylinder fuel injection, which would forestall the generally most troublesome fuel-type icing; however, the induction system with this might still be subject to impact icing and throttle icing.

It is theoretically possible to design an engine installation that would not be subject to ensure terifing. In practice, however, this ideal is not attained, for one or more reasons of a practical nature. Consequently, it is still incumment upon and pilots to remain alert to the possibility of

carburetor icing, and take preventive action as appropriate for the equipment and conditions.

Recognizing the fact that some installations require less carbureter ice concern than others, the procedural rules listed at the end of this section can only be generally applicable in guarding against carbureter ice troubles.

Prevention Procedures

To prevent accidents due to carburetor icing, there should be routine use of carburetor heat under certain operational conditions, plus awareness and appraisal at other times of possible icing conditions in the friuction system, and the consequent need for carburetor heat as appropriate.

Carburstor icing troubles can be avoided by practicing the following procedures:

- 1. Periodically check carburetor heat systems and controls for proper condition and operation.
- tart engine with carburetor heat control in the "cold" position, to avoid possible damage to the carburetor heat system.
- 3. As preflight item, check carburetor heat availability by noting heat "on" power drop.
- 4. When the relative humidity is above 50 percent and the ambien, temperature is below 60° F., use carturator heat immediately before takeoff. In general, carburator heat should not be used during taxi because of possible firely matter that when intake air is unfiltered in the "alternate" or parterator heat "on" position.
- 5. Consum takeoff without carburator peat unless extrame early to or leing conditions are present, when carburater heat may be used if approved by alreraft manufacturer, and when conditions are such that there will still be angle power for takeoff without incurring engine overhead impact.
- 6. Remain alert after takeoff for indications of carbureter icing, especially when the relative humidity is accressed percent, or when visible moisture is present.
- 7. With supplemental instrumentation, such as a carour for eir temperature gage, partial carbureter hear should be used as necessary to maintain safe temperatures to forestall icins. Wi hout such instrumentation, use full heat but only intermittently if considered necessary.

- 8. If carburetor ice is suspected of causing a power loss, immediately apply full heat. Do not disturb throttle initially, since throttle movement may kill engine if heavy icing is present. Watch for further power loss to indicate effect of carburetor heat, then rise in power artice melts.
- 9. In case carburetor ice persists after a period of full heat, gradually move throttle to full open position and climb aircraft at maximum rate available in order to obtain greatest amount of carburetor heat. If equipped with mixture control, adjust for leanest practicable mixture, (approach this remedy with caution although carburetor ice generally serves to enrich mixture, the reverse can be true; if the engine is lost through excessive leaning, an airstart might be impossible with an iced induction system).
- 10. Avoid clouds as much as possible.
- 11. In severely iced conditions, and when equipped with mixture control, backfiring the engine can constimes be effective in dislodging induction system ice. With carburetor heat control "off," lean engine while at full throttle (observe caution note in No. 9, above).
- 12. Consider that carburetor icing can occur with ambient temperature as high as 100° F. and humicity as low as 50 percent. Remain especially alert to carburetor icing possitities with a combination of ambient temperature below 70 F. and relative humidity above 80 percent. However, the possibility of carburetor ice decreases in the range below 32° F. This is because of (a) lessened humidity as the temperature decreases, and (b) at around 15° F. any entrained moisture ice mas ice crystals which pass through the induction system harmlessly. It should be remembered that if the intake vir does contain these ice crystals, carburetor heat might actually cause carburetor icing by melting the crystals and raising the moisture-lader air to the carburetor icing temperature range.
- 13. Prior to closed-throttle operation, such as for a descent, apply full heat and leave on threshout throttles requence. Periodically, open throttle suring extended closed throttle operation so that enough explications will be produced to prevent leing. Be prepared to remove confused to go-around is initiated.

14. Return control to "cold" position immediately after landing. If carburetor heat should be further required, observe ground operation precaution in (4), above.

III. CONCLUSIONS AND FINDINGS

The National Transportation Safety Board, from its study of the carburetor ice problem area in general aviation, has concluded that:

- 1. Many accidents induced by carburetor ice continue to occur, despite the fact that the means of preventing carburetor ice are available to use at the pilot's discretion.
- 2. The incidence of carburetor icing can and should be reduced by further pilot education.
- 3. Distribution of an Advisory Circular on carburetor ice must be made to all pilots in order to reduce significantly carburetor ice involvement in alreraft accidents. This broad coverage is required because, even though only a very small percentage of all pilots can be expected to get into serious trouble with carburetor ice, there is no way of predicting which particular ones these will be.

IV. RECOMMENDATIONS

In view of these findings, the Safety Board recommends that:

- 1. The Federal Aviation Administration prepare an Advisory Circular on the prevention of carburetor ising in reciprocating engines used on general aviation aircraft.
- 2. The FAA mail this publication to all general aviation pilots, flight instructors, and flight schools.

BY THE NATIONAL TRANSPORTATION SAFETY BOARD:

/s/	JOHN H. REED	Chairman
/s/	OSCAR M. LAUREL	Member
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January 19, 1972