NTSB ID: LAX98FA106 Alrcraft Registration Number: N257NW

Occurrence Date: 03/05/98 Most Critical Injury: FATAL

Occurrence Type: Accident Investigated By: NTSB

Location/Time

Nearest City/Place	State	Zip Code	Local Time	Time Zon	е
NOVATO	CA	94947	1905	PST	
Accident Location: Off Airport	Distance	From Landing	Facility: UNK/NA	Direction	From Airport: UNK/NA

Aircraft Information Summary

Aircraft Manufacturer	Model/Series	Type of Aircraft
Piper	PA-31-350	Airplane

Sightseeing Flight: No Air Medical Transport Flight: No

Narrative

Brief narrative statement of facts, conditions, and circumstances pertinent to the accident/incident:

HISTORY OF FLIGHT

On March 5, 1998, about 1905 hours Pacific standard time, Airpac Flight 263, a Piper PA-31-350, N275NW, collided with trees and terrain near Novato, California. The aircraft was destroyed and the airline transport rated pilot, the sole occupant, received fatal injuries. The aircraft was being operated by Airpac Airlines, Inc., as a positioning flight under 14 CFR Part 91 when the accident occurred. The aircraft departed Santa Rosa, California, at 1848. Marginal visual meteorological conditions prevailed at the departure airport and no flight plan was filed.

The airplane was part of a flight of two company aircraft that departed the Sonoma County airport en route to the Oakland Metropolitan International airport. Airpac Flight 1961 was in the lead and Airpac Flight 263 was in trail, about 1 minute behind. The pilot of Flight 1961 stated that they were both navigating by a combination of dead reckoning and pilotage. They both had VOR navigation aids on their routes of flight that could be used to verify their positions and monitor progress.

During the flight, the pilot of Flight 1961 radioed the pilot of Flight 263 and asked about her progress. She reported that she was "still back here." Later, while he was about 20 miles north of the Oakland airport, he began to encounter instrument meteorological conditions. He called to advise the accident pilot, but this time he was not successful in establishing radio contact. Flight 1961 landed at the Oakland airport at 1920.

When Flight 263 did not arrive, the flight was reported overdue and a missing aircraft report was issued. At 0148 on March 6, a U.S. Coast Guard search and rescue helicopter, homing on an ELT signal, located the wreckage near the northern crest of a hill. The accident site was approximately 1,500-feet mean sea level (msl), about 40 miles north of San Francisco.

The pilot of Flight 1961 reported that the accident site was about 5 miles west of the route that Airpac pilots routinely flew between Santa Rosa and Oakland. While the westerly route is a more direct course to Oakland, it does, however, cross higher terrain. The pilot of Flight 1961 also reported that during the flight he found it necessary to descend between 1,200 and 1,500 feet msl in order to maintain VFR conditions.

PILOT INFORMATION

The chief pilot reported that the pilot had flown between Santa Rosa and Oakland 2 to 3 times a week for

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the past 2 years. A review of her training records showed a Part 135 check dated September 19, 1997, and revealed that she had successfully passed a flight check in the aircraft and route flown as single pilot IFR. No training deficiencies or weaknesses were recorded in her file.

On the evening of the accident, she was scheduled to give a speech as her final examination in an evening college course. She had informed the instructor that she might be late, and had been told that he could not hold the class past its scheduled dismissal time to accommodate her late arrival.

AIRCRAFT INFORMATION

Both altimeters on the accident aircraft were set to 29.69 inHg. The current altimeter setting at 1850 at the departure airport was 29.78 inHg. Since each .01 equals 10 feet, an error of this magnitude and direction would cause the altimeter to indicate that the aircraft was 90 feet higher than it would be with the current setting.

The aircraft records reflected a current pitot-static system check.

The aircraft had last been refueled on the day of the accident. The fuel log shows that it was filled with 92.8 gallons of 100LL aviation fuel by truck No. 2 at the Hayward Jet Center, Hayward Air Terminal, Hayward, California. The operator estimated that 70 gallons were onboard at the time of the accident.

METEOROLOGICAL CONDITIONS

The 2100 surface analysis chart showed a low pressure approaching the coast from the west. The 1904 Weather Surveillance Radar showed returns of 5 to 25 dBZ or video integrator processor (VIP) level 1 thunderstorm (weakest of 5 levels). At the time of the accident, Napa (APA) the nearest reporting station, reported a 1,300-foot broken ceiling with 5- to 6-mile visibility in light rain and mist.

Sun and moon illumination tables for the time and location of the accident reported nautical twilight with 58 percent moon illumination at an altitude of 69.3 degrees above the horizon.

AIDS TO NAVAGATION

The navigation radios, NAV 1 and 2, were set to 116.80 and 116.60. Oakland visual omni range tactical air navigation (VORTAC) is 116.80; however, 116.60 does not correlate to any very high frequency (VHF) navigational aid in the area. The omni bearing selector (OBS) settings were destroyed on both NAV radios.

The automatic direction finder (ADF) was set to 341 which is the frequency for the RORAY nondirectional beacon (NDB) locator middle marker (LMM). RORAY is associated with the OAK instrument lading system (ILS) 27R approach.

These NAVAIDS were reported to have been in service at the time of the accident. The aircraft maintenance records indicated that all navigation radios were serviceable.

COMMUNICATIONS

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The only radio communication reported from the accident pilot was in response to a request for an en route progress check from the pilot of Flight 1961.

The communication radios, COM 1 and 2, were set to the Sonoma County airport tower frequency and the company frequency, respectively.

The pilot had last used the Sonoma County tower frequency on her departure and the company frequency while en route. The aircraft maintenance records indicated that both communication radios were serviceable.

WRECKAGE AND IMPACT INFORMATION

The accident site was located in an area of fully-grown oak trees. The geographic location was 38 degrees 9.19 minutes east latitude and 122 degrees 35.48 minutes west longitude, about the 1,500-foot level of Mount Burdell. The wreckage distribution was along a bearing of 144 degrees and extended about 300 feet from the initial point of impact to the main wreckage. A survey of the debris field revealed that all of the major flight control components were present at the accident site.

The trees on the northwest end of the accident site exhibited a series of broken limbs of up to 10 inches in diameter. The width of the initial swath through the trees was consistent with the tip-to-tip wingspan of the aircraft.

A section of the left wing was found to the left side of the swath near the initial impact point. The aileron was attached and the counterweight was in place. The wing had separated outboard of the fuel tank near the outboard flap hinge.

On the right side of the swath a section of the right wing was found, also near the initial impact point. The section was 3 feet in length measured inboard from the wingtip. A few feet beyond another section of the right wing with the right fuel tank and outboard engine nacelle were found.

The empennage was located about 50 feet beyond the initial impact point. Found nearby was one blade from the left propeller, two propeller hub springs, a "do not push" placard, and a section of burned wing tank with the fuel sender float still attached. Although the tank was sooted, the metal structure did not exhibit any evidence of heat distress

The majority of the right horizontal stabilizer remained attached to empennage with the elevator and trim tab in place. The major portion vertical and rudder also remained attached and exhibited leading edge crushing. The upper section of the vertical and rudder had separated. A portion of the left horizontal and elevator remained attached to the empennage. The left horizontal had separated about 2 feet outboard of the tip. At the point of separation the spar was bent aft.

The next 5 feet of the left wing contained the left main landing gear assembly in the "up" position. Wing walk material partially covered the upper skin.

The cabin and fuselage section came to rest inverted about 75 feet forward of the tail section. It exhibited extensive crushing and fragmentation. Numerous fragments could be seen along the debris path extending forward from the tail section. Evidence of a ground fire was found that had consumed portions of the

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remaining structure.

The throttle, mixture, and propeller levers were found in the aft position. The electronic trim indicators were both at bottom of their scales. The fuel control panel was still attached to the cabin. The left selector was found in the "inboard" tank position, the right fuel lever was found between the "off" and "inboard" tank detents, and the cross-feed valve was found in the "off" position. Both cross-feed lockout levers were examined. The first lever was bent while the second was secure.

The electrical circuit panel had been destroyed by fire. The master switch was found in the "on" position.

A fuel gage cluster had separated from the aircraft and was in the debris path between the fuselage and the empennage. The left and right fuel gauges read about 0.5 to 0.75 full. The ammeter was reading about plus 5 amps.

The inboard section of the right wing with the engine nacelle still attached and right main landing gear partially extended was found leaning against a tree. The nacelle exhibited evidence of ground fire.

The left front seat was separated from the aircraft. The seat exhibited downward crushing and the seat back was bent aft. The seat belt was attached to the seat frame although the webbing was distorted and torn. The shoulder harness was separated about 6 inches from the buckle. The webbing was torn, stretched, and distorted at the point of separation.

The right seat was also separated from the aircraft. The shoulder harness did not appear stretched or distorted

The left engine was separated from the airframe and was found on the left side of the primary debris path. The accessories, baffles, propeller, and induction system had separated from the case, although portions of the exhaust system were still attached. There was no evidence of catastrophic mechanical malfunction/separation. The No. 2 cylinder rocker box casting was damaged and the two valves were protruding from their respective ports without support.

An attempt to hand-rotate the crankshaft was unsuccessful. The rocker covers from the No. 1, 3, 5, 4, and 6 cylinders were removed to examine the rocker assemblies. The cavities were free of foreign debris and showed indications of lubrication.

The turbocharger impellers and spark plug electrodes were undamaged.

The propeller governor was separated from the engine and exhibited fractures at the point it separated from its mounting pad. Control system continuity could not be determined. The gasket and screen were destroyed.

The single drive magneto was separated from the engine and exhibited fractures in the area of its mounting lobes. The engine timing could not be determined and the magneto could not be tested. The drive was saftied and secure.

The fuel injection servo remained attached to its mounting pad on the plenum casting. The plenum, itself,

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exhibited multiple fractures. Control continuity could not be determined. The throttle and mixture control cables were severed but the rodends were still attached to the servo arms.

The injector nozzles were present at each cylinder but exhibited evidence of impact damage.

The engine-driven fuel pump remained attached to its mounting pad that itself had separated from the accessory section.

The exhaust system showed evidence of crushing and bending. The pipes remained attached at each cylinder and the clamps were snuggly attached. The exhaust bypass valve was undamaged. The differential controller was destroyed. The exhaust system gas path coloration was unremarkable.

The left 3-bladed constant speed propeller was found along the left side of the debris path with 1 blade detached. The separated blade was found in the vicinity of the initial impact point. The propeller was separated from the engine and the propeller hub was fractured. A portion of the hub remained attached to the crankshaft flange. All three blades were equipped with de-icing boots.

The right engine was separated from the airframe and was also found on the left side of the primary debris path. The accessories, baffles, propeller, and the complete induction system had separated from the case. The engine case still had portions of the exhaust system attached. The case was separated near the front exposing the crankshaft, camshaft, and connecting rods for the forward cylinders. The camshaft was separated aft of the No. 2 cylinder exhaust lobe. The No. 1 cylinder exhibited impact damage and was partially displaced. The No. 6 cylinder exhibited impact damage in the area of the rocker box casting. There was no evidence catastrophic mechanical malfunction.

An attempt to hand-rotate the crankshaft was unsuccessful. The rocker covers from the No. 1, 3, 5, 2, and 4 were removed to examine the rocker assemblies. The cavities were free of foreign debris and showed indications of lubrication.

The turbocharger impellers and spark plug electrodes were undamaged.

The propeller governor was separated from the engine and exhibited fractures at the point it separated from its mounting pad. Control system continuity could not be determined. The gasket and screen were destroyed.

The single drive magneto was separated from the engine and was destroyed. The engine timing could not be determined and the magneto could not be tested. The drive was saftied and secure.

The fuel injection servo was separated from the engine and had been destroyed. The data plate was missing and was not recovered. Control continuity could not be determined. The throttle and mixture control cables were severed and the control arms were detached.

The injector nozzles were present at each cylinder but exhibited evidence of impact damage. The engine driven fuel pump was not recovered.

The exhaust system showed evidence of crushing and bending. The pipes remained attached at each

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cylinder but were distorted. The clamps were secure at each location. The exhaust bypass valve was undamaged. The differential controller was destroyed. The exhaust system gas path coloration was unremarkable.

The right 3-bladed constant speed propeller was found near the main fuselage with all 3 blades attached to the hub. The propeller was separated from the engine and the propeller hub was fractured. A portion of the hub remained attached to the crankshaft flange. All three blades were equipped with de-icing boots.

The three blades exhibited various combinations of S-bending, torsional twisting, midspan bending and curling with leading and trailing edge damage. Of the three blades, two were turned in the hub while one was found at low pitch.

MEDICAL AND PATHOLOGICAL INFORMATION

An autopsy was conducted on March 6, 1998, by the Marin County Coroner's Office, with specimens retained for toxicological examination. The toxicological tests were negative for alcohol and all screened drug substances.

FIRE

Evidence of ground fire was found most notably in the area of the right wing and fuselage.

TESTS AND RESEARCH

Utilizing radar data provided Federal Aviation Administration Quality Assurance office a radar plot was generated depicting profile altitude and distance information. Radar data showed the aircraft as it departed Santa Rosa and climbed to about 2.000 feet msl. It maintained 2.000 feet plus or minus 100 feet until beginning a descent. The last three returns showed the aircraft in a descent from to 2,000 feet until radar contact was lost about 1,500 feet msl. During the last three returns, the aircraft traveled about 1,94 nautical miles. Radar returns are sequenced once every 12 seconds.

ADDITIONAL INFORMATION

The aircraft wreckage was released to Peter Gordon of the Thomas Howell Group, a representative of the registered owner on April 21, 1998.

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TACTUAL REFO	1/1	L											
AVIATION		Occı	ırrence ⁻	Туре	: Accide	ent							
Landing Facility/Approa	ch Inform	ation											
Airport Name			Airport II	$A \mid A$	irport Ele	vatio	n Runw	/ay Used	Runw	ay Le	ength	Run	way Width
UNK/NA			U/NA	\	J/NA Ft	. MS	L UN	IK/NA	UN	IK/N/	۸	U	NK/NA
Runway Surface Type: UNK/I	٧A												
Runway Surface Condition :	UNK/NA				•							٠	
Type Instrument Approach:	UNK/NA											-	
VFR Approach/Landing :	UNK/NA												
Aircraft Information													
Aircraft Manufacturer			Mod	del/S	eries					Seria	al Nun	nber	
Piper			F	PA-3	1-350					,	31-79	5201	4
Airworthiness Certificate :	Normal												
Landing Gear Type : Tricyc	le-Retractab	ole											
Homebuilt Aircraft? No N	umber of Sea	its: 2		Cer	tified Ma	x Gro	oss Wt.	Number	of Eng	gines			
Stall Warning System Installed	? Yes			L	7045		LBS	2	2				
Engine Type			Engine	Mar	ufacture	r		Model/S	eries			Rate	ed Power
Recip-Fuel Injected				Lycoming				TIO-54	10-J2E	3D		350	HP
- Aircraft Inspection Informat	ion												
Type of Last Inspection			Date of	Last	Inspection	on	Time Sir	nce Last I	nspect	tion	Airfra	ame T	otal Time
AAIP			02/12	2/98			3		Н	ours	688	31	Hours
- Emergency Locator Transm	itter (ELT) li	nform	ation							***************************************			
ELT Installed? Yes	ELT Oper	ated?	Yes			EL	Γ Aided	in Locatin	g Acci	dent	Site?	Yes	
Owner/Operator Informa	tion												
Registered Aircraft Owner			Stre	et Ac	ldress								
AIRPAC AIRLINES, INC.					7277	PER	IMETE	ROAD	SOU.	TH			
			City		SEAT	TI E					Sta [*]		Zip Code 98108
Operator of Aircraft			Stre	et Ac	Idress	1					, vv	<u> </u>	30100
•	mor					as F	Registe	red Aircra	aft Ov	vner			
Same As Reg'd Aircraft Ow	riei		City								Sta	te	Zip Code
								<u> </u>					
Operator Does Business As:							Ope	erator Des	signato ———	or Co	de: AF	PCA	<u></u>
- Type of Certificate(s) Held:	- long	oman	nd Taxi		<u> </u>				-				
Air Carrier Operating Certificat	e : OII-D				L_								
Operating Certificate:					Operato	r Cer	tificate:						
Regulation Flight Conducted U		FR 91											
Type of Flight Operation Cond	ucted: Posit	ioning]										
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AVIAT	ION		,	Occur	renc	е Туре:	Acc	ident									
First Pilot Informa	tion																
Name							City		_			Sta	te	Date	e of Birth	7	Age
ISABELLE ARSONI	NEAU					ALAMEDA CA 04/09/						/09/64		33			
Sex: F Seat Occupi	ed: Left		Princ	ipal P	pal Profession: Pilot-Civilian Certificate Nur								umb	er: 2473	046		
Certificate(s) : Con	nmercial			ATP													
Airplane Rating(s):	SE Land			N	1E L	and.											
Rotorcraft/Glider/LTA :	None)												[
Instrument Rating(s):	Airpla	ane															
Instructor Rating(s):		irplane)		ME	E Airpla	ne		Insti	rume	nt						
		•				•											
Type Rating Endorsement for Accident/Incident Aircraft? UNK/NA Current Biennial Flight Review? Yes																	
Months Since Last BFR BFR Aircraft Make BFR Aircraft Model Medical Certificate: Class 1																	
7	PIPE	ER				PA-31			Da	ate of	Last	Med	ical E	xam	n: 08/05	 /97	
Medical Certificate Status: Valid Medical-No Wai						 ⊃ Waivers/Limitations			L							- ·	
Source of Pilot Flight Time Information : FAA Pilot						t Rpt				5////							
- Flight Time Matrix	All A/C	This Make		Airplane ngle Engi		Airplane lulti-Engine	Nig	jht	Ir Actual	nstrume	it Simulate		Rotorcra	ft	Glider		ghter an Air
Total Time	4300																
Pilot in Command (PIC)					_		<u> </u>					_					
Instructor Last 90 Days	00	40			+									\dashv			
Last 30 Days	63 23	19 10				3 23				_		+	_				
Last 24 Hours	1	1			1	.0		-				_					
Seatbelt Used? Yes	Shoulder	Hames	s Use	ed? Ye	es	Autops	sy Per	forme	d? Ye	s	Toxi	colog	ју Ре	rforr	ned? Ye	s	
Person at Controls of A	ircraft at 7	Time of	Accid	dent/Ir	ncide	ent: Firs	t Pilo	t			Seco	nd F	Pilot?	No			
Flight Plan/Itinera	ry			-													
Type of Flight Plan File	ed: None																
Departure Point								State	A	irport	lden	lifier	Dep	artur	re Time	Time	Zone
SANTA ROSA								CA		STS	3			185	60 l	PS	ST
Destination						_		State	A	irport		tifier					
OAKLAND								CA		OA	K						
Type of Clearance : None									, L								
Type of Airspace :	Class G	······································								-							
Weather Informati	on																
Source of Briefing :	UNK/NA																
Method of Briefing :	UNK/NA																
			F	ACT	UAI	REPO	ORT .	- AVI	ATIC	N						P	age 3

AMATION

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l	AVIATION		00	currer	псе Ту	rpe: Ac	cident							
Weathe	er Information											2.2		
WOF ID	Observation Time	Time Zone	WOF	Eleva	ation	WOF	Distance Fi	rom	Accident 8	Site D	irectic	on From A	ccident	Site
STS	1950	PST	125	Ft.	. MSL		24		NM		317		Deg. N	Mag.
Sky/Low	est Cloud Condition:						Ft. AG	}L ∣	Condition	of Lig	jht: [Jusk		
Lowest C	Ceiling: Broken		3000	Ft.	AGL	Visit	oility: 5		SM	Altim	eter:	29.79		"Hg
Tempera	ature: 42 F	Dew Point:	41	F	Winc	d Directi	ion: 330			Dens	ity Al	titude: UN	1K/NA	Ft.
Wind Sp	eed: 5	Gusts: N	lone		Wea	ther Co	nditions at	Acc	ident Site:	Instru	umen	nt Conditio	วทร	
. Visibility	(RVR): Ft.	Visibility	(RVV)		SM	Intens	ity of Preci	pitat	tion:					
Restriction	ons to Visibility:	None											. * .	
Type of	Precipitation :	None												
Accide	ent Information													
Aircraft [Damage: Destroye	d	Air	craft F	ire: C	On Gro	und		Aircraft E	xplosi	ion: N	lone		
Classific	ation: US Register	red on US	Soil, T	erritor	ries or	Posse	ssions, or	r Int	l Waters					
Injury	Summary Matrix	Fatal S	Serious	Mine	or	None _.	TOTAL							
First	Pilot	1					1							
Seco	nd Pilot													
Dual	Student													
Chec	k Pilot													
Flight	t Engineer													
Cabir	n Attendants													ı
Other	r Crew													ļ
Passo	engers													
TOTA	AL ABOARD	1					1							
Other	r Aircraft													
Other	r Ground								•					
GRAI	ND TOTAL	1					1							

FACTUAL REPORT AVIATION

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Occurrence Date: 03/05/98

Occurrence Type: Accident

Administrative Information

Investigator-In-Charge (IIC)

ROBERT R. CRISPIN

Additional Persons Participating in This Accident/Incident Investigation:

RICHARD M. BROWN

WP-OAK-FSDO

OAKLAND

CA

94621

MARK W. PLATT

TEXTRON LYCOMING

VAN NUYS

CA

91406

CHARLES R. LITTLE

PIPER AIRCRAFT

CHINO HILLS

CA

91709

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NTSB Accident/Incident Number

	AV	ATION				1 1 1 1 1 1 1899		
E)				t	_ A X 9	8 F A 1 0 6		
Supplement A	A e Documentation, Si	ngle and Twin F	Reciprocating E	ngine and l	Unpowered .	Aircraft		
1 Engine #1 Serial				charger Install		er 6 Propeller		
RL-9570-61A	1	1 Yes		Yes	Manufac			
A Other	A Other	2 🔀 No	2 [No		rizeli <u>HCE3YR-2ATF</u>		
7 Oller	7 Odio	A Other	A Oth	er	A Other _	A Other		
7 Propeller Type	(Multiple Entry)				B Aircraft STC	OL Modification Installed		
1 Wood		_	Ground adjustable/vai	riable pitch	1 🗌 Yes			
2 Metal			Reversible	ulus 4.	2 🔀 No			
3 Compos	site nt speed controllable pitch		Full automatic Feathe Full manual feathering		A Other			
3	n apara and common process	A Other _						
Landing Gear	9 Nose/Tail	10 Left Mair	n	11 Right Ma	in	For Rotorcraft or		
Positions	1 🔯 Up	1 <u>⊠</u> ∪p	· •	1 ⊠ Մթ		Balloon accidents, go		
(If fixed gear, go to block 12)	2 Down		wn	2 Do		to block 20		
,	3 Intermediate	3 Into		3.		· -		
	A Other	A Other		A Other				
Control Surface Positions	12 Left Trailing Edge Fla		iling Edge Flap	14 Speed Br	1	15 Spoiler 1 Not Installed		
	1 ⊠ Up	1 🔯 Up		-	wed	2 Stowed		
	A Extended deg.	ì	led deg.	3 De	1	3 Deployed		
	B Other	B Other		A Other		4 Deployed Asymmetrically		
						A Other		
Trim Tab Positions	16 Left Aileron	17 Right Ail		18 Rudder		19 Elevator/Stabilator/ Ruddervator		
(Multiple Entry)	1 Not Installed 2 Neutral	=	ot Installed eutral	2 × Ne	ot Installed	1 Not installed		
	3 Up	3 U		3 🗍 U		2 Neutral		
	4 Down	4 D	own	4 🔲 D	own	3 Up		
	Adegrees	A _	degrees	A <u>.5</u>	degr ees	4 Down A Odegrees		
	B Other <u>Destroyed</u>	B Other De	<u>stroyed</u>	B Other	_	B Other		
Cargo Restraint	20 Cargo Restraint Install	ed (Multiple entry)	21 Cargo Restraint	Used (Multiple	entry) 22 Car	go Restraint Failed (Multiple entry)		
Systems	1 None (Go to bid	ock 26)	1 🔯 None (G	o to block 26)		None (Go to block 26)		
	2 Cargo net		2 Cargo no			Cargo net		
	3 Straps/tie down		3 ☐ Straps/ti A Other	e down		Straps/tie down		
	feight and Balance Ir			veight and/a		avity limitations are		
Computed Y	Aldin and parame ii		exceeded on acc					
Takeoff								
26 Weight	27 Center of Gravity	28 CG Range (Multi	ple entry)		<u>, </u>			
Lbs.	A % MAC or	1 At takeoff we	eight	Α	% MAC to	% MAC or		
	B inches	2 At max gross	s weight	В	inches to	inches		
Accident						32 Fuel On Board At Accident		
29 Weight	30 Center of Gravity	1 CG Range (Multipl	le entry)			1 🗵 Estimated		
Lbs.	A % MAC or	1 🔲 At takeoff w	eight A%	MAC to	_ % MAC or	2 Verified		
	B inches	2 At max gros	s weight B ir	nches to	_ inches	A Total Gallons 70		
						B Other		
MEGR	Cd 20 4 Complement							

NTSB Form 6120.4 Supplement A (1-84)

NTSR	Acciden:	l/Incident	Number

3									L	A	(9	8 F	A	1 0	6
Supplement A		kage Dod tinued)	cumenta	ation, S	Single a	nd Twi	n Reci	proca	ting E	Engine	and (Unpov	wered /	Aircrat	ft
Fuel Tanks .	Fuel on	Board at Acc	ldent		Tank Co	atruction	1	F Spli	F Spillsafe Fittings H Fuel Leakage/Rupture						
	A Gallons	B Gallons	C	1 Wet	2	3	E	1	2	G	1	2	3	4	1
	Estimated	Verified	Other	Wing	Bladder	Metai	Other	Yes	No	Other	None	Line	Fitting	Tank	Other
33 Left Wing	30				\boxtimes				\boxtimes	7000		X		X	
34 Right Wing	30				☒				X			Ø	X	X	
35 Left Tip	5				\square				\boxtimes			\boxtimes	\boxtimes	\boxtimes	
36 Right Tip	5				\boxtimes				Ø			\boxtimes	\boxtimes	×	
37 Fuselage		<u></u>													
38 (Specify)										·					
41 Fuel Found in	#1 Engine	! (Multiple er	otry)		<u> </u>	41 F	uel Found	in #2 E	naine (Multiple	entry)	<u> </u>	<u> </u>	<u> </u>	L
1 None 2 Lines 3 Gascol 4 Carbur	1 None 7 Filter(s) 1 None 7 Filter(s) 2 Lines 8 ⊠ Selector Valve 2 Lines 8 ⊠ Selector Valve 3 □ Gascolator/strainer 9 □ Fuel manifold/spider 3 □ Gascolator/strainer 9 □ Fuel manifold/spider 4 □ Carburetor/Fuel Injector 10 □ Accumulator Tank									/spider					
6 Auxiliary fuel pump A Other 6 Auxiliary fuel pump A Other															
43 Flight Controls, Evidence of Operational Failure or Malfunction (Multiple entry) 1 None 1 None 2 Helicopter (Complete Supp. G) 8 Vertical fin/rudder 3 General disintegration 9 Canard 1 None 2 Pitch Control 3 Rolf Control 4 Propeller, Evidence of In-Flight Malfunction 1 None 2 Helicopter (Complete Supp. G) 8 Vertical fin/rudder 3 General disintegration 9 Canard 2 No A Other A Other A Other 4 Paw Control 4 Paw Control 4 Paw Control 4 Paw Control 4 None 4 Other 4 Paw Control 5 Right wing 6 Left stab/elevator 6 In-Flight Malfunction 1 Yes 2 No A Other A Other Multiple Entry) 48 Oil, Evidence of Improper Grade or Contamination (Multiple entry)							nation								
A Other		1 No lmg	ne oroper grad	ie	3 L (Contamin	ation		1 🔀 2 🔲	None Improp	er grade	3 • A	Other _	ntamina ——	CION
Emergency L	ocator T	ransmitte	r (ELT)	inform	ation			•							
51 ELT Manufact				52 ELT		***************************************				55 Prei	mpart	ELTI~	cation(s)	(Multin	e entru\
A Other					ther	ELT 1				1 [2 [3 [Coci	kpit in cone	5 [] 6 []	Raft Surviv	
53 ELT Battery		7	ľ	54 ELT	Battery Ex	piration	Date (No	s for M _s	(ץ,ט	4 L	_ Emp	ennage	,		
1															
3 Nicad				A & Jr. 1						L					
1 Operated 2 Insufficie 3 Improper	2 Insufficient G's 7 Incorrect battery 12 Unit not armed 17 Signal direction aftered by terrain 3 Improper installation 8 Fire damage 13 Shielded by wreckage 18 Packing device still installed														
5 Battery c	оггодед	10 🗌	Antenna b	roken/dis	connected	15	Intern	al failure	·	A Oth	er	_			

NTSB Accident/Incident Number

·		L A	X 9 8 F A 1 0 6 🎇
Supplement B — Cockp Aircra	oit Documentation, Single a	ind Twin Reciprocating E	ngine and Unpowered
1 Cockpit Secured, Readings No	et Pertinent 1 Yes (Go to block 3)	2 Cockpit/Instrument Panel	Destroyed 1 Yes (Go to block 3)
Cockpit Instrument Indication	enter direct in appropria	ite category	
Flight	Instruments	Engine/S	ystem Instruments
Item	Reading/Setting	Item	Reading/Setting
Altimeter	29.69°	Left Tachometer	1,100 Rpm
Standby Altimeter	29.69", 2,440 Feet	Right Tachometer	400 Rpm
Airspeed	140 Kts	Left Main Fuet Gauge	1/2
Hsi	135 Degrees	Right Main Fuel Gauge	1/2
Rmi	350 Degrees		
			ł
	ļ		
	Nav Equipment		scellaneous
Item	Frequency/Remark	Item	Remark
Com #1	118.50	Clock	3:30
Nav #1	116.80		
Com #2	128.90		
Nav #2	116.60		
Adf	341		
	ļ.		
	1		·
NTSB Form 6120.4 Supple	ement B (1-84)		Page

NTSB	Accident/	neident	Number

pl	,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,				LAX	9 8 F A 1 0 6
Supplement B — Cockpi	t Documer	tation. Si	nale and	Twin Recipr	ocating En	
	(Continue		.g.o a.r.o	· · · · · · · · · · · · · · · · · · ·	oddding Eiri	girle area oripowered
3 Navigational Equipment/Display			A Au	topilat	6 Digital Elect	ronic/ 6 Primary Altimeter Type
	LORAN/On		1	Not Installed	Nav/Com Di	
	⊠ DME		2	Engaged	1 Not in	
	⊠ ADF		3 🗆		2 🔀 Install	
= ""	Marker bea	cons	A OH	her Destroyed	A Other	4 2-pointer
	Other		// •	101 200110100		A Other
6 X RNAV					,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,	
	Radar Altime	ter Installed	9 Transpo			10 Attitude Indicator Installed
1 🛛 Yes	1 X Yes		. =	ot Installed		1 🔀 Yes
2 🔲 No	2 🔲 No	•	, =	stalled - not used stalled - used		2 No
A Other	A Other	_	. =	stalled - used - Alt	itude encodina	A Other
			A Other			
11 Attitude Indicator Power Source	e (Multiple entry)	<u> </u>	17 Type	of Stall Warning	Indicator 13	Weather Radar/Detection Equipment
1 Pressure/vacuum system		,	1 [-	, ,	1 Not Installed
2 Pressure/vacuum system		ower source	2	š		2 Installed – on
3 Electrical	.,		3 🗔		1	3 🔲 Installed – off
4 Standby indicator with alte	rnate power sou	ırce		Aural	1	4 🔲 Installed, on/off unknown
A Other			5 _		Į.	A Other
			A 0	ther		
14 Type Weather Radar/Detection			_			
1 Storm Scope 2	Black and	white radar	3 📙 (Color radar	Α	Other <u>N/A</u>
Electrical/System Switche	9			18 🛛 Switch	nes Destroyed/li	naccessible (Go to block 56)
***	,		*******		Positions Not	Pertinent (Go to block 56)
Switch/Item	1 Not installed	2 On	3 Off	A Other		Pertinent Setting/Remark
20 Electrical Master		\boxtimes				
21 Battery				Destroyed		
22 #1 Gen/Alternator				Destroyed		
23 #2 Gen/Alternator				Destroyed		
24 Inverter						
25 Avionics Master						
26 Pitot Heat				Destroyed		
27 Ice Detection				Destroyed		
28 Propeller Deice/Anti-ice	1 —			The street and	1	
	 			Destroyed		
29 Windshield Deice						
29 Windshield Deice 30 Windshield Anti-Ice	X			Destroyed		
29 Windshield Deice 30 Windshield Anti-tce 31 Airframe Deice				Destroyed Destroyed		
29 Windshield Deice 30 Windshield Anti-fce 31 Airframe Deice 32 Cabin Air/Fan				Destroyed Destroyed Destroyed		
29 Windshield Deice 30 Windshield Anti-Ice 31 Airframe Deice 32 Cabin Air/Fan 33 Cabin Heater				Destroyed Destroyed		
29 Windshield Deice 30 Windshield Anti-tce 31 Airframe Deice 32 Cabin Air/Fan 33 Cabin Heater 34 Air Conditioning				Destroyed Destroyed Destroyed		
29 Windshield Deice 30 Windshield Anti-tice 31 Airframe Deice 32 Cabin Air/Fan 33 Cabin Heater 34 Air Conditioning 36 Cabin Pressure Altitude				Destroyed Destroyed Destroyed		
29 Windshield Deice 30 Windshield Anti-Ice 31 Airframe Deice 32 Cabin Air/Fan 33 Cabin Heater 34 Air Conditioning 36 Cabin Pressure Altitude 36 Cabin Pressure Temperature				Destroyed Destroyed Destroyed		
29 Windshield Deice 30 Windshield Anti-Ice 31 Airframe Deice 32 Cabin Air/Fan 33 Cabin Heater 34 Air Conditioning 36 Cabin Pressure Altitude 36 Cabin Pressure Temperature 37 Crew Oxygen				Destroyed Destroyed Destroyed		
29 Windshield Deice 30 Windshield Anti-Ice 31 Airframe Deice 32 Cabin Air/Fan 33 Cabin Heater 34 Air Conditioning 36 Cabin Pressure Altitude 36 Cabin Pressure Temperature 37 Crew Oxygen 38 Cabin/Passenger Oxygen				Destroyed Destroyed Destroyed Destroyed		
29 Windshield Deice 30 Windshield Anti-Ice 31 Airframe Deice 32 Cabin Air/Fan 33 Cabin Heater 34 Air Conditioning 36 Cabin Pressure Altitude 36 Cabin Pressure Temperature 37 Crew Oxygen 38 Cabin/Passenger Oxygen 39 Taxi Lights'				Destroyed Destroyed Destroyed Destroyed		
29 Windshield Deice 30 Windshield Anti-Ice 31 Airframe Deice 32 Cabin Air/Fan 33 Cabin Heater 34 Air Conditioning 35 Cabin Pressure Altitude 36 Cabin Pressure Temperature 37 Crew Oxygen 38 Cabin/Passenger Oxygen				Destroyed Destroyed Destroyed Destroyed		

NTSB Form 6120.4 Supplement B (1-84)

NTSR	Accident/Incider	t Blumbar

T .	AVIATIO	N			العلمانا	
Cumplement B. Cooksi	t Dagumontal	ion Si	nalo er	ad Twin Reginer		9 8 F A 1 0 6
Supplement B — Cockpi	(Continued)	.1011, 51	ngie ai	id Twiii Reciprod	aung Engir	ie and unpowered
www.		uung erren saasaa saasaa saasaa saasaa saasaa saasaa				
Electrical/System Switches		7.02 N				
Switch/Item	Not 2 1 Installed	On	3 Of	f A Other	Pe	rtinent Setting/Remark
42 Strobes	, matalied			Destroyed		
43 Navigation Lights				Destroyed		
44 Instrument Panel Lights				Destroyed		
45 Cockpit/Storm Lights				Destroyed		
46 Cabin Lights			<u> </u>	Destroyed		·
47 ELT Remote				Destroyed		
Engine Controls-No. 1 Eng	ine			48 L Engine Contro	ol Positions Not	Pertinent (Go to block 57)
49 Throttle Position	50 Propeller			51 Mixture		52 Carburetor Heat
1 Not Installed	1 Not Inst	alled		1 Not Installed	l ·	1 Not Installed
2 Full Forward	2 D Full Inci	rease (Lov	v Pitch)	2 Full Rich		2 Full On
3 Midrange	3 Midrang	je		3 Midrange		3 Partial
4 🔀 idle	4 🛛 Full Dec	crease		4 🔀 Idle Cutoff		4 🗌 Off
A Other	5 🔲 Feather					
	A Other			A Other		A Other
53 Alternate Air	54 Cowl Flaps			55 Magneto Switch P	osition	56 Throttle Friction
1 Not Installed	t 🔲 Not inst	alled		1 Not Installed	1	1 Not Installed
2 Dpen	2 D Open			2 D Both		2 🔲 Tight
3 Closed	3 Closed			3 Left		3 Loose
4 🔝 Midrange	4 Midrang	je	•	4 Right		
				5 🗌 Off		
				6 Start		A Other Destroyed
A Other <u>Destroyed</u>	A Other <u>Dest</u>	royed		A Other Destroyed	<u>1</u>	A Other <u>Destroyed</u>
Engine Controls-No. 2 Eng	gine			67 Engine Contr	ol Positions Not	Pertinent (Ge to block 57)
58 Throttle Position	59 Propeller			60 Mixture		61 Carburetor Heat
1 Not Installed	1 Not Inst			1 Not Installed	d .	1 Not installed
2 Full Forward	1 =	rease (Lov	v Pitch)	2 Full Rich		2 🔲 Full On
3 Midrange	3 Midrang	•		3 Midrange		3 Partial
4 🔀 Idle	4 🔀 Full De	crease		4 🛛 Idle Cutoff		4 🗌 011
	5 Feather					
A Other	A Other	_		A Other		A Other
62 Alternate Air	63 Cowl Flaps			64 Magneto Switch F		65 Throttle Friction
1 Not Installed	1 Not Ins	tailed		1 Not Installed	d	1 Not Installed
2 Open	2 Open			2 L Both		2 Tight
3 Closed	3 Closed			3 Left		3 Loose
4 Midrange	4 Midran	ge		4 Right		
A DN Back or I				5 🗌 Off		A Other Dectroyed
A Other <u>Destroyed</u>	A Other Dest	royed		6 Start		A Other <u>Destroyed</u>
				A Other <u>Destroye</u>	<u>d</u>	
MTCD Form \$110 4 Cuppler	(4 04)					

v	TSR	Accident/Incident Numbe	
٠	120	wernelining and in Mailine	•

ı b	AVIATION			! _
			L A X 9 8	33088
		, Single and Twin Rec	iprocating Engine an	d Unpowered
	craft (Continued)			¥11.
Landing Gear Control	s/Indicators, Flight Cor	trois/indicators, and Fu	el Selectors/Pumps	
66 Landing Gear Control	67 Landing Gear Indicator	68 Trailing Edge Flap System	69 Trailing Edge Flap	70 Trailing Edge Flap
1 Not Installed	1 Not Installed	1 Not Installed	Control	Indicator
2 🔲 Up	2 🔲 Up	2 Manual	1 Not installed	1 Not Installed
3 Down	3 Down	3 🔀 Electric	2 🔲 Up	2 □ Up
4 ∐ Off	4 Transit/unsafe	4 Hydraulic	A Down degrees	A Downdegrees
A Other <u>Destroyed</u>	A Other <u>Destroyed</u>	A Other	B Other <u>Destroyed</u>	B Other <u>Destroyed</u>
71 Speed Brake Control	72 Spoiler Control	73 Dual Controls	74 Throwover Control/Yo	ke Position
1 Not Installed	1 Not Installed	1 Not Installed	1 X Not installed	
2 Stowed	2 Stowed	2 X Installed	2 Left	
3 Deployed	3 Deployed		3 Right	
A Other	A Other	A Other	4 Intermediate	
7 Other	A Other		A Other	
76 Elev/Stab Trim Control	76 Elev/Stab Trim Indicator	77 Aileron Trim Control	78 Aileron Trim Indicator	79 Rudder Trim Indicator
(Multiple Entry)	1 Not Installed	1 Not Installed	1 Not installed	1 Not Installed
1 Not Installed	2 Up	2 Manual	2 Left	2 Left
2 🔀 Manual	3 Down	3 Electric	3 Right	3 Right
3 🔀 Electric	4 Neutral		4 Neutral	4 Neutral
A Other	A Other <u>Destroyed</u>	A Other	A Other Destroyed	A Other <u>Destroyed</u>
80 Fuel Selector Position(s	(Multiple entry)	<u> </u>	81 Fuel Boost Pump, I	Engine #1
1 🛛 Left main	7 Forward	13 On-engine #1	1 Not Installed	
2 Right main	8 🔲 Aft	14 Off-engine #1	2	
3 🔲 Both	9 External tank	15 🔲 On-engine #2	3 High	
4 Left auxiliary	10 🔀 Between tanks	16 🔲 Off-engine #2	4 Low	
5 Right auxiliary	11 X-feed left to rigi	nt	5 🗌 Off	
6 Center	12 X feed right to le	ft A Other	A Other Destroyed	
81 Fuel Boost Pump, Engir		· '	83 Primer, Engine #1	84 Primer, Engine #2
1 Not Installed	1 Not Insta	iled	1 🛛 Not Installed	1 Not Installed
2 🔲 On	2 🗌 Off		2 Locked	2 Locked
3 🗌 High	A On (ank to tank)	3 Unlocked	3 Unlocked
4 Low				
5 Off	Other		A Other	A Other
A Other <u>Destroyed</u>			A Other	
1				

NTSB Accident/Incident Number National Transportation Safety Board **FACTUAL REPORT AVIATION** L A X 9 8 F A 1 0 6 Supplement K - Occupant, Survival and Injury Information 2 Position 3 Age 1 Seat No. 4 Height 5 Weight 1 Pilot in command For non-A ____ Yrs ____ Inches ____Lbs B If Seat Unknown Enter survivable 2 Second Pilot B Under 24 mos. enter accident. Persons Name A Other ____ A Other _ 3 Other Crewmember months go to 4 Passenger block 36 C Other __ C Other ____ A Other 8 Physically Handicapped 6 Injury Index 7 Condition Prior to Accident 9 Seat Belt Adjustment 9 Shoulder Harness (Multiple entry) Adjustment 1 Smoker Not Fastened 1 None 1 No 1 Not Fastened 2 Language difficulty 2 Loose 2 Minor 2 Blind Loose 3 Pre-existing disease 3 Snug 3 Mobility impaired 3 Serious 3 Snug 4 Prosthesis 4 Tight 4 Deaf 4 Tight Fastened-4 Fatal 5 Fastened-Tightness Unknown A Other ___ Tightness Unknown Not Seated A Other __ 6 Seat not equipped Seat not equipped A Other A Other 11 Knew Impact/Accident Coming 12 Braced for Impact 13 Direction of Movement at Impact Yes 1 Yes 1 Forward 3 Dupward 5 Left 2 Rearward 4 Downwards 6 Right A Other ____ A Other A Other 15 Escape Hampered by 14 Exit Used Exit Diagram (Multiple entry) Did not escape Use Following codes for overhead Not Hampered hatches 2 Split fuselage 2 Smoke ÇL Cockpit CR A Exit number (use diagram) Cockpit 99 Heat 14 Injuries Cabin 88 Trapped 2L 2R 8 Other __ Darkness Tailcone 77 ЗL 2E Cabin Debris Disorientation Difficulty Using Exit Specify B Other _ 17 Evacuation Aided by 18 Injured During Evacuation 16 Briefed on Emergency Procedures (Multiple entry) (Multiple entry) ☐ No Passenger Yes Crew Before takeoff 3 Before impact/accident Bystander CFR personnel A Other ____ A Other ____ Unaided A Other Complete this section if oxygen was used. 21 Type of Equipment 22 Difficulty In Use 23 Type of Oxygen System

1 Yes

2 No

A Other _

NTSB Form 6120.4 Supplement K (1-84)

1 Supplemental

Portable

A Other ___

1 Solid state

2 Gaseous

Specify

B Other

NTSB Accident/Incident Number

											<u>L]</u>	<u> </u>	9	8	<u> </u>	1	0 [6	<u>}</u>
Supplement K — Occ	ирап	it, Su	rvival a	and In	jury	Inforn	nation	(cont	linued	l)								
Complete this section for	or acc	ident	s involv	ing fire	•						24		lo fire i	nvolve	d (Goto	block 2	(9)	
Complete this section for accidents involving fire 24 No fire involved (Go to to to the section for accidents involving fire 25 Fire First Sighted (Location) 1 Inside aircraft 2 Outside aircraft 3 Both A Other						/) face) s)	ire											
Complete this section for	or acc	ident	s involv	ing dit	ching	/water					29				ct (Go			<u> </u>
Flotation Devices	A	Availab			Used		E Wit		,	G II	blems Use		I	alfunct With	Jse	K Da		
	1 Yes	No No	8 Other	1 Yes	No No	D Other	1 Yes	2 No	F Other	1 Yes	2 No	H Other	1 Yes	No.	J Other	1 Yes	No No	L Ot
30 Liferaft																		
31 Vest-inflatable																		
32 Vest-Non-Inflatable																		Γ
33 Cushion																		
34 Time in Water		,	·	3	6 Res	cued b	у	· ·	<u> </u>	·				·	•	-	<u> </u>	
A Hrs B Mins					1 [Boar				3 🗌	Helic	-						
D Willha					2 [Airp	lane			5 📙	None				Other			
Occupant Injuries —	Con	nplete	applic	able p	arts f	or sun	vivors a	nd no	nsurvi	vors.								
Items 36 thru 39 apply	ONL'	Y to fl	ight cre															· .
36 Medication Prescribed 1 ⊠ No				37		ation Be No	ing Tak	en			38	Medic 1 ⊠	ation/I No	Drugs	Found			
A Yes (Specify:)				A.		pecify:					A	Yes (S			١			
B Other				1		her							her _		_			
39 Pre-existing Disease Fo	und at	Autor	sy		·											. .		
1 No Autopsy perfor	med				Α.	Yes Sp	ecify _			•			В	Other				
2 None reported											_,							
Results of Toxicolog		Analy	ses —	Comp	olete a	as app	licable	for su	ırvivor	s and n	onsu	vivors.						
40 Toxicology (Multiple ent	ry)		_ ,															
1 Not ordered			3 [_		- perfore		5	=	mbalmed		-11-4.4				er	-	
2 Not ordered — p			4 [iered –	- not pe	normed	6		Specimen ———	not av	allable/u	nsuital	ble for	analysis			

NTSB Accident/Incident Number

L A X 9 8 F A 1 0 6

Suppler	nent K — Occupant, Surviv	al and Inju	ıry inform	ation (c	ontin	nued)	
Results	of Toxicological Analyses -	— (Comple	ete as appl	icable for	r survi	vors and nonsurvivors.)	(continued)
		Α	Test Results				
Substanc	es	1 Positive	2 Negative	3 Other		C Level of Sub	estances Found
41 Ethar	nol (Alcohol)		\boxtimes			W	Mg
42 CO (C	Carbon Monoxide)			N/A			% Saturati
43 hb (H	emoglobin)			N/A			gm
44 HCN	(Hydrogen Cyanide)			N/A		·	Microgram/
45 Acidi	c and Neutral Drugs		\boxtimes				
46 Basic	: Drugs		\boxtimes				
47 Marij	uana		\boxtimes				
48 (Spec	cify)						
	List any additional toxicole	gical substar	nces discover				
A Substance Code	B Level of Substan	ces Found		A Go	de	B Level of S	Substances Found
49		· · · · · · · · · · · · · · · · · · ·		56			
50				57			
- 51				58			
52				59			
63				60			
54					cify)		
	· .		· <u>, </u>	(Spe	cify)		,,
55			Toxicological	8ubstance	s/Codes		Menthol
Acetone Amoxapine Aminptyline Amobarbital Amphetamine Benzoylecgoni Brompheniram Butalbital Butsbarbital Caffene Cannabinoids Chiorazepate Chlordiazepoxi.		mine m podeinone ydramine hydantoin flurazepam pen vyrtol epam am azine nide doi	019 020 021 022 023 024 025 026 027 028 029 030 031		isopropai Ketimine Lidocaini Loxapine Mekloqu Meperidi Mephent Methona Methona Methagu Methagu Methagu Methagu Methagu Methagu Methagu	ne. 035 nol. 036 t. 037 e. 037 e. 038 e. 047 talone. 039 ne 040 termine 041 tmate 042 tl. 043 one 044 phetamine 045 talone. 045 talone. 048 ne 049 termidate. 050	Morphine 053 Medazepan 054 Nicotine 055 Normphyline 056 Oxazepam 057 Pentazocine 058 Phenobarbital 059 Procaine 060 Propoxyphene 061 Secobarbital 062 Thiniidazine 063 Temazepam D64 Nordiazepam 065 Pentobarbital 066 Phenoyclidine 067 Phondimetrazine 068

NTSB Accident/Incident Number

L A X 9 8 F A 1 0 6

Supplement K —	 Occupant, Surviv 	al and Injury I	nformation	(continued)
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53 For multiple extreme traumatic injuries, check box, and go to next applicable supplement.

Occupant Injury Coding Chart (Complete for survivors and non survivors as applicable.)

	A Body Region	B Aspect	C Lesion	D System/Organ	E A.I.S. Severity	F 6 Injury Source	7 Source of Da
64							
65							
66						·	
67							
68							
69							
70	,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,				,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,		
71						**	
72							
73							
Body Region – A		88 Injured a 99 Other	spect unknown	05 Liver 06 Nervous S	vetem	Source of Data	_G

1			
73	•		
Body Region - A	88 Injured aspect unknown	05 Liver	Source of Data — G
	99 Other	06 Nervous System	
		07 Brain	Official
01 Head (Skull, scalp, ears)		08 Spinal cord	01 Autopsy records with or without
02 Face (Forehead, nose, eyes, mouth) Lesion - C		09 Ears	hospital/medical records
03 Neck (Cervical spine, C1-C7)			02 Hospital/medical records
04 Shoulder (Clavicle, scapula, joint)	01 Laceration	11 Heart	03 Emergency room records
05 Upper limb (Whole arm)	02 Contusion	12 Spleen	04 Private or treating physicians
06 Arm (Upper)	03 Abrasion	13 Urogenital	
07 Elbow	04 Fracture	14 Kidneys	Unofficial
08 Forearm	05 Concussion	15 Respiratory	
09 Wrist	Q6 Avulsion	16 Eye	05 Lay coroner
10 Hand—Fingers	07 Rupture	17 Pulmonary/lungs	06 E.M.S. personnel
11 Chest (Anterior and posterior ribs)	08 Sprain	18 Airway	07 Interviewee
12 Abdomen (Diaphragm and below)	09 Dislocation	19 Muscles	08 Police

06	Avulsion	16	Eye	05	Lay coroner
07	Rupture	17	Pulmonary/lungs	06	E.M.S. personne
80	Sprain	18	Airway	07	Interviewee
09	Dislocation	19	Muscles	08	Police
10	Crush	20	Integumentary	09	Other source
11	Amputation	21	Thyroid or other endocrine gland)		
12	Burn	88	Injured, unknown system or organ		

Abbreviated Injury Scale - E

00	Not injured
01	Minor Injury
02	Moderate injury
03	Serious injury (Not life-threatening)
04	Severe injury (Life-threatening survival probable)
05	Critical injury (Survival uncertain)
06	Maximum (untreatable)
Q7	Injured (Unknown severity)
88	Unknown if injured

Aspect Of Injury

22 Whole body

13 Back (Thoracic spine T1-T12)

14 Back (Lumbar L1-L5)

16 Lower limb (Whole leg)

88 Injured, unknown region

15 Peivis-hip

18 Knee

20 Ankle 21 Foot-toes

99 Other

17 Thigh (Femur)

19 Leg (Below knee)

01 Rìght 02 Left

System/Organ - D

15 Strain

99 Other

13 Fracture and dislocation

14 Severence (Transection)

16 Detachment (Separation) 17 Perforation (Puncture)

88 Injured unknown lesion

01 Skeletal 02 Vertebrae alnioL 80 04 Digestive

NTSB Form 6120.4 Supplement K (1-84)

NTSB Accident/Incident Number

L A X 9 8 F A 1 0 6

Supplement K — Occupant, Survival and Injury Information (continued)

Injury Source List	- F	13/1-1		
01	Windshield	25	Ground/runway	
02	Windshield frame	26	Unsecured seat(s)	
03	Window	27	Outside object(s) entering aircraft	
04	Window frame	28	Galley item(s)	
05	Instrument panel	29	Food/beverage item(s)	
06	Side Console	30	Other interior objects	
07	Center console	31	Other exterior objects	
08	Control stick/cyclic stick	32	Evacuation slide/slide raft	
. 09	Collective	33	Escape rope/tape	
10	Control yoke/column	34	Escape inertia device	
11	Throttle quadrant/levers	35	Ejected from aircraft	
12	Rudder pedals	36	Propeller/rotor blades	
13	Ceiling	37	Exterior aircraft surface	
14	Sidewall	38 39	Engine Wheel/tires	
15 16	Floor Fuselage framing/structure	39 40	Ground vehicle	
17	Table	41	Toxic/noxious/irritant fumes	
18	Seat	42	Fire/radiant heat	
19	Seatback tray	43	Flying glass	
20	Restraints—seatbelt/tiedown	44	Door/hatches	
21	Restraints—shoulder harness	45	Acceleration forces	
22	Unsecured item(s) in cockpit	46	Exposure	
23	Unsecured item(s) in cabin	47	Glare Shield	
24	Other occupants	48	Eyeglasses	
		88	Unknown	
<u>.</u>		99	Other	
			•	
		· · · · · · · · · · · · · · · · · · ·		
74 Death Due	To Fire/Smoke	75 Deat	th Due To Drowning	
1 L Yes	i	1 [Yes	
2 🔀 No		2	⊠ №	
A Other _		A	 Other	
		1		
			/	
NTSB Form 61	20.4 Supplement K (1-84)			P

FACTUAL REPORT AVIATION

N	ITSB	Acciden	L	Incident Number

	1.00									-	AIN	17	D	$r_{\parallel}r_{\parallel}$	1 14	6
Supplement N-I	Fire/E	xplos	sion													
1 Fire Started 1 ln-flight 2 During ground		on 4	■ Duri		-		5 🗆 G A Othe		r than 1	minute al	ter aircr	aft can	ne to re	est		
Location of Initial Fire or	r Explos	ion (Sp	ecify)		A	В		С	In-Fligh	nt	E Or	i Grou	nd	G	After Imp	act
				1 F	ire 1	Explos	sion	Yes	2 No	D Other	1 Yes	2 No	F Ot	her 1 Yes	2 No	H Other
2 Fushlas	6 LE				X				×		*			×		
3 R1647	ייון נא	<u></u>			x.				Х		እ			X		
4																
5 Fire/Explosion Ignition 1	ı İy				A Othe	'es lo (Go to	n Direct	ion (M	(Multiple e	Oil Hydrauli her entry)	() C		е Агеа	(M 1		try) al gas ag oil ne ene
11 ☐ Ground vehick 12 ☐ Ground structe 13 ☐ Aircraft occupa A Other	ure ant				з 🔣	Rearward Upward Downwa	. Ч	☐ Rig Other	ght to le	ft			her	Percent N/A		
11 Ground Structure B 1 Single family h 2 Multifamily ho 3 Commercial b	nouse	4 [5 [∏ Farm □ Trees □ Vehic	building	8	Other		't			12	1 C 2 1 3 C	Yes No (g and Ex stalled/A Omit bloo pertinent it blooks	v ailable cks 13-34 to accide)
Sensors				Sensor							Exti	nguist				
Extinguishers	A Ava		г		e of Sen			-	vailable	T	G Nur			I Type of		
13 Engine #1	1 Yes	2 No	B Other	1 Heat	2 Smok	e 3 Optic	Othe D	r 1 Ye	s 2 No	P F Othe	r J Numi	per H	Other	1Manual	2 Automa	itic J Other
14 Engine #2																
15 Engine #3																
16 Engine #4																
17 APU													ļ			
NTSB Form 6120.	.4 Sup	plem	nent N	(1-84)												Page

FACTUAL REPORT AVIATION

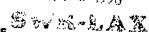
NTSB Accident/Incident Number

LAX19181F1A1110161

Supplement N—I	Fire/	Expl	osi	ion	(co	ntin	ued)											
Sensors Extinguishers	ļ				Г	nsors				-				Extingu				
	 	Availa			+-		Type of 8		1			Available		G Nun	,		of Extingu	0.0
	1 Yes	2 No) <u>E</u>	3 Othe	≥r 1 ⊦	leat	2 Smoke	3Optio	D C)ther	1 Yes	2 No	F Other	1Number	H Other	1Manual	2 Automatic	; Julium
20 Galley																		
21 Lavalory						mu										· 		
22 Heater																		
23 Battery																		
24 Electrical System																		
25 Specify																		
26 Specify																		
27 Specify																		
Sensor Operation	Fraine #1	Frome #2	2	Engine #3	Engine #4	AP 5	U Cabi		argo 7	C	alley 8		atory	Heater	Hattery		ical Sys.	Other A
28 Activated																		
29 Did Not Activate																		
Extinguisher Operation	Enoine #1	V V	B	Engine #3	Engine #4 0	AP E		in C	argo G		Galley H		atory	Heater J	Battery K	Elect	rical Sys	Other M
30 M	d an																	
Activated 31 A	Auto																	
32 ! # Did Not	Vlan																	
Activate 33 A																		
34 Extinguisher System 1 Fixed equipm 2 Portable equipment	ent			з 🔲	CO ₂	(Ca	rbon Diox probroyet				Halon ME (Me		omide)	7 ☐ H A Othe	alon 1301 r			

FORM APPROVED FOR USE THROUGH, 14/30(90, BY OMB NO. 3147-0X SAFETY BOARD

NATIONAL TRANSPORTATION SAFETY BOARD This form To Be Used For Reporting Civil Aircraft Accidents Involving Commercial and General Aviation Aircraft



Date of Accident Local Time Zone Elevation At Accident Stile Zone Zone Elevation At Accident Stile Zone Zo	Naverset City/Place, State, Zip Gode				888000000		(7/////			
If the Accident Occurred On Approach, Takeoff Or Within 3 Miles Of An Airport, Complete The Following Information				[Date of Acc	ident	Local Time	Zone	Fievatio	n At Accident Site
The Accident Occurred on Approach, Takeoff Or Within 3 Miles of An Airport, Complete The Following Information	MT	Mediest Ottyrriace, State	, tip code	1	· ·			OCK)		
The Accident Occurred On Approach, Takeoff Or Within 3 Miles Of An Airport, Complete The Following Information	IT The Accident Occurred On Approach, Talkeoff Or Within 3 Miles Of An Airport 1.	MT Burdell	Nobato La	a.	3 5-	47	1956 1470	PST		
1.	1.				s Of An Ain					
Airport Name	A Within ½ Mile	Proximity To Airport			<u> </u>					
Airport Name	A Within ½ Mile	1 Ca Airmort	3 [7]	Mithin 14 Mile		s 🗆	Within 1 Mil		7 🗀 V	Mithin 3 Milae
Airport Name	Airport Name	<u> </u>							_	•
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2	2	Phase of Operations								
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Aircraft Information Registration Mark Aircraft Manufacturer Aircraft Typer/Model Serial Number Cerf Max Gross W A 25 7 N V P - Y P P P P P P P P P	Aircraft Information	<u> </u>						<u> </u>		
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2.	Helicopter		- (7)					_		Amateur Built
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2.	2.		4 🗀 Tailua	ani Datrantah	No.		7 M Chia			
3. ☐ Tailwheel—Fixed 5. ☐ Amphiblan 9. Specify Pax ← Stall Warning System Installed Installed Installed Installed In Ministry Installed Instal	3.	•	=					baal		1 "
Stall Warning System IFR Equipped Installed 1. Yes	FREquipped Engine Type Installed I. Yes I. Reciprocating—Carburetor S. Turbo Prop S. Turbo Fan S. No S. Reciprocating—Fuel Injected A. Turbo Jet G. Turbo Shaft Turbo Manufacturer Engine Model/Series Engine Rated Power I. Mone System Used I. Mone Secolety I. Mone I.				VE MIDILIZ		_	licai		
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Time Since Inspection Time Since Overhaul	Time		Engine	Model/Series		Engi	ne Rated Pov	ver		
Engine(s)	Engine(s)			_		1, 7,	で、			
Engine (s)	Engine(s) Date of Mfg. Mfg. Serial No. Total Time Time Since Inspection Time Since Overhaul Engine No. 1	Lycoming	1710.	540-72BI)					
Engine No. 1	Engine No. 1 10-12-94 RL-95-70-WIA 3-90-94 Hours 3.0 Hours 5-40-94 Hours Engine No. 2 10-17-94 RL-1400-60-A 41-9-2 Hours 3.0 Hours 41-9-2 Hours Engine No. 3 Hours Hours Hours Hours Engine No. 4 Hours Hours Hours Type of Maintenance Program Type of Last Inspection 1	<u> </u>								
Engine No. 2 10-17-44 RL-1400 - 60 A 419 2 Hours 3.0 Hours 4 19.2 Hours Engine No. 3 Hours Hours Hours Hours Hours Engine No. 4 Hours Hours Hours Hours Type of Maintenance Program Type of Last Inspection Date Last Inspection Performed 1. ☐ Annual 1. ☐ Annual 2. ☐ 100-Hour Time Since Last Inspection 3. ☑ Other Approved Inspection Program (AAIP) 3. ☑ AAIP Airframe Total Time 4. ☐ Continuous Airworthiness 4. ☐ Continuous Airworthiness Airframe Total Time 5. Specify Image: Continuous Airworthiness Serial Number Battery Date Locator Un Known Un Known (M/DM) 10-31-9 Transmitter Switch Operated Aided in Accident Location (ELT) 1. ☐ On 2. ☐ Off 3. ☑ Armed 1. ☐ Yes 2. ☐ No	Engine No. 2 C-17-94 R-1400 - 60 A 419 A Hours 3.0 Hours 419 A Hours Engine No. 3 Hours Hours Hours Engine No. 4 Hours Hours Hours Engine No. 4 Hours Hours Type of Maintenance Program Type of Last Inspection 1									
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Engine No. 4 Hours Hours Hours Type of Maintenance Program Type of Last Inspection Date Last Inspection Performed 1. □ Annual 2 - 12 - 93 (M/D/N) 2. □ Manufacturer's Inspection Program 2. □ 100-Hour Time Since Last Inspection 3. ☑ Other Approved Inspection Program (AAIP) 3. ☑ AAIP Airframe Total Time 4. □ Continuous Airworthiness 4. □ Continuous Airworthiness Airframe Total Time 5. Specify Locator Model/Series Serial Number Battery Date Locator Un Known Un Known (M/D/N) /2 - 31 - 9 Transmitter Switch Operated Aided in Accident Location (ELT) 1. □ On 2. □ Off 3. ☑ Armed 1. □ Yes 2. □ No 1. □ Yes 2. □ No	Engine No. 4		10-11-44	RL-1400	- 60 A	414 1		J.O		
Type of Maintenance Program 1. Annual 2. Manufacturer's Inspection Program 3. Other Approved Inspection Program (AAIP) 4. Continuous Airworthiness 5. Specify Emergency ELT Manufacturer Locator Unit No and N	Type of Maintenance Program 1. Annual 2. Manufacturer's Inspection Program 3. Other Approved Inspection Program (AAIP) 4. Continuous Airworthiness 5. Specify Emergency Locator Transmitter Witch (ELT) 1. On 2. Off 3. Armed Aircraft Owner Air Puc Aircraft Operator of Aircraft Operator of Aircraft Node/Series Locator Address 1. Same As Registered Owner Address 1. Same As Registered Owner 3. Annual 1. Annual 2. 12 - 93 (M/DN) Airword Time Since Last Inspection Performed 2. 100-Hour 3. AAIP Air Continuous Airworthiness Serial Number Locator Locator Locator Locator Locator Locator Airword Airwo		<u>,, , , , , , , , , , , , , , , , , , ,</u>							
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3. Other Approved Inspection Program (AAIP) 4. Continuous Airworthiness 5. Specify Emergency ELT Manufacturer Locator Un Known Model/Series Vinctor Switch 1. On 2. Off 3. Armed 1. Yes 2. No 1. Yes 2. No	3. Other Approved Inspection Program (AAIP) 4. Continuous Airworthiness 5. Specify Emergency ELT Manufacturer Locator UnKnown Witch (ELT) 1. On 2. Off 3. Armed 1. Yes 2. No 1. Yes 2. No Registered Aircraft Owner Air puc Airlines Inspection Program (AAIP) 3. AAIP 4. Continuous Airworthiness 6. 2 3 1. 3 Hour 6. M/D/N 10 31-4 7. Aided in Accident Location 7. Aided in Accident Location 7. Air puc Airlines Inc. 8. Address 7. 2 7 Perima fer Rd Sc. 9. Address 7. 2 7 Perima fer Rd Sc. 9. Address 7. 2 7 Perima fer Rd Sc. 9. Address 7. 2 7 Perima fer Rd Sc. 9. Address 1. Same As Registered Owner 2. Name 3. DBA:	" =			\ <u></u>					
4. Continuous Airworthiness 5. Specify Emergency ELT Manufacturer Locator Un Known Transmitter (ELT) 1. On 2. Off 3. Armed Airframe Total Time Location Model/Series Serial Number Battery Date (M/DN) 10 - 31 - 4 Operated 1. Yes 2. No No Airframe Total Time Location (M/DN) 10 - 31 - 4 Airframe Total Time Location (M/DN) 10 - 31 - 4 Airframe Total Time Location (M/DN) 10 - 31 - 4 Airframe Total Time Location (M/DN) 10 - 31 - 4 Airframe Total Time Location (M/DN) 10 - 31 - 4 Airframe Total Time Location (M/DN) 10 - 31 - 4 Airframe Total Time Location Location (M/DN) 10 - 31 - 4 Airframe Total Time Location Location (M/DN) 10 - 31 - 4 Airframe Total Time Location Location Location (M/DN) 10 - 31 - 4 Airframe Total Time Location	4. Continuous Airworthiness Airframe Total Time 4. 2 3 1 . 3 Hour (M/DY) /0 . 31 - 4 Transmitter (M/DY) /0 . 31 - 4 Transmitter (ELT)			N	4. L 2 ⊠					
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Locator Un Known Un Known (M/DM) /2 · 31 · 4 Transmitter (ELT) Switch Operated Aided in Accident Location 1. □ On 2. □ Off 3. ☒ Armed 1. □ Yes 2. □ No 1. □ Yes 2. □ No	Locator UNKNOWN UNKNOWN (M/DM) 10-31-4 Transmitter Switch Operated I. On 2. Off 3. Armed I. Yes 2. No I. Yes 2. No Registered Aircraft Owner Air Pac Airlines Inc. Operator of Aircraft Owner Address 1. Same As Registered Owner 2. Name 3. DBA:		ufacturer	М	odel/Series		Si	erial Number	В	attery Date
Transmitter Switch Operated Aided In Accident Location (ELT) 1. □ On 2. □ Off 3. ☒ Armed 1. □ Yes 2. □ No 1. □ Yes 2. □ No	Transmitter Switch (ELT) 1. On 2. Off 3. Armed 1. Yes 2. No Registered Aircraft Owner Air pac Airlines Fric Operator of Aircraft 1. Same As Registered Owner 2. Name 3. DBA: Aided In Accident Location 1. Yes 2. No Address 7.2.7 Perimater Rd 50 Address 1. Same As Registered Owner 2. Same As Registered Owner 2.									MOM 10.31-4
	Address 727 Perimater Rd Sci Air pac Airlines Inc Operator of Aircraft 1. Same As Registered Owner 2. Name 3. DBA: Address 7.277 Perimater Rd Sci 8. Same As Registered Owner 2. Same As Registered Owner 2. Same As Registered Owner	Transmitter Switch			· -		_ [~]			
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	Operator of Aircraft 1. Same As Registered Owner 2. Name 3. DBA:	_				Addi		77 Yerin	12 <u>7er</u>	
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	1. Same As Registered Owner 2. Name 2. DBA:	\$				Addi	ess		***************************************	· · · · · · · · · · · · · · · · · · ·
	2. Name 2 2 2 2		ed Owner					egistered Owner		
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2. Name										

Owner/Operator informatio Operator (Certificate Number		Designator (4 Le	etter Designator)					
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Type Ratings/Student End			Date of Bienni		Review	BFR Aircraft			
unKnown		1	or Equivalent テーカー			1. Make 2. Model	P1Per		
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1. None 1. 2. Minor 2. [Front 1	rson at Contro Pilot In Con Second Pil	mmand	ae of Accid 3. ☐ Bo 4. ☐ N	oth Pilots 5. 🔲 No	o One	Seat Belt A I. 🔯 Yes 2. 🔲 No	.vailable
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This Make/Model Last 90 Days Last 30 Days Last 24 Hours	632 19. 22.5 10.	4	7.	3 2	=				
Second Pilot Information Second Pilot Responsibili	ties at the Time of A	ccident		٠. ن				I	I
1. Co-Pilot 2. Pilot Name	Dual Student Pilot Certific	3. ☐ Safet eate No.	y Pilot 4	4. Che ess	eck Pilot	5. None (P		senger) tionality	
	3. Commercial 4. Airline Transport		Flight Instructor		7. Mi 8. Fo	-	9. None D. Specify —		

Owner/Operator Informatic Rating(s) 1. None 2. Single Engine Land 3. Single Engine Sea 4. Multiengine Land 5. Multiengine Sea Type Ratings/Student End	6. 7. 8. 9. 10.	Helico Glider Free B	oter alloon	Instrume 1. No 2. Air 3. He Date of Equiv	ent Rati one rplane elicopter	ing(s)	Review	1. [2. [3. [BFR Aircraft 1. Make	6, [. 7, [. 8, [9, 9	Instrume	ent Airplane nt Helicopter nstructor
Medical Certificate		··· »···	Date of La	st Medical	T	Limitati	ons		2. Model	Date	of Birth (A	A/D/Y)
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Pilot in Command (PIC)	•							-				
Instructor This Make/Model					***	•	l				1	
Last 90 Days												
Last 30 Days									<u></u> -			
Last 24 Hours												
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For NTSB Use Only

NTSB Accident No. Review By NTSB Office Located At Name of Investigator Date Report Received

1515 W. 190th Street, Suite 555

LAX--98-F-A106 Gardena, California 90248-4319 Robert R. Crispin, EdD

3/30/98

Signature of Person Filing Report Other Than Pilot/Operator

1. Signature _



National Transportation Safety Board Office of Aviation Safety Washington, D.C. 20594-2000

September 14, 1998

METEOROLOGY FACTUAL REPORT LAX98FA106

A. ACCIDENT

Location: Novato, California

Date: March 5, 1998

Time: 1905 Pacific standard time (0305Z)

Aircraft: Piper PA-31-350, Registration: N257NW

B. METEOROLOGICAL SPECIALIST

Donald E. Eick Meteorologist National Transportation Safety Board Operational Factors Division, AS-30 Washington, DC

C. SUMMARY

On March 5, 1998, at about 1905 Pacific Standard Time (PST), Airpac flight 263, a Piper PA-31-350, N275NW, collided with trees and terrain near Novato, California. The aircraft was destroyed and the pilot, the sole occupant, received fatal injuries. The airplane was operated by Airpac Airlines, Inc. as a positioning flight under 14 CFR Part 91 when the accident occurred. The airplane departed Santa Rosa, California, at 1850 PST. Visual meteorological conditions prevailed at the departure airport and no flight plan was filed.

The airplane was part of a flight of two company airplanes that departed the Sonoma County airport en route to the Oakland Metropolitan International airport. Airpac flight 1961 was in the lead and Airpac flight 263 was in trail about 5 minutes behind. The pilot of flight 1961 stated that although they were both navigating by a combination of dead reckoning and pilotage, they also had VOR navigation aids available to verify their position while en route.

During the flight, the pilot of flight 1961 radioed the pilot of flight 263 to ask about her progress. She reported that she was "still back here." At about 20 miles north of the Oakland airport, flight 1961 began to encounter instrument meteorological conditions and again attempted to contact the pilot; however, this attempt was not successful in establishing contact. Flight 1961 landed at the Oakland airport at 1920 PST.

When flight 263 did not arrive, it was reported as overdue and a missing aircraft report was issued. At 0148 PST on March 6, a U.S. Coast Guard search and rescue helicopter located the wreckage near the northern crest of a hill. The approximate location of the accident site was about 40 miles north of San Francisco at an elevation of 1,500 feet.

The pilot of flight 1961 reported that the accident site was about 10 miles west of the route that Airpac pilots routinely flew between Santa Rosa and Oakland. The westerly route is a more direct course to Oakland; however, it crosses higher terrain. He said that during the flight he had to descend to 1,200 to 1,500 feet above mean asea level (MSL) to remain in VFR conditions.

According to Airpac, the accident pilot had reportedly flown this route two to three times a week for the past 2 years.

D. DETAILS OF INVESTIGATION

All data and products in this report were obtained from the National Weather Service (NWS) headquarters and the National Climatic Data Center (NCDC) to reconstruct the weather conditions. All times are reported in Universal Time Coordinated (UTC) based upon the 24 hour clock. Pacific Standard Time (PST) is + 8 hours to UTC and UTC=Z. Directions are referenced to true north and distances are in nautical miles. Heights are above mean sea level (MSL) unless otherwise noted. Visibility is reported in statute miles and fractions of statute miles.

1.0 Synoptic Situation

The NWS Surface Analysis Chart for 0300Z on March 6, 1998 (attachment 1), provided the main synoptic situation on the evening of the accident. The chart depicted a low pressure system with a central pressure of 999 millibars (mb)

over the California and Nevada border near Las Vegas. Extending southward from the low pressure center was a cold front that stretched across southern Nevada, southeastern California, into Baja California and the Pacific Ocean. To the east of the low, a warm front stretched across Nevada and southern Utah to the four corners of Utah, Colorado, New Mexico and Arizona. To the north and northwest of the low pressure center, two troughs of low pressure were also indicated. The troughs were depicted with one across Nevada and the other along the Nevada-California border.

A second low pressure system with a central pressure of 1004 mb was also depicted approaching the California coastline near the accident location. A trough of low pressure extended southwestward from the low across the eastern Pacific Ocean to 26.5 degrees north latitude and 135 degrees west longitude.

Except for extreme northern California, the station models depicted on the 0300Z Surface Analysis Chart indicated overcast skies across the state. Over central and northern California several stations reported continuous light to moderate rain or fog and mist. Over the northern Sierra Mountains of California and across Nevada, several stations also reported continuous light to heavy snow with temperatures in the mid twenties (degrees Fahrenheit).

The NWS computer-generated Weather Depiction Chart (attachment 2) for 0100Z on March 6, 1998, depicted a large area of Marginal Visual Flight Rules (MVFR) conditions over northern California and Nevada, with an embedded area of Instrument Flight Rules (IFR) over the northern Sierra Mountains of California and over western Nevada. An enlarged western region Weather Depiction Chart is included (attachment 3). The 0100Z Weather Depiction Charts indicate that stations over the route from Santa Rosa to Oakland were all reporting MVFR conditions due to low ceilings and restricted visibility in rain and fog.

2.0 Weather Observations

The surrounding area was documented by meteorological aerodrome reports or METARs for conditions likely encountered en route and in the vicinity of the accident site. All cloud heights reported in this section are above ground level (AGL).

2.0.1 Santa Rosa, Sonoma County Airport, California (KSTS)

Sonoma County Airport, the departure station, is located approximately 28 miles northwest of the accident site at an elevation of 125 feet MSL. Santa Rosa reported the following conditions at departure:

KSTS at 0145Z, wind 320 degrees at 6 knots, visibility 4 miles in light rain and mist, ceiling broken at 2,900 feet, overcast at 4,400 feet, temperature 7 degrees

Celsius (C), dew point temperature 5 degrees C, altimeter 29.77 inches of mercury (Hg).

KSTS at 0250Z, wind 320 degrees at 6 knots, visibility 5 miles in light rain an mist, ceiling broken at 1,000 feet, overcast at 3,000 feet, temperature 7 degrees C, dew point 5 degrees C, altimeter 29.78 inches Hg.

2.0.2 Napa, California (KAPC)

Napa was the closest weather reporting station to the accident location, located approximately 15 miles east northeast at an elevation of 33 feet MSL. The following conditions were reported about the time of the accident:

KAPC at 0245Z, wind 340 degrees at 12 knots, visibility 6 miles in light rain, ceiling overcast at 1,300 feet, altimeter 29.75 inches Hg.

KAPC at 0345Z, wind 010 degrees at 10 knots, visibility 5 miles in light rain, ceiling broken at 2,100 feet, overcast at 3,100 feet, altimeter 29.76 inches of Hg. Remarks last observation of the day.

2.0.3 Concord Buchanan Airport (KCCR)

Concord Buchanan Airport is located approximately 25 miles east southeast of the accident site at an elevation of 23 feet MSL, reported the following:

KCCR at 0245Z, winds variable at 5 knots, visibility 1 1/2 miles in light rain and mist, ceiling broken at 400 feet, overcast at 2,000 feet, temperature 8 degrees C, dew point 7 degrees C, altimeter 29.73 inches of Hg.

KCCR at 0345Z, winds 010 degrees at 10 knots, visibility 1 1/2 miles in light rain and mist, ceiling broken at 400 feet, overcast at 1,000 feet, temperature 8 degrees C, dew point 7 degrees C, altimeter 29.24 inches Hg.

2.0.4 Oakland International Airport (KOAK)

Oakland international Airport, the planned destination of the flight, is located approximately 27 miles southeast of the accident site at an elevation of 6 feet MSL. The following conditions were recorded:

KOAK at 0247Z, winds 140 degrees at 18 knots, visibility 10 miles, scattered clouds at 800 feet, ceiling overcast at 2,000 feet, temperature 11 degrees C, dew point 6 degrees C, altimeter 29.73 inches of Hg.

KOAK at 0345Z, winds 150 degrees at 14 knots, visibility 10 miles, scattered clouds at 800 feet, ceiling overcast at 2,000 feet, temperature 11 degrees C, dew point 7 degrees C, altimeter 29.74 inches of Hg.

2.0.5 San Francisco International Airport (KSFO)

San Francisco International Airport is located approximately 31 miles south southeast of the accident site at an elevation of 11 feet MSL, and reported the following conditions:

KSFO at 0256Z, winds 110 degrees at 7 knots, visibility 10 miles, a few clouds at 2,300 feet, ceiling broken at 2,900 feet, overcast at 5,000 feet, temperature 9 degrees C, dew point 8 degrees C, altimeter 29.71 inches of Hg. Remarks: rain began at 0220Z and ended at 0230Z, sea level pressure 1005.9 mb.

Special KSFO at 0325Z, winds 340 degrees at 8 knots, visibility 10 miles in light rain, a few clouds at 900 feet, ceiling broken at 2,000 feet, overcast at 3,400 feet, temperature 9 degrees C, dew point 8 degrees C, altimeter 29.72 inches of Hg. Remarks: rain began at 0306Z, precipitation total 0.04 inches since last observation.

Special KSFO at 0345Z, winds 330 degrees at 14 knots, visibility 5 miles in light rain and mist, ceiling broken at 900 feet, broken at 1,700 feet, overcast at 2,300 feet, temperature 9 degrees C, dew point 8 degrees C, altimeter 29.72 inches of Hg. Remarks: ASOS, rain began at 0306Z, ceiling 500 variable 1,300 feet, precipitation since last hours observation 0.05 inches.

2.0.6 Sacramento Executive Airport (KSAC)

Sacramento Executive Airport approximately 55 miles east northeast of the accident site at an elevation of 21 feet MSL, reported the following conditions:

KSAC at 0247Z, winds calm, visibility 3 miles in light rain and mist, ceiling overcast at 1,100 feet, temperature 7 degrees C, dew point 6 degrees C, altimeter 29.75 inches of Hg.

KSAC at 0352Z, winds 360 degrees at 10 knots, visibility 3 miles in light rain and mist, ceiling broken at 1,000 feet, overcast at 1,600 feet, temperature 7 degrees C, dew point 6 degrees C, altimeter 29.75 inches of Hg.

2.0.7 Sacramento International Airport (KSMF)

Sacramento International Airport located approximately 58 miles east northeast of the accident site, and 12 miles north of KSAC at an elevation of 25 feet MSL, reported the following conditions:

KSMF at 0252Z, wind 350 degrees at 6 knots, visibility 3 miles in light rain and mist, a few clouds at 500 feet, ceiling broken at 900 feet, overcast at 2,000 feet, temperature and dew point 6 degrees C, altimeter 29.76 inches of Hg.

Special for KSMF at 0324Z, winds 350 degrees at 8 knots, visibility 2 ½ miles in moderate rain and mist, a few clouds at 300 feet, ceiling broken at 1,100 feet, overcast at 1,800 feet, temperature 7 degrees C, dew point 6 degrees C, altimeter 29.76 inches of Hg.

3.0 Upper Air Data

The upper air sounding from Oakland, California, station number 72493 was examined to provide the temperature and wind structure near the accident site. The balloon launch occurred at 2303Z on March 5, 1998, and provided the following information:

Height MSL (FT)	Temp. (C)	Dew point(C)	RH % Wind	d direction & Speed (KT)
20	9.0	6.8	86.0	120 degrees at 12
220	8.0	5.8	86.0	123 degrees at 14
690	6.2	4.9	90.7	130 degrees at 17
1,600	4.3	3.3	93.3	144 degrees at 17
2,310	2.7	2.1	95.3	164 degrees at 16
3,040	1.8	1.2	95.4	187 degrees at 18
4,550	0,1	- 0.5	95.5	218 degrees at 22
4,870	0.0	- 0.6	95.4	222 degrees at 23
6,130	- 1.8	- 2.4	95.0	232 degrees at 22
7,810	- 4.5	- 5.3	94.7	235 degrees at 24
8,460	- 5.5	- 6,3	94.5	232 degrees at 25
9,580	- 8,5	-11.9	76.5	235 degrees at 27
11,460	-11.4	-12.5	91.7	236 degrees at 21
13,460	-15.6	-17.2	87.7	237 degrees at 16

4.0 Weather Radar Information

The closest Weather Surveillance Radar 1988, Doppler (WSR-88D) was located at Santa Clara, California (KMUX), approximately 72 miles southeast of the accident site. The Level III archive Composite Reflectivity product was obtained from the NWS for the accident investigation. The 0304Z Composite Reflectivity image at an elevation angle of 0.5 degrees (attachment 4) indicated several areas of precipitation across California. The band of greatest radar returns was located south of the Oakland/San Francisco area and was oriented in an east northeast to west southwest direction (ENE-WSW band). The reflectivity levels are 25 decibels (dBZ) or above in this ENE-WSW band with a maximum of 56 dBZ noted off the coast. Several separate areas of return are noted off the coast and south of the ENE-WSW band ranging from

20 to 40 dBZ. In the immediate area of the accident site (northwest of the OAKLAND label) radar returns of 5 to 25 dBZ or VIP level 1 is indicated. A larger more defined area to the north over the Santa Rosa area, extended eastward and then southeast across the Sacramento Valley to the Sierra Mountains.

The Composite Reflectivity image at 0310Z (attachment 5) continues to indicate several small areas of 5 to 25 dBZ returns in the vicinity of the accident site and south over the San Francisco Bay area. However, at the 72 mile range, the 0.5 degrees radar elevation would put the center of the WSR-88D's beam at approximately 10,900 feet, with the base of the beam at 7,200 feet and the top at 14,500 feet.

5.0 Pilot Reports

Within a 75-mile radius of the accident location, pilot report bulletins (UA/UUA) below 18,000 feet from 0000Z to 0600Z were documented for flight conditions reported by other aircraft. The reports within the area and time constraints are listed below in plain language and in chronological order:

Sacramento (SAC) pilot report, 50 miles north of Linden VORTAC (LIN) at 0016Z, altitude 8,500 feet, type aircraft Cessna 414 multiengine airplane, air temperature minus 1 degree C with light rime ice.

Stockton (SCK) pilot report, 6 miles southwest of Linden VORTAC (LIN) at 0019Z, altitude 6,000 feet, type aircraft Beechcraft Bonanza BE35, air temperature + 5 degrees C with light rime ice.

Sacramento (SAC) urgent pilot report, over SAC at 0030Z from a Cessna 414 multiengine airplane, sky cover in Instrument Meteorological Conditions (IMC), moderate to severe rime ice between 7,000 and 11,000 feet, remarks during climb out of SAC southbound, unable to maintain altitude descended to 6,000 feet.

Sacramento (SAC) pilot report, from 10 miles north to 10 miles west of Hangtown VOR at 0103Z, altitude 7,000 feet, type aircraft Trinidad TB21, light rime ice from 10,500 to 7,000 feet.

Sacramento (SAC) pilot report, route from Reno to Sacramento at 0115Z, altitude 6,000 feet, type aircraft not reported, light rime ice between 10,000 and 8,000 feet during descent into SAC.

Sacramento (SAC) pilot report, location 15 miles northeast of Auburn (AUN) at 0120Z, at 8,000 feet, type aircraft Douglas DC-10 jet, light rime icing between 12,000 and 8,000 feet.

Sacramento (SAC) pilot report, location 15 miles northeast of SAC at 0128Z, altitude 8,000 feet, type aircraft McDonald Douglas MD-80 jet, icing light rime 15,000 feet and moderate at 12,000 feet, remarks freezing rain and sleet from 8,000 to 7,000 feet.

San Francisco (SFO) pilot report, 20 miles northwest of Point Reyes VORTAC (PYE) at 0205Z, altitude 15,000 to 9,000 feet, Type aircraft Boeing B-737 jet, light rime icing.

Stockton (SCK) pilot report, from 15 miles east of Livermore (LVK) at 0300Z, altitude 6,000 feet, type aircraft Cessna C-172, air temperature minus 1 degree C with a trace of rime ice.

Santa Rosa (STS) pilot report, over Point Reyes VORTAC (PYE) at 0400Z, report from multiple jet aircraft, light to moderate icing between 9,000 to 11,000 feet.

Marysville/Yuba airport (MYV) at 0407Z, altitude 17,000 feet, type aircraft Boeing B737, moderate rime ice between 17,000 to 8,000 feet.

Oakland (OAK) urgent pilot report, over Oakland at 0430Z, type aircraft British Aerospace HS-125 executive jet, remarks low level wind shear plus/minus 10 knots between 750 and 300 feet landing on runway 27 at Oakland.

San Francisco (SFO) urgent pilot report, 25 miles southeast of Woodside VORTAC at 0547Z, altitude 12,000 feet, type aircraft McDonald Douglas MD80 jet, moderate to severe rime ice between 10,000 to 12,000 feet.

San Jose (SJC) urgent pilot report, over SJC at 0539Z, altitude 300 feet, type aircraft Boeing B737 jet, low level wind shear +10 knots at 300 feet during descent to final approach to runway 12R.

6.0 Satellite Data

Geostationary Operational Environmental Satellite 9 (GOES-9) data was obtained from the NCDC and displayed on the National Transportation Safety Board's Man Computer Interactive Data Access System (McIDAS) workstation. The infrared band 4 GOES-9 satellite imagery at 0245Z, 0300Z, 0330Z and 0400Z on March 6, 1998, were examined. Attachment 6 is the 0300Z infrared GOES 9 satellite image with a resolution of 4 kilometers with the accident site located at the red square. The image was centered over the accident site and magnified (4X) on attachment 7. The radiative temperatures obtained from the imagery at 0300Z provided a temperature of 260 degrees Kelvin (K) or -13.16 degrees C at the accident site, with a minimum of 258.7 degrees K, maximum of 268.0 degrees K, and a mean of 261.9 degrees K. By utilizing

the upper air data, the cloud tops can be estimated based on the radiative temperatures obtained.

The following table provides the observed satellite derived temperatures at 0300Z on March 6, 1998 and approximate cloud heights based upon the Oakland sounding.

Location	Degrees K	Degrees C	Cloud tops feet
Accident site	260.0	-13.16.	12,000
Minimum	258.7	-14.46	12,800
Maximum	268.0	- 5.16	8,000
Mean	261.9	-11.26	11,500

The area of precipitation noted in the radar sector above was examined to determine its satellite characteristics. An enhancement curve used in identifying thunderstorms was applied to the 0300Z infrared image at 4X magnification (attachment 8). The area in the immediate location of the accident did not show any enhancement, nor did the area of radar returns south of the San Francisco/Oakland area. The area north of Santa Rosa and northeast of the accident site was the location of coldest temperatures and highest cloud tops.

7.0 Area Forecast

Information contained in the area forecast issued at 2045Z on March 5, 1998, from the Aviation Weather Center (AWC) located in Kansas City, Missouri and valid from 2045Z to 0900Z on March 6, 1998, indicated the following:

The header began with a flag to see the AIRMET Sierra series for IFR conditions and mountain obscuration that was current at issuance.

The synopsis section followed the flag and discussed the current features influencing the weather over the western states. The features were an upper level low centered over northwestern Oregon moving southeastward to central Oregon by 0000Z, with dry northerly wind flow behind the low over Washington, Oregon and northern California. West to southwesterly wind flow over central and southern California was to become northwesterly by 1200Z. At the surface, a trough of low pressure located over western Oregon to northern California would move quickly eastward into Idaho and Nevada by 0600Z with isolated showers of rain and snow. A coastal low pressure system would be moving ashore over Central California by 2300Z, and then move eastward to eastern Nevada by 0600Z. The system would be over central Utah by 1200Z. A trailing cold front would extend southwestward from the low across northern Arizona and southern California. With couds and precipitation

spreading eastward ahead of the system. Dry northwesterly wind flow would then fill behind both systems.

The forecast for central California and the coastal section indicated that after 2045Z, the visibility would be unrestricted, greater than 6 miles, with scattered to broken clouds at 2,000 feet with a second broken layer at 6,000 feet. Cloud tops would extend to 20,000 feet with occasional light rain. From 0200Z, visibility would be unrestricted, greater than 6 miles, with scattered clouds at 3,000 feet and broken clouds at 8,000 feet with cloud tops to 15,000 feet with isolated light rain showers. From 0400Z, visibility would be unrestricted, greater than 6 miles, with scattered clouds at 3,000 feet and at 8,000 feet. The outlook from 0900Z to 1500Z, called for visual flight rules (VFR).

The Area Forecast was updated on March 6, 1998, at 0345Z. The forecast for central California changed to unrestricted visibility, a few clouds at 1,500 feet, broken to scattered clouds at 3,000 to 4,000 feet and broken clouds at 6,000 feet with tops to 10,000 feet. Scattered rain showers were forecasted through the period. At 0800Z, the forecast called for scattered clouds at 4,000 feet and 6,000 feet. The outlook called for VFR conditions to prevail.

8.0 In-Flight Weather Advisories

There were three separate in-flight weather advisories that were current for the area before and up to the time of the accident. The advisories included AIRMET SIERRA for mountain obscuration, AIRMET ZULU for icing conditions and AIRMET TANGO for moderate turbulence.

AIRMET SIERRA 3 was issued for IFR conditions and mountain obscuration at 2045Z and was valid until 0300Z on March 6, 1998. The advisory covered portions of California, Nevada and Utah. The position of the advisoru was from Medford, Oregon (MFR) to 80 miles northeast of Salt Lake City, UT (SLC) to Dove Creek, CO (DVC) to 80 miles south of Wilson Creek, NV (ILC) to 40 miles east of San Diego. CA (SAN) to San Diego to 40 miles west of Santa Barbara, CA (SBA) to Fortuna, CA (FOT) and back to Medford. The advisory was issued for mountains occasionally obscured in clouds and precipitation. Conditions over northern and central California were spreading eastward over the remaining area by 0200Z to 0600Z and were expected to continue through 0900Z. The AIRMET was updated at 0245Z on March 6, 1998, and was valid until 0900Z. The area was extended to cover portions of Oregon, California, Idaho, Wyoming, Nevada, Utah, Colorado and Arizona. The area was defined from Eugene, OR (EUG) to Rock Springs, WY (RKS) to Hayden, CO (CHE) to Gunnison, CO (GUC) to Needles, CA (EED) to 40 miles east of San Diego, CA (SAN) to 25 miles southwest of Ukiah, CA (UKI) to Fortuna, CA (FOT) and back to Eugene, OR (EUG). The advisory called for occasional mountains obscured in clouds and precipitation with conditions continuing beyond 0900Z through 1500Z.

AIRMET ZULU 3 was issued at 2045Z and was valid until 0300Z on March 6, 1998. The advisory was issued for icing conditions over California, Nevada and the coastal waters of California. The advisory was issued for occasional moderate rime to mixed icing in clouds and in precipitation between the freezing level and 18,000 feet. The freezing level altitudes were provided and over the central California area was estimated at 6,000 feet. The advisory was updated at 0245Z on March 6, 1998, and was valid until 0900Z. The updated advisory extended the area that was affected to include portions of California, Nevada, Utah, Colorado, Arizona and coastal waters. Occasional moderate rime and mixed icing in clouds and in precipitation between the freezing level and 18,000 feet. The advisory stated that the freezing level had lowered over central California to 4,500 to 7,000 feet.

AIRMET TANGO 3 was issued at 2045Z and included California, Nevada, Utah and Arizona. The coordinates of the advisory extended south of a line from San Francisco to Reno, and excluded the accident site but included the destination airport Oakland. The advisory was issued for occasional moderate turbulence below 15,000 feet, with conditions expected to develop over California from 2100Z to 0000Z and spread over the remainder of the area between 0200Z to 0600Z on March 6, 1998. The turbulent conditions were expected to continue through 0900Z. AIRMET TANGO was updated at 0245Z on March 6, 1998, and was valid until 0900Z.

Attachment 9 depicts the in-flight weather advisories that were issued at 2045Z on March 5, 1998. These advisories were current until 0300Z on March 6, 1998. Attachment 10 depicts the AIRMETs that were issued at 0245Z on March 6, 1998, and valid until 0600Z.

There were no SIGMETs or Convective SIGMETs issued for the period from 2200Z on March 5, 1998, through 0500Z on March 6, 1998.

9.0 Terminal Aerodrome Forecast (TAF)

The following terminal forecasts were issued for the departure and destination airports:

9.0.1 Santa Rosa Airport (KSTS)

The TAF for KSTS was issued March 5, 1998, at 2330Z and was valid for a 24-hour period beginning at 0000Z on March 6, 1998. At 0000Z, the winds were forecasted to be from 140 degrees at 8 knots, visibility better than 6 miles in light rain showers. The forecast included a few clouds at 1,500 feet, ceiling broken at 3,000 feet, and broken at 5,000 feet. The forecast also included temporarily conditions between 0000Z and 0200Z of 5 miles visibility in mist and a ceiling broken at 2,500 feet. From 0200Z, the winds were forecasted to be variable at 6 knots with visibility

better than 6 miles. The forecast included a few clouds at 1,000 feet, scattered clouds at 3,000 feet, and ceiling broken at 6,000 feet. The forecast included a temporary condition between 0200Z and 0600Z of light rain showers.

9.0.2 Oakland International Airport (KOAK)

The TAF for KOAK was issued on March 5, 1998 at 2330Z and was valid for a 24-hour period beginning at 0000Z on March 6, 1998. At 000Z, the winds were forecasted to be from 120 degrees at 12 knots with visibility better than 6 miles in light rain showers. Sky conditions included scattered clouds at 1,000 feet, ceiling broken at 2,000 feet, and broken at 4,000 feet. The forecast included temporary conditions between 0000Z and 0300Z of visibility 5 miles in mist, scattered clouds at 2,000 feet, ceiling broken at 3,500 feet. From 0400Z, the winds were forecasted from 280 degrees at 10 knots and visibility better than 6 miles. Sky conditions included a ceiling broken at 4,000 feet and broken at 6,000 feet. The forecast included a temporary conditions of light rain showers and scattered clouds at 4,000 feet between 0400Z and 0600Z.

There were no forecast amendments issued from 0000Z through 0400Z on March 6, 1998, for either Santa Rosa or Oakland.

10.0 Astronomical Data

Accident location:

Novato, CA at N38° 06.61' W122° 33.43'

Sunset:

1808 PST (0208Z)

End of civil twilight: 1834 PST (0234Z)

Altitude of the sun: 12 degrees below the horizon

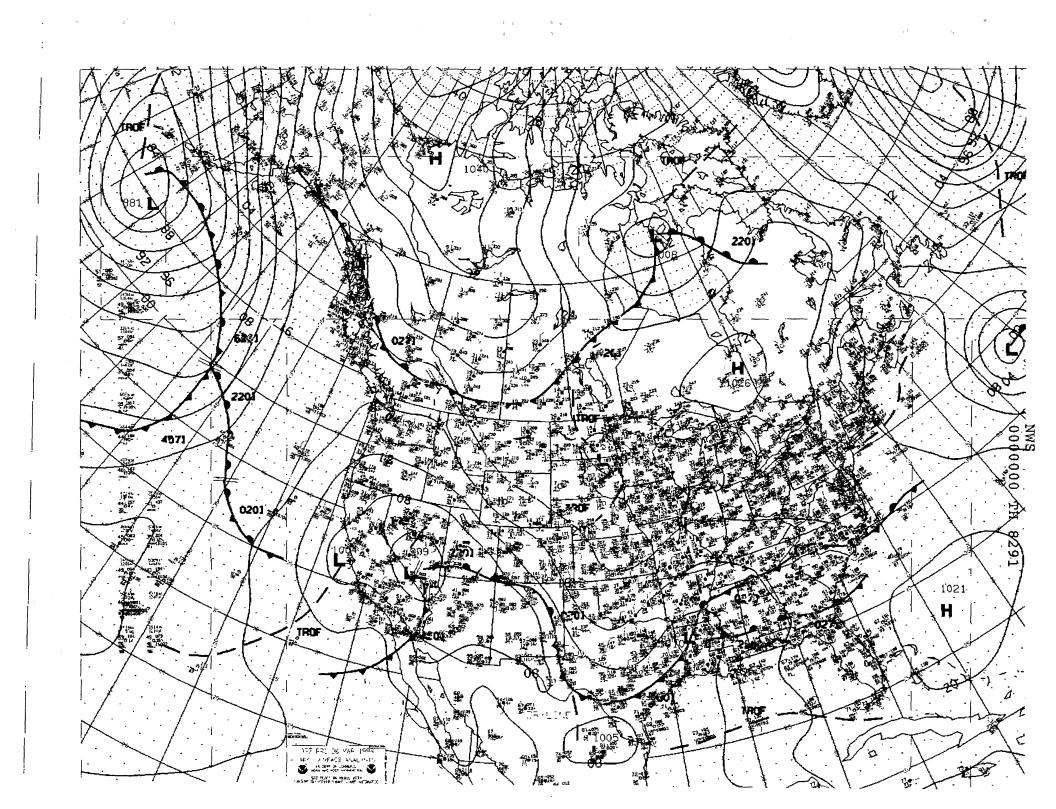
True bearing of the sun: 272 degrees

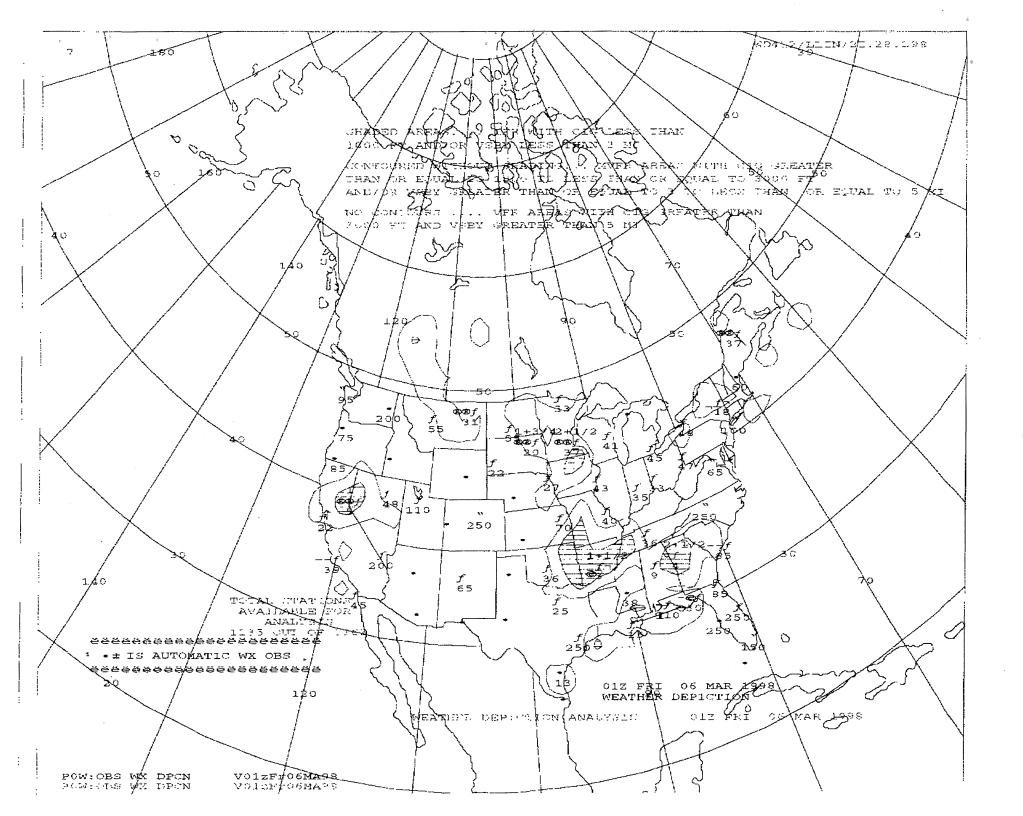
Altitude of the moon: 69.3 degrees above the horizon Percent illumination of moon: 58 % above the clouds

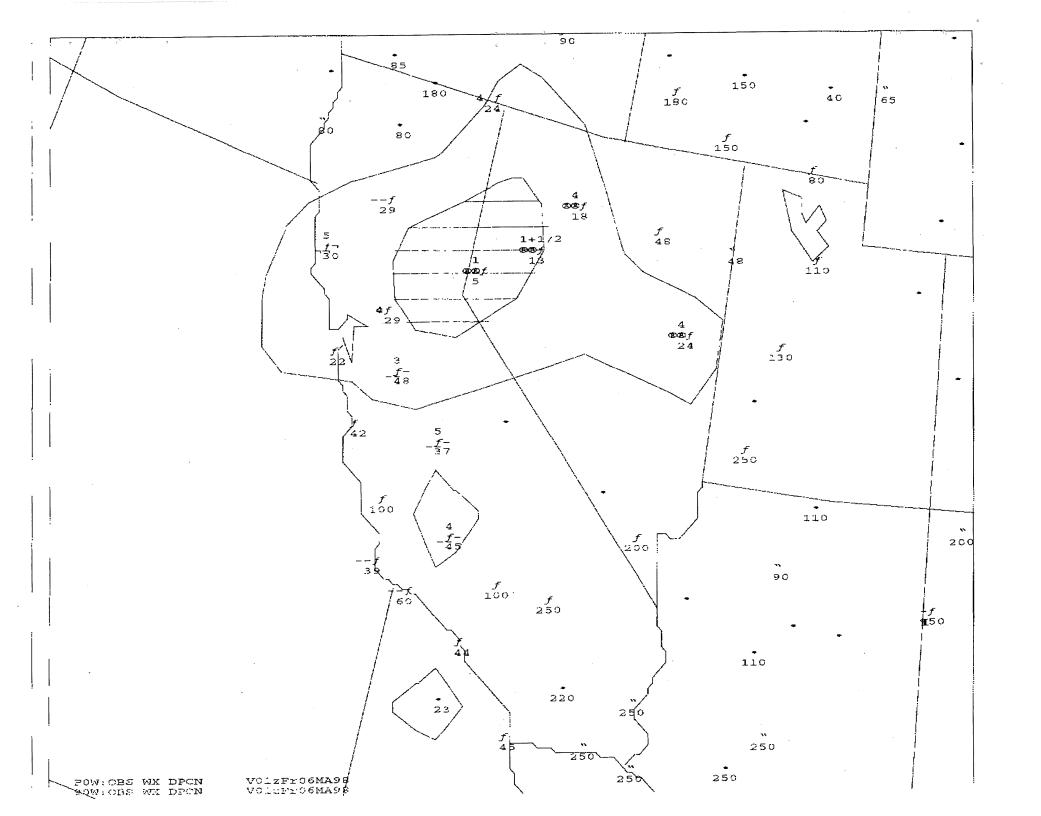
Phase of the moon: First quarter occurred March 5, 1998 at 12:41 PST

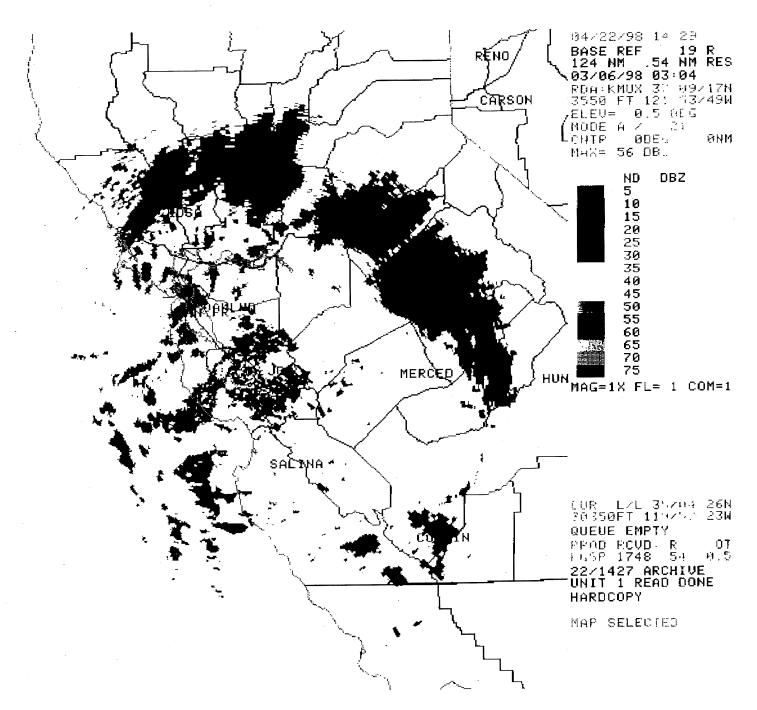
Donald E. Eick

NTSB Meteorologist

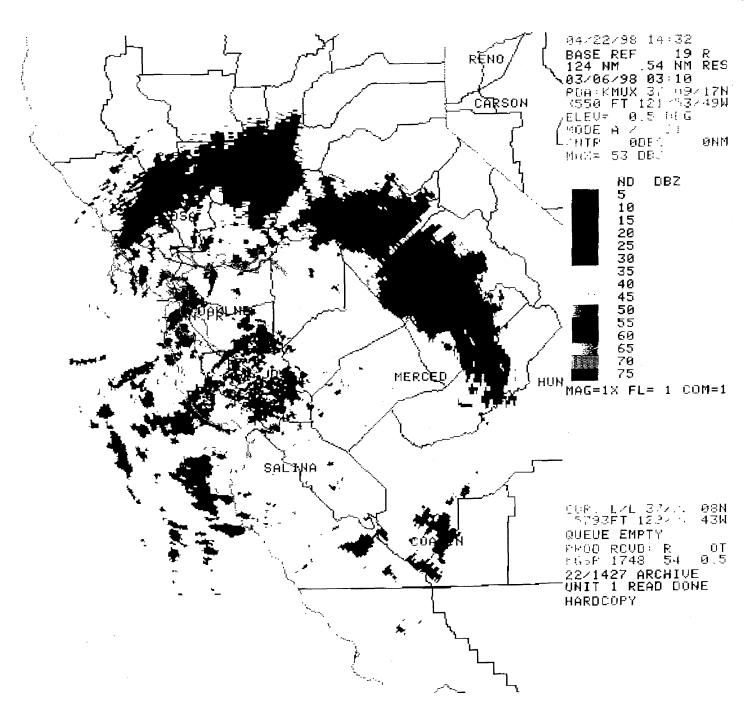




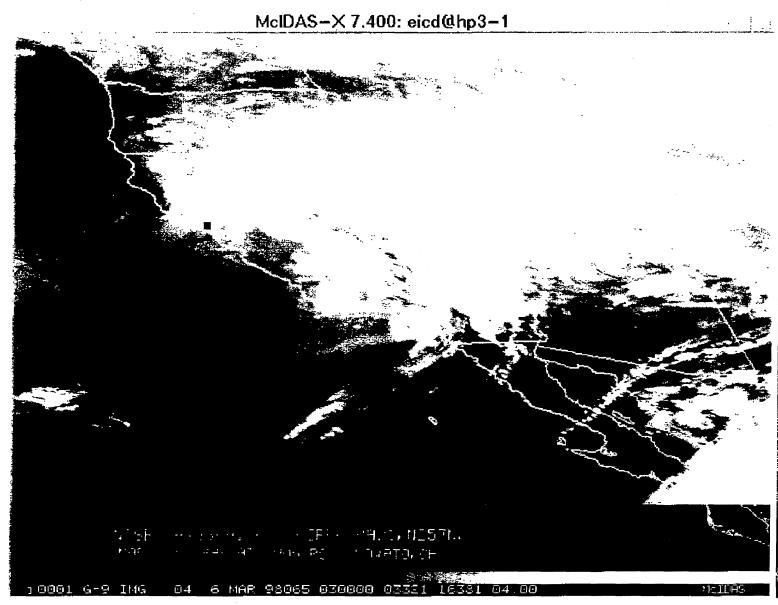




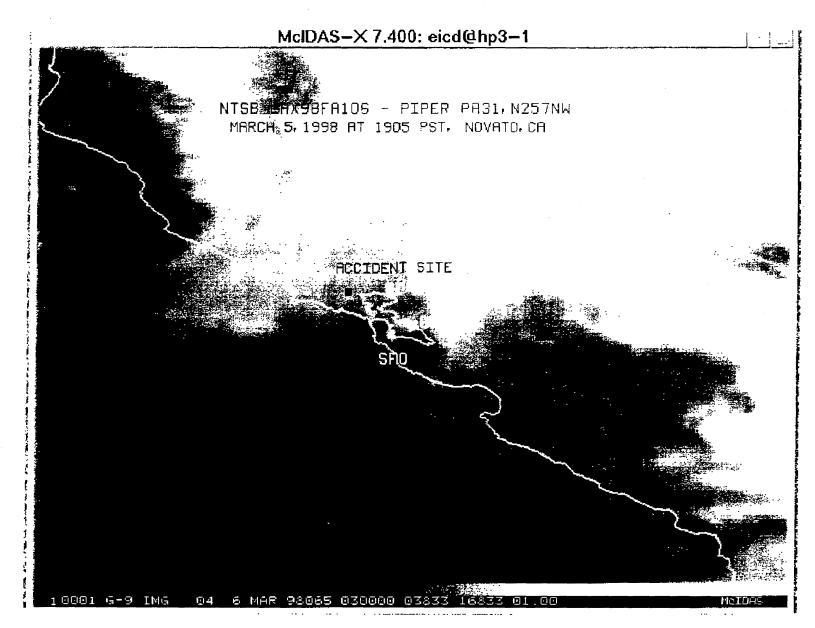
ATTACHMENT 4 - WSR-BBD AT 0304Z FROM KMUX

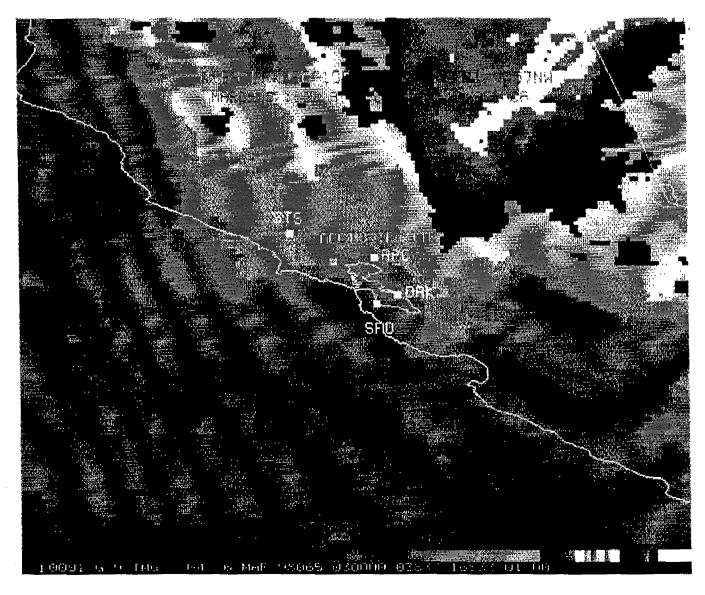


ATTACHMENT 5 - KMUX AT BRIDE

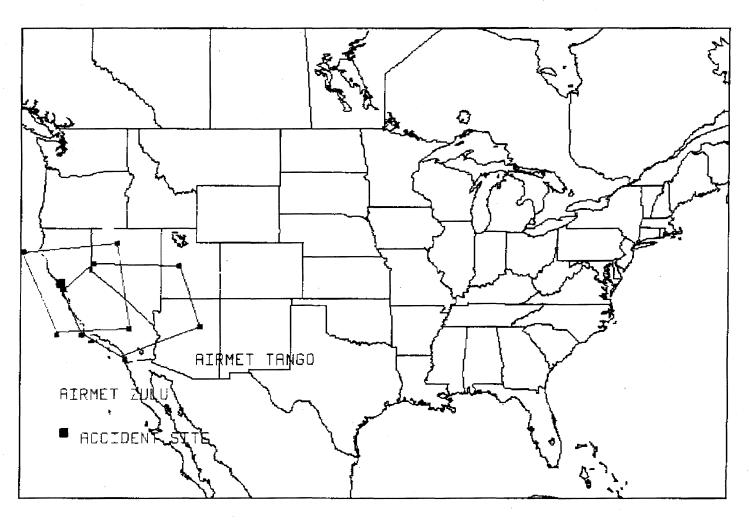


ATTACHMENT G

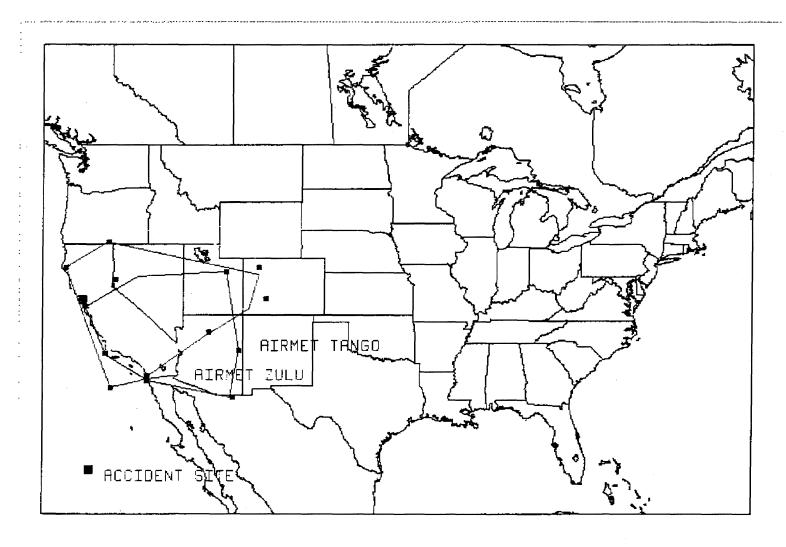




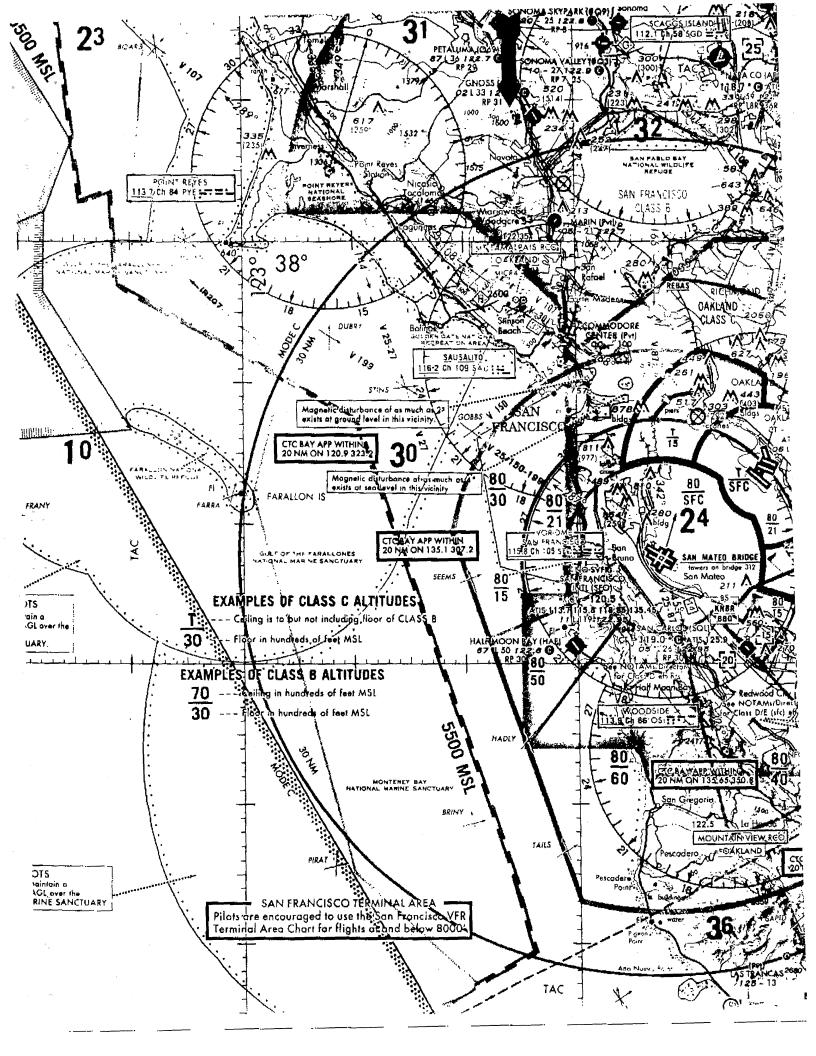
ATTACHMENT 8

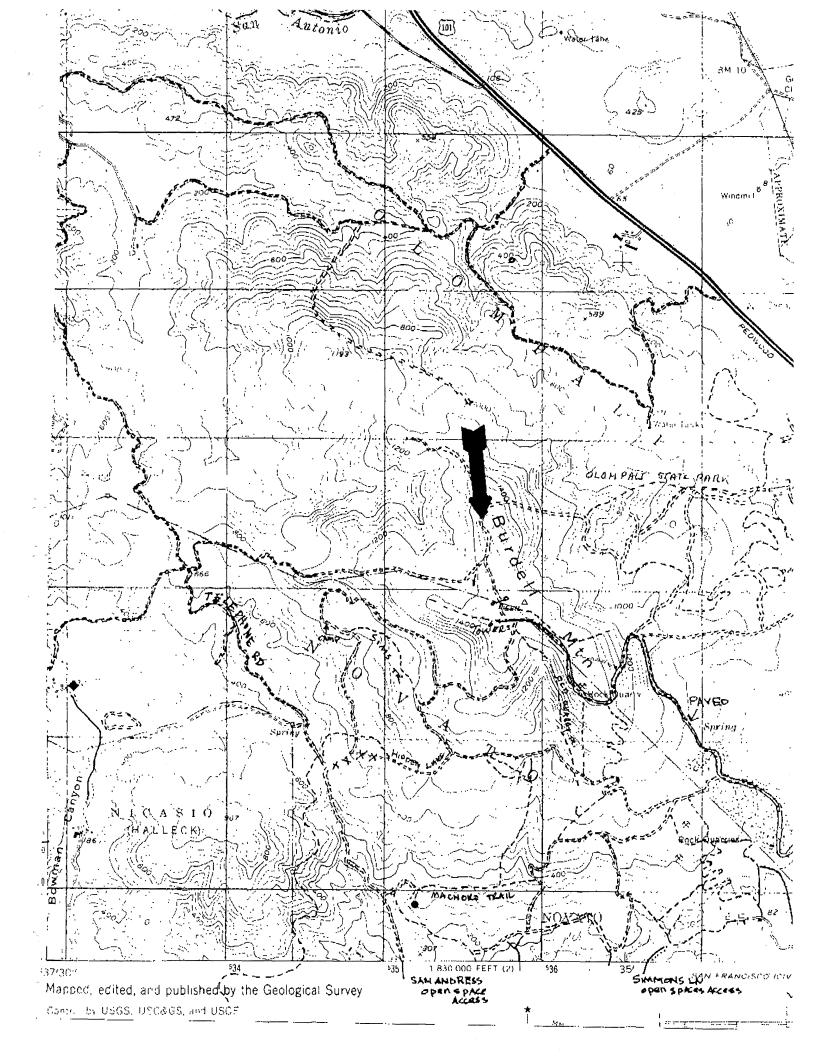


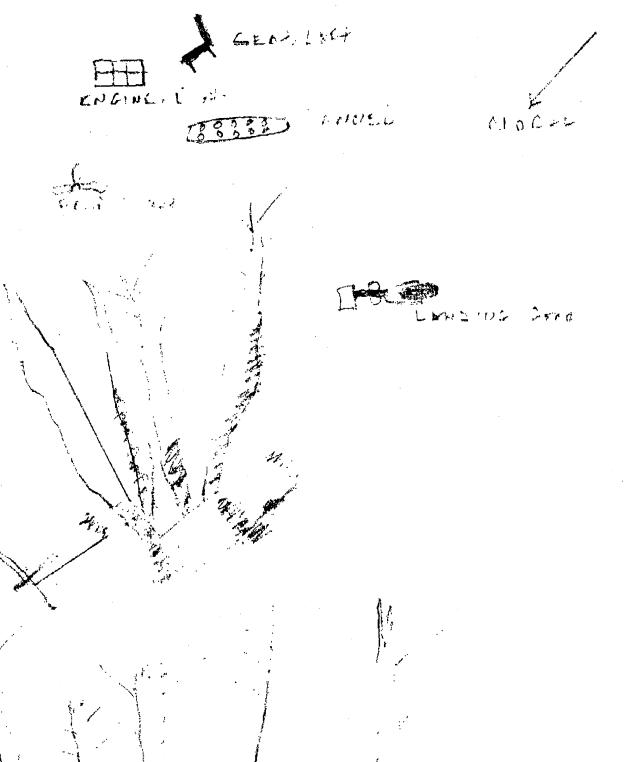
IN FLIGHT WEATHER ADVISORIES ISSUED AT 2045Z 5 MAR 98
ATTACHMENT 9



IN FLIGHT WEATHER ADVISORIES ISSUED AT 0245Z 6MAR9B







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MEMORANDUM

Date:

March 6, 1998

To:

Lt. R. Russell

From:

Stephen Lawrence

Captain, Air Patrol

Subject:

Search For Downed Aircraft On Mount Burdell

We were provided with Lat./Long. Coordinates of 38 degrees, 09 minutes, 2 seconds and 122 degrees, 35 minutes, 7 seconds as the coordinates established by the crew of the United States Coast Guard helicopter, which was on scenc in the early morning hours. This location was found to be off Mount Burdell and was not further pursued.

Martin Sprick and I flew a total of 1.6 hours over the Mount Burdell area.

Given that we did not descend below approximately two thousand feet (2,000') in the vicinity of the point at which the aircraft impacted the terrain, Mount Burdell being one thousand five hundred eighty feet (1,580') MSL (mean secreted), my observations can best be described as impressions or estimations.

The scene of the crash was located on the northerly portion of Mount Burdell at a point where the mountain has risen from north to south to a somewhat level area. As this somewhat level area continues in a southerly direction, the terrain begins to rise again to another area which is also somewhat level and which constitutes the highest point on Mount Burdell.

The aircraft appears to have impacted in either level flight or perhaps while descending, with the terrain in the impact area rising slightly to the south. Given our altitude, I am unable to positively state what heading the aircraft was flying at the time of impact. It is however my impression that the aircraft was travelling on an approximate heading of 340 degrees magnetic. In any event, it is evident that the aircraft was travelling in a southerly

direction and impacted the trees first, either while descending or while the terrain was rising, and disintegrated in a straight line to the south. From my observations it is reasonable to say that the aircraft burned after the initial impact.

I would have to estimate that had the aircraft been perhaps as little as one hundred feet (100') higher, it would have cleared the highest point of Mount Burdell.

The Federal Aviation Administration office in Oakland reports that the only official weather observation for the area occurred at 1950 Hours on March 5th and was made by the Sonoma County Airport at Santa Rosa as follows:

Winds 330 degrees at 5 knots
Visibility 5 statute miles
Moderate rain/mist
Ceiling 3,000 feet broken and 8,000 overcast
Temperature 6 degrees Celsius
Dewpoint 5
Altimeter 29.79 inches of mercury

I am informed by a pilot, that the weather in the Marin County area during the night of March 5th/6th was rain with a low ceiling. What specifically the weather was in the vicinity of Mount Burdell at the time of the crash may never be k. Jwn.

STATEMENT OF PARTY REPRESENTATIVES TO NTSB INVESTIGATION

Aircraft Identification

Registration Number 1557NW

Make and Model Piece Piece

Location Number OA

Date Transport

The undersigned hereby acknowledge that they are participating in the above-referenced aircraft accident or incident investigation (including any component tests and teardowns or simulator testing) on behalf of the party indicated adjacent to their name, for the purpose of providing technical assistance to the National Transportation Safety Board.

The undersigned further acknowledge that they have read the attached copy of 49 C.F.R. Part 831 and have familiarized themselves with 49 C.F.R. § 831.11, which governs participation in NTSB investigations and agree to abide by the provisions of that regulation.

It is understood that a party representative to an investigation may not occupy a legal position or be a person who also represents claimants or insurers. The placement of a signature hereon constitutes a representation that participation in this investigation is not on behalf of either claimants or insurers and that, while any information obtained may ultimately be used in litigation, participation is not for the purposes of preparing for litigation.

By placing their signatures hereon, all participants agree that they will neither assert, nor permit to be asserted on their behalf, any privilege in litigation, with respect to information or documents obtained during the course of and as a result of participation in the NTSB investigation as described above. It is understood, however, that this form is not intended to prevent the undersigned from participating in litigation arising out of the accident referred to above or to require disclosure of the undersigned's communications with counsel.

MANUAL MARK W. PLATT LYCOMING 03/06/98
(Continued on reverse side)

NAME (Print)

PARTY

DATE

DATE

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NATIONAL TRANSPORTATION SAFETY BOARD

RELEASE OF AIRCRAFT WRECKAGE

ACCIDENT IDENTIFICATION NUMBER

LAX-98-F-A106

PART I—RELEASE OF AIRCRAFT WRECKAGE

REGISTERED OWNER (name and address)		REGISTRATION NUMBER—N257NW MAKE Piper				
Airpac Airlines, Inc. 7277 Perimeter Road South Seattle, WA 98108						
MODEL	DATE OF ACCIDENT	LOC	ATION			
PA-31-350	3/5/98		Novato, CA			
The National Transportation Safety Board has wreckage except that listed on the reverse side disposition. (If no parts are retained, insert NC NONE	is hereby released to the registered of					
	TITLE	,	Dize			
SIGNATURE OF NTSH REPRESENTATIVE	IIILE .		DATE			
		.114	4/21/98			
	Air Safety Inves	tigator	4721700			
Receipt of the above described aircraft Removal of the parts, if any, listed on the parts.						
SIGNATURE	TITLE		DATE			
Pet Sol	adjuster		APR 2 7 1998			
REMARKS:						
NTSB FORM 6120.15 {Rev. 5/79)			STATE OF THE STATE			
M19B LORM 0150'12 (K66' 21, 18)						



THESE RECORDS MAY BE RELEASABLE UNDER THE FOIA REQUEST 15 DAYS AFTER SIGNATURE DATE UNLESS WE HEAR OTHERWISE FROM FAA OR NTSB COUNSEL.

U.S. Department of Transportation Mike Monroney Aeronautical Center P.O. Box 25082 Oklahoma City, Oklahoma & 3125

100

Federal Aviation Administration

April 20, 1998

National Transportation Safety Board 1515 W. 190th St., Suite 555 Gardena, CA 90248

CASE#: 9800052001 NAME: ARSONNEAU, ISABELLE Putrefied: No

DATE OF INCIDENT: 030698 DATE RECEIVED: 031198

LOCATION OF ACCIDENT: NOVATO, CA

SPECIMENS RECEIVED: Vitreous fluid, Bile, Liver, Lung, Muscle

FORENSIC TOXICOLOGY FATAL ACCIDENT REPORT

CARBON MONOXIDE:

Carbon monoxide analysis was not performed due to a lack of suitable specimen.

CYANIDE:

Cyanide analysis was not performed due to a lack of suitable specimen.

VOLATILES: The volatile concentrations were determined by headspace gas chromatography at a cutoff of 10 mg/dl. All positive ethanols were confirmed by Radiative Energy Attenuation.

--> NO Ethanol detected in Vitreous fluid

DRUGS: Immunoassay and chromatography are used to screen for abused drugs such as amphetamine (0.010), opiates (0.010), marihuana (0.001), cocaine (0.020), phencylidine (0.002), benzodiazepines (0.030), barbiturates (0.060), and other drugs such as antidepressants (0.100), antihistamines (0.020), meprobamate (0.100), methaqualone (0.100), and nicotine (0.050). The values in () are the threshold values in ug/ml used to report positive results. Values below this concentration are normally reported as not detected.

GC/Mass Spec, or GC/FTIR, is used to confirm most positive results.

--> NO Drugs detected in Muscle

Dennis V. Canfield, Ph.D.

Manager Toxicology and Accident

Research Laboratory

page 1/1

Airpac Airlines, Inc.

Pilot Assignment

Name	ARSONNEAU.
Date Employed	9-17.93
Date of Assignment	Aircraft 9031-750 2-25-99 3-20-95
	Chambe Base $\left(\frac{3}{10-23-96}\right)$
Current Duties	TERIVER CAVE CALLES
•	

Airpac Airlines, Inc.

Certificate of Completion

Company Recurrent

Flight Crew Member/Recurrent Training/Pilot-in-command (Except Specific Aircraft)

This is to certify that
Hours of training completed: 5.0
Date of completion: 10/20/97
Instructor's Signature:
l acknowledge having received the Specific Aircraft Flight training for Pilot-in- Command duties as set forth in the Company Training Program as certified
Signature of PIC trainee:

Form: 160



GROUND TRAINING LOG

Name: Inkelle	HESCAN
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Type of Training

() In	itial	(X)Red	urrent	() Differences	() Transi	tion () Other
Date	A/C#	Type		Operations S. SPECS, OPS MANUAL		Time	Instruct
10/20/97			COLD WE	S. SPECS, OPS MANUAL ATHER OPS., HAZ-MAT	. METEOROLOGY , EMERGENCY PROC	5.0	G. SMI
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Jr.							
Form 141			Rev 1	0-16-92			

Airpac Airlines, Inc.

Certificate of Completion

Aircraft **GROUND** Training and Emergency Drills/Pilot-in-Command

Type Aircraft PA 31	1-350	
This is to certify that ARSONNEA completed all segments, modules, and		has
	Initial New Hire Transition Recurrent	
ground training program for the aircraft Company Training Program, and is pro- pertinent to operation Certificate No. Al Administration.	oficient and knowledgeable in s	subjects
Hours of training completed:	<u>``O</u>	
Date of completion: 10/23/96		
Instructor's Signature:	much)	
And has completed the emergency drill equipment and procedures.	ls, indicated below, using the p	proper
Emergency Evacuation Use of Fire Extinguisher Smoke Control Use of Crew & Passenger Oxygen* Use of Flotation Equipment* Ditching*	Date: Instructors S 10/23/96 Sand 10/23/96 Sand 10/23/96	
*If applicable		
I acknowledge having received the Spe Command duties as set forth in the Cor above.		
Signature of PIC trainee:	1 Jan	
Form: 140 Revised 10-29-93		

FAR 135				OA k	9-19-	97				
MAME OF AIRMAN						TYPE OF CHECK				-
	_		11.			FAR 135.293 🖾 FAR 135.297	İΣ EΔΙ	 2 135 2	go Da	
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\LOT _ÆRTIFICATION	Grade	<u> </u>	<u>-</u>			DATE OF		۔ ' ـــ ' ،		
ÆRTIFICATION NFORMATION:		247				Date of Birth 4.94	Class 1431-350	سر ر سر	7	
EMPLOYED BY		BASED AT				TYPE AIRPLANE (Make/Model)	1431.75	<u></u>		
AMPACA	12/INES	CAK	<u> </u>	<u>سر</u> . ۱۸	· .	Simulator/Training Device (Make/Model)				
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<u> </u>	minh	- Jela	-	49-1	<u> </u>	/ /				
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· · · · · · · · · · · · · · · · · · ·	TAKE	OFFS	1			5, Sim. Engine Failure		1		-
5. Normal						5. Confined Areas, Slopes, & Pinnacles				
6. instrument			Ú			7. Rapid Deceleration (Quick Stops)				
7. Crosswind						8. Autorotations (Single Engine)		1	ļ <u>-</u>	L
	l Powerplant Failure					9. Hovering Autorotations (Single Engin	ne)	-	<u> </u>	_
9. Rejected Taked		SA. K			<u> </u>	10. Tail Rotor Failures (Oral)		┼	ļ	L
IO Page Torre	INFLIGHT M	IANEUVERS	 		1	11. Settling With Power (Oral or Flight) SEAPLANE OP	EDATIONS	<u> </u>	<u> </u>	
10. Steep Turns 11. Approaches to	Stalle	<u></u>	-		 -	1. Taxiing, Sailing, Docking	ERATIONS	1:	Τ	Γ
12. Specific Flight			- 1.5	+		2. Step Taxi & Turns		++	<u> </u>	H
B. Powerplant Fai			िंद	+	 	3. Glassy/Rough Water T.O./Landings		11		Τ
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14. Normal			5			5. Crosswind T.O. & Landings	-			
15, From an ILS						ОТНЕ	R	ļ.		
16. Crosswind						6. Ski Plane Ops. (when applicable)				L
	Powerplant(s) Fail	ure			ļ.,	GENER	RAL		·	<i>,</i> -
to, Rejected Landi					+-	7. Judgment			 -	
19, From Circling		CNOICE	<u></u>			8. Crew Coordination		1	<u> </u>	L.
20 Normal and At	EMERG pnormal Procedures		1.5	- 	T	AIRMAN COMPETENC	Y INFORMATIO	IN:		
21. Emergency Pro				+-	+	Demonstrated Current Knowledge FAR	175 207/-1			
er, Ellergency 110	INSTRUMENT	PROCEDUR	ES	ш		Make/Model Expires 14931-250		e) [/	0/90	-
22. Area Departure		7317	7'-			Demonstrated Competency FAR 135,29				
23. Holding	M+	5,11460	_5			Make/Model Expires PP31 358		s) (/	w/90	U
24. Area Arrival			<u>حـ</u> تــــا			Satisfactority Demonstrated Line Checks			,	
25, ILS Approache	rs		Š			FAR 135,299 Expires	(12 month	1) (/	به و الرت	<u>ې</u>
26, Other Instrume	ent Approaches					Satisfactority Demonstrated IFR Proficie	ncy	,	ىر . ،	_
Approaches:						FAR 135,297 Expires	(6 month	s)" (Z	1.9	<u> </u>
	VOR		25			Use of Autopilot (is) (is not) Authorized.			·*	
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	Other (Specify)	(05	- <u>S</u>			REMARKS DONE IN Th	rin GRAD	حد		
27, Circling Appro- 28, Missed Approa			3		+	1				
29. Comm./Nav. Pr			3	+	+	1				
30. Use of Auto, P				4		1				
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ESULT OF		Disap)	proved			PERFORMANCE (FAA Only)	□ Unsa			
ESULT OF										
			DISTRIC	OFFI	CE	FAA INSPE	CTOR's SIGNAT	URE		

Airpac Airlines, Inc.

Certificate of Completion

Company Basic Emergency Training

(Except Specific Aircraft)

This is to certify that Samuellas completed all segments, modules, and elements of the
Initial New Hire Recurrent
Emergency Training as required by the Company Training Program, and the Federal Aviation Regulations, and is proficient and knowledgeable in subjects pertinent to operation Certificate No. APCA208C issued by the Federal Aviatio Administration.
Hours of training completed: 1.6
Date of completion: 10 20 97
Instructor's Signature:
I acknowledge having received the Specific Aircraft Flight training for Pilot-in- Command duties as set forth in the Company Training Program as certified above. Signature of PIC trainee:

Form: 135 Revised 10-29-93

2 3	33	23	23	43	43	:												
AUC #	1100	VEE 53	208ks	170578	1325		*******											
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Augus Track #2

SUN & MOON INFORMATION

Location : NAVATO, CALIFORNIA

Ref. No.: LAX98FA106

Time/Date: 1905 PST, MARCH 5, 1998

Latitude: 38,09 N Longitude: 122,35 W

UTC Differential : 8 Hours

Magnetic Variation : 16 East

All data computed for an altitude of 1500 Feet MSL

0634	SUNRISE	SUNSET	1811
0607	CIVIL T	WILIGHT	1837
0537	NAUTICAL	TWILIGHT	1908
0506	ASTRONOMICA	L TWILIGHT	1938
	de of Sun		-11.5
Magneti	c Bearing to	Sun	256.1
	le of Moon		1
	ic Bearing to		277.1
Percent	: Illumination	of Moon	57 %

No Compensation For Non-Level Horizons
Standard Corrections for Atmospheric Refraction and Parallax

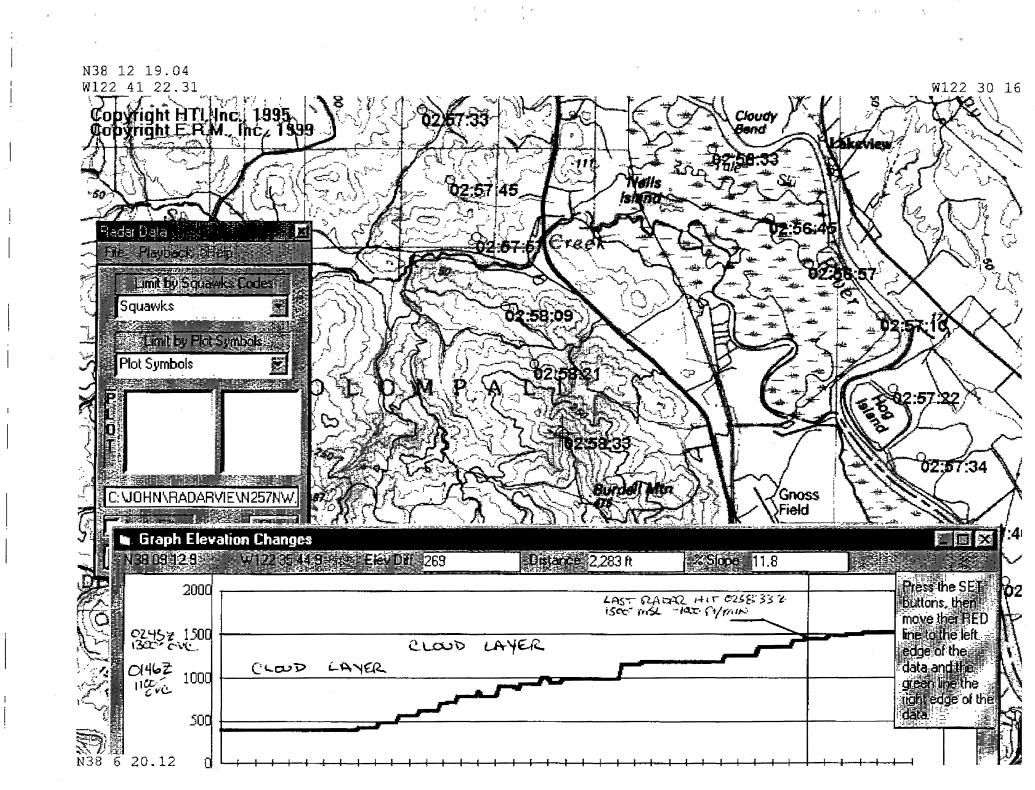
Time Latitude Longi	itude Altitud	le knots	V.ft/m	inbearing	Nmiles	
0250:20 N38 30 14	W122 48 25	100	115	1000	162.2	0.38
0250:32 N38 29 52	W122 48 16	300	141	1500	148.1	0.47
0250:44 N38 29 28	W122 47 57	600	158	1000	131.7	0.53
0250:56 N38 29 07	W122 47 27	800	137	500	147,0	0.46
0251:08 N38 28 44	W122 47 08	900	131	1000	143,2	0.44
0251:20 N38 28 23	W122 46 48	1100	143	1000	146.8	0.48
0251:32 N38 27 59	W122 46 28	1300	159	500	151.9	0.53
0251:44 N38 27 31	W122 46 09	1400	168	500	135,5	0.56
0251:56 N38 27 07	W122 45 39	1500	180	500	140,8	0,60
0252;08 N38 26 39	W122 45 10	1600	155	500	165.3	0,52
0252:20 N38 26 09	W122 45 00	1700	159	500	167.1	0.53
0252;32 N38 25 38	W122 44 51	1800	171	500	151.2	0.57
0252:44 N38 25 08	W122 44 30	1900	194	0	168.3	0.65
0252:56 N38 24 30	W122 44 20	1900	150	0	180.0	0,50
0253:08 N38 24 00	W122 44 20	1900	189	. 0	168,0	0.63
0253:20 N38 23 23	W122'44 10	1900	167	500	150.3	0.56
0253:32 N38 22 54	W122 43 49	2000	188	-500	169.2	0.63
0253;44 N38 22 17	W122 43 40	1900	174	0	153.1	0.58
0253;56 N38 21 46	W122 43 20	1900	165	0	151.5	0.55
0254:08 N38 21 17	W122 43 00	1900	201	0	156.9	0.67
0254:20 N38 20 40	W122 42 40	1900	141	500	148.1	0.47
0254:32 N38 20 16	W122 42 21	2000	184	-461	158.0	0.66
0254:45 N38 19 39	W122 42 02	1900	167	500	150.3	0.56
0254;57 N38 19 10	W122 41 41	2000	188	0	169.2	0.63
0255:09 N38 18 33	W122 41 32	2000	165	0	134.2	0.55
0255:21 N38 18 10	W122 41 02	2000	201	0	156.9	0.67
0255:33 N38 17 33	W122 40 42	2000	199	0	158.0	0,66
0255:45 N38 16 56	W122 40 23	2000	150	0	180.0	0.50
0255:57 N38 16 26	W122 40 23	2000	195	-500	142.7	0.65
0256:09 N38 15 55	W122 39 53	1900	182	500	142.7	0.61
0256:21 N38 15 26	W122 39 25	2000	201	0	156.9	0.67
0256:33 N38 14 49	W122 39 05	2000	137	0	146.9	0.46
0256;45 N38 14 26	W122 38 46	2000	206	0	157.5	0.69
0256;57 N38 13 48	W122 38 26	2000	169	0	152.3	0.56
0257:09 N38 13 18	W122 38 06	2000	195	500	157,4	0.65
0257:21 N38 12 42	W122 37 47	2100	169	-500	152.3	0.56
0257;33 N38 12 12	W122 37 27	2000	201	0	156.9	0.67
0257;45 N38 11 35	W122 37 07	2000	168	0	153.4	0.56
0257;57 N38 11 05	W122 36 48	2000	201	-500	156.9	0.67
0258:09 N38 10 28	W122 36 28	1900	174	-1000	153.0	0.58
0258;21 N38 09 57	W122 36 08	1700	206	-1000	157.4	0.69
0258;33 N38 09 19	W122 35 48	1500				

N257NW ALTITUDE PROFILE

Altitude

→ Altitude

Page 1



Time	Latitude	Longitude	Alt.	knots	ft/min	bearing	NMiles
0250:20.00	N38-30-14"	W122-48-25'	' 100) 115	1000	162	0.38
	N38-29-52"						0.47
		W122-47-57					0.53
	N38-29-07"						0.46
0251:08.00	N38-28-44"	W122-47-08					0.44
0251:20.00	N38-28-23"						0.48
	N38-27-59"						0.53
0251:44.00	N38-27-31"						0.56
0251:56.00	N38-27-07"	W122-45-39'					0.60
0252:08.00	N38-26-39"	W122-45-10'					0.52
0252:20.00	N38-26-09"						0.53
0252:32.00	N38-25-38"	W122-44-51	1800				0.57
0252:44.00	N38-25-08 ¹¹	W122-44-30'	1900	194			0.65
0252:56.00	N38-24-30"	W122-44-20'	1900				0.50
0253:08.00	N38-24-00"	W122-44-20'	1900				0.63
0253;20.00	N38-23-23"	W122-44-10'	1900	167	7 500		0.56
0253:32.00	N38-22-54"	W122-43-49	2000				0.63
0253;44.00	N38-22-17"	W122-43-40'	1900				0.58
0253:56.00	N38-21-46"	W122-43-20'	1900	165			0,55
	N38-21-17"						0.67
0254:20.00	N38-20-40"	W122-42-40'	1900				0.47
0254:32.00	N38-20-16"	W122-42-21	2000	184	-461		0.66
0254:45.00	N38-19-39"	W122-42-021	1900	167	7 500		0.56
0254:57.00	N38-19-10"	W122-41-41'	2000	188	3		0.63
0255:09.00	N38-18-33"	W122-41-32	2000	165	5 (134	0.55
	N38-18-10"			201		157	0.67
		W122-40-42	2000	199) (0.66
0255:45.00	N38-16-56"	W122-40-23	2000	150) (0.50
0255:57.00	N38-16-26"	W122-40-23	2000	195	-500	143	0.65
	N38-15-55"		1900	182	500	143	0.61
	N38-15-26"			201		157	0.67
		W122-39-05		137	7 (147	0.46
		W122-38-46		206	5 (157	0.69
	N38-13-48"		2000	169) (152	0.56
	N38-13-18"						0.65
	N38-12-42"						0.56
	N38-12-12"			201	L		0.67
	N38-11-35"				-		0.56
		W122-36-48'					0.67
	N38-10-28"						0.58
		W122-36-08			-1000	157	0.69
0258:33.00	N38-09-19"	W122-35-48	" 1500)			

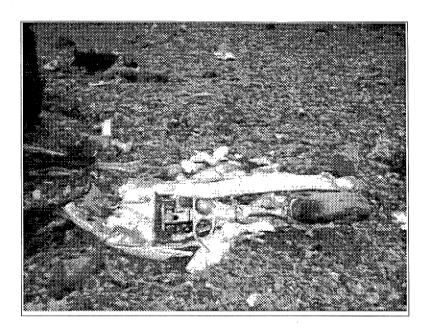


Photo 01: View of Nose Gear with Nose Gear Door (NTSB)

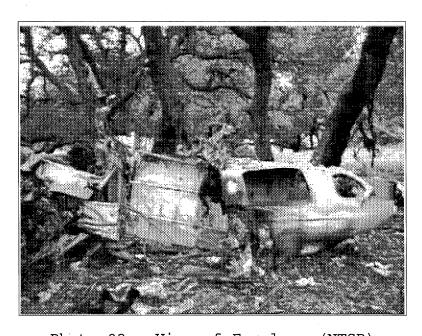


Photo 02: View of Fuselage (NTSB)

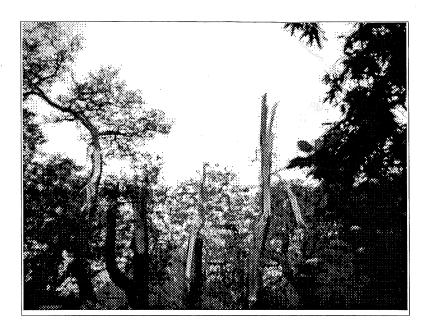


Photo 03: Initial Impact Point Looking in the Direction of Approach (NTSB)

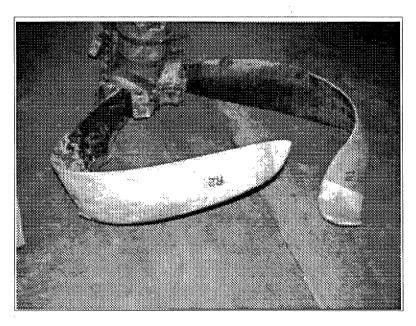


Photo 04: View of Right Propeller (NTSB)

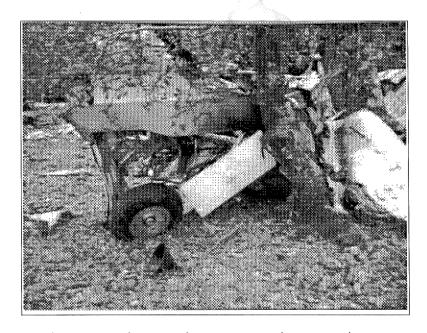


Photo 05: View of Right Wing and Main Landing Gear (NTSB)

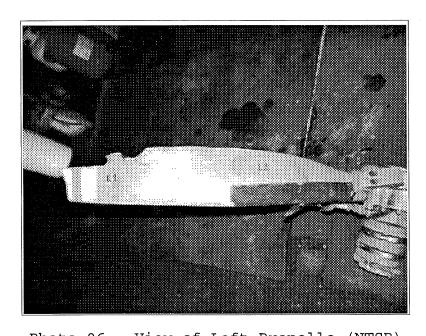


Photo 06: View of Left Propelle (NTSB)